


QJ-MED

CA Form 15

REPUBLIC



OF ZAMBIA

Now  QJ Doc

AIRFRAME LOG-BOOK

Im also N701 8/71

11-2003-11-3004

QJN 9 QJ 1045

AIRFRAME LOG-BOOK

INSTRUCTIONS FOR USE

1. The entries in this log-book shall be made and signed in accordance with the provisions of the Air Navigation Regulations for the time being in force under the Aviation Act, 1954, or any Act amending or replacing the same.
2. All entries are to be made in ink or indelible pencil as soon as is practicable after completion.
3. The following requirements in connection with the completion of column 4 should be noted:
 - (a) MAJOR COMPONENTS:

When removal, renewal or major repair is involved part numbers and serial numbers must be quoted.
 - (b) OTHER COMPONENTS:

The part number must be quoted, or, if this cannot be ascertained an adequate description should be entered for easy reference.
 - (c) MODIFICATIONS:

The modification number, together with a brief title of the modification should be entered in column 4 and in the 'Modification Summary' on the last pages of this log-book.
 - (d) TESTS:

The complete designation of tests undergone by equipment must be stated.
 - (e) RELEASE NOTES:

Release notes or other certificates relating to the overhaul/repair of components and ancillary equipment fitted must be quoted.
 - (f) LOOSE CERTIFICATES:

Loose log-certificates are to be attached to the log-book by gluing only.
4. This log-book should be kept in safe custody since its loss may involve the owner in a long delay in bringing the aircraft into operation, in addition to possible extensive physical examination of the aircraft before a duplicate log-book is issued.
5. This log-book must not be carried in the aircraft to which it relates.
6. This log-book shall be produced, on demand, for the inspection of any authorised person.
7. This log-book shall be preserved until a date two years after the aircraft to which it relates has been destroyed or permanently withdrawn from service.

AMO 104

CERTIFICATE RELATING TO MAINTENANCE OF AN AIRCRAFT
CRMA No: 1800

Aircraft Type: Beechcraft King Air B200	Registration Number: 9J-MED	Serial Number: BB1046
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1. Phase 2 inspection carried out as per Beechcraft Maintenance Manual No: 101-590010-19B, Revision B7, Revised 31-Jan-03.
2. Hours corrected to 7146.7 Total Time – addition error discovered from flight folio – corrections made to flight folio by pilot.
3. New forward evaporator filter fitted.
4. New forward blower vent filter fitted.
5. New LH ice vane indication system internal separator switch fitted – P/N: 101HS1.
6. New seat belts fitted to bench seats.
7. New cargo net manufactured and fitted.
8. New interior dome lights fitted.
9. Spare tyres supplied.
10. Door micro switch mechanisms repaired.
11. Two new fire extinguishers fitted – new brackets bolted to floor panels.
12. LH landing gear micro switch adjusted.
13. Flush patch fitted to fuselage in non pressure area IAW AC 43.
14. Garmin GPS 150 repositioned – sheet metal work carried out.
15. New #3 main tyre fitted.
16. #3 and #4 brake assembly overhauled IAW Manual 99-36816, A2, Revised 30/84.
17. LH wing white LD light – bad connection at wing tip – repaired.
18. LH wing ice light – bad connection inside nacelle – repaired.
19. Wing spar cap inspections carried out as per Component TBO Listing.
20. Nacelle fuelcell and probe inspections carried out as per Component TBO Listing.
21. Fitted new drain fittings to LH and RH wing tanks.

Records of all work carried out held under Job Card 1800.

I hereby certify that in carrying out the foregoing specified maintenance all the requirements prescribed in the Civil Aviation Regulations, 1997, as amended, which are applicable hereto, have been complied with.



License or Approval No: (Stamp): -----

G.Burrill

Date: 17-Dec-03

1.		PART OR MATERIAL CERTIFICATION FORM				3. System Shipping Ret No. 01-80145556-30			
2. Seller's Name		AIRCRAFT BRAKING SYSTEMS CORPORATION				5. Customer P. O. No. PO #: 119661 Item: 000003			
4. Address Aircraft Braking Systems Corporation 1204 Massillon Road Akron, Ohio 44306-4186 Phone: (330)796-4400 SITA CODE: CAKLLCR Fax: (330)796-9805		6. Item		7. Description	8. Part Number*	9. Eligibility**	10. Qty	11. Ser./Batch No.	12. Status/Work
		1		RIVET, TUBULAR	GY18B-7	*See Below	9,500 EA	N/A	New
13A. Remarks		*8. All part mfg./Insp. by ABSC **9. Eligibility (Consult ABSC manuals & service bulletins for additional eligibility) FAA PMA to the following model(s): DORNIER DO28D, DO28D-1 to 9543989-1 TSO Article, N/A TSO-C26c to 9544482 TSO-C26b to 5002376 TSO-C26c to 5000059-1 TSO-C26c to 9543163 TSO-C26g to 5003096-5							
13B. Obtained from: N/A ABSC Manufactured/Inspected.		13C. Last Cert. Agency - ABSC Certification							
14. New - Unused Parts/Material Certification: The following signature attests that the part(s) or material(s) identified above was (were) manufactured by an FAA approval holder and inspected in accordance with established industry or U.S. specifications, and/or the airworthiness regulations of the country of origin. FAH No. PT2064CE-D(TSO)		18. Used, Repaired or Overhauled Part Certification: The following signature attests that the documentation specified above or attached is accurate with regard to the item(s) described. Repaired parts conform and were inspected in accordance with the Code of Federal Regulations 14 and were approved for return to service.							
15. Authorized Signature: <i>Robert Salisbury</i>		19. Authorized Signature:							
16. Name (Typed or Printed) ROBERT SALISBURY		17. Date 30 Sep 01		20. Name (Typed or Printed)				21. Date	

CONFORMANCE CERTIFICATE FOR MATERIAL SHIPPED

Parker O-Ring Division certifies that the material used in the manufacture of the parts identified below and called for on Purchase Order Number 114925 received by us from VALTEC INC is traceable to C.B.I Number 80007173 and that the material used conforms to the requirements of the specifications indicated below. The tests were performed in accordance with these specifications, and data supporting the indicated results are on file.

CUSTOMER PART NUMBER	PARKER PART NUMBER	COMPOUND	QUANTITY	CURE DATE	SHIP DATE
MS28775-220	2-220	N0304 -75	2998	3Q02	09/18/02
TESTED AS: 2-214					
PHYSICAL PROPERTIES		REQUIREMENTS	VARIATION	PARKER FINDINGS	
SPECIFICATION AMS-P-25732	TYPE	CLASS	TEST DT 09/09/02		
SPECIFIC GRAVITY			(1.26 - 1.30)	1.2800	
HARDNESS SHORE A		(68 - 80)	(70 - 80)	76	
TENSILE STRENGTH, PSI		1350 MIN	(1440 - 2160)	1856	
ELONGATION, %		160 MIN	(160 - 226)	223	
MODULUS AT 100% ELONG.		500 MIN	(500 - 9999)	573	
TR-10 50% ELONG. 10% RETURN		-49 MAX		-54.0	
FLUID AGEING 70 HRS AT 275F IN MIL H-5606					
COMPRESSION SET % MAX		55		31.6	
TR-10		-49 MAX		-53.0	
HARDNESS CHANGE		+5 -15		-13.0	
VOLUME CHANGE		1 +20		15.5	

RECEIVED

SEP 26 2002

INSPECTOR 7

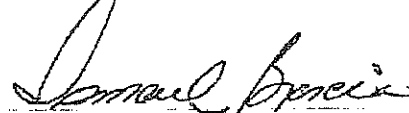
No Mercury, Mercury compounds or Mercury bearing instruments, equipment or apparatus which might cause contamination has been used in the manufacture or subsequent processing of material covered by this order.

Purchaser use only. Reproduce only in full. Data pertains to items referenced.

The information contained herein is submitted as the privileged and confidential property of the Parker Hannifin Corporation pursuant to 5 U.S.C. Section 552 (b)(3) and (b)(4), the Freedom of Information Act, exemption Number 3 and Number 4. This information shall not be duplicated, used or disclosed in whole or in part outside the recipient organization without express written consent of the Parker Hannifin Corporation.

O-Ring Division
PARKER HANNIFIN CORPORATION

By




Ismael Garcia

Quality Assurance Manager

1. Approving National Aviation Authority / Country FAA / United States		2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, Airworthiness Approval Tag		3. Form Tracking Number: AVI-ATL-02-10-050	
4. Organization Name and Address: Aviall Services Inc. 574 Airport South Parkway, Suite 200, Atlanta, GA 30349				5. Work Order / Contract / Invoice Number: PO # PO10271	
6. Item: 1	7. Description: WEAR PAD, BRAKE	8. Part Number: 5002105 101-8002-33	9. Eligibility: TSO ARTICLE N/A TSO-C26b	10. Quantity: 56EA	11. Serial/Batch Number: NSN
13. Remarks: EXPORT TO: SOUTH AFRICA Part Condition: NEW				12. Status / Work:	NEW

MANUFACTURED BY AIRCRAFT BRAKEING SYSTEMS CORP., 1204 MASSILLON RD., AKRON, OH 44306-4186 USA

14. Certifies the items identified above were manufactured in conformity to:		19. <input type="checkbox"/> 14 CFR 43.9 Return to Service <input type="checkbox"/> Other Regulation Specified in Block 13 Certifies that unless otherwise specified in Block 13, the work identified in Block 12 and described in Block 13 was accomplished in accordance with Title 14 of the Code of Federal Regulations, Part 43 and in respect to that work, the items were approved for return to service.	
<input checked="" type="checkbox"/> Approved design data and are in a condition for safe operation <input type="checkbox"/> Non-approved design data specified in Block 13		15. Authorized Signature:  16. Approval/Authorization No.: DART-20018-SO 17. Name (Typed or Printed): Charles R. Hubbard 18. Date (m/d/y): October 28, 2002	
		21. Approval/Certification: N/A 23. Date (m/d/y): N/A	

User / Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the part/component/assembly.

Where the user/installer work in accordance with the national regulations of an Airworthiness Authority different than the Airworthiness Authority of the country specified in block 1 it is essential that the user/installer ensures that his/her Airworthiness Authority accepts parts/components/assemblies from the Airworthiness Authority of the country in block 1.

Statements in block 14 and 19 do not constitute an installation certification. In all cases aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

AIRCRAFT BRAKING SYSTEMS

Aircraft Braking Systems 1204 Massillon Rd. Akron OH 44306-4188 USA

Packing list

Page 1 of

Ship To: 1072
AVIALL INC. DISTRIBUTION DIV
CENTRAL WAREHOUSE
2750 REGENT BLVD
DFW AIRPORT, Texas 75261
USA

Sold To: 340
AVIALL SERVICES
ATTN: PURCHASING
PO BOX 619048
DALLAS, Texas 75261-9048
USA

Delivery no: 80198584

Group no: 2000096500

Customer P.O.: 194667



Delivery no: 80198584

All items listed on this document are pr
subject to the provisions of ABSC Gene
GN-1 as to the use of unauthorized par

Manufacturer Part Number	Customer Part Number	Description	PO line Item	Qty Shipped	Unit of Measure	Condition	Cure/Expiration Date
5002105		PAD, WEAR ABSC FAA PAH# PT2064CE-D(TSO) & PO2064CE-D(PMA)	< none >	400	EA		

Date Shipped: 03-Oct-2003
Method of Shipment: ROADWAY EXPRESS
Waybill No.: 2000096500
Number of Pallets: Number of Crates:
Number of Boxes: 1
Total Weight: 45 LB

Dimensions:
12X12X6

(TRA) TRAC #: 80198



PO 0194667

23

PART OR MATERIAL CERTIFICATION FORM

1. Seller's Name

AIRCRAFT BRAKING SYSTEMS CORPORATION

4. Address

Aircraft Braking Systems Corporation

Phone: (330)796-4400 SITA CODE: CAKLICR

1204 Massillon Road Akron, Ohio 44306-4186

Fax: (330)796-9805

6. Item	7. Description	8. Part Number*	9. Eligibility**	10. Qty	11. Ser/ Batch No.	12. Status/Work
1	PAD, WEAR	5002105	*See Below	400 EA	N/A	New

13A. Remarks

*8. All Part mfg./Insp. by ABSC

**9. Eligibility (Consult ABSC manuals & service bulletins for additional eligibility)

FAA PMA to the following model(s):

DORNIER DO28D, DO28D-1 to 9543999-1

TSO Article, N/A

TSO-C26b to 5003065-2

TSO-C26b to 5003065

TSO-C26c to 9543163

TSO-C26c to 5000059-1

TSO-C26b to 5002374

13B. Obtained from: N/A ABSC Manufactured/Inspected.

13C. Last Cert. Agency - ABSC Certification

14. New - Unused Parts/Material Certification:

The following signature attests that the part(s) or material(s) identified above was (were) manufactured by an FAA approval holder and inspected in accordance with established industry or U.S. specifications, and/or the airworthiness regulations of the country of origin.

FAH No. PT2064CE-D(TSO)

FAH No. PQ2064CE-D(PMA)

15. Authorized Signature:

Mark A. Matheny

16. Name (Typed or Printed)

MARK A. MATHENY

17. Date

01 Oct 02

20. Name (Typed or Printed)

21. Date

19. Authorized Signature:


18. Used, Repaired or Overhauled Part Certification:

The following signature attests that the documentation specified above or attached is accurate with regard to the item(s) described. Repaired parts conform and were inspected in accordance with the Code of Federal Regulations 14 and were approved for return to service.

1. Approving National Aviation Authority / Country FAA / United States		2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, Airworthiness Approval Tag		3. Form Tracking Number: AVI-ATL-03-11-027	
4. Organization Name and Address: Aviall Services Inc. 574 Airport South Parkway, Suite 200, Atlanta, GA 30349				5. Work Order / Contract / Invoice Number: PO # 10939	
6. Item: 1	7. Description: DISK, STATOR BRAKE	8. Part Number: 5002259	9. Eligibility: N/A TSO ARTICLE TSO-C26b	10. Quantity: 4EA	11. Serial/Batch Number: NSN, MARKED OB9R9
13. Remarks: EXPORT TO: Part Condition: NEW				12. Status / Work: NEW	

MANUFACTURED BY AIRCRAFT BRAKING SYSTEMS CORP., 1204 MASSILLON ROAD, AKRON, OHIO 44306-4186

NOTE: (BEECH PN 101-8002-15)

14. Certifies the items identified above were manufactured in conformity to:		19. <input type="checkbox"/> 14 CFR 43.9 Return to Service <input type="checkbox"/> Other Regulation Specified in Block 13 Certifies that unless otherwise specified in Block 13, the work identified in Block 12 and described in Block 13 was accomplished in accordance with Title 14 Code of Federal Regulations, Part 43 and in respect to that work, the items approved for return to service.	
<input checked="" type="checkbox"/> Approved design data and are in a condition for safe operation <input type="checkbox"/> Non-approved design data specified in Block 13			
15. Authorized Signature: 	16. Approval/Authorization No.: DART-20018-SO	20. Authorized Signature: N/A	
17. Name (Typed or Printed): Charles R. Hubbard	18. Date (m/d/y): November 19, 2003	21. Approval/Certification: N/A	
		22. Name (Typed or Printed): N/A	
		23. Date (m/d/y): N/A	

User / Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the part/component/assembly.

Where the user/installer work in accordance with the national regulations of an Airworthiness Authority different than the Airworthiness Authority of the country specified in block 1 it is essential that the user/installer ensures that his/her Airworthiness Authority accepts parts/components/assemblies from the Airworthiness Authority of the country in block 1.

Statements in block 14 and 19 do not constitute an installation certification. In all cases aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

AIRCRAFT BRAKING SYSTEMS

Aircraft Braking Systems 1204 Massillon Rd Akron OH 44306-4186 USA

Packing list

Page 1 of

Ship To: 1072
AVIALL CENTRAL WAREHOUSE
2750 REGENT BOULEVARD
2750 REGENT BLVD
DFW AIRPORT, Texas 75261
USA

Sold To: 340
AVIALL SERVICES
ATTN: PURCHASING
PO BOX 619048
DALLAS, Texas 75261-9048
USA

Delivery no: **80248646**
Group no: 2000119000

Customer P.O.: 251699



All items listed on this document are
subject to the provisions of ABSC Ge
GN-1 as to the use of unauthorized p

Delivery no: **80248646**

Manufacturer Part Number Customer Part Number	Description	PO line item	Qty Shipped	Unit of Measure	Condition	Cure/Expira Date
5002259	DISK STA PN 101-8002-15(1/10) ABSC FAA PAH# PT2064CE-D(TSO)	000003	20	EA		
5004608	DISC STA (1/5) ABSC FAA PAH# PT2064CE-D(TSO)	000004	5	EA		

ENTERED
OCT 03 2003
SP

(TRA) TRAC #:80248646

PO 0251699

LOT AV1003604491

Dimensions:
1 @ 14X14X4, 2 @ 11X11X5 EACH

Date Shipped: 25-Sep-2003
Method of Shipment: ROADWAY EXPRESS
Waybill No.: 2000119000
Number of Pallets: Number of Crates:
Number of Boxes: 3
Total Weight: 89 LB


1. PART OR MATERIAL CERTIFICATION FORM				3. System Shipping Ref No. 03-80248646-30		
2. Seller's Name AIRCRAFT BRAKING SYSTEMS CORPORATION				5. Customer P. O. No. PO #: 251699 Item: 000003		
4. Address Aircraft Braking Systems Corporation 1204 Massillon Road Akron, Ohio 44306-4186 Phone: (330)795-4400 SYTA CODE: CAKLLCR Fax: (330)795-9805						
6. Item	7. Description	8. Part Number*	9. Eligibility**	10. Qty	11. Ser/Batch No.	12. Status/Wor
1	STATOR	5002259	*See Below	20 EA	N/A	New
13A. Remarks *8. All Part mfg./Insp. by ABSC *9. Eligibility (Consult ABSC manuals & service bulletins for additional eligibility) TSO Article, N/A TSO-C26b to 5002256						
13B. Obtained from: N/A ABSC Manufactured/Inspected.			13C. Last Cert. Agency - ABSC Certification			
14. New - Unused Parts/Material Certification: The following signature attests that the part(s) or material(s) identified above was (were) manufactured by an FAA approval holder and inspected in accordance with established industry or U.S. specifications, and/or the airworthiness regulations of the country of origin.			18. Used, Repaired or Overhauled Part Certification: The following signature attests that the documentation specified above or attached is accurate with regard to the item(s) described. Repaired parts conform and were inspected in accordance with the Code of Federal Regulations 14 and were approved for return to service.			
PAH No. PT2064CE-D(TSO)						
15. Authorized Signature: <i>Jerome F. Hughes</i>			19. Authorized Signature:			
16. Name (Typed or Printed) JEROME F. HUGHES			17. Date 15 Sep 03		20. Name (Typed or Printed) 21. Date	

1. Approving National Aviation Authority / Country FAA / United States		2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, Airworthiness Approval Tag		3. Form Tracking Number: AVI-ATL-03-11-026	
4. Organization Name and Address: Aviall Services Inc. 574 Airport South Parkway, Suite 200, Atlanta, GA 30349					
6. Item:	7. Description:	8.	Part Number:	9.	Eligibility:*
1	DISK, ROTATING BRAKE		5002260	N/A TSO ARTICLE	10. Quantity: 16EA
			11. Serial/Batch Number: NSN, MARKED F4J3815		
			12. Status / Work: NEW		

13. Remarks:	EXPORT TO:	SOUTH AFRICA
Part Condition:	NEW	

MANUFACTURED BY AIRCRAFT BRAKING SYSTEMS CORP., 1204 MASSILLON ROAD, AKRON, OHIO 44306-4186

NOTE: (BEECH PN 101-8002-13)

14. Certifies the items identified above were manufactured in conformity to:		19.	
<input checked="" type="checkbox"/> Approved design data and are in a condition for safe operation <input type="checkbox"/> Non-approved design data specified in Block 13		<input type="checkbox"/> 14 CFR 43.9 Return to Service <input type="checkbox"/> Other Regulation Sp Block 13 Certifies that unless otherwise specified in Block 13, the work identified in 12 and described in Block 13 was accomplished in accordance with Title Code of Federal Regulations, Part 43 and in respect to that work, the item approved for return to service.	
15. Authorized Signature:	16. Approval/Authorization No.:	20. Authorized Signature:	21. Approval/Certific
	DART-20018-SO	N/A	N/A
17. Name (Typed or Printed):	18. Date (m/d/y):	22. Name (Typed or Printed):	23. Date (m/d/y):
Charles R. Hubbard	November 19, 2003	N/A	N/A

User / Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the part/component/assembly.

Where the user/installer work in accordance with the national regulations of an Airworthiness Authority different than the Airworthiness Authority of the country specified in block 1 it is essential that the user/installer ensures that his/her Airworthiness Authority accepts parts/components/assemblies from the Airworthiness Authority of the country in block 1.

Statements in block 14 and 19 do not constitute an installation certification. In all cases aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

AIRCRAFT BRAKING SYSTEMS

Aircraft Braking Systems 1204 Massillon Rd Avon OH 44306-4136 USA

Packing list

Page 1 of

Ship to: 1072
AVIAL CENTRAL WAREHOUSE
2750 REGENT BOULEVARD
2750 REGENT BLVD
DFW AIRPORT, Texas 75261
USA

Sold to: 340
AVIAL SERVICES
ATTN: PURCHASING
PO BOX 619048
DALLAS, Texas 75261-9048
USA

Delivery no: 80253486

Group no: 2000120900

Customer P.O.: 259743



All items listed on this document are
subject to the provisions of ABSC G
GN-1 as to the use of unauthorized

Delivery no: 80253486

Manufacturer Part Number Customer Part Number	Description	PO line Item	Qty Shipped	Unit of Measure	Condition	Cure/Expir Date
5002260	DISK ROT PN 101-8002-13(1/10) ABSC FAA PAH# PT2064CE-D(17SO)	000004	50	EA		

ENTERED

NOV 07 2003

418

Date Shipped: 31-Oct-2003
Method of Shipment: ROADWAY EXPRESS
Waybill No.: 2000120900
Number of Pallets: 1 Number of Crates:
Number of Boxes:
Total Weight: 144 LB

Dimensions:
28X28X15

PO 0259743

LOT AV1103618867

(TRA) TRAC #:80253486



PART OR MATERIAL CERTIFICATION FORM

AIRCRAFT BRAKING SYSTEMS CORPORATION

2. Seller's Name

4. Address

Aircraft Braking Systems Corporation
1204 Massillon Road Akron, Ohio 44306-4186

Phone: (330)796-4400 SITA CODE: CAKLLCR
Fax: (330)796-9805

5. Customer P. O. No.

PO #: 259743
Item: 000004

3. System Shipping Ref No.

03-80253486-40

6. Item	7. Description	8. Part Number*	9. Eligibility**	10. Qty	11. Ser./Batch No.	12. Status/Wo
1	DISK, ROTATING, BRAKE	5002260	*See Below	50 EA	N/A	New

13A. Remarks

*8. All part mfg./insp. by ABSC

**9. Eligibility (Consult ABSC manuals & service bulletins for additional eligibility)

TSO Article, N/A
TSO-C28b to 5002256

13B. Obtained from:	13C. Last Cert. Agency - ABSC Certification
N/A ABSC Manufactured/Inspected.	
14. New - Unused Parts/Material Certification: The following signature attests that the part(s) or material(s) identified above was (were) manufactured by an FAA approval holder and inspected in accordance with established industry or U.S. specifications and/or the airworthiness regulations of the country of origin.	18. Used, Repaired or Overhauled Part Certification: The following signature attests that the documentation specified above or attached is accurate with regard to the item(s) described. Repaired parts conform and were inspected in accordance with the Code of Federal Regulations 14 and were approved for return to service.

PAH NO. PT2064CE-D(TSO)

15. Authorized Signature:

Mark A. Matheny

19. Authorized Signature:

16. Name (Typed or Printed)

MARK A. MATHENY

17. Date

23 Oct 03

20. Name (Typed or Printed)

21. Date

CERTIFIED
BELOW

REPAIRS, ADDENDUMS, ETC.

(4)

PHASE 2 INSPECTION CARRIED OUT. REFER CRMA NO: 1800.

AIRCFT UNLTD
TWO
AMO-104

18/12/03

*NOTE—A Signature (the date of which must be added) in this column (5) will be taken as a certificate that, in carrying out the overhaul, repair, modification, or replacement to which it relates, all the conditions and requirements of the Air Navigation Regulations for the time being in force applicable thereto under the Aviation Act, 1954, as amended by any other enactment have been complied with.

AIRCRAFT NATIONALITY AND REGISTRATION MARKS

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		LANDINGS	DETAILS OF MODIFICATION
	Hrs.	Mins.	Hrs.	Mins.	Hrs.	Mins.		
Brought Forward								
19.12.2003	1	50	1	50	7150	06	8203	
					7151	56	8204	
3								
30.1.2004	2	00	3	50	7153	56	8205	
11.2.2004	6	00	9	50	7159	56	8212	
12.2.2004	4	42	14	32	7164	38	8217	
13.2.2004	3	30	18	02	7168	08	8221	
15.2.2004	2	00	20	02	7170	08	8223	
16.2.2004	1	06	21	08	7171	14	8224	
18.2.2004	1	12	22	20	7172	26	8225	
12.2.2004	3	18	25	38	7176	44	8229	
23.2.2004	2	30	28	08	7179	14	8233	
24.2.2004	2	00	30	08	7181	14	8235	
29.2.2004	2	06	32	14	7183	20	8237	
Carried Forward								

AIRCRAFT NATIONALITY AND REGISTRATION MARKS

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		SPRINGS	DETAILS OF MODIFICATIONS,
	Hrs.	Mins.	Hrs.	Mins.	Hrs.	Mins.		
Brought Forward								
3 MARCH 04	6	54	32	14	7183	20	8237	
4 " " "	1	06	39	08	7190	14	8243	
5 " " "	1	12	40	14	7191	20	8244	
7 " " "	2	30	41	26	7192	32	8245	
9 " " "	2	54	43	56	7195	02	8247	
12 " " "	2	36	46	50	7197	56	8249	
13 " " "	2	36	49	26	7200	32	8251	
19 " " "	2	36	52	02	7203	08	8253	
20 " " "	3	42	55	44	7206	50	8256	
23 " " "	3	54	59	38	7210	44	8262	
8 APRIL 04	3	00	62	38	7213	44	8266	
10 " " "	2	18	64	56	7216	02	8268	
13 " " "	5	00	69	56	7221	02	8270	
16 " " "	3	36	73	32	7224	38	8274	
Carried Forward	3	00	76	32	7227	38	8275	

AIRCRAFT NATIONALITY AND REGISTRATION MARKS

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		DETAILS OF MODIFICATIONS,
	Hrs.	MINS.	Hrs.	MINS.	Hrs.	MINS.	
Brought Forward			76	32	7227	38	LANDINGS
25 APRIL 04	2	06	78	38	7229	44	
29 " " "	-	54	79	32	7230	38	
2 MAY 2004	1	48	81	20	7232	26	LANDINGS
3 " " "	1	36	81	56	7234	02	
4 " " "	-	48	82	44	7234	50	
5 " " "	2	24	85	08	7237	14	
7 " " "	3	12	88	20	7240	26	
10 " " "	2	48	91	08	7243	14	LANDINGS
12 " " "	-	36	91	44	7243	50	
15 " " "	-	54	92	38	7244	44	
16 " " "	1	18	93	56	7246	02	
18 " " "	2	30	96	26	7248	32	
21 " " "	-	42	97	08	7249	14	LANDINGS
25 " " "	-	36	97	44	7249	50	
Carried Forward							

REPAIRS,
(4)

AIRCRAFT NATIONALITY AND REGISTRATION MARKS

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		Landing	DETAILS OF MODIFICATIONS,
	HRS.	MINS.	HRS.	MINS.	HRS.	MINS.		
Brought Forward								
6 MAY 2004	8	18	97	44	7249	50	8297	
0 " "	4	24	106	02	7258	08	8302	
0 " "	2	00	110	26	7262	32	8308	
0 " "	2	00	112	26	7264	32	8312	
1 " "								
1 " JUNE 2004	2	12	114	38	7266	44	8314	
2 " 06.2004	1	48	116	26	7268	32	8317	
4 " 06.2004	1	30	117	56	7270	02	8319	
13 " 06.2004	4	18	122	14	7274	20	8320	
14 " 06.2004	5	42	127	56	7280	08	8322	
16 " 06.2004	3	24	130	20	7283	26	8325	
18 " 06.2004	0	12	131	32	7283	38	8326	
17 " 06.2004	2	12	133	44	7285	50	8329	
18 " 06.2004	2	06	135	50	7287	56	8331	
25 " 07.2004	1	06	136	56	7289	02	8332	
26 " 07.2004	1	12	138	08	7290	14	8333	
Brought Forward			138	08	7290	14	8333	

PURGE 3 INSPECTION CAPRED



Buck Rogers Aviation co
P.O. BOX 1368, HALFWAY HOUSE 1685

2550

CUSTOMER DETAILS

Customer: **9J-MED Aircraft Unlimited**
Address: **Hanger 24 Rand Airport 1419**
Customer Order No.: **75-30/2010** Works Order No.:

PROCESS DETAILS

Description of Process: **Fuel Cell Repair**
Associated Documents: **N/A**
Data Pack Ref. No.: **SS05**
Part Number: **101-381002-8RH**

Serial Number: **81-25951**

REMARKS:

**Cell Rerubberised.
One leaking nipple
reinforced**

CERTIFICATION

I hereby certify that the item/s mentioned above has/have been processed in accordance with the requirements prescribed in the Manual of Procedure submitted for the issue of Certificate of Approval No. MP52. Repairs carried out I.a.w. Engineered Fabrics Manuals AP-368 and AP-472.

Sign

Responsible Person or Alternate

**BUCK ROGERS
AVIATION CC
CAA APP. NO.
MP 52
INSPECTOR NO. 1**

Date

6/1/04

AMO 104

CERTIFICATE RELATING TO MAINTENANCE OF AN AIRCRAFT
CRMA No: 2010

Aircraft Type: Beechcraft King Air B200	Registration Number: 9J-MED	Serial Number: BB1046
---	---------------------------------------	---------------------------------

AIRFRAME

1. Phase 3 inspection carried out as per Beechcraft Maintenance Manual No: 101-590010-19B, Revision B7, Revised 31-Jan-03.
2. Hours correctly recalculated as requested from 18-Dec-03 to 16-Jun-04.
3. Starboard fuel tank removed for repairs and fitted thereafter. Refer CRMA No: 2550.
4. Both inertial separator systems inspected. LH actuator system rebushed, RH actuator system found satis. Systems adjustment and chain tension carried out.
5. Oxygen cylinder/regulator removed for overhaul and refitted thereafter. Refer CRMA No: 106459.
6. Oxygen bottle removed for hydrostatic test and refitted thereafter.
7. New seat belt manufactured and fitted to toilet station.
8. New floor panels manufactured and fitted. Holes and fasteners fitted.
9. Starboard wing aft upper wing bolt access panel painted.
10. Landing gear squat switch replaced. System inspected IAW above Maintenance Manual and found satis.
11. WX radar system removed for repairs and refitted thereafter.
12. Windshield defrost mechanisms dismantled, cable lubricated, refitted and rigged.
13. New RH current limiter fitted.
14. Sunvisor hooks manufactured and fitted.
15. Co-pilot AH RMI removed for repairs and refitted thereafter.
16. Radar wave guide & pressurization tubes cleaned.
17. Toilet bucket removed to provide more storage.
18. New port Nav light fitted.
19. New strobe light lamp fitted.
20. 2x new clamps fitted to rotating beacon.
21. New port landing light lamp fitted.
22. Stall warning system power supply removed, stall warning system removed for repairs and refitted thereafter.
23. 2 new airconditioning flexible hoses fitted at RH engine.
24. New cup and bearing fitted to #1 main wheel outboard.
25. New cup and bearing fitted to #2 main wheel inboard.
26. Shimmy dampener removed for overhaul and refitted thereafter.
27. Air conditioning filters replaced.
28. Both co-pilots master cylinder overhauled.
29. New nose gear tyre fitted.
30. New #1 main wheel fitted.
31. Cabin safety valve diaphragm repositioned.
32. Both LH fuel tank vent check valves removed, cleaned and refitted.
33. New upper door safety springs fitted.
34. Elevator trim cable tension re-rigged.
35. Elevator cable tension re-rigged.
36. New rivnuts fitted to cabin door upholstery.
37. New rotating beacon lamp fitted and unit sealed.

SCHEDULED INSPECTIONS

The following scheduled inspections were carried out:

- 400hr lubrication.
- 1 yr ELT operational check.
- Annual lubrication.
- 1 yr Wing c/section upper surface bonded panel inspection.
- 3 yr Oxygen cylinder hydrostatic test.
- 3 yr Oxygen regulators overhaul.
- 1 yr Wing fuel bay upper skins inspection.
- 2 yr Passenger oxygen shut off valve o-ring replacement.
- 600 hr Pneumatic flow control unit solenoid valve filter replacement, P/N: 1520161.
- 800 hr Pneumatic flow control unit solenoid valve filter clean, P/N: 1520173.

AIRWORTHINESS DIRECTIVES – One Time

- AD 87-04-24 – Elevator trim cable system (SB 2028 R3) – Found Embodied
- AD 89-22-14 – Aileron (SB 2256) – Carried Out
- AD 92-10-12 – Aft cowling doors (SB 2416 R1) – Found Embodied
- AD 92-15-01 – Failure of truss / firewall bolts (SB 2432) – Carried Out
- AD 96-09-13 – Icing conditions – Carried Out
- AD 97-06-06 – Pilot or co-pilot chair – Carried Out
- AD 97-25-01 – Outflow / Safety valve – N/A due to P/N
- AD 97-25-03 – AFM limitations section – Carried Out
- AD 98-15-13 – Rubber fuel hose (SB 2718 R1) – Carried Out
- AD 98-20-38 – AFM revision – Carried Out
- AD 2002-23-11 – Balance weight attachment screws (SB 3187 R1) – Carried Out
- AD 2003-13-16 – Aft pressure bulkhead of fuselage (SB 3513 R1) – Carried Out

AIRWORTHINESS DIRECTIVES – Recurring

- AD 87-22-01 R1 – Nose landing gear (SB 2102 R1) – Improved type fork found fitted – AD Fully Complied With – No further action necessary.
- AD 89-19-04 R1 – Wing fuel bay skin panels (SB 2040 R3) – 1 yr/600 hr inspection carried out.
- AD 93-25-07 – Fuselage stringers (SB 2472 R1) – 600 hr inspection carried out.

SERVICE BULLETINS – One Time

- SB 2266 – Hartzell Service Bulletin 159 – Found Embodied.
- SB 2422 – Rudder torque shaft – Found Embodied.
- 2681 – Environmental / air conditioning relief valve installation – Found Embodied.
- SB 3038 – Rudder torque shaft support bearing retention nut inspection – Carried Out
- SB 3433 – MLG retract actuator nut assembly – N/A by S/N

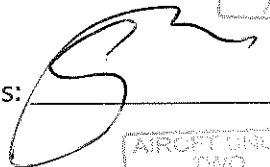
SERVICE BULLETINS – Recurring

- SB 2156 – Fuel boost pumps inspection – Carried Out
- SB 2212 – Flight control cable tensions inspection – Carried Out
- SB 2213 – Pilot & co-pilot seat track wear – 100 hr inspection Carried Out
- SB 2254 – Sealing of horizontal stabilizer & dorsal fin for moisture protection and installation of bonding jumper – SB found fully embodied – no further action required.

Records of all work carried out held under Job Card 2010.


I hereby certify that in carrying out the foregoing specified maintenance all the requirements prescribed in the Civil Aviation Regulations, 1997, as amended, which are applicable hereto, have been complied with.

1st Insp C/O on all controls:



AIRCFT UNLTD
TWO
AMO 104

2nd Insp C/O on all controls:


G. Burrill

AIRCFT UNLTD
THREE
AMO 104

License or Approval No: (Stamp): -----

Date: 16-Jul-04

Airframe Total Time: 7284.92 hrs

Airframe Total Cycles: 8328

PCN011652

1. Approving National Aviation Authority/Country: FAA/United States		2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG			3. Form Tracking Number: 0070228406-00001	
4. Organization Name and Address		RAYTHEON AIRCRAFT PARTS INVENTORY AND DISTRIBUTION (RAPID) DALLAS DISTRIBUTION CENTER 801 INDUSTRIAL BLVD, SUITE 100 GRAPEVINE, TX 76051				5. Work Order/Contract/ Invoice Number 0000703583-000610
6. Item	7. Description	8. Part Number	9. Eligibility *	10. Quantity	11. Serial/Batch Number	12. Status/Work
0001	BELT (SEE MD04 NOTES)	115-389009-3	300	2.000	N/A	NEW

13. Remarks
EXPORT TO South Africa

May be eligible for installation on models not listed in block 9, installer is responsible to verify eligibility for additional model:

14. Certifies the items identified above were manufactured in conformity to: <input checked="" type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 13.		19. <input type="checkbox"/> 14 CFR 43.9 Return to Service <input type="checkbox"/> Other regulation specified in Block 13 Certifies that unless otherwise specified in Block 13, the work identified in Block 12 and described in Block 13 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.	
15. Authorized Signature: <i>David L. Burke</i>	16. Approval/ Authorization No: DOA230339CE	20. Authorized Signature :	21. Approval/ Certificate No.:
17. Name (Typed or Printed): David Burke	18. Date (m/d/y): 03/11/2004	22. Name (Typed or Printed) :	23. Date (m/d/y):

USER/INSTALLER RESPONSIBILITIES

It is important to understand that the existence of this Document alone does not automatically constitute authority to install the part/component/assembly.

Where the user/installer work in accordance with the national regulations of an Airworthiness Authority different than the Airworthiness Authority of the country specified in block 1 it is essential that the user/installer ensure that his/her Airworthiness Authority accepts parts/components/assemblies from the Airworthiness Authority of the country specified in block 1.

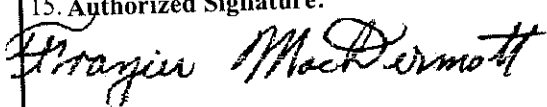

Statements in block 14 and 19 do not constitute installation certification. In all cases aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

Authority/Country: FAA/United States	AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG	Form Tracking Number: 0070175902-00001
4. Organization Name and Address RAYTHEON AIRCRAFT PARTS INVENTORY AND DISTRIBUTION (RAPID) DALLAS DISTRIBUTION CENTER 801 INDUSTRIAL BLVD, SUITE 100 GRAPEVINE, TX 76051	5. Work Order/Contract/ Invoice Number 0000652459-000970	

6. Item	7. Description	8. Part Number	9. Eligibility *	10. Quantity	11. Serial/Batch Number	12. Status/Work
0001	BELT (SEE MD04 NOT ES)	115-389009-3	300	5.000	N/A	NEW

13. Remarks
 THIS IS A REPLACEMENT FAA FORM 8130-3. ISSUED ON (11/05/2003)
 EXPORT TO South Africa

May be eligible for installation on models not listed in block 9, installer is responsible to verify eligibility for additional models

14. Certifies the items identified above were manufactured in conformity to: <input checked="" type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 13.	19. 14 CFR 43.9 Return to Service Other regulation specified in Block 13 Certifies that unless otherwise specified in Block 13, the work identified in Block 12 and described in Block 13 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
15. Authorized Signature: 	16. Approval/ Authorization No: DOA230339CE
17. Name (Typed or Printed): Frazier Macdermott	18. Date (m/d/y): 12/01/2003
20. Authorized Signature : 	21. Approval/ Certificate No.:
22. Name (Typed or Printed) :	23. Date (m/d/y):

USER/INSTALLER RESPONSIBILITIES

It is important to understand that the existence of this Document alone does not automatically constitute authority to install the part/component/assembly.

Where the user/installer work in accordance with the national regulations of an Airworthiness Authority different than the Airworthiness Authority of the country specified in block 1 it is essential that the user/installer ensure that his/her Airworthiness Authority accepts parts/components/assemblies from the Airworthiness Authority of the country specified in block 1.

Statements in block 14 and 19 do not constitute installation certification. In all cases aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

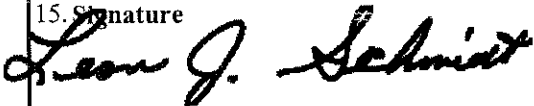
		Airworthiness Approval Tag U.S. Department of Transportation Federal Aviation Administration		0000146601-00112	
4. Organization RAYTHEON AIRCRAFT COMPANY 9709 EAST CENTRAL P O BOX 85 WICHITA, KANSAS 67201-0085				5. Work Order Contract or Invoice Number 560000281-00016	

6. Item	7. Description	8. Part Number	9. Eligibility *	10. Quantity	11. Serial/Batch Number	12. Status/Work
0001	COVER DUCT	101-389008-1	TBV by Installer	1.000	N/A	NEW

13. Remarks
 AIRWORTHINESS APPROVAL - PARTS. THIS FORM IS NOT AN EXPORT APPROVAL

Limited Life parts must be accompanied by maintenance history including total time/total cycles/time since new.

14. New <input checked="" type="checkbox"/> Newly overhauled <input type="checkbox"/> Certifies that the new or newly overhauled part(s) identified above, except as otherwise specified in block 13 was (were) manufactured in accordance with FAA approved design data and airworthiness. NOTE: In case of parts to be exported, the special requirements of the importing country have been met.	19. Return to service in accordance with FAR 43.9 Certifies that the work specified in block 13 (or attached) above was carried out in accordance with FAA airworthiness regulations and in respect to the work performed the part(s) is (are) approved for return to service.
---	---

15. Signature 	16. FAA Authorization No. DOA230339CE	20. Authorized Signature	21. Certificate Number
17. Name (Typed or Printed) LEON SCHMIDT	18. Date: 06-29-2001	22. Name (Typed or Printed)	23. Date:

USER/INSTALLER RESPONSIBILITIES

It is important to understand that the existence of this Document alone does not automatically constitute authority to install the part/component/assembly.

Where the user/installer work in accordance with the national regulations of an Airworthiness Authority different than the Airworthiness Authority of the country specified in block 1 it is essential that the user/installer ensure that his/her Airworthiness Authority accepts parts/components/assemblies from the Airworthiness Authority of the country specified in block 1.

Statements in block 14 and 19 do not constitute installation certification. In all cases aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

The FAA form 8130-3 and JAA Form One are equivalent. Other countries such as Canada also have equivalent acceptable documents.

Phn: 918-234-7771 Fax: 918-234-7771

Invoice No. ITXOBRUL

Invoice Date. 03/29/00

Box ID. 4820536

S
O RAYTHEON AIRCRAFT
L P. O. BOX 44529
D ATLANTA, GA 30336
T
O

S
H RAPID
I HERITAGE BUSINESS PARK BLDG #4
P 801 INDUSTRIAL BLVD
T GRAPEVINE, TX 76051
O

P.O. #:4500085552

SHIP MODE:UPS GROUND

OUR SALE #:STX09PSV

NO. CARTONS:XXXXXX

TERMS:NET 30

F.O.BSHIP POINT

QUANTITY ORDERED	U/M	QUANTITY SHIPPED	BALANCE DUE	PART NO. & DESCRIPTION	UNIT PRICE	TOTAL
750	EA	750	0	MS14145L5 NUT, S/L, CASTELLATED 33-328-277-00 ITEM #4 Lot. 4056051-000 Manf. KAYNAR	1.10000 EA	825.00
T O T A L ----->						825.00 =====



CERTIFICATE OF CONFORMANCE DRAWING

We certify that the material delivered on this order is new and conforms to applicable blueprint and specification requirements. Evidence of such is on file with us and/or our suppliers and is available for examination upon request. TriStar Aerospace, Inc. does not manufacture or produce modification or replacement parts. TriStar certifies that compliance with the requirements of the Fastener Quality Act, Public Law 101-592, 15 United States Code §5401 is maintained.

By: [Signature]
JON A. COHEN
DIRECTOR OF QUALITY

For return of material, shortage, or damage must be made within 7 days after receipt of goods. Our liability is limited to the material or refunding the invoice value of the material sold. Returns must have a return label number issued by TriStar Aerospace, Inc. and must be affixed to all return containers. Returns are subject to a special handling charge if not requested. If a part number/description or our recital of a customer-designated part number/ description is not satisfactory, please contact the TriStar Sales Department within 7 days after receipt of goods for return.

Phn: 918-234-7771 Fax: 918-234-7771

 Invoice No. ITX08RUL
 Invoice Date 03/29/00
 Box ID 4820536

S
O
L
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T
O
 RAYTHEON AIRCRAFT
 P O BOX 44529
 ATLANTA GA 30336

S
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 RAPID
 HERITAGE BUSINESS PARK BLDG #4
 801 INDUSTRIAL BLVD
 GRAPEVINE, TX 76051

P.O. #:4500085552
 OUR SALE #:STX09PSV
 TERMS NET 30

SHIP MODE UPS GROUND
 NO. CARTONS:XXXXXX
 FOB SHIP POINT

QUANTITY ORDERED	U/M	QUANTITY SHIPPED	BALANCE DUE	PART NO. & DESCRIPTION	UNIT PRICE	TOTAL
750	EA	750	0	MS14145L5 NUT, S/L, CASTELLATED 33-328-277-00 ITEM #4 Lot. 4056051-000 Manf. KAYNAR	1.10000 EA	825.00
T O T A L ----->						825.00 =====

33-328-277-00



Warranty for error, shortage, or damage must be made within 7 days after receipt of goods. Our liability is limited to replacing the material or refunding the invoice value of the material sold. Returns must have a return authorization number issued by TriStar Aerospace, Inc. and must be affixed to all return containers. Manufacturer's certifications are subject to a special handling charge if not requested. Use of a part number/description or our recital of a customer-designated part number/manufacturer does not of itself commit TriStar Aerospace, Inc. to supply parts of a specific manufacturer. If the manufacturer named is not satisfactory, please contact the TriStar Sales Department within 7 days after receipt of goods for return.

CERTIFICATE OF CONFORMANCE DRAWING
 We certify that the material delivered on this order is new and conforms to applicable blueprint and specification requirements. Evidence of such is on file with us and/or our suppliers and is available for examination upon request. TriStar Aerospace, Inc. does not manufacture or produce modification or replacement parts. TriStar certifies that compliance with the requirements of the Fastener Quality Act, Public Law 101-592, 15 United States Code §5401 is maintained.

By: *[Signature]*
 JON A. COHEN
 DIRECTOR OF QUALITY

AIRCRAFT TEST REPORT/RELEASE CERTIFICATE

REPORT/CERTIFICATE NO: *101931*

CLIENT: *SKYLAM 31 CC T/A AIRCRAFT UNLIMITED*

ADDRESS: *RAND AIRPORT*

ORDER NO: *75/02/2010*

JOB CARD NO AND DATE : *106459 29.6.2004*

DESCRIPTION OF ITEM FOR TEST, SERVICE REQUIRED AND IDENTIFICATION
MARKS (IF ANY):

AIRCRAFT CYLINDER

ORIGINAL EQUIPMENT MANUFACTURERS DOCUMENT NUMBER AND REVISION
STATUS *SABS 1475 - REVISED 1/1981 26-21-40*

WORK DONE : *HYDROSTATIC TEST, RECHARGED, SERVICED AND CERTIFIED*

TEST/PROCESS PERFORMED BY: *R. VAN DER WESTHUIZEN*

INSPECTED BY: *A. BIRD* DATE : *29.6.2004*

SERIAL NO.	PART NO.	QTY	TYPE EXT	MASS
46096X	DOT-3AA1800	1	OXYGEN	6,5KG

I, hereby certify that the item/s mentioned above has/have been tested/processed in accordance with the requirements prescribed in the Manual of Procedure submitted for the issue of Certificate of Approval No. P4 (1)

DATE : *29.6.2004*

1ST RESPONSIBLE PERSON *[Signature]* *[Signature]*

AIRCRAFT NATIONALITY AND REGISTRATION MARKS

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		DETAILS OF MODIFICATIONS,
	Hrs.	Mins.	Hrs.	Mins.	Hrs.	Mins.	
Brought Forward							
04.08.2004	1	00	138	08	7290	14	8333
05.08.2004	1	12	139	08	7291	14	8334
06.08.2004	1	12	140	20	7292	26	8335
07.08.2004	1	42	141	32	7293	38	8336
12.08.2004	3	24	143	14	7295	20	8337
14.08.2004	1	06	146	38	7298	44	8339
16.08.2004	1	06	147	44	7299	50	8340
20.08.2004	1	06	148	50	7300	56	8341
21.08.2004	1	24	150	14	7302	20	8343
23.08.2004	0	48	151	02	7303	08	8344
24.08.2004	1	06	152	08	7304	14	8345
25.08.2004	1	00	153	08	7305	14	8346
27.08.2004	8	12	161	20	7313	26	8350
04.09.2004	6	48	168	08	7320	14	8354
04.09.2004	2	42	170	50	7322	56	8357
Carried Forward			170	50	7322	56	

AIRCRAFT NATIONALITY AND REGISTRATION MARKS

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		DETAILS OF MODIFICATIONS,
	Hrs.	Mins.	Hrs.	Mins.	Hrs.	Mins.	
Brought Forward							
05.09.2004	1	24	170	50	7322	56	8357
06.09.2004	8	00	172	14	7324	20	8359
10.09.2004	2	00	180	14	7332	20	8363
17.09.2004	2	00	182	14	7334	20	8365
20.09.2004	3	42	184	14	7336	20	8367
23.09.2004	2	30	187	56	7339	02	8371
24.09.2004	2	18	190	26	7341	32	8374
25.09.2004	6	42	192	44	7343	48	8376
28.09.2004	2	36	199	26	7350	30	8380
02.10.2004	7	54	202	02	7353	06	8382
08.10.2004	2	06	209	56	7361	00	8386
09.10.2004	1	06	212	02	7363	06	8388
09.10.2004	1	06	213	08	7364	12	8389
09.10.2004	-	-	210	36	7360	42	Corrected hours
13.10.2004	1	60	211	36	7361	42	8390
18.10.2004	4	48	216	24	7366	30	8393
Carried Forward			216	24	7366	30	

REPAIRS
(4)

Aircraft Technical Services

Lusaka International Airport

COMPASS SWING/ADJUSTMENT FORM

AIRCRAFT TYPE: KECHICRAFT 200
A/C SN: 881046

REGISTRATION: 9J-MED
EQUIPMENT: AIRRAITH

POINTS	COMPASS READING	AIRCRAFT COMPASS READINGS	DEVIATION	DEVIATION CORRECTED FOR "C" OR "B"	AIRCRAFT COMPASS READING	DEVIATION	ALL DEVIATION CORRECTED FOR "A"	FINALLY CORRECTED READING FOR CARD	HDG	CARD
N	080	081	080	+1	-1	-1	-1	080	081	
E	090	088	089	-1		-1	-1	090	091	
S	180	178	175	+3	+1		+1	180	179	
W	270	270						270	270	
NW	315							315	316	
NE	045							045	044	
SE								135	135	
SW								225	225	

"C" = DEV ON N - DEV ON S =

$$1 - (+3)$$

$$= 1 - -3$$

$$= 4$$

"D" = DEV ON E - DEV ON W =

$$1 - 0$$

$$= -1$$

$$= -1$$

"A" = ALGEBRAIC SUM OF DEVIATIONS ON ALL POINTS =

$$+2 - 1 + 2 - 1 + 1$$

$$5 - 2 = 3$$

NO CORR

DATE: 10/12/2004

CERTIFIED:

[Signature]

AUTHORITE: 345 Acc



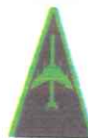
FMS 800 (Rev. 10-1-70)

TECHNICAL SERVICES

DETAILS OF MODIFICATIONS,

DATE (1)	TIME IN AIR (2)		C. OF A. RENEWAL (3)		TIME		
	Hrs.	Mins.	Hrs.	Mins.	Hrs.	Mins.	
Brought Forward							
19.10.2004	3	30	216	24	7366	30	8393
20.10.2004	1	00	220	54	7370	00	8395
27.10.2004	1	42	222	36	7372	42	8397
28.10.2004	3	48	226	24	7376	30	8402
04.11.2004	2	12	228	36	7378	42	8404
06.11.2004	9	12	237	48	7387	54	8409
14.11.2004	6	36	244	24	7394	40	8415
15.11.2004	3	30	247	54	7398	10	8419
17.11.2004	5	00	252	54	7403	10	8422
18.11.2004	4	48	257	42	7407	58	8426
20.11.2004	2	30	260	12	7410	28	8428
21.11.2004	6	12	266	24	7416	40	8432
25.11.2004	1	24	267	48	7418	04	8434
29.11.2004	6	30	274	18	7424	34	8437
19.12.2004	1	24	275	42	7425	58	8439
Carried Forward							

RE



LOG BOOK ENTRY

Beech Craft King Air B200

AIRFRAMES

AIRCRAFT REG: 9J-MED

S/N: BB1046

DATE: 17-01-05

Total Airframe Hours: 7440Hours 54Min

Landings...8452

Engine hours= R.H—7440Hours 54Min

L.H—7440Hours 54Min

COMPLETED CHECK -Phase 4 Inspection as per Maintenance Manual P/N-101-590010-19B

AIRFRAMES:

Maintenance Manual P/N-101-590010-19B

Inspection on airframes-: wings, fuse large,
Tail section, nose section and Landing Gear System.

1-WINGS

- Inspected Fuel probes points of attachment for leaks-satisfactory
- inspected fuel drain points-found blocked-removed and unblocked-satisfactory
- checked lights for broken lenses-satisfactory
- exterior fuel vents -found one blocked -unblocked-satisfactory
- Ant siphon valves removed and checked for damage -satisfactory
- checked general condition of wing including de-icer boots and access panels-satisfactory
- replaced 4 bonding jumper terminals between wing and flaps
- Flap tracks -cleaned and lubricated, checked flaps for full up and down movement
- checked Ailerons for free full up and down movement.

2-FUSELAGE

- Inspected the entire external fuselage for damage, missing screws and rivets, cracked skin, flaking paint
- inspected windows for damage and sealing
- Inspected cabin door- cables, steps, seal, latches for condition and checked door for proper locking.
- Internal fuselage-Removed all cabin seats and floor panels-cleaned the floor and all seat tracks -checked all equipment for cleanliness, security of attachment tightened all loose screws and nuts.
- Replaced aft evaporator filter and checked brushes-satisfactory
- Inspected control cables for freedom of movement-satisfactory
- inspected floor for corrosion and damage
- L/H Cabin Table repaired satisfactory

3-TAIL SECTION

- Inspected empennage for damage, paint blistering, paint flaking, missing screws or rivets, loose items-satisfactory
- checked free full travel of control surfaces (rudder and elevator, including trim tabs)-satisfactory
- cleaned control surface attach points and hinges-lubricated
- Removed tail cone and checked rudder torque shaft support bearing retention nut-cleaned and lubricated and no corrosion found.
- checked rudder and elevator control cables for wear, corrosion and freedom of movement-satisfactory
- Inspected oxygen plumbing for security of attachment
- checked elevator and rudder trim tab free play-satisfactory
- inspected ELT battery for leakage, corrosion and loose leads

4-LANDING GEAR

- Inspected nose and main landing gear attachments, actuators and drive shafts for corrosion, wear and cracks and lubricated the system- satisfactory
- Retraction checks carried out- satisfactory
- Replaced blown out Landing Light P/N GE 4596

The following AD'S were inspected and found incorporated:

- AD 97-06-06-Pilot or co-Pilot chair
- AD 97-25-01-Outflow/Safety valve
- D 97-25-03-AFM limitation section
- AD 87-04-24-Elevator trim cable system
- AD 89-22-14-Aileron
- AD92-10-12-Aft cowling door
- AD 92-15-01 Failure of truss
- AD 96-09-13-Icing conditions
- AD 98-15-13 - Rubber fuel hose
- AD 98-20-38—AFM revision
- AD 2002-23-11—Balance weight attachment screws
- AD 2002-13-16—Aft pressure bulkhead of fuselage

The following recurring AD's were checked:

- AD 87-22-01 R1 - Nose landing gear
- AD 89-19-04 R1 Wing fuel bay skin panels
- AD 93-25-07 - Fuselage stringers

The following SB's were inspected and found incorporated:

- SB 2266 -- Hartzell Service Bulletin 159
- SB 2422 - Rudder torque shaft
- SB 2681 - Environmental/air conditioning relief valve installation
- SB 3038 - Rudder torque shaft support bearing retention nut inspection
- SB 3433 - MLG retract actuator nut assembly

The following recurring SB's were checked:

- SB 2156 - Fuel boost pumps inspection
- SB 2212 - Flight control cable tensions
- inspection SB 2213 - Pilot and Co-pilot seat tract wear
- SB 2254 - Sealing of horizontal stabilizer & dorsal fin for moisture protection and installation of bonding jumper

I hereby certify that in carrying out the
overhaul/Repair/Replacement/Modification/Inspection specified above all the
conditions and requirements Of the Air Navigation Regulations for the time being in
force which are applicable thereto have been complied with.

Signature: *[Signature]*

auth: date: 17/1/05



All airworthiness Directives issued affecting this Airframe to date. have been complied with.
No mandatory Service Bulletins have been issued affecting the Aircraft are pending.
Duplicate Inspection OF the Flight Controls carried out in accordance with Engineers Notice
No 10 satisfactory

1st INSPECTION

SIGN: *[Signature]*

LIC. NO.

DATE: 17/1/05



2nd INSPECTION:

SIGN: *[Signature]*

LIC. NO. 345 Acc

DATE: 17/01/05



AIRCRAFT TECHNICAL SERVICES

LOG BOOK ENTREY

AIRCRAFT TYPE: BEECHCRAFT KING AIR B200

Control surface movements

	UP	DOWN
ELEVATOR:	20Deg	15Deg
ELEVATOR TRIM	3.5Deg	14Deg
AILERONS	24Deg	17Deg
AILERON TRIMS	15Deg	15Deg
FLAPS	UP- 0Deg	APPROACH -14Deg DOWN -33Deg
RUDDER	Left-25Deg	Right-25Deg



AIRCRAFT TECHNICAL SERVICES

LOG BOOK ENTREY

9J-MED

CABLE TENSIONS

	Right	Left
AILERONS	30lbs	30lbs
AILERON TRIMS	10lbs	10lbs
ELEVATOR	40lbs	40lbs
ELEVATOR TRIMS	9lbs	9lbs
RUDDER	80lbs	80lbs

Duplicate Inspection OF the Flight Controls carried out in accordance with Engineers Notice
No 10 satisfactory

1st INSPECTION

SIGN:.....

LIC. NO. ZIA 269-



DATE:

17/1/2005

2nd INSPECTION:

SIGN:.....

LIC. NO.

345 Acx

DATE:

17/01/2005

P.O BOX AP10, CH Lusaka, Zambia. Tell: 097887247

AFT NATIONALITY AND REGISTRATION MARKS

DETAILS OF MODIFICATIONS,

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		
	HRS.	MINS.	HRS.	MINS.	HRS.	MINS.	
Forward							
12.04	3	24	225	42	7425	58	Compass being carried out
12.04	3	00	282	06	7438	22	3844Z.1.A. 345 Acc.
01.05	2	48	284	54	7435	10	4844B
01.05	3	06	288	00	7438	16	18447
01.05	0	48	288	48	7439	04	28449
01.05	2	00	290	48	7441	04	18450
			292	00	7440	54	28452
Corrected hours							
Ranks 25/01/05							
Checked for C 9A.							
St. Maurice							
26/07/2005.							
000	00	7440	55	8452			

*NOTE - A Signature
the conditions and
complied with.

REPAIRS, ADJUSTMENT
(4)

AIRCRAFT NATIONALITY AND REGISTRATION MARKS

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		DETAILS OF MODIFICATIONS,
	Hrs.	Mins.	Hrs.	Mins.	Hrs.	Mins.	
rought Forward			0	00	7440	55	8452
21-01-05	2	06	2	06	7443	01	2 8452
23-01-05	2	06	4	12	7445	07	2 8454
27-01-05	5	42	9	54	7450	49	6 8460
28-01-05	1	18	11	12	7452	07	1 8461
30-01-05	2	30	13	42	7454	37	3 8464
06-02-05	1	12	14	54	7455	49	2 8466
11-02-05	1	35	16	29	7457	24	2 8468
12-02-05	1	06	17	35	7458	30	1 8469
16-02-05	1	48	19	28	7460	18	2 8471
20-02-05	5	54	25	17	7466	12	2 8473
23-02-05	2	18	27	35	7468	30	2 8475
25-02-05	1	42	29	17	7470	12	1 8476
01-03-05	2	36	31	53	7472	48	2 8478
10-03-05	2	42	34	35	7475	30	2 8480
12-03-05	2	30	37	05	7478	00	2 8482
Carried Forward			37	05	7478	00	8482

REPAIRS, A
(4)

NOTE-A
the com
complete

AIRCRAFT NATIONALITY AND REGISTRATION MARKS

DETAILS OF MODIFICATIONS,

DATE (1)	Time in Air (2)		Time Since Last C. of A. RENEWAL (3)		Total Time		
	Hrs.	Mins.	Hrs.	Mins.	Hrs.	Mins.	
Brought Forward							
13-03-05	1	54	37	05	7478	00	8482
14-03-05	01	00	38	59	7479	54	8484
15-03-05	01	54	39	59	7480	54	8485
18-03-05	01	54	41	53	7482	38	8484
19-03-05	01	54	43	47	7484	38	8489
20-03-05	01	18	45	05	7485	50	8491
22-03-05	01	18	46	23	7484	08	8493
24-03-05	01	18	47	35	7488	20	8494
25-03-05	04	54	52	29	7493	14	8496
29-03-05	00	54	53	23	7494	08	8497
08-04-05	08	54	62	17	7503	02	8505
13-04-05	02	36	64	53	7505	38	8508
14-04-05	01	24	66	17	7507	02	8510
15-04-05	06	24	72	41	7513	26	8513
17-04-05	01	12	73	53	7514	38	8515
19-04-05	07	18	81	11	7521	56	8518
Carried Forward			81	11	7621	56	8518

ZAMBIA SKYWAYS LTD

AMO:- 026

SERVICEABLE

95-MED

BATCH NO.	71RE		B850
DESCRIPTION	265K08-1 42780119		B90
P/N	42750615	TSL0	
SHELF LIFE	EXPIRY DATE		
ROTABLE	<input type="checkbox"/>	EXPENDABLE	<input type="checkbox"/>
I hereby certify that in carrying out the Inspection / Repair / O/H / Replacement / Modification specified above, all the conditions and requirements of the AIR NAVIGATION REGULATIONS for the time being in force which are applicable thereto have been complied with.			
SIGN.	AUTH.		DATE
INSTALLED ON	97-MED		2/3/05
A/C	384		
POS	ENGINEER		
DATE	ENGINEER		
REMOVED FROM	A/C		
POS	ENGINEER		
DATE	ENGINEER		

LIMITS REPLACED WITH
265K08-1
RESPECTIVELY.
him and NO1 also replaced

14.03.05.

INSPECTED AND CERTIFIED
AS IN NOTE* BELOW
(5)

*NOTE-A Signature (the date of which must be added) in this column (5) will be taken as a certificate that, in carrying out the overhaul, repair, modification, or replacement to which it relates, all the conditions and requirements of the Air Navigation Regulations for the time being in force applicable thereto under the Aviation Act, 1964, as amended by any other enactment have been complied with.

REPAIRS, ADJUSTMENTS, ETC.

(4)

INSPECTED AND CERTIFIED
AS IN NOTE* BELOW
(5)

NO 1 & NO 4 TIRES WORN TO LIMITS REPLACED WITH
NEW ONES FROM STORES P/N 265K08-1

S/N 42780119 & S/N 42750615 RESPECTIVELY.

NO 2 wheel installed on NO 4 position and NO 1 wheel installed
on NO 2 position.

14-03-05.

Thyler



*NOTE—A Signature (the date of which must be added) in this column (5) will be taken as a certificate that, in carrying out the overhaul, repair, modification, or replacement to which it relates, all the conditions and requirements of the Air Navigation Regulations for the time being in force applicable thereto under the Aviation Act, 1954, as amended by any other enactment have been complied with.

AIRCRAFT NATIONALITY AND REGISTRATION MARKS

DETAILS OF MODIFICATIONS,

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		
	HRS.	MINS.	HRS.	MINS.	HRS.	MINS.	
Brought Forward							
21.04.05	04	48	81	11	7581	56	3 8518
28.04.05	04	48	85	53	7586	38	5 8581
29.04.05	01	06	90	41	7581	88	1 8524
01.05.05	04	18	91	49	7538	34	3 8525
03.05.05	01	00	97	05	7536	58	1 8528
10.05.05	04	48	101	49	7537	58	2 8529
13.05.05	05	30	109	11	7548	04	2 8531
18.05.05	01	54	109	11	7549	58	3 8534
20.05.05	01	48	110	59	7549	46	2 8536
24.05.05	08	24	113	23	7551	46	2 8538
26.05.05	08	30	121	53	7554	10	4 8540
02.06.05	01	06	122	59	7568	40	1 8544
06.06.05	08	00	124	59	7563	46	2 8545
10.06.05	01	06	126	05	7564	46	1 8547
12.06.05	01	36	127	41	7565	52	1 8548
Carried Forward					7567	28	1 8549

TURBO PROP SERVICE CENTRE c.c.

HANGAR NO. 7, LANSERIA AIRPORT
P.O. BOX 348, LANSERIA 1748

0710

Reg. No. CK92/02393/23

CERTIFICATE OF RELEASE TO SERVICE

Nationality and registration marks 95 - MED.

Aircraft type B200.

Serial No BB/1046.

I hereby certify that I am satisfied that the abovementioned aircraft and all its equipment are in every way serviceable for flight and that all maintenance has been carried out in accordance with the Civil Aviation Regulations, 1997, and is Approved Maintenance Schedule..

This certificate lapses at a total of 8701.50 hours of flight time

or on 29 - July 2010. (date), whichever occurs first, unless the aircraft is involved in an accident or becomes unserviceable, in which case the certificate is invalid for the duration of the period.

Signed _____

Licence No _____



*Aircraft maintenance engineer organisation: _____

Date 18/08/2009.

Time 13.00 H.

*Delete whichever is not applicable

1. Approving National Aviation
Authority/Country:
FAA/UNITED STATES

2.

AUTHORIZED RELEASE CERTIFICATE

FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG

3. Form Tracking Number:

31639

4. Organization Name and Address:

THE AVIATION GROUP, INC. 211 PROSPECT PARK, PEACHTREE CITY, GA 30269

5. Work Order/Contract/Invoice
Number:
31639

6. Item:	7. Description:	8. Part Number:	9. Eligibility:*	10. Quantity:	11. Serial/Batch Number:	12. Status/Work:
1	PUMP	RG34720A	VARIOUS	1	B-4713	OVERHAULED

13. Remarks: THE ABOVE APPLIANCE WAS OVERHAULED AND, INSPECTED IN ACCORDANCE WITH CURRENT REGULATIONS OF THE FEDERAL AVIATION ADMINISTRATION AND IS APPROVED FOR RETURN TO SERVICE. Manual: RG34720 Revision#:3 Revision Date:9/91 Full details held on W.O. 31639.) Comments:

CERTIFIES THAT THE WORK SPECIFIED IN BLOCK 12/13 WAS CARRIED OUT IN ACCORDANCE WITH EASA PART 145, AND WITH RESPECT TO THAT WORK, THE COMPONENT IS CONSIDERED READY FOR RELEASE TO SERVICE UNDER EASA APPROVAL NUMBER: EASA.145. 4208.
(*INSTALLER MUST CROSS CHECK ELIGIBILITY WITH APPLICABLE TECHNICAL DATA.)

LIFE LIMIT PARTS MUST BE ACCOMPANIED BY MAINTENANCE HISTORY INCLUDING TOTAL TIME/TOTAL CYCLES

14. Certifies the items identified above were manufactured in conformity to:

- ☐ Approved design data and are in a condition for safe operation.
☐ Non-approved design data specified in Block 13.

19. ☒ 14 CFR 43.9 Return to Service

☒ Other regulation specified in Block 13

Certifies that unless otherwise specified in block 13, the work identified in Block 12 and described in Block 13 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.

15. Authorized Signature:	16. Approval/Authorization No.:	20. Authorized Signature: <i>LR Hunt</i>	21. Approval/Certificate No.: VGOR728L
17. Name (Typed or Printed):	18. Date (m/d/y):	22. Name (Typed or Printed): Lawrence R. Hunt	23. Date (mm/dd/yyyy): JUN/30/2009

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the part/component/assembly.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts parts/components/assemblies from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 14 and 19 do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

THE FAA form 8130-3 and JAA Form One are equivalent. Other countries such as Canada also have equivalent acceptable documents.



THE AVIATION GROUP, INC.
211 Prospect Park
Peachtree City, GA 30269
CRS# VGOR728L

WORK ORDER

1. WORK ORDER NUMBER	2. PO NUMBER	3. CUSTOMER
31639	SAS10490	SPECIALIZED A/C
4. DROP SHIP TO	5. WARRANTY	6. WARR. ORIG. DATE/TECH
	<input type="checkbox"/>	

7. COMMENTS: FROM STOCK AOG AOG

8. UNIT TYPE	9. UNIT MFG NAME	10. MFG/VENDOR PART #	11. OEM PART #
PUMP	ROMEC	RG34720A	
12. UNIT S/N	13. HOURS/CYCLES	14. AIRCRAFT MODEL #	15. AIRCRAFT S/N
B-4713			
16. AIRCRAFT REG. #	17. DATE RECEIVED	18. DATE NEEDED BY	19. DATE COMPLETED
	6/30/2009		6/30/2009
20. DISPOSITION	21. ACTION TAKEN	22. ACTUATOR	23. ACTUATOR OTHER
OVERHAUL	OVERHAULED		
24. # OF ITEMS ON PO	25. CORE DUE		
1	YES <input checked="" type="checkbox"/> NO <input type="checkbox"/>		

26. COMPONENT DATA					
PART #	DESCRIPTION	QTY	PART #	DESCRIPTION	QTY
MS29513-011	ORING	1	M83248/1-114	ORING	1
M83248/1/112	ORING	1	RA5326	DIAPHRAGM	2
RA8987	GASKET	1	RA8297	SPRING	1
PARTS ABOVE AND BEYOND STANDARD OVERHAUL					

27. CUSTOMER COMPLAINT/REPAIR INSTRUCTIONS: DUE OVERHAUL

28. INCOMING INSPECTION: NHD INITIALS: LRH

29. PREASSEMBLY INSPECTION: ASSEMBLED USING SERVICEABLE PARTS. NEW PARTS IN #26

30. FINAL ASSEMBLY & TEST: OVERHAULED AND TESTED PER MANUAL LISTED BELOW.

31. MANUAL: RG34720 MANUFACTURER: ROMEC REVISION #: 3 REVISION DATE: 9/91

32. MOD STATUS IN: MOD STATUS OUT:

33. CORROSIVE TREATMENT: BEAD BLAST ☐ CAD PLATING ☐ ALODINE ☐ ZINC CHROMATE ☐ PAINT ☐

THIS ARTICLE IS CERTIFIED AIRWORTHY

34. SIGNATURE (MECHANIC/REPAIR MAN) DRH

35. Name: Larry Hunt

36. SIGNATURE (INSPECTOR) [Signature]

37. SHIPPING INSPECTOR _____ DATE: ____/____/____

IN-HOUSE INFORMATION

38. DROP SHIP: YES ☐ NO ☒

39. SHIPPING METHOD:

UPS RED

40. WEIGHT:

2.00 LBS.

TAG101



CRS# VGOR728L

**The Aviation Group, Inc.
Overhauled Component
Warranty Information**

IMPORTANT INFORMATION

1. Approving National Aviation
Authority/Country:

2.

3. Form Tracking Number:

FAA/UNITED STATES

AUTHORIZED RELEASE CERTIFICATE

FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG

30992

4. Organization Name and Address:

THE AVIATION GROUP, INC. 211 PROSPECT PARK, PEACHTREE CITY, GA 30269

5. Work Order/Contract/Invoice
Number:
30992

6. Item:	7. Description:	8. Part Number:	9. Eligibility:*	10. Quantity:	11. Serial/Batch Number:	12. Status/Work:
1	PUMP	RG34720A	VARIOUS	1	B-9714	OVERHAULED

13. Remarks: THE ABOVE APPLIANCE WAS OVERHAULED AND, INSPECTED IN ACCORDANCE WITH CURRENT REGULATIONS OF THE FEDERAL AVIATION ADMINISTRATION AND IS APPROVED FOR RETURN TO SERVICE. Manual: RG34720 Revision#:3 Revision Date:9/91 Full details held on W.O. 30992.) Comments:

CERTIFIES THAT THE WORK SPECIFIED IN BLOCK 12/13 WAS CARRIED OUT IN ACCORDANCE WITH EASA PART 145, AND WITH RESPECT TO THAT WORK, THE COMPONENT IS CONSIDERED READY FOR RELEASE TO SERVICE UNDER EASA APPROVAL NUMBER: EASA.145. 4208.
(*INSTALLER MUST CROSS CHECK ELIGIBILITY WITH APPLICABLE TECHNICAL DATA.)

LIFE LIMIT PARTS MUST BE ACCOMPANIED BY MAINTENANCE HISTORY INCLUDING TOTAL TIME/TOTAL CYCLES

14. Certifies the items identified above were manufactured in conformity to:

19. ☒ 14 CFR 43.9 Return to Service

☒ Other regulation specified in Block 13

- ☐ Approved design data and are in a condition for safe operation.
☐ Non-approved design data specified in Block 13.

Certifies that unless otherwise specified in block 13, the work identified in Block 12 and described in Block 13 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.

15. Authorized Signature:

16. Approval/Authorization No.:

20. Authorized Signature:

21. Approval/Certificate No.:

VGOR728L

17. Name (Typed or Printed):

18. Date (m/d/y):

22. Name (Typed or Printed):

23. Date (mm/dd/yyyy):

Lawrence R. Hunt

JUN/30/2009

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the part/component/assembly.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts parts/components/assemblies from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 14 and 19 do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

THE FAA form 8130-3 and JAA Form One are equivalent. Other countries such as Canada also have equivalent acceptable documents.

FAA Form 8130-3 (6-01)

*Installer must cross-check eligibility with applicable technical data.

NSN: 0052-00-012-9005



THE AVIATION GROUP, INC.
211 Prospect Park
Peachtree City, GA 30269
CRS# VGOR728L

WORK ORDER

1. WORK ORDER NUMBER 30992	2. PO NUMBER SAS10490	3. CUSTOMER SPECIALIZED A/C
4. DROP SHIP TO	5. WARRANTY <input type="checkbox"/>	6. WARR. ORIG. DATE/TECH

7. COMMENTS: FROM STOCK AOG AOG

8. UNIT TYPE PUMP	9. UNIT MFG NAME ROMEC	10. MFG/VENDOR PART # RG34720A	11. OEM PART #
12. UNIT S/N B-9714	13. HOURS/CYCLES	14. AIRCRAFT MODEL #	15. AIRCRAFT S/N
16. AIRCRAFT REG. #	17. DATE RECEIVED 6/30/2009	18. DATE NEEDED BY	19. DATE COMPLETED 6/30/2009
20. DISPOSITION OVERHAUL	21. ACTION TAKEN OVERHAULED	22. ACTUATOR	23. ACTUATOR OTHER 1
24. # OF ITEMS ON PO 1	25. CORE DUE YES <input checked="" type="checkbox"/> NO <input type="checkbox"/>		

26. COMPONENT DATA					
PART #	DESCRIPTION	QTY	PART #	DESCRIPTION	QTY
MS29513-011	ORING	1	M83248/1/114	ORING	1
M83248/1/112	ORING	1	RA5326	DIAPHRAGM	2
RA8987	GASKET	1	RA8297	SPRING	1
PARTS ABOVE AND BEYOND STANDARD OVERHAUL					

27. CUSTOMER COMPLAINT/REPAIR INSTRUCTIONS: DUE OVERHAUL

28. INCOMING INSPECTION: NHD INITIALS: MM

29. PREASSEMBLY INSPECTION: ASSEMBLED USING SERVICEABLE PARTS. NEW PARTS IN #26

30. FINAL ASSEMBLY & TEST: OVERHAULED AND TESTED PER MANUAL LISTED BELOW.

31. MANUAL: RG34720 MANUFACTURER: ROMEC REVISION #: 3 REVISION DATE: 9/91

32. MOD STATUS IN: MOD STATUS OUT:

33. CORROSIVE TREATMENT: BEAD BLAST ☐ CAD PLATING ☐ ALODINE ☐ ZINC CHROMATE ☐ PAINT ☐

THIS ARTICLE IS CERTIFIED AIRWORTHY

34. SIGNATURE (MECHANIC/REPAIR MAN) [Signature]

35. Name: Mark Meenan

36. SIGNATURE (INSPECTOR) [Signature]

37. SHIPPING INSPECTOR _____ DATE: ____/____/____

IN-HOUSE INFORMATION

38. DROP SHIP: YES ☐ NO ☐

39. SHIPPING METHOD:

40. WEIGHT:
4.00 LBS.

1. Approving National Aviation Authority/Country: FAA/UNITED STATES		2. AUTHORIZED RELEASE CERTIFICATE FAA FORM 8130-3, AIRWORTHINESS APPROVAL TAG			3. Form Tracking Number: ALS-9559	
4. Organization Name and Address: Kelly Aerospace Turbine Rotables 3414 West 29th Street South Wichita KS 67217				5. Work Order/Contract/Invoice Number: ALS-9794		
6. Item: 1	7. Description: BRACE	8. Part Number: 101-820024-5	9. Eligibility: N/A	10. Quantity: 1	11. Serial/Batch Number: AJ930105 ALS-9559	
12. Status/Work: OVERHAULED						
13. Remarks: SEE W/O. OVERHAULED & CERTIFIED IAW CMM 101-590097-13, REV.B2, 11/01/08 FULL DETAILS HELD ON WORK ORDER						
<p>"Certifies that the work specified in block 12/13 was carried out in accordance with EASA part - 145 and with respect to the work the aircraft component is considered ready for release to service under EASA Approval Certificate Number: EASA.145.4033"</p>						
14. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 13.			19. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in block 13 Certifies that unless otherwise specified in Block 13, the work identified in block 12 and described in Block 13 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
15. Authorized Signature: N/A		16. Approval/Authorization No.: N/A		20. Authorized Signature: <i>William E Briggs</i>		
17. Name (Typed or Printed): N/A		18. Date (m/d/y): N/A		21. Approval/Certificate No.: NV2R045L		
				22. Name (Typed or Printed): <i>William E Briggs</i>		
				23. Date (m/d/y): <i>May 21, 2009</i>		
<p align="center">User/Installer Responsibilities</p> <p>It is important to understand that the existence of this document alone does not automatically constitute authority to install the part/component/assembly. Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts parts/components/assemblies from the airworthiness authority of the country specified in Block 1</p> <p>Statements in Blocks 14 and 19 do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.</p>						
<p>FAA Form 8130-3 (8-04) *Installer must cross-check eligibility with applicable technical data. NSN: 0052-00-012-9005</p>						

Cust...: 86300-QUEST AVIATION, INC.

Part...: 101-820024-5

Repr: 0



Desc...: BRACE

AE W/O.: ALS-9794

AE S/N: ALS-9559



Mfg S/N: AJ930105



P/O....: 825822



**KELLY
AEROSPACE**
Turbine Rotables

3414 West 29th Street South
Wichita, Kansas 67217

FAA CERTIFIED REPAIR STATION NV2R045L



WARRANTY APPLICATION

3414 West 29th Street South
Wichita, KS 67217
(316) 943-6100 • Fax: (316) 943-2917
FAA Certified Repair Station NV2R045L

ALS-9794

WORK ORDER NO.:



LIMITED WARRANTY

Turbine Rotables warrants each product to be free from defects in material and workmanship under normal use and service; Turbine Rotables obligation under this warranty will be limited to repairing or replacing at its option any defective parts which fail within the time limit specified by this warranty. In the event a product requires warranty service, the product shall be returned to Turbine Rotables transportation and handling charges prepaid; along with a statement briefly describing the malfunction or reason for returning such product. Turbine Rotables will at its option repair or replace the unit at the nearest Turbine Rotables Dealer or Distributor. THIS WARRANTY DOES NOT INCLUDE ANY LABOR CHARGES. THERE IS NO IMPLIED WARRANTY OF MERCHANTABILITY, IMPLIED WARRANTY OF FITNESS FOR A PARTICULAR PURPOSE OR ANY OTHER WARRANTIES; EXPRESS OR IMPLIED, PROVIDED WITH THE WRITTEN WARRANTY SET OUT ABOVE.

A/E SN: ALS-9559

MFG SN: AJ930105

101-820024-5

PART NUMBER
UPON INSTALLATION ONLY

SERIAL NO.

WARRANTY PERIOD

Customer has 18 months from date of
certification to install unit and 5 days after
installation to register warranty.

Return this copy to KELLY AEROSPACE TURBINE ROTABLES



WARRANTY APPLICATION

3414 West 29th Street South
Wichita, KS 67217
(316) 943-6100 • Fax: (316) 943-2917
FAA Certified Repair Station NV2R045L

ALS-9794

WORK ORDER NO.:



LIMITED WARRANTY

Turbine Rotables warrants each product to be free from defects in material and workmanship under normal use and service; Turbine Rotables obligation under this warranty will be limited to repairing or replacing at its option any defective parts which fail within the time limit specified by this warranty. In the event a product requires warranty service, the product shall be returned to Turbine Rotables transportation and handling charges prepaid; along with a statement briefly describing the malfunction or reason for returning such product. Turbine Rotables will at its option repair or replace the unit at the nearest Turbine Rotables Dealer or Distributor. THIS WARRANTY DOES NOT INCLUDE ANY LABOR CHARGES. THERE IS NO IMPLIED WARRANTY OF MERCHANTABILITY, IMPLIED WARRANTY OF FITNESS FOR A PARTICULAR PURPOSE OR ANY OTHER WARRANTIES; EXPRESS OR IMPLIED, PROVIDED WITH THE WRITTEN WARRANTY SET OUT ABOVE.

A/E SN: ALS-9559

MFG SN: AJ930105

101-820024-5

PART NUMBER
UPON INSTALLATION ONLY

SERIAL NO.

WARRANTY PERIOD

Customer has 18 months from date of
certification to install unit and 5 days after
installation to register warranty.

RETAIN THIS COPY

1. Approving National Aviation Authority Country FAA / UNITED STATES		2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG				3. System Tracking Ref. No. 33013	
4. Organization Name and Address Rocky Mountain Propellers, Inc., 2865 Airport Drive, Erie, CO 80516 USA FAA FR6R545N - EASA.145.4827						5. Work Order, Contract, or Invoice Number: 33013	
6. Item:	7. Description:	8. Part Number:	9. Eligibility:	10. Qty:	11. Serial / Batch Number:	12. Status / Work:	
1	Woodward Governor	210638	N / A	1	2240134 F	Overhauled	
13. Remarks: Overhauled governor in accordance with Woodward Service Manual #33048G, Rev. 9. Complied with Service Bulletins thru 33580M. No AD's currently apply. Overhaul due 6 years or 6500 hours if unit is operated 120 hours per year continuously.							
<p>A complete Description of work performed is on file at the above referenced organization under the work order and system tracking reference number indicated in blocks 3 and 5. Notice: An Airworthiness Directive may apply to the article(s) described hereon. The installer is responsible for ensuring complete compliance with any applicable Airworthiness Directives. *Certifies that the work specified in block 12 / 13 was carried out in accordance with FAR Part-145 and EASA Part-145 and in respect to that work the component is considered ready for release to service under FAA FR6R545N and EASA Approval Number: EASA.145.4827"</p>							
14. Certifies the items identified above were manufactured in conformity to:				19.			
<input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 13.				<input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input type="checkbox"/> Other regulation specified in Block 13 Certifies that unless otherwise specified in Block 13, the work identified in Block 12 and described in Block 13 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
15. Authorized Signature:		16. Approval / Authorization No.:		20. Authorized Signature:		21. Approval / Certificate No.:	
				<i>[Signature]</i>		FR6R545N	
17. Name (Typed or Printed):		18. Date (m/d/y):		22. Name (Typed or Printed):		23. Date (m/d/y):	
				Dave Hampel		Feb 16, 2009	
User / Installer Responsibilities							
<p>It is important to understand that the existence of this document alone does not automatically constitute authority to install the part / component / assembly. Where the user / installer performs work in accordance with the national regulations of an Airworthiness Authority different than the airworthiness authority of the country specified in block 1, it is essential that the user / installer ensures that his / her airworthiness authority accepts parts / components / assemblies from the airworthiness authority of the country specified in block 1. Statements in block 14 and 19 do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user / installer before the aircraft may be flown.</p>							

REV. 4 02-23-00

SERVICE MANUAL **33048G**SERIAL NUMBER **2240134 F**

- ☐ RESEAL
☐ FLUSH
☒ SERVICE BULLETIN COMPLIANCE
☐ WARRANTY REPAIR
☐ OTHER
☐ AD COMPLIANCE
☐ LIGHTNING STRIKE
☐ METAL IN OIL (MIO)

GOVERNOR INSPECTION:

- ☒ Cleaning
☒ De-mag of all steel parts
☒ Inspect idler/drive gear for wear
☒ Dye-penetrant body
☒ Magnaflux drive gear
☒ Drive gear dimensional check

Pre-Assembly Inspection

Chief Inspector: *[Signature]*

Service Bulletin Compliance:

- 33531B .004" max. gear wear - **3** Max "M"s per S.B. by part number, inspection marks on drive gear.
 33532B Safety wire on the solenoid screws (P/N 1039-240).
 33559 Add 3004-936 strainer screen on 1310-105 reset solenoid.
 33569 Replace body if the reset solenoid is leaking (units with solid casting).
 33602 Replace 1310-160 solenoid if affected by serial number (hang-up due to burr).

33580 M TRO

CUSTOMER Professional AviationDATE 2-12-09WORK ORDER 33013SERVICE MANUAL 33048GGOVERNOR MODEL 210638SERIAL NUMBER 2240134FRESET SOLENOID MODEL # 1310-105 SERIAL # 2936RAUTO-FEATHER SOLENOID MODEL # 1310-118 SERIAL # 3382RCHANGE LETTER BEFORE OVERHAUL OR REPAIR D

WORK TO BE ACCOMPLISHED:

- ☒ OVERHAUL
- ☐ RESEAL
- ☐ FLUSH
- ☒ SERVICE BULLETIN COMPLIANCE
- ☐ WARRANTY REPAIR
- ☐ OTHER
- ☐ AD COMPLIANCE
- ☐ LIGHTNING STRIKE
- ☐ METAL IN OIL (MIO)

Normal OVH

GOVERNOR INSPECTION:

- ☒ Cleaning
- ☒ De-mag of all steel parts
- ☒ Inspect idler/drive gear for wear
- ☒ Dye-penetrant body
- ☒ Magnaflux drive gear
- ☒ Drive gear dimensional check

Pre-Assembly Inspection

Chief Inspector: [Signature]

Service Bulletin Compliance:

- 33531B .004" max. gear wear - 3 Max "M"s per S.B. by part number, inspection marks on drive gear.
- 33532B Safety wire on the solenoid screws (P/N 1039-240).
- 33559 Add 3004-936 strainer screen on 1310-105 reset solenoid.
- 33569 Replace body if the reset solenoid is leaking (units with solid casting).
- 33602 Replace 1310-160 solenoid if affected by serial number (hang-up due to burr).

33580 M TBO

TEST RECORD FOR OVERSPEED GOVERNORS PER WOODWARD TSP-847

A. SUPPLY Pressure	(150 +/- 10 PSI)	150	INSP	<i>PMI</i>
B. Maximum RPM Specified <u>4047</u> (+/- 10 RPM)	Actual RPM	4047		
Flow specified <u>9.2</u> QPM	Flow QPM	9.2	INSP	<i>PMI</i>
C. Reset RPM Specified <u>3570</u> (+/- 40 RPM)	Actual RPM	3566		
Flow specified <u>8.2</u> QPM	Flow QPM	8.2	INSP	<i>PMI</i>
(Energize Solenoid with 18 VDC)				
D. Internal Leakage (60 PSI Inlet Pressure @ <u>3548</u> RPM)	Flow QPH	40	INSP	<i>PMI</i>
E. Feather Solenoid Check (Zero Stand Speed)	Flow QPM	1109	INSP	<i>PMI</i>
(Set inlet for 60 PSI)				
F. Feather Leak Check (Zero Stand Speed)				
(Energize with 22 VDC set inlet for 60 PSI Delta P)				
(Reduce voltage to 1.0VDC no under shoot)				
(2.0 QPM Maximum Flow)				
G. Magnetic Pickup	(Voltage .7 RMS min)	1.55		
3200 RPM	(Indicated RPM)	1.27	INSP	<i>PMI</i>
H. External Unfeather Adapter (210625 ONLY)	Voltage	N/A		
(4 qts./min. minimum flow @ 200 PSI inlet)	RPM	N/A	INSP	-
I. Unfeather Adapter Leak Check	Flow	N/A	INSP	-
(1 oz./min. max leakage @ 150 PSI inlet)	Leakage	N/A	INSP	-
J. External Leakage Check (None Allowed)		0	INSP	<i>PMI</i>
K. Torque Check		0	INSP	<i>PMI</i>

INSPECTOR: *[Signature]*

INSPECTION FINAL:

Work Accomplished:

Overhaul governor per Woodward Service Manual 330486 revision 9

☐ GENERAL APPEARANCE
☐ SAFETY WIRE AND LEAD SEAL
☐ TORQUE SEAL
☐ SHIPPING COVER & MTING GASKET
☐ NAMEPLATE CHG LETTER F

ASSEMBLER / TECHNICIAN *[Signature]*

DATE 2/16/09

OVERHAUL DUE 6500 hours

CHIEF INSPECTOR *[Signature]*

DATE 2-16-09

AD'S Thru ns

SB'S Thru 33580M

SL'S Thru ns

1. Approving National Aviation Authority (Country) FAA / UNITED STATES		2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG		3. System Tracking Ref. No. 33380	
4. Organization Name and Address Rocky Mountain Propellers, Inc., 2865 Airport Drive, Erie, CO 80516 USA FAA FR6R545N - EASA.145.4327				5. Work Order, Contract, or Invoice Number. 33380	
6. Item:	7. Description:	8. Part Number:	9. Eligibility:	10. Qty:	11. Serial / Batch Number:
1	Woodward Governor	210638	N / A	1	2482821 F
12. Status / Work: Overhauled					
13. Remarks: Overhauled governor in accordance with Woodward Service Manual #33048G, Rev. 9. Complied with Service Bulletins thru 33580M. No AD's currently apply. Overhaul due 6 years or 6500 hours if unit is operated 120 hours per year continuously. A complete Description of work performed is on file at the above referenced organization under the work order and system tracking reference number indicated in blocks 3 and 5. Notice: An Airworthiness Directive may apply to the article(s) described hereon. The installer is responsible for ensuring complete compliance with any applicable Airworthiness Directives. "Certifies that the work specified in block 12 / 13 was carried out in accordance with FAR Part-145 and EASA Part-145 and in respect to that work the component is considered ready for release to service under FAA FR6R545N and EASA Approval Number: EASA.145.4827"					
14. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 13.			19. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input type="checkbox"/> Other regulation specified in Block 13 Certifies that unless otherwise specified in Block 13, the work identified in Block 12 and described in Block 13 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
15. Authorized Signature:		16. Approval / Authorization No.:		21. Approval / Certificate No.: FR6R545N	
17. Name (Typed or Printed):		18. Date (m/d/y):		23. Date (m/d/y): May 19, 2009	
20. Authorized Signature: <i>[Signature]</i>		22. Name (Typed or Printed): Dave Hampel			
User / Installer Responsibilities					
It is important to understand that the existence of this document alone does not automatically constitute authority to install the part / component / assembly. Where the user / installer performs work in accordance with the national regulations of an Airworthiness Authority different than the airworthiness authority of the country specified in block 1, it is essential that the user / installer ensures that his / her airworthiness authority accepts parts / components / assemblies from the airworthiness authority of the country specified in block 1. Statements in block 14 and 19 do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user / installer before the aircraft may be flown.					

FAA Form 8130-3 (6-01) * Installer must cross check eligibility with applicable technical data.

MSN: 0052-00-012-9005

- ☐ FLUSH
- ☐ SERVICE BULLETIN COMPLIANCE
- ☐ WARRANTY REPAIR
- ☐ OTHER
- ☐ AD COMPLIANCE
- ☐ LIGHTNING STRIKE
- ☐ METAL IN OIL (MIO)

2-033
FLUG WAS OVERTORQUED AND DISTURBED AND DAMAGE DURING REMOVAL

GOVERNOR INSPECTION:

- SL* Cleaning
- SL* De-mag of all steel parts
- SL* Inspect idler/drive gear for wear
- SL* Dye-penetrant body
- SL* Magnaflux drive gear
- SL* Drive gear dimensional check

Pre-Assembly Inspection

Chief Inspector: *[Signature]*

Service Bulletin Compliance:

- SL* 33531B .004" max. gear wear - *4* Max "M"s per S.B. by part number, inspection marks on drive gear.
- SL* 33532B Safety wire on the solenoid screws (P/N 1039-240).
- SL* 33559 Add 3004-936 strainer screen on 1310-105 reset solenoid.
- SL* 33569 Replace body if the reset solenoid is leaking (units with solid casting).
- SL* 33602 Replace 1310-160 solenoid if affected by serial number (hang-up due to burr).

SL ~~33580M~~

TBO

REV. 4 02-23-00

SERVICE MANUAL **33048G**

SERIAL NUMBER **2482821 F**

INSPECTION REPORT #W-TSP-847

(WOODWARD OVERSPEED GOVERNORS)

REV. 4 02-23-00

CUSTOMER Professional Aviation Associates

DATE 5-13-09

WORK ORDER 33380

SERVICE MANUAL 33048G

GOVERNOR MODEL 210638

SERIAL NUMBER 2482821 F

RESET SOLENOID MODEL # 1310-105 SERIAL # 40814

AUTO-FEATHER SOLENOID MODEL # 1310-118 SERIAL # 4879

CHANGE LETTER BEFORE OVERHAUL OR REPAIR _____

WORK TO BE ACCOMPLISHED:

☒ OVERHAUL

☐ RESEAL

☐ FLUSH

☐ SERVICE BULLETIN COMPLIANCE

☐ WARRANTY REPAIR

☐ OTHER

☐ AD COMPLIANCE

☐ LIGHTNING STRIKE

☐ METAL IN OIL (MIO)

NEEDS NEW BODY PLUG, P/N: 3612-033

PLUG WAS OVERTORQUED AND DISTORTED AND DAMAGE DURING REMOVAL

GOVERNOR INSPECTION:

92 Cleaning

92 De-mag of all steel parts

92 Inspect idler/drive gear for wear

92 Dye-penetrant body

92 Magnaflux drive gear

92 Drive gear dimensional check

Pre-Assembly Inspection

Chief Inspector: [Signature]

Service Bulletin Compliance:

52 33531B

92 33532B

92 33559

92 33569

92 33602

92 ~~33580~~ N

.004" max. gear wear - 4 Max "M"s per S.B. by part number, inspection marks on drive gear.

Safety wire on the solenoid screws (P/N 1039-240).

Add 3004-936 strainer screen on 1310-105 reset solenoid.

Replace body if the reset solenoid is leaking (units with solid casting).

Replace 1310-160 solenoid if affected by serial number (hang-up due to burr).

TBO

TEST RECORD FOR OVERSPEED GOVERNORS PER WOODWARD TSP-847

A. SUPPLY Pressure	(150 +/- 10 PSI)	150	INSP	B
B. Maximum RPM Specified <u>4047</u> (+/- 10 RPM)	Actual RPM	4047		
Flow specified <u>9.2 +/- .1 QPM</u>	Flow QPM	9.2	INSP	B
C. Reset RPM Specified <u>3570</u> (+/- 40 RPM)	Actual RPM	3577		
Flow specified <u>8.2 +/- .1 QPM</u> (Energize Solenoid with 18 VDC)	Flow QPM	8.2	INSP	B
D. Internal Leakage (60 PSI Inlet Pressure @ <u>3818</u> RPM)	Flow QPH	18	INSP	B
E. Feather Solenoid Check (Zero Stand Speed) (Set inlet for 60 PSI)	Flow QPM	9.77	INSP	B
F. Feather Leak Check (Zero Stand Speed) (Energize with 22 VDC set inlet for 60 PSI Delta P) (Reduce voltage to 1.0VDC no under shoot) (2.0 QPM Maximum Flow)	Shut-Down Voltage Flow	1.82 1.0	INSP	B
G. Magnetic Pickup (Voltage .7 RMS min) 3200 RPM (Indicated RPM)	Voltage RPM	NA NA	INSP	-
H. External Unfeather Adapter (210625 ONLY) (4 qts./min. minimum flow @ 200 PSI inlet)	Flow	NA	INSP	-
I. Unfeather Adapter Leak Check (1 oz./min. max leakage @ 150 PSI inlet)	Leakage	NA	INSP	-
J. External Leakage Check (None Allowed)		52	INSP	B
K. Torque Check		92	INSP	B

INSPECTOR: John P. H.

INSPECTION FINAL:

Work Accomplished:

OVERHAULED GOVERNOR IAW WOODWARD SERVICE MANUAL 61-20-37 REVISION 9.

- ☐ GENERAL APPEARANCE
- ☐ SAFETY WIRE AND LEAD SEAL
- ☐ TORQUE SEAL
- ☐ SHIPPING COVER & MTING GASKET
- ☐ NAMEPLATE CHG LETTER F

ASSEMBLER / TECHNICIAN [Signature]

DATE 5/18/09

OVERHAUL DUE 6 YEARS OR 6500 HOURS IF UNIT
IS OPERATED 120 HOURS PER YEAR CONTINUOUSLY

CHIEF INSPECTOR [Signature]

DATE 5-19-09

AD'S Thru NA

SB'S Thru 33580m

SL'S Thru NA

AERO ELECTRICAL SOLUTIONS

AMO 1011

AES

Cert. : 5743

CERTIFICATE RELATING TO MAINTENANCE OF AN AIRCRAFT

Aircraft Type: KING AIR 200

Registration: 9J-MED

Component: VENT BLOWER

Part Number: 603-7

Particulars of work done: OVERHAULED

Serial Number: 74

UNIT STRIPPED, CLEANED AND INSPECTED. ALL HARDWARE
DECARBONISED, SANDBLASTED AND ALLODINED. ARMATURE
SKIMMED, UNDERCUT AND POLISHED. ARMATURE LIMITS
1,239 INCH. REPLACED PARTS. UNIT RE-ASSEMBLED
AND BENCH TESTED. BRUSHES SEATED UP TO 100%.
UNIT FOUND SERVICEABLE.

Records held on Job Card: 8067

Modifications embodied:

NIL

Maintenance Manuals P/N's used & Revision Status:

P/N: OHM 603

REV: 24 JAN 1986

**AERO ELECTRICAL
SOLUTIONS**
AMO 1011
SERVICEABLE ITEM
Nomenclature: VENT BLOWER
Registration: 9J-MED
Work Done: OVERHAULED
Part No: 603-7
74

Parts used: BEARINGS
BRUSHES

I hereby certify that in carrying out the foregoing specified maintenance, all the requirements prescribed in the Civil Aviation Regulations, 1997, as amended, which are applicable hereto, have been complied with.

Date: 03-07-2009

Signature: 

Approval No. (Stamp):



AERO ELECTRICAL SOLUTIONS

AES

AMO 1011

Cert. : 5797

CERTIFICATE RELATING TO MAINTENANCE OF AN AIRCRAFT

Aircraft Type: KING AIR 200

Registration: 9J-MED

Component : CONDENSOR BLOWER

Part Number : M7921C-2B

Particulars of work done: OVERHAULED

Serial Number: 1718

UNIT STRIPPED, CLEANED AND INSPECTED.
ALL HARDWARE DECARBONISED, SANDBLASTED
AND ALLODINED. REPLACED PARTS. ARMATURE
SKIMMED, UNDERCUT AND POLISHED.

UNIT RE-ASSEMBLED AND BENCH TESTED.
BRUSHES SEATED UP TO 100%. UNIT
FOUND SERVICEABLE.

Records held on Job Card:

Modifications embodied:

NIL

Maintenance Manuals P/N's used & Revision Status:

P/N: 101-590010-17

REV: 31 OCT 2003

Parts used:

BRUSHES

BEARINGS

FILTER ASSY

I hereby certify that in carrying out the foregoing specified maintenance, all the requirements prescribed in the Civil Aviation Regulations, 1997, as amended, which are applicable hereto, have been complied with.

Date: _____ Signature: _____

License or Approval No. (Stamp): _____



AERO ELECTRICAL SOLUTIONS
AMO 1011
SERVICEABLE ITEM
Nomenclature: CONDENSOR BLOWER
Registration: 9J-MED
Work Done: OVERHAULED
Part No: M7921C-2B
Serial No: 1718
TTSN: N/K TCSN: _____
TTSO: 0100Hrs TCSSO: _____
CRMA No: 5797
AES AMO 1011 INSP 4

S A N A S

ACCREDITED
INSPECTION BODY

ISO/IEC 17020

SANS/SABS 1825

ACCREDITATION NUMBER:-

IG50048

DEPARTMENT OF LABOUR

REFILLABLE CONTAINER TEST STATION

APPROVAL NUMBER:-

CI0015

P.J. Edwards cc

Reg. No.: CK94/07071/23 / VAT No.: 4130162748
 SANAS Accredited TYPE A Testing Station No.: IGS 0048
 Department of Labour Approval number: CI015
 Unit 7, Wynpol, cnr 4th Street & Thora Crescent, Wynberg.
 PO Box 783874, Sandton City, 2146
 Tel: +27 (0) 11 887-1410, Fax: +27 (0) 11 507-5046
<http://www.pjedwards.net> E-mail: support@pjedwards.net



CYLINDER TEST CERTIFICATE

Cylinder owner/agent Turbo Pco Telephone _____ Certificate No. 24159

Postal Address _____ Code _____

Work done: ☐ Visual Inspection ☐ Hydrostatic Test ☐ Chemical Clean ☐ Eddy Current or Dye-Pen. Pest ☐ Valve Service ☐ Fill

CYLINDER INFORMATION

Gas service type ☐ Air ☐ Oxygen Other _____Make PST Serial Number 46096 Specification DOTMaterial ST Water Volume 7 (Litres) Mass _____ (kg)Service Pressure 124 (Bar) Test Pressure 180 (Bar) Manufacturer's Hydro Date 06/81Colour of Body Black Colour of Neck Black

EXTERIOR

☒ Good ☐ Fair ☐ Poor ☐ Fail

Comments _____

INTERIOR

☒ Good ☐ Fair ☐ Poor ☐ FailOdour ☐ Yes ☒ No Corrosion ☐ Yes ☒ No

Comments _____

THREADS

☒ Good ☐ Fair ☐ Poor ☐ FailCorrosion and/or Galling ☐ Yes ☒ No Damaged Threads _____

Comments _____

HYDROSTATIC TEST RESULTS (If applicable)

☐ Pass ☐ Fail

Expansion due to applied pressure _____ (ml) Permanent Set _____ (ml) % Permanent set _____ %

TARE MASS CHECK (If applicable)

☐ Pass ☐ Fail

Comments _____

CYLINDER STATUS

☒ Accept ☐ Condemn ☐ Accept with NotationComments H109107

1. Tested at Wynberg, Sandton, on this 02 day of 07 2009 and found to comply with the Occupational Health and Safety Act 85, 1993, along with the Vessels Under Pressure Regulations read with code of practice SABS 1825 and SABS 019.
2. The abovementioned cylinder is required to be periodically re-inspected/tested in accordance with the above mentioned acts.
3. Cylinders are cleaned and certified for the stated service gas. Any subsequent filling by any party other than PJ Edwards cc may contaminate the cylinder. PJ Edwards cc accepts no responsibility whatsoever for damage, injury or death of whatever nature arising from the handling and/or filling of the above cylinder.
4. This certificate does not imply any warranty in respect of any cylinder as to its fitness for use once it has left the company's premises.
5. This test report should be retained by the cylinder owner and might be required by a cylinder filler before filling of the cylinder.

FORM #3

RELEASE CERTIFICATE

CERTIFICATE NUMBER

C 8506

AEROSPACE ELECTROPLATING C.C.

POSTAL ADDRESS :

P.O. BOX 2628
KEMPTON PARK
1620

REG.NO. 1995/032811/23

C A A APPROVED PROCESS AND
TESTING ORGANISATION P T 7

TELEPHONE :011-827-7535

FAX NUMBER:011-827-9896

E-mail:info@aeroplate.co.za

COMPANY NAME

CASH SALES

CUSTOMER I.D.

71

DATE

13/07/2009

POSTAL ADDRESS

TURBO PROP SERVICE CENTRE

DELIVERY ADDRESS

HANGER 7

LANSERIA

SECTION

POSTAL CODE

PHONE NUMBER

011 701 3210

FAX NUMBER

011 701 3485

ORDER NUMBER

3058

AIRCRAFT REG.

9J-MED

TIE UP JOB

5709

PART NUMBER

SERIAL NUMBER

DESCRIPTION

AGS

PROCESS

CADMIUM PLATE

SPECIFICATION

SPOP 21

QUANTITY

1 BAG

NOTE

I HEREBY CERTIFY THAT THE ITEM / S DESCRIBED ABOVE HAS / HAVE BEEN TESTED /
PROCESSED IN ACCORDANCE WITH THE REQUIREMENTS PRESCRIBED IN THE MANUAL OF
PROCEDURE SUBMITTED FOR THE ISSUE OF CERTIFICATE OF APPROVAL NUMBER P T 7.

CHIEF INSPECTOR / ALTERNATE CHIEF INSPECTOR

SIGNED :

INSP. STAMP

AEROPLATE

Insp. #1

CAA No. PT7

AIRCRAFT NATIONALITY AND REGISTRATION MARKS

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		DETAILS OF MODIFICATIONS,
	Hrs.	Mins.	Hrs.	Mins.	Hrs.	Mins.	
Brought Forward							
21.04.05	04	48	85	53	7581	56	3 8518
28.04.05	04	48	90	41	7586	38	3 8521
29.04.05	01	06	91	49	7531	88	1 8524
01.05.05	04	18	96	05	7538	34	3 8525
03.05.05	01	00	97	05	7536	52	1 8528
10.05.05	04	48	101	49	7537	52	2 8529
13.05.05	05	30	104	19	7542	34	3 8531
18.05.05	01	54	109	11	7548	04	2 8534
20.05.05	01	48	110	59	7549	58	2 8536
24.05.05	08	24	113	23	7551	46	2 8538
26.05.05	08	30	118	53	7554	10	4 8540
08.06.05	01	06	122	59	7562	40	1 8544
06.06.05	02	00	124	59	7563	46	2 8545
10.06.05	01	06	126	05	7564	46	1 8547
18.06.05	01	36	127	41	7565	52	1 8548
Carried Forward					7567	28	1 8549
					7567	28	8549

AIRCRAFT NATIONALITY AND REGISTRATION MARKS

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		DETAILS OF MODIFICATIONS,
	Hrs.	Mins.	Hrs.	Mins.	Hrs.	Mins.	
Brought Forward							
21.09.05	2	00	163	42	9601	08	8576
23.07.05	2	06	165	48	7603	08	8579
25.07.05	8	06	173	48	7605	34	8581
26.07.05	5	42	179	30	7613	34	8584
27.07.05	8	36	188	06	7619	16	8586
28.07.05	1	06	189	06	7627	52	8590
30.07.05	3	18	192	24	7628	52	8591
31.07.05	3	18	195	42	7632	10	8594
AUG.03.08	2	12	197	54	7635	28	8595
04.08.05	3	12	201	06	7637	40	8597
07.08.05	1	54	203	00	7640	52	8599
08.08.05	3	18	206	08	7642	46	8600
09.08.05	2	00	208	18	7646	04	8603
10.08.05	4	48	213	06	7648	04	8605
11-0-05	2	6	215	12	7652	52	8609
Carried Forward					7654	58	8611

Airworthiness Approval Tag
U.S. Department of Transportation
Federal Aviation Administration

0000128421-00027

4. Organization

RAYTHEON AIRCRAFT COMPANY
9709 EAST CENTRAL
P O BOX 85
WICHITA, KANSAS 67201-0085

5. Work Order Contract or
Invoice Number

5609000001-00520

6. Item	7. Description	8. Part Number	9. Eligibility *	10. Quantity	11. Serial/Batch Number	12. Status/Work
0001	INSTAL OF STRESS F WINDSHIELD (COMM	101-5041-3 S	TBV by Installer	1.000	23180	NEW

13. Remarks

AIRWORTHINESS APPROVAL - PARTS. THIS FORM IS NOT AN EXPORT APPROVAL

Limited Life parts must be accompanied by maintenance history including total time/total cycles/time since new.

14. New



Newly overhauled



Certifies that the new or newly overhauled part(s) identified above, except as otherwise specified in block 13 was (were) manufactured in accordance with FAA approved design data and airworthiness.

NOTE: In case of parts to be exported, the special requirements of the importing country have been met.

19.

Return to service in accordance with FAR 43.9

Certifies that the work specified in block 13 (or attached) above was carried out in accordance with FAA airworthiness regulations and in respect to the work performed the part(s) is (are) approved for return to service.

15. Signature



16. FAA
Authorization No.

DOA230339CE

20. Authorized Signature

21. Certificate
Number

17. Name (Typed or Printed)

B. J. COTTER

18. Date:

05/04/2001

22. Name (Typed or Printed)

23. Date:

USER/INSTALLER RESPONSIBILITIES

It is important to understand that the existence of this Document alone does not automatically constitute authority to install the part/component/assembly.

Where the user/installer work in accordance with the national regulations of an Airworthiness Authority different than the Airworthiness Authority of the country specified in block 1 it is essential that the user/installer ensure that his/her Airworthiness Authority accepts parts/components/assemblies from the Airworthiness Authority of the country specified in block 1.

Statements in block 14 and 19 do not constitute installation certification. In all cases aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

The FAA form 8130-3 and JAA Form One are equivalent. Other countries such as Canada also have equivalent acceptable documents.

FAA Form 8130-3 (11-93)

*(Optional) Installer must cross check eligibility with applicable technical data.



WORKS NO. 22 HUNTSVILLE, AL.
AIRCRAFT PRODUCTS



PLEASE UNPACK AND CHECK CAREFULLY. The contents of this case were DOUBLE CHECKED, and correspond with the invoice they were packed and shipped in apparent good order. Most SHORTAGES & DISCREPANCIES arise from the fact that glass is not carefully checked and measured by consignee. Most Breakages are due to improper care in handling in transit or at destination. All shipments are F.O.B. Factor. Shipments should be carefully examined at time of delivery, and if damage is found, notation to that effect should be made on delivery receipt or carrier's agent. When damage is concealed, it is absolutely necessary that carrier be notified within fifteen days from date of delivery in order to be in position to present claim. If a discrepancy should occur which seems clearly the fault of the Packer, this ticket must be returned with claim within 5 days after receipt of shipment, with full particulars.

ISN _____ LEVEL / PROTECTION _____ ITEM DESC _____
JUST PO / ITEM H0762634 0001 GOVT CONT _____
TD CASE QTY 0 WT. 54 LBS. CU. FT. 0.00 CUST PART/REV 101-384025-18

SHIP TO: 00015000
RAYTHEON AIRCRAFT COMPANY
BEECH HAWKER
9709 EAST CENTRAL AVENUE
WICHITA KANSAS 67206

MISCELLANEOUS INFORMATION:

'PARAGRAPH A.' APPLICABLE BELOW

CASE #

SHIPPER #

\$4287

QUANTITY

1

PG ORDER NO.
/ ITEM

6-H-AC-2574

0001

SERIAL NUMBER

97086H3378

LICENSE NO. _____

PPG PART

BA-002821

SHIPPING ADDRESS

PARAGRAPH A.

CERTIFICATE OF CONFORMANCE

PPG IND. INC. HEREBY CERTIFIES THAT THE EQUIPMENT DESCRIBED ABOVE IS OF THE QUANTITY LISTED, IS MANUFACTURED OF NEW MATERIAL AND IS IN CONFORMITY WITH APPLICABLE CONTRACTUAL SPECIFICATIONS. IT IS FURTHER CERTIFIED THAT THIS EQUIPMENT IS MANUFACTURED TO THE SAME GENERAL STANDARDS WHICH APPLY TO IDENTICAL EQUIPMENT BEING SUPPLIED TO THE U.S. GOVERNMENT AND OTHER CUSTOMERS. TEST AND INSPECTION HAS BEEN PERFORMED AS REQUIRED TO ASSURE PRODUCT CONFORMANCE WITH APPLICABLE EQUIPMENT SPECIFICATIONS.

R.D. Linton
MANAGER, QUALITY CONTROL

PARAGRAPH B.

CERTIFICATE OF REPAIR

THIS WILL CERTIFY THAT REPAIR HAS BEEN PERFORMED UPON THIS PRODUCT AND THAT THE PRODUCT CURRENTLY COMPLIES WITH THE MANUFACTURER'S SPECIFICATIONS FOR THE CHARACTERISTICS OF THE PRODUCT WHICH WERE REPAIRED. ONLY THOSE CHARACTERISTICS INCLUDED IN THE RECORD OF REPAIR ARE COVERED BY THIS CERTIFICATE. THIS CERTIFICATE SHALL IN NO WAY COMPRISE OR BE INTERPRETED AS A REPRESENTATION OF ANY KIND WITH REGARD TO ANY OTHER CHARACTERISTICS OF THE PRODUCT.

PARAGRAPH C.

CERTIFICATE OF CONFORMANCE - REBUILT PRODUCT

PPG INDUSTRIES, INC. CERTIFIES THAT THIS PRODUCT HAS BEEN FACTORY REBUILT TO THE LATEST REVISION PART NUMBER AND IS IN CONFORMITY WITH APPLICABLE CONTRACTUAL SPECIFICATIONS. IT IS FURTHER CERTIFIED THAT TEST AND INSPECTION HAS BEEN PERFORMED AS REQUIRED TO ASSURE PRODUCT CONFORMANCE WITH APPLICABLE PPG SPECIFICATIONS.

03/31/97

PARAGRAPH D.

NON-REPAIRABLE PRODUCT

PPG INDUSTRIES, INC. HAS INSPECTED THIS PRODUCT AND DETERMINED THE PRODUCT TO BE NON-REPAIRABLE / OR NOT OF PPG MANUFACTURE.

REPAIRS, ADJUSTMENTS, ETC.
(4)

INSPECTED AND CERTIFIED
AS IN NOTE* BELOW
(5)

2 TYRE WORN TO LIMITS REPLACED WITH P/N 020-524-0
S/N 9278000904

H. J. J.
16/8/05

Co Pilot Wind shield elements and glass burnt and
internal damage - Removed P/N 101-384025-8
S/N 88-H-1127-429 and fitted with P/N 101-384025-18
S/N 9708643378 used kit N° 101-5041-35
adnant P/N C P S 890 B 1/2 Pressurization
satisfactory after cure. A/c released into service.

H. J. J.
22/9/2005

4 TYRE WORN TO LIMITS REPLACED WITH P/N 265F86-8
S/N 0 41420124. COPIED REPLACED N° 528775-267

P/N REMOVED 265K08-1 S/N 00FF42780119 (TYRE)

FWD CIRCULATION BLOWER MOTOR BRUSHES REPLACED PART
NUMBER ON EM 35-2-3001 FUNCTION CHECKED FOR PROPER OPERATION
SATISFACTORY.

H. J. J.
25/9/2005

H. J. J.
29/9/2005

*NOTE - A Signature (the date of which must be added) in this column (5) will be taken as a certificate that, in carrying out the overhaul, repair, modification, or replacement to which it relates, all the conditions and requirements of the Air Navigation Regulations for the time being in force applicable thereto under the Aviation Act, 1954, as amended by any other enactment have been complied with.

AIRCRAFT NATIONALITY AND REGISTRATION MARKS

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		DETAILS OF MODIFICATIONS,
	Hrs.	Mins.	Hrs.	Mins.	Hrs.	Mins.	
Brought Forward							
07-10-5	2	36	260	58	7700	34	Q650
08-10-5	2	18	263	34	7703	10	Q656
10-10-5	2	12	265	52	7705	23	Q658
10-10-5	2	12	268	4	7707	40	Q660
10-10-5	1	24	269	28	7709	04	Q662
17-10-5	0	48	270	16	7709	52	Q663
22-10-05	2	00	272	16	7711	52	Q665
26-10-05	2	18	274	34	7714	10	Q667
28-10-05	1	54	276	28	7716	04	Q668
31-10-05	1	12	276	40	7716	16	Q669
05-Nov-05	4	00	280	40	7720	16	Q673
07-11-05	4	12	284	52	7724	28	Q677
10-11-05	0	24	285	16	7724	52	Q678
11-11-05	0	24	285	40	7725	16	Q679
12-11-05	3	42	289	24	7729	00	Q681
18-11-05	3	24	292	48	7731	24	Q683
Carried Forward							

REPAIRS, ADJUSTMENTS, ETC.
(4)

INSPECTED AND CERTIFIED
AS IN NOTE* BELOW
(5)

- PHASE III INSPECTION COMPLETED AT 7709 THRS T.T.

ON 21/10/5

- NO 1 MAIN WHEEL TYRE REPLACED P/N 00F-265K08-1

P/N 00N 265F86-8 S/N 00N 4420125 S/N 00F 42750615

- NO 1 MAIN WHEEL INNER BEARING REPLACED P/N 00N 1368

- EVAPORATION FILTER FWD POSITION REPLACED P/N 00N 101-635171-1

- PORT ENGINE INTAKE ICING VANE HINGE REPAIRED BY RIVETING VANE TO THE HINGE

- INSTRUMENT AIR FILTER REPLACED P/N 00N B3-5-1

I hereby certify that in carrying out the Overhaul, Repair, Replacement, Modification specified above all the conditions and requirements of the Air Navigation Regulations for the time being in force which are applicable thereto have been complied with.

Signature: *[Signature]*
Licence: 269 ATX
Date: 21.10.1955
Hawthornditch



*NOTE - A Signature (the date of which must be added) in this column (5) will be taken as a certificate that, in carrying out the overhaul, repair, modification, or replacement to which it relates, all the conditions and requirements of the Air Navigation Regulations for the time being in force applicable thereto under the Aviation Act, 1954, as amended by any other enactment have been complied with.

AIRCRAFT NATIONALITY AND REGISTRATION MARKS

DETAILS OF MODIFICATIONS,

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		
	Hrs.	Mins.	Hrs.	Mins.	Hrs.	Mins.	
Brought Forward							
20.11.2005	1	24	292	48	7731	24	8683
22.11.2005	1	30	295	42	7732	48	8685
23.11.2005	2	00	297	42	7736	18	8687
24.11.2005	4	12	301	54	7740	30	8691
8.12.2005	1	48	303	42	7742	18	8692
11.12.2005	1	48	305	30	7744	06	8693
?	1	48	307	18	7745	54	8694
22-12-2005	2	12	309	30	7748	6	8698
1-1-2006	2	6	311	36	7750	12	8700
4-1-2006	4	36	316	12	7754	48	8703
5-1-2006	3	18	319	30	7758	6	8705
Carried Forward			000	00	7758	06	

Survey

checked for CofA

21.02.06

ZAMBIA SKYWAYS LTD

AMO:- 026

SERVICEABLE

STAR OF AFRICA

BATCH NO.

DESCRIPTION WIND SHIELD

LET

TEMP CONTROL

B90

P/N 100-364295-3 S/N 120304171CT

200

TBO..... TSLD.....

SHELF LIFE..... EXPIRY DATE.....

ROTABLE ☐ EXPENDABLE ☐ REPAIRABLE ☐

I hereby certify that in carrying out the Inspection / Repair / O/H / Replacement / Modification specified above, all the conditions and requirements of the AIR NAVIGATION REGULATIONS for the time being in force which are applicable thereto have been complied with.

SIGN. [Signature] AUTH. 227 DATE 9/11/02

INSTALLED ON

A/C MED

POS. 2

DATE 30/12/15

ENGINEER

REMOVED FROM

A/C.....

POS.....

DATE.....

ENGINEER

INSPECTED AND CERTIFIED
AS IN NOTE* BELOW
(5)

30/12/05

NTROUVER REPLACED

100 81F-723-222-81

120304171CT

9L COFA RENEWAL CARRIED

(Signature) 26/1/2006



*NOTE—A Signature (the date of which must be added) in this column (5) will be taken as a certificate that in carrying out the overhaul, repair, modification, or replacement to which it relates, all the conditions and requirements of the Air Navigation Regulations for the time being in force applicable thereto under the Aviation Act, 1954, as amended by any other enactment have been complied with.

REPAIRS, ADJUSTMENTS, ETC.

(4)

CO-PILOT WINDOW HEAT CONTROLLER REPLACED

P/N^o OFF 100-3642895-3 S/N^o 8F-723-222-81

P/N^o ON 100-364285-3 S/N^o 120304171CT

PHASE IV INSPECTION & ANNUAL COFA RENEWAL CARRIED

UOT AT T.T. 7758 HR

INSPECTED AND CERTIFIED
AS IN NOTE* BELOW
(5)

30/12/05

269/1762

20/1/2006



*NOTE-A Signature (the date of which must be added) in this column (5) will be taken as a certificate that, in carrying out the overhaul, repair, modification, or replacement to which it relates, all the conditions and requirements of the Air Navigation Regulations for the time being in force applicable thereto under the Aviation Act, 1954, as amended by any other enactment have been complied with.

UNIT 1 NATIONALITY AND REGISTRATION MARKS

DETAILS OF MODIFICATIONS,

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		CYCLES
	Hrs.	MINS.	Hrs.	MINS.	Hrs.	MINS.	
Brought Forward							
80-01-06	1	00	319	30	7758	06	83705
24-01-06	0	54	321	24	7759	06	83706
28-01-06	1	30	322	54	7761	30	83707
03-02-06	2	00	2	00	7763	30	83711
04-02-06	2	12	4	12	7766	42	83712
07-03-06	2	06	6	18	7768	48	83714
14-03-06	1	06	7	24	7769	54	83715
17-03-06	1	06	8	30	7771	00	83716
18-03-06	1	06	9	36	7772	06	83717
19-03-06	0	18	9	54	7772	24	83718
19-03-06	0	24	10	18	7772	48	83719
20-03-06	4	12	14	30	7777	00	83722
21-03-06	0	42	15	12	7777	42	83723
22-03-06	0	42	15	54	7778	24	83724
22-03-06	0	48	16	42	7779	12	83725
Carried Forward	0	48	16	42	7779	12	83725

SINCE COP A HE ZEROED
TO MAKE CORRECTION
CORRECTION TO TOTAL TIME + 1hr 10st
20-18-11-5 will read 7766 42

REPAIRS, ADJUSTMENTS, ETC.
(4)

INSPECTED AND CERTIFIED
AS IN NOTE* BELOW
(6)

20/3

No 2 MAIN WHEELTYRE REPAIRED P/W OFF 026-524-0

S/W OFF 927800904 P/W ON 265K08-8 S/W ON 6087139

20/3/06



*NOTE—A Signature (the date of which must be added) in this column (6) will be taken as a certificate that, in carrying out the overhaul, repair, modification, or replacement to which it relates, all the conditions and requirements of the Air Navigation Regulations for the time being in force applicable thereto under the Aviation Act, 1954, as amended by any other enactment have been complied with.

95-meb

AIRCRAFT NATIONALITY AND REGISTRATION MARKS

DETAILS OF MODIFICATIONS,

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		Cycles	
	Hrs.	Mins.	Hrs.	Mins.	Hrs.	Mins.		
Brought Forward	0	48	16	42	7779	12	8325	
28-03-06	0	48	17	30	7780	00	8326	
02-04-06	3	18	20	48	7783	18	8330	
04-04-06	2	18	23	06	7785	36	8333	
05-04-06	1	30	24	26	7787	06	8335	
06-04-06	2	06	26	42	7789	12	8338	
07-04-06	2	06	28	48	7791	18	8340	
07-04-06	0	48	29	36	7792	06	8341	
07-04-06	0	42	30	18	7792	48	8342	
08-04-06	0	06	31	24	7793	54	8343	
10-04-06	1	00	32	24	7794	54	8344	
12-04-06	1	00	34	24	7796	54	8346	
16-04-06	2	12	37	36	7800	06	8348	
23-04-06	3	12	41	48	7804	18	8352	
24-04-06	4	00	44	48	7807	18	8356	
25-04-06	3	54	46	42	7809	12	8358	
27-04-06	1	34	46	42	7809	12	8358	
Carried Forward	1	34	46	42	7809	12	8358	

REPAIRS, ADJUSTMENTS, ETC.
(4)

INSPECTED AND CERTIFIED
AS IN NOTE* BELOW
(5)

INSPECTION PHASE I COMPLETED WITH IN ACCEPTANCE WITH MAINTANCE

[Signature]

SCHEDULE ON CALENDAR TIME AT 718 TOTAL TIME, NEXT CHECK AT 7958 HRS

21-4-6

*NOTE—A Signature (the date of which must be added) in this column (5) will be taken as a certificate that, in carrying out the overhaul, repair, modification, or replacement to which it relates, all the conditions and requirements of the Air Navigation Regulations for the time being in force applicable thereto under the Aviation Act, 1934, as amended by any other enactment have been complied with.

REPAIRS, ADJUSTMENTS, ETC.
(4)

INSPECTED AND CERTIFIED
AS IN NOTE* BELOW
(5)

AS RESULT OF ADDITION ERRORS FROM 03102106 TO 30105106.

269

ZH 269/1798



*NOTE—A Signature (the date of which must be added) in this column (5) will be taken as a certificate that, in carrying out the overhaul, repair, modification, or replacement to which it relates all the conditions and requirements of the Air Navigation Regulations for the time being in force applicable thereto under the Aviation Act, 1954, as amended by any other enactment have been complied with.

AIRCRAFT NATIONALITY AND REGISTRATION MARKS

95-MED

DETAILS OF MODIFICATIONS,

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		
	HRS.	MIN.	HRS.	MIN.	HRS.	MIN.	
Brought Forward	1	25	80	20	7840	40	2 8784
22.06.06	2	20	82	40	7843	00	2 8786
26.06.06	1	25	84	05	7844	25	2 8788
28.06.06	1	20	85	25	7845	45	2 8790
30.06.06	1	20	86	45	7847	05	3 8791
04.07.06	3	25	90	10	7850	30	3 8794
11.07.06	2	30	92	40	7853	00	3 8797
13.07.06	0	50	93	30	7853	50	3 8798
14.07.06	0	50	94	20	7854	40	3 8799
14.07.06	0	50	95	10	7855	30	3 8800
15.07.06	0	50	96	00	7856	20	3 8801
16.07.06	3	00	99	00	7859	20	3 8804
17.07.06	1	00	100	00	7860	20	3 8805
18.07.06	2	00	102	00	7862	20	3 8807
19.07.06	1	00	103	00	7863	20	3 8808
03.08.06	2	05	105	05	7865	25	3 8810
Carried Forward	2	05	105	05	7865	25	3 8810

INSPECTION PHASE II CARRIED ON
7863.20 HRS TOTAL TIME.

NO 3 main wheel tyre replaced
cleaned, inspected and lubricated
NO 2 main wheel tyre replaced
cleaned, inspected and

REPAIRS, ADJUSTMENTS, ETC.
(4)

INSPECTED AND CERTIFIED
AS IN NOTE* BELOW
(5)

AND APPROVED MAINTENANCE SCHEDULE ON CALENDAR TIME AT
Next Check due IS Check III at 8063.

PM 265K08-1, O-ring replaced PM MS875-267 and bearings
with grease S. Setis
PM 265K08-1, O-ring replaced PM MS875-267 and bearings
lubricated with grease S. Setis.

I hereby certify that in carrying out the
Overhaul, Repair, Replacement,
Modification specified above all
the conditions and requirements of the
Air-Navigating Regulations for the
time being in force which are applicable
herein have been complied with.
Signature: *Phillips*
License: 2671798
Date: 22.7.2006
Transferring: 22.07.06



*NOTE-A Signature (the date of which must be added) in this column (5) will be taken as a certificate that, in carrying out the overhaul, repair, modification, or replacement to which it relates, all the conditions and requirements of the Air Navigation Regulations for the time being in force applicable thereto under the Aviation Act, 1954, as amended by any other enactment have been complied with.

AIRCRAFT NATIONALITY AND REGISTRATION MARKS

95-MED

DETAILS OF MODIFICATIONS,

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		
	Hrs.	Mins.	Hrs.	Mins.	Hrs.	Mins.	
Brought Forward	2	05	105	05	7865	25	2710 New Battery installed P/L 9-6381E
05.08.06	3	05	108	10	7868	30	2813
05.08.06	3	15	111	25	7871	45	2817
06.08.06	2	45	114	10	7874	30	2820
08.08.06	0	40	114	50	7875	10	2821
09.08.06	0	55	115	45	7876	05	2822
08.06	0	55	116	40	7877	00	2823
16.06.06	0	55	117	35	7877	55	2824
18.08.06	0	50	118	25	7878	45	2826
21.06.06	1	05	119	30	7879	50	2827
22.06.06	3	10	122	40	7883	00	2830
25.08.06	1	40	124	20	7884	40	2831
27.08.06	1	30	125	50	7886	10	2832
29.08.06	2	10	128	00	7888	20	2833
30.08.06	2	25	130	25	7890	45	2835
31.08.06	0	50	131	15	7891	35	2836
Carried Forward	0	50	131	15	7891	35	2836

REPAIRS, ADJUSTMENTS, ETC.
(4)

INSPECTED AND CERTIFIED
AS IN NOTE * BELOW
(5)

W. J. 269 ACF
27/July 2006

*NOTE—A Signature (the date of which must be added) in this column (5) will be taken as a certificate that, in carrying out the overhaul, repair, modification, or replacement to which it relates, all the conditions and requirements of the Air Navigation Regulations for the time being in force applicable thereto under the Aviation Act, 1954, as amended by any other enactment have been complied with.

AIRCRAFT NATIONALITY AND REGISTRATION MARKS

97-men

DETAILS OF MODIFICATIONS,

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		
	Hrs.	Mins.	Hrs.	Mins.	Hrs.	Mins.	
Brought Forward	0	50	181	15	1891	35	8836
03.09.06	1	10	132	25	1892	45	8837
04.09.06	1	00	133	25	1893	45	8838
05.09.06	1	48	135	13	1895	33	8839
08.09.06	4	36	136	49	1900	09	8841
16.09.06	2	12	139	01	1902	21	8843
19.09.06	0	48	139	49	1903	09	8844
20.09.06	0	48	140	37	1903	57	8845
20.09.06	-	-	143	40	1904	00	8845
21.09.06	2	30	146	10	1906	30	8848
22.09.06	1	10	147	20	1907	40	8849
08.10.06	4	05	151	25	1911	45	8853
10.10.06	0	55	152	20	1912	40	8854
11.10.06	0	50	153	10	1913	30	8855
12.10.06	4	25	157	35	1917	55	8860
13.10.06	0	50	158	25	1918	45	8861
Carried Forward	0	50	158	25	1918	45	8861

TS - 4A HOURS and TOTAL
OF ERRORS DATED 03.09.06

REPA

TO
Time

REPAIRS, ADJUSTMENTS, ETC.
(4)

INSPECTED AND CERTIFIED
AS IN NOTE* BELOW
(5)

TIME HOWAS COLLECTED TO THE NEAREST FIVE MINUTES AS A RESULT
TO 20.09.06.

(Hylus) 20/9/06
209/1796 ACX

*NOTE—A Signature (the date of which must be added) in this column (5) will be taken as a certificate that, in carrying out the overhaul, repair, modification, or replacement to which it relates, all the conditions and requirements of the Air Navigation Regulations for the time being in force applicable thereto under the Aviation Act, 1934, as amended by any other enactment have been complied with.

AIRCRAFT NATIONALITY AND REGISTRATION MARKS

93-MFA

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		DETAILS OF MODIFICATIONS,
	Hrs.	MINS.	Hrs.	MINS.	Hrs.	MINS.	
Brought Forward	0	50	158	25	7918	45	19165
18.10.06	2	25	160	50	7921	10	² 8863
26.10.06	1	10	162	00	7922	20	² 8864
01.10.06	1	20	163	20	7923	40	² 8866
28.10.06	1	10	164	30	7924	50	³ 8869
24.10.06	1	00	165	30	7925	50	² 8871
25.10.06	2	30	168	00	7928	20	¹ 8872
27.10.06	3	40	171	40	7932	00	² 8874
28.10.06	3	25	175	05	7935	25	³ 8877
06.11.06	2	20	177	25	7937	45	² 8879
09.11.06	1	10	178	35	7938	55	³ 8882
10.11.06	3	00	181	35	7941	55	² 8884
01.12.06	3	20	184	35	7944	55	¹ 8885 (8997 landing)
12.06	3	05	187	40	7948	00	² 8887
19.12.06	3	40	191	30	7951	40	² 8889
20.12.06	1	25	192	55	7958	05	² 8891
Brought Forward	1	25	192	55	7953	05	

Phase 3 inspection carried out
Next inspection due at 03.10.06

REPAIRS, ADJUSTMENTS, ETC.
(4)

INSPECTED AND CERTIFIED
AS IN NOTE* BELOW
(5)

out 1. And. Approved Maintenance Schedule of 7932:00 hrs
8132:00 FAI

(Signature)
3rd Nov 2006

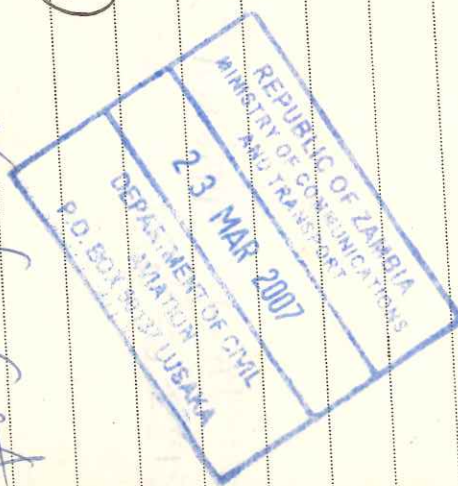
*NOTE - A Signature (the date of which must be added) in this column (5) will be taken as a certificate that, in carrying out the overhaul, repair, modification, or replacement to which it relates, all the conditions and requirements of the Air Navigation Regulations for the time being in force applicable thereto under the Aviation Act, 1954, as amended by any other enactment have been complied with.

93-meb.

AIRCRAFT NATIONALITY AND REGISTRATION MARKS

DETAILS OF MODIFICATIONS,

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		
	Hrs.	Mins.	Hrs.	Mins.	Hrs.	Mins.	
Brought Forward							
21.12.06	3	25	192	56	7953	05	8891
22.12.06	3	20	196	30	7956	30	8895
27.12.06	2	25	202	15	7962	15	8897
29.12.06	3	25	205	40	7965	40	8902
02.01.07	0	50	206	30	7966	30	8906
09.01.07	2	40	209	10	7969	10	8908
13.01.07	2	40	211	50	7971	50	8910
01.07	1	10	213	00	7973	00	8912
23.01.07	1	10	214	10	7974	10	8913
24.01.07	1	05	215	15	7975	15	8914
25.01.07	1	35	216	50	7976	50	8915
26.01.07	0	50	217	40	7977	40	8916
31.01.07	2	40	220	20	7980	20	8917
02.02.07	2	55	223	15	7983	15	8919
05.02.07	0	50	224	05	7984	05	8921
Carried Forward			000	00	7984	05	8922



Checked for CoFA Rem
Hudson
20.03.07

Aircraft Technical Services Ltd.

P.O. Box AP 10, Lusaka, Zambia. Tel: 271143

LOGBOOK ENTR Y

DATE 23RD FEB, 2007

Aircraft Registration : 9J-MED

SERIAL NO: BB1046 MAKE BEECHCRAFT 200

AIRFRAMES.

OPERATION: FOURTH DETAIL INSPECTION

For the Certificate of Airworthiness renewal the fourth detailed inspection was carried out on the above mentioned aircraft in accordance with the maintenance manual part number as stipulated by the manufacturers procedures.

All cabin floor panels were opened up to inspect the under floor structures for corrosion, cracks, deformation, loose or failed fasteners satisfactory.

Inspected the front bulkhead pressure seals, instrument mountings and electrical wire bundles.
Inspected the rear bulk head for cracks, pressure seals for deterioration and found satisfactory.

Inspected the stringers, frame attachments for corrosion and cracks and found satisfactory.
Inspected the cabin front wing spar and rear spar cracks and corrosion and found satisfactory.
Checked left and right wing structures for integrity and found satisfactory. Inspected the fuel panels for leaks, de ice boots and failed rivets and found satisfactory.

Inspected the left and right aileron cables for broken strands and corrosion and cables were lubricated with a corrosion protective compound LPS-3 checked cable guards, pulleys and pins, inspected pressure seals at wing attachments satisfactory.
Cleaned and serviced the safety and outflow valves, checked pressurization and vacuum lines for looseness.

Inspected the tail planes : elevator, rudder skins, for loose and failed fasteners.
Oxygen lines leaking repaired and replenished to 1800 psi as per procedure. cleaned the oxygen crew and passenger masks and stowed satisfactory.

Inspected the nose landing gear emergency retraction chain and sprockets for corrosion, wear and looseness, cleaned and lubricated with molly coat Dow conning 33 satisfactory. Inspected the landing gear actuators, actuator hoses for leaks and hoses for chafing and found satisfactory.
Inspected the landing gear attachment bolts and mounting structure for cracks and lubricated the landing gear retraction mechanism with grease 22. Retraction and extension sequences performed satisfactory. Cable tensions and ranges of movements recorded refer to the attached sheets.
Compass swing carried out and compensated as per on the work compass card which is attached. No Airworthiness directives or mandatory service bulletins are pending affecting the air frames at this inspection see AD listing attached. *on 17/1/05*
Duplicate inspection of all flight controls carried out in accordance with engineering notice No.10 as follows:

1st inspection.....*[Signature]*.....license No. *559*.....date 22nd February, 2007

2nd inspection.....*[Signature]*.....silence No. *227*.....date 22nd February, 2007

FILED
BROW
(5)

I hereby certify that in carrying out the
Overhaul, Repair, Replacement,
Modification specified above all
the conditions and requirements of the
Air Navigation Regulations for the
time being in force which are applicable
thereto have been complied with
Signature: *PHH*
Licence: *PHH*
Date: *23/2/2007*
Compl. ending: *23/2/2007*

*NOTE—A Signature (the date of which must be added) in this column (5) will be taken as a certificate that, in carrying out the overhaul, repair, modification, or replacement to which it relates, all the conditions and requirements of the Air Navigation Regulations for the time being in force applicable thereto under the Aviation Act, 1954, as amended by any other enactment have been complied with.

AIRCRAFT NATIONALITY AND REGISTRATION MARKS

DETAILS OF MODIFICATIONS,

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		
	Hrs.	Mins.	Hrs.	Mins.	Hrs.	Mins.	
Brought Forward							
22.08.07	1	15	000	00	7984	05	8922
23.08.07	2	15	3	30	7985	20	8924
06.09.07	2	00	5	30	7989	35	8928
12.09.07	2	40	8	10	7992	15	8930
16.09.07	4	55	13	05	7997	10	8932
17.09.07	3	18	16	23	8000	28	8934
19.09.07	3	22	19	46	8003	50	8936
31.09.07	0	15	20	01	8004	05	8938
01.10.07	0	20	20	21	8004	25	8940
	7	35	27	56	8008	00	8942
	2	35	30	31	8014	35	8944
15.06.07	2	20	32	51	8016	55	8946
18.06.07	2	35	35	26	8019	30	8948
22.06.07	1	00	36	26	8020	30	8950
24.06.07	1	00	37	26	8021	30	8951
Carried Forward			37	26	8021	30	8951

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8950
8951

REPAIRS, ADJUSTMENTS, ETC.
(4)

INSPECTED AND CERTIFIED
AS IN NOTE* BELOW
(5)

FWD BLOWER FILTER REPLACED P/N 0N 101-555171-1
AFT BLOWER FILTER REPLACED P/N 0N 101-555084-3

28/5/7

[Signature] 28/5/7
[Signature] 28/5/7

in accordance with Beachcraft Maintenance Manual
T.A.T. ~~was~~ due to calendar time next inspection is due

29.05.07

[Signature] 29/1/78

replaced with new tyre P/N 265F86-8 due to skid marks.
replaced with new tyre P/N 265F86-8 - worn to limits
replaced. P/N 622-1879-001 S/N 20071.

269/1796 ACX

13.07.07

13.07.07

27.07.07

*NOTE-A Signature (the date of which must be added) in this column (5) will be taken as a certificate that, in carrying out the overhaul, repair, modification, or replacement to which it relates, all the conditions and requirements of the Air Navigation Regulations for the time being in force applicable thereto under the Aviation Act, 1954, as amended by any other enactment have been complied with.

AIRCRAFT NATIONALITY AND REGISTRATION MARKS

DETAILS OF MODIFICATIONS,

DETAILS OF MODIFICATIONS,						
DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME	
	Hrs.	MINS.	Hrs.	MINS.	Hrs.	MINS.
Brought Forward						
29.06.07	1	45	37	26	8021	30
29.06.07	2	00	39	11	8023	15
30.06.07	2	00	41	11	8025	15
10.07.07	1	55	43	11	8027	15
11.07.07	1	05	45	06	8029	10
07.07	4	45	46	11	8030	15
21.07.07	0	35	50	56	8035	00
24.07.07	0	35	51	31	8035	35
30.07.07	2	20	52	06	8036	10
31.07.07	2	00	54	26	8038	30
04.08.07	3	00	56	26	8040	30
07.08.07	2	35	59	26	8043	30
08.08.07	2	20	62	01	8046	05
09.08.07	1	25	64	21	8048	25
10.08.07	2	05	65	46	8050	50
			67	51	8052	55
Carried Forward						

AIRCRAFT NATIONALITY AND REGISTRATION MARKS

DETAILS OF MODIFICATIONS,

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		
	Hrs.	Mins.	Hrs.	Mins.	Hrs.	Mins.	
Brought Forward							
11.08.07	5	45	73	36	8052	55	8983
12.08.07	2	40	76	16	8058	40	8986
16.08.07	2	00	78	16	8061	20	8990
08.07	2	10	80	26	8063	20	8993
22.08.07	6	20	82	46	8065	30	Phase II inspection completed
06.09.07	4	35	87	21	8071	50	8998
12.09.07	4	20	91	41	8076	25	89001
17.09.07	6	45	98	26	8080	45	4 9005
21.10.07	2	55	161	21	8087	30	3 9008
3.10.07	2	45	104	06	8090	25	4 9012
5.10.07	5	55	110	00	8093	10	2 9014
16.10.07	2	10	112	10	8099	05	1 9015
20.10.07	1	00	113	10	8001	15	3 9018
26.10.07	2	40	115	50	8102	15	2 9020
27.10.07	2	40	118	30	8104	55	4 9024
Carried Forward	2	40	118	30	8107	35	4 9028
					8107	35	9028

REPAIRS, ADJUSTMENTS, ETC.

(4)

INSPECTED AND CERTIFIED
AS IN NOTE* BELOW
(5)

in accordance with AMS/OS-MED. at 74, 8071 Hrs.

31.08.07

Philip 259

I hereby certify that in carrying out the
Overhaul, e air Replacement,
Modification specified above all
the conditions and requirements of the
the conditions and requirements of the
therein have been complied with

Signature

Licence

Date

Hours/Landing

03/9/2007

8071

*NOTE—A Signature (the date of which must be added) in this column (5) will be taken as a certificate that, in carrying out the overhaul, repair, modification, or replacement to which it relates, all the conditions and requirements of the Air Navigation Regulations for the time being in force applicable thereto under the Aviation Act, 1934, as amended by any other enactment have been complied with.

AIRCRAFT NATIONALITY AND REGISTRATION MARKS

DETAILS OF MODIFICATIONS,

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		
	HRS.	MINS.	HRS.	MINS.	HRS.	MINS.	
Brought Forward 20	2	40	118	30	8107	35	9028
20/11/2007	3	20	121	50	8110	55	9032
21/11/2008	3	50	124	52	8113	55	9037
09 FEB 08	-	50	2414	20	8114	45	9038
10 "	2	00	2416	20	8116	45	9041
15 "	3	00	2419	20	8120	05	9047
18 "	3	00	2422	20	8123	05	9053
20 "	5	50	2428	10	8128	55	9059
22 "	3	35	2431	45	8132	30	9065
25 "	3	05	2436	50	8134	30	9066
26 "	1	30	2438	20	8137	35	9072
27 "	1	25	2439	45	8139	05	9074
28 "	2	10	2440	55	8141	30	9077
Carried Forward	2	10	2440	55	8143	40	9080
					8143	40	9080
					8148	40	9080

(4) REPAIRS, ADJUSTMENTS, ETC.

INSPECTED AND CERTIFIED
AS IN NOTE* BELOW
(5)

Operation 3 inspection completed. at 8113 hrs.

23/1/2008



N^o 2 MAN WHEEL TYRE WORK TO LIMITS AND REPLACED WITH NEW ONE. P/JS OFF 026-524-0 AND P/JS ON 026-524-0.

27/2/2008



N^o 4 MAIN WATER TYLE WORK TO LIMITS AND REPEATED WITH NEW ONE. P/M ON 265F86-8.

11/3/2008



*NOTE—A Signature (the date of which must be added) in this column (5) will be taken as a certificate that, in carrying out the overhaul, repair, modification, or replacement to which it relates, all the conditions and requirements of the Air Navigation Regulations for the time being in force applicable thereto under the Aviation Act, 1964, as amended by any other enactment have been complied with.

AIRCRAFT TECHNICAL SERVICES

P.O. Box AP10 Lusaka, Zambia. Tel: 271143

LOGBOOK ENTRY

Date 23rd January 2008

AIRCRAFT REGISTRATION: 9J-MED

SERIAL NO: BB1046

MAKE: BEECHCRAFT 200

AIRFRAMES.

OPERATION: THIRD DETAIL INSPECTION.

For the Certificate of Airworthiness renewal, the third detailed inspection was carried out on the above mentioned Aircraft in accordance with the maintenance manual as stipulated by the manufacturer's procedures.

All cabin floor panels were opened up to inspect the under floor structures for corrosion, cracks, deformation, loose or failed fasteners satisfactory.

Inspected the front bulkhead pressure seals, instrument mountings and electrical wire bundles. Inspected the rear bulkhead for cracks, pressure seals for deterioration and found satisfactory.

Inspected the stringers, frame attachments for corrosion and cracks and found satisfactory.

Inspected the cabin front wing spar and rear spar cracks and corrosion and found satisfactory.

Checked left and right wing structures for integrity and found satisfactory. Inspected the fuel panels for leaks, de ice boots and failed reverts and found satisfactory.

Inspected the left and right aileron cables for broken strands and corrosion and cables were lubricated with a corrosion protective compound LPS-3, checked cable guards, pulleys and pins, inspected pressure seals at wing attachments satisfactory.

Cleaned and serviced the safety and out flow valves, checked pressurization and vacuum lines for looseness.

Inspected the tail planes, elevator, rudder skins, for loose and failed fasteners.

Inspected the nose landing gear and emergency retraction chain and sprockets for corrosion, wear and looseness, cleaned and lubricated with molly coat Dow conning 33 satisfactory. Inspected the landing gear actuators, actuator hoses for leaks and chaffing and found satisfactory.

Inspected the landing gear attachment bolts and mounting structure for cracks and lubricated the landing gear retraction mechanism with grease 22. Retraction and extension sequences performed satisfactory. Cable tension and ranges of movements recorded. Refer to the attached sheets.



269 ACX/NDT
UT/ED.

Compass swing carried out and compensated as per on the work compass card which is attached. No Airworthiness Directive or mandatory service bulletins are pending affecting the Airframe at this inspection. See AD listing

AD 87-22-01 R1 -NOSE LANDING GEAR FOUND EMBODIED

AD 89-19-04 R1 -WING FUEL BAY SKIN PANNELS FOUND EMBODIED

AD 93-25-01 -FUSELAGE STRINGERS -FOUND EMBODIED

AD97-23-17-Inspection of tubing found satisfactory

AD99-09-15 -Flight control mechanism found satisfactory

AD2005-01-04 FUEL HOSE REPLACEMENT P/N MIL-H-6000B COMPLIED WITH

AD2006-23-02 DETECT AND CORRECT IMPROPERLY ASSEMBLED FLIGHT CONTROLS -CONTROLS FOUND CORRECTLY ASSEMBLED

SB27-3038-STABILIZERS-INSPECTION OF RUDDER TORQUE SHAFT SUPORT BEARINGS FOUND SATISFACTORY

SB 27-3232 -FLIGHT CONTROLS- INSPECTION OF FLIGHT CONTROLS FOUND SATISFACTORY

SB 35-3020 -INSPECTION OF PURITAN BENNET OXYGEN MASKS FOUND SATISFACTORY



SB2676-INSPECTION OF AIRCON PRESSURIZATION PNEUMATIC TUBING FOUND SATISFACTORY


SB2607-R1 - INSPECTION OF MAIN GEAR DRAG LEG COMPLIED WITH SATISFACTORY

SB 2472-R2 -INSPECTION OF AFT FUSELAGE STRINGERS CHECK AND FOUND SATISFACTORY

SB2312 INSPECTION OF AFT FUSELAGE MOISTURE COMPLIED WITH SATISFACTORY

Duplicate inspection of all flight controls carried out in accordance with engineering notice No.10 as follows:

1st inspection..........license No..........Date.....23/1/08.....

2nd inspection..........license No.....227.....Date.....23/1/08.....



DATE
(1)

Carried Forward

Checked for GFA Revenue
In L8 DL
26.03.08

AIRCRAFT NATIONALITY AND REGISTRATION MARKS

DETAILS OF MODIFICATIONS,

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		
	Hrs.	MINS.	Hrs.	MINS.	Hrs.	MINS.	
Brought Forward			000	00	8162	15	9113
14/03/08	3	10	3	10	8165	25	9114
17/03/08	4	00	7	10	8169	25	9124
19/03/08	01	00	8	10	8170	25	9125
20/03/08	02	30	10	40	8172	55	9127
21/03/08	3	05	13	45	8176	00	9132
24/03/08	3	10	16	55	8179	10	9136
26/03/08	3	00	19	55	8182	10	9141
28/03/08	3	10	23	05	8185	20	9147
01/04/08	3	05	26	10	8188	25	9152
04/04/08	3	10	29	20	8191	35	9158
07/04/08	3	05	32	25	8194	40	9163
09/04/08	3	10	35	35	8197	50	9169
11/04/08	3	05	38	40	8200	55	9174
12/04/08	3	10	41	50	8204	05	9180
18/04/08	3	10	45	00	8207	15	9185
Carried Forward	3	10	45	00*	8207	15	9185

3003

PLEASE START ENTERING
"TIME SINCE LAST C of A
RENEWAL" CORRECTLY.
26-03-08

REPAIRS, ADJUSTMENTS, ETC.
(4)

INSPECTED AND CERTIFIED
AS IN NOTE* BELOW
(5)

Phase II inspection completed at 8207 hrs.
satisfactory.



18/4/08

*NOTE—A Signature (the date of which must be added) in this column (5) will be taken as a certificate that, in carrying out the overhaul, repair, modification, or replacement to which it relates, all the conditions and requirements of the Air Navigation Regulations for the time being in force applicable thereto under the Aviation Act, 1954, as amended by any other enactment have been complied with.

AIRCRAFT NATIONALITY AND REGISTRATION MARKS

9J-MED

DETAILS OF MODIFICATIONS,

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		
	Hrs.	Mins.	Hrs.	Mins.	Hrs.	Mins.	
Brought Forward	3	10	45	00	9207	15	9185
21/04/08	3	10	48	10	8210	25	9191
23/4/08	3	10	51	20	8213	35	9197
24/4/08	2	35	53	55	8216	10	9202
25/4/08	-	45	54	40	8216	55	9204
28/04/08	3	05	57	45	8220	00	9209
30/04/08	2	20	60	05	8222	20	9211
02/05/08	3	20	63	25	8226	40	9217
03/05/08	06	05	67	30	8228	45	9219
05/05/08	03	05	72	35	8235	50	9224
07/05/08	03	05	75	40	8238	55	9229
09/05/08	03	10	78	50	8242	05	9235
12/05/08	03	10	82	00	8245	15	9241
14/05/08	03	10	85	10	8248	25	9247
16/05/08	03	10	88	20	8251	35	9253
19/05/08	03	05	91	25	8254	45	9258
Carried Forward	3	5	91	25	8254	45	9258

AIRCRAFT NATIONALITY AND REGISTRATION MARKS

DETAILS OF MODIFICATIONS,

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		
	HRS.	MIN.	HRS.	MIN.	HRS.	MIN.	
Brought Forward	03	05	91	25	8254	45	9258
20.05.08	06	00	97	25	8250	45	9260
21.05.08	03	05	100	30	8253	50	9265
23.05.08	03	10	103	40	8257	00	9271
26.05.08	03	10	106	50	8260	10	9277
28.05.08	03	05	109	55	8263	15	9282
29.05.08	08	45	118	40	8272	00	9285
30.05.08	03	10	121	50	8275	10	9291
02.06.08	03	10	125	00	8278	20	9297
04.06.08	03	10	128	10	8279	30	9303
07.06.08	03	10	131	20	8282	40	9309
09.06.08	03	10	134	30	8285	50	9315
11.06.08	03	10	137	40	8289	00	9321
13.06.08	03	05	140	45	8292	05	9326
16.06.08	06	20	147	05	8298	25	9332
17.06.08	03	10	150	15	8301	35	9333
Carried Forward	03	10	150	15	8301	35	9333

AIRCRAFT NATIONALITY AND REGISTRATION MARKS


DETAILS OF MODIFICATIONS,

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		
	Hrs.	Mins.	Hrs.	Mins.	Hrs.	Mins.	
Brought Forward							
18.06.08	03	10	150	15	8301	35	9333
20.06.08	03	10	153	25	8304	45	9338
23.06.08	03	10	156	35	8307	55	9342
25.06.08	03	10	159	45	8311	05	9348
27.06.08	03	10	162	55	8314	15	9354
29.06.08	03	10	166	05	8317	25	9360
01.07.08	03	10	169	15	8320	35	9366
03.07.08	03	10	172	25	8323	45	9372
05.07.08	03	10	177	55	8329	15	9380
07.07.08	03	10	181	05	8332	25	9385
09.07.08	03	10	186	25	8337	45	9387
11.07.08	03	10	190	10	8341	30	9392
13.07.08	03	10	193	20	8344	40	9398
15.07.08	03	10	196	45	8348	05	9404
17.07.08	03	10	200	20	8351	40	9410
19.07.08	03	10	200	20	8351	40	9416
21.07.08	03	10	200	20	8351	40	9416
Carried Forward							

REPAIRS, ADJUSTMENTS, ETC.
(4)

INSPECTED AND CERTIFIED
AS IN NOTE* BELOW
(5)

Operation inspection complied with at 8348:00
141 satisfactory as it replaced on both
engines.

20/7/200


*NOTE—A Signature (the date of which must be added) in this column (5) will be taken as a certificate that, in carrying out the overhaul, repair, modification, or replacement to which it relates, all the conditions and requirements of the Air Navigation Regulations for the time being in force applicable thereto under the Aviation Act, 1954, as amended by any other enactment have been complied with.

AIRCRAFT NATIONALITY AND REGISTRATION MARKS

DETAILS OF MODIFICATIONS,

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		
	Hrs.	Mins.	Hrs.	Mins.	Hrs.	Mins.	
Brought Forward	03	35	200	20	8351	40	9416
24.07.04	03	50	204	10	8355	30	9422
25.07.04	03	10	207	20	8358	40	9428
26.07.04	03	10	210	30	8361	50	9434
27.07.04	03	00	213	30	8364	50	9441
28.07.04	01	00	214	30	8365	50	9442
29.07.04	03	60	217	30	8368	50	9447
30.07.04	03	10	220	40	8372	00	9453
31.07.04	03	10	223	50	8375	10	9459
01.08.04	04	30	228	20	8379	40	9466
02.08.04	04	35	132	55	8384	15	9469
03.08.04	04	00	135	55	8387	15	9474
04.08.04	03	10	139	05	8390	25	9480
05.08.04	03	10	142	15	8393	35	9486
06.08.04	02	55	145	10	8396	30	9488
07.08.04	03	10	148	20	8399	40	9494
08.08.04	03	10	148	20	8399	40	9494
Carried Forward	03	10	148	20	8399	40	9494

AIRCRAFT NATIONALITY AND REGISTRATION MARKS

97 MED.

DETAILS OF MODIFICATIONS,

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		
	Hrs.	Mins.	Hrs.	Mins.	Hrs.	Mins.	
Brought Forward							
20.08.08	03	10	148	20	8399	40	9494
22.08.08	03	05	151	25	8402	45	9499
25.08.08	03	10	154	35	8405	55	9505
26.08.08	02	30	160	15	8411	35	9511
27.08.08	03	05	163	20	8414	40	9521
28.08.08	03	18	166	35	8417	55	9525
30.08.08	01	30	168	05	8419	25	9526
31.08.08	01	25	169	30	8420	50	9527
01.09.08	03	10	172	40	8424	00	9533
03.09.08	03	20	178	00	8427	20	9539
05.09.08	03	10	179	10	8430	30	9545
08.09.08	03	10	182	20	8433	40	9551
10.09.08	03	10	185	30	8436	50	9557
12.09.08	03	20	188	50	8440	10	9563
13.09.08	02	25	191	15	8442	35	9565
Carried Forward			191	15	8442	35	9565

AIRCRAFT NATIONALITY AND REGISTRATION MARKS

9J MEB

DETAILS OF MODIFICATIONS,

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		
	Hrs.	Mins.	Hrs.	Mins.	Hrs.	Mins.	
Brought Forward							
15.09.08	02	25	191	15	8442	35	9565
16.09.08	03	10	194	25	8445	45	9571
17.09.08	08	00	202	25	8453	45	9575
18.09.08	03	10	205	35	8456	55	9581
19.09.08	03	10	208	45	8460	05	9587
19.09.08	02	10	210	55	8462	15	9589
20.09.08	03	10	214	05	8465	25	9595
21.09.08	03	10	217	15	8468	35	9601
22.09.08	08	00	225	15	8476	35	9605
23.09.08	03	10	228	25	8479	45	9611
24.09.08	05	00	233	25	8464	45	9613
25.09.08	05	10	238	35	8469	55	9621
26.09.08	03	10	241	45	8473	05	9627
27.09.08	03	10	244	55	8476	15	9633
28.09.08	03	10	248	05	8479	25	9639
29.09.08	03	10	251	15	8482	35	9645
30.09.08	03	10	251	15	8482	35	9645
Brought Forward							

REPAIR
(4)

REPAIRS, ADJUSTMENTS, ETC.
(4)

INSPECTED AND CERTIFIED
AS IN NOTE* BELOW
(5)

Operation Phase 2 inspection completed
at 8400:35 satisfactory next phase is
a Phase 3 detailed inspection + wing bolt replacement.

4/10/2008



*NOTE—A Signature (the date of which must be added) in this column (5) will be taken as a certificate that, in carrying out the overhaul, repair, modification, or replacement to which it relates, all the conditions and requirements of the Air Navigation Regulations for the time being in force applicable thereto under the Aviation Act, 1934, as amended by any other enactment have been complied with.

AIRCRAFT NATIONALITY AND REGISTRATION MARKS

DETAILS OF MODIFICATIONS,

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		DETAILS OF MODIFICATIONS,
	HRS.	MINS.	HRS.	MINS.	HRS.	MINS.	
Brought Forward	03	10	251	15	8482	35	9645
12.10.08	02	50	254	05	8485	25	9649
13.10.08	03	05	257	10	8488	30	9654
15.10.08	03	05	260	15	8491	35	9659
17.10.08	03	10	263	25	8494	45	9665
20.10.08	03	10	266	35	8497	55	9671
22.10.08	-	40	267	15	8498	35	9672
23.10.08	3	15	270	30	8501	50	9674
27/07/2009	-	-	270	30	8501	50	9674
06.08.2009	-	50	271	20	8502	40	9675
21.08.2009	2	30	273	50	8505	10	9676
02.09.2009	-	-	342	55	8505	10	9676
Carried Forward	00	00	8505	10			

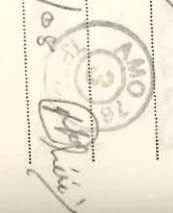
CHECKED FOR
RENEWAL



03.08.04

PHASE 3 INSPECTION & NEW
ENGINES INSTALLED (LH & RH SIDES)

Corrected Hours 02/13/15/08



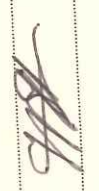






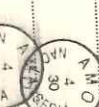




AIRCRAFT NATIONALITY AND REGISTRATION MARKS

95-MED

DATE (1)	TIME IN AIR (2)		TIME SINCE LAST C. OF A. RENEWAL (3)		TOTAL TIME		DETAILS OF MODIFICATIONS,
	HRS.	MINS.	HRS.	MINS.	HRS.	MINS.	
Brought Forward	0	00			8505	10	Landing.
2-9-2009	1	05	1	05	8506	15	'9677
4-9-2009	1	10	2	15	8507	25	'9678
7-9-2009	1	05	3	20	8508	30	'9679
11-9-2009	3	50	7	10	8512	20	'9680
LOGBOOK CLOSED, NEW ONE OPENED WITH DE-REGISTRATION							
From 95-MED TO 5X-MED. HOURS AND CYCLES TRANSFERRED							
TO A NEW LOGBOOK							
SIGN: Mark.							
NAME: TIBATWA ALEX							
Auth. Mark 21/8/09 10/03/10/9							
Carried Forward							

SUMMARY OF MODIFICATIONS EMBODIED SINCE MANUFACTURE

MODIFICATION NUMBER (1)	DESCRIPTION OF MODIFICATION (2)	CLASSIFICATION (3)	DATE EMBODIED (4)	CERTIFIED AS IN NOTE* BELOW (5)
STC 5A3519NM	RAISBECK DASH AFT BODY STRAKES	OPT	8/06/2009	 
STC 5A3366NM	FITTED RAISBECK RAMP AIR RECOVERY SYSTEM	OPT	8/06/2009	 
STC 5A33831NM	INSTALLED RAISBECK ENHANCED PERFORMANCE LEADING EDGE SYSTEM INSTALLED	OPT	8/06/2009	 
STC 4A175NM	RAISBECK FAULTY ENCLOSED HIGH FROTATION GEAR DOOR SYSTEM INSTALLED	OPT	8/06/2009	 
STC 5A01946KA	ENGINE INDICATORS INSTALLED	OPT	8/06/2009	 
STC 5A1082KSL	INSTALLED	OPT	8/06/2009	 

*NOTE.—A Signature (the date of which must be added) in this column will be taken as a certificate that, in carrying out the modification to which it relates, all the conditions and requirements of the Air Navigation Regulations for the time being in force applicable thereto under the Aviation Act, 1954, as amended by any other enactment, have been complied with.

MODIFICATION
NUMBER
(1)

DESCRIPTION OF MODIFICATION (2)

CLASSIFICATION (3)

DATE
EMBODIED
(4)

CERTIFIED AS IN NOTE*
BELOW
(5)

AD204-01-04

fuel hose replacement

Ad 205. 07. 08

(mu-4-6208)

22.02.05

269/1796 Act

AD2005-01-18R

check and repair crled furlage

01.03.05

269/1796 Hct

FD-200b-23-02

Detect and correct in-flight anomalies

13.12.06

File 259/1796 AX

*NOTE.—A Signature (the date of which must be added) in this column will be taken as a certificate that, in carrying out the modification to which it relates, all the conditions and requirements of the Air Navigation Regulations for the time being in force applicable thereto under the Aviation Act, 1954, as amended by any other enactment, have been complied with.

SUMMARY OF MODIFICATIONS EMBODIED SINCE MANUFACTURE

MODIFICATION NUMBER (1)	DESCRIPTION OF MODIFICATION (2)	CLASSIFICATION (3)	DATE EMBODIED (4)	CERTIFIED AS IN NOTE* BELOW (5)
AB 87-22-01R1	New landing gear		24.01.01	
AD 89-19-04R1	Wing fuel bay air panels		"	
AD 93-35-01	Fuselage struts		"	
AD 97-23-17	Inspection of landing		"	
AD 99-09-15	Flight control mechanism		"	
SB 27-3038	Stabilizer - drop struts	Manufactured	"	
SB 27-3038	torque shaft support bearing.	Manufactured	"	
SB 27-3038	Flight controls - inspection	"	"	
SB 35-3020	Puritan Beumel - Oxygen Drop	Manufactured	"	
SB 2676	Air-con: Pressurized central	Manufactured	"	
SB 2676	oxygen pneumatic tubing.	Manufactured	"	
SB 2607R1	Main landing gear drag leg	Manufactured	"	
SB 2607R1	door link inspection	Manufactured	"	
SB 2472 R2	Fuselage - drop modification	Manufactured	"	
SB 2472 R2	of aft fuselage struts	Manufactured	"	

*NOTE.—A Signature (the date of which must be added) in this column will be taken as a certificate that, in carrying out the modification to which it relates, all the conditions and requirements of the Air Navigation Regulations for the time being in force applicable thereto under the Aviation Act, 1954, as amended by any other enactment, have been complied with.

TRANSFERRED FROM
ZIMBABWE AIRFRAME LOG
BOOK FOR
27-APR-60
95-MED
17/11/03
606 AC

SB 2310
SB 2311
SB 2312
SB 2313
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SUMMARY OF MODIFICATIONS EMBODIED SINCE MANUFACTURE

MODIFICATION NUMBER (1)	DESCRIPTION OF MODIFICATION (2)	CLASSIFICATION (3)	DATE EMBODIED (4)	CERTIFIED AS IN NOTE* BELOW (5)
SB 2312	Swop of aft fuselage moisture drainage system			
SB 2254 R1	Sealing of longitudinal slab & door sill for moisture protection			
SB 2220	Swop or replacement of the nose landing gear steering link.			
SB 2213 R3	Swop of pilot & copilot seat track wear			
SB 2213	"	"	"	
SB 2208 R1	Replacement of cabin, storm & baggage side windows (cast)	"	"	
SB 2150 R2	Installation of re-inforced elevator pulley bracket supports			
SB 2109 R5	Swop replacement of NLG lower shock absorbers			
SB 2044	Swop replacement of rivets in fuselage	"	"	

*NOTE - A Signature (the date of which must be added) in this column will be taken as a certificate that, in carrying out the modification to which it relates, all the conditions and requirements of the Air Navigation Regulations for the time being in force applicable thereto under the Aviation Act, 1954, as amended by any other enactment, have been complied with.

and aft spar of vertical stabilizer

TRANSFERRED FROM
ZIMBABWE AIRFRAME
LOG BOOK FOR SJ-MED.
17/11/03
606 AC

1. 2004