US Department
of Transportation
Federal Aviation
receral Aviation

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance

OMB No. 2120-0020 Exp: 04/30/2023	Electronic Tracking Number
	or FAA Use Only

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A. Ag	gency's I	Name and A	ddress				B. 1	Kind of Agend	СУ				
Name		<u>Ropg</u>	Hohh				X		ated Mechanic				facturer
Address City)3 S. MESA	ngnu		149 TX	—	Н	<u></u>	ificated Mechanic Repair Station		C. (Certific	ate No.
Zip			untry U.S				\vdash		Maintenance Organ	nization	A	ėρ:	3173828 IA
D:	have b	een made in	accordance	e with the	made to the e requiremen ne best of my	ts of	Part	entified in item 43 of the U.S		scribed on	the re	everse	or attachments hereto
	nded ran 4 CFR F B			Signatur	Te/Date of Au	thori	zed I	Individual	sed				04-21-201
			· · · · · · · · · · · · · · · · · · ·	· · · · · ·				al for Return	/ / 				,
					specified bistration and		, the	unit identif	ied in item 5 v	was inspec		n the	manner prescribed by the
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NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished (If more space is required, attach and registration mark and date work completed.)

Inspected Rirplans to Comply With Nationally and Regension Mark
AD 90-08- 14 Phod Mandatory Service Bulletin SB53-3269 R3.

Found crack From Rivet to rivet about John Sping in length, on the alluminum Cover. Installed in AcCord Ance with Installed kit.

PN 58-4008-1-5, installed in AcCord Ance with Installation

Instructions included in Joubler Kit. Right Stringer Was aluminum Cover. ** \$0.83-B56-TE-387-01, revision IR.

Engineering Order ** \$0.83-B56-TE-387-01, revision IR.

Braineering Order ** \$0.83-B56-TE-387-01, revision IR.

Acted 39 March 30.83 Signed by Der Raub Echevarria.

Der 6.3253768 Copy Attrackdy, Substantiation

Acted 39 March 30.83 Signed by Der Raut Echevarria.

Der 6.3253768 Copy Attrackdy, Substantiation

Echevarria, Lighthouse Point, FIAN Riveraft Registration

Echevarria, Dighthoma City, OK, signed by Der Raul Echevarria.

Der 6.3253788. Copy Attached.

FAR Form 8110-3, Signed by RAUD Echevarria, DER 633537688 Engineering Order 2023-BES-TE-387-01, REVISION IR 69007 for Stringer Repair # JSV-2033-003 ALB FORM 8110-3, Signed by RAUD Echevarria, DER 633537688

Additional Sheets Are Attached

DETERMINATION	U.S. DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION DETERMINATION OF COMPLIANCE WITH AIRWORTHINESS STANDARDS								
AIRCRAFT OR AIRCRAFT COMPONENT IDENTIFICATION									
2. MAKE Textron Aviation Inc.	3. MODEL NO. 95-c55	4. TYPE (Aircraft, Engine, Propell Actuator etc.) Aircraft							
		PURPOSE OF DATA		·					
6. IN SUPPORT OF:	TC/ATCSTC	PMA X Major Repair	Major Alteration	Other (Explain)					
PROJECT SPECIFIC INF	Stringer Rep ORMATION:	pair							
PURPOSE OF SUBMITTA	AL: In support of FAA for	rm 337 for Major Repair for airc	raft S/N TE-387						
		LIST OF DATA igs, material specifications, and process ulirements listed in block 9. A reference to							
7. IDENTIFICATION	8. TITLE OF DATA		<u> </u>						
EO 2023-855-TE-387-01 Rev IR	Stringer Repair								
Date: Mar 29, 2023									
JSV-2023-003 Rev IR	BEECHCRAFT BARON 95-C55	STRINGER REPAIR STATIC SUBSTANT	PIATION						
Date: Mar 30, 2023									
		:							
	Notes:								
	This approval is for engineering design data only. It indicates the data listed above demonstrates compliance only with the regulations specified by paragraph and subparagraph listed below as 'Applicable Requirements'.								
	1. This is not a Damage 2. This approval covers	Tolerance Approval. structural detais only.							
9. APPLICABLE REQUIR	L EMENTS (List specific sec	tions and amendment levels)							
14 CFR23.601 [Amdt 23-0 23.613 (all) [Amdt 23-2		-23]; 23.605 (a) [Amdt 23-23]; 2	23.611 [Amdt 23-7];						
23.013 (a11) (Aud. 23-2.	, ,								
10. FAA DESIGNEE APPI 14 CFR, Part 183, data lis therefore	ROVAL - As directed by the ted above, and on attached	Administrator and in accordance was sheets numbered, have be	ith the conditions and en examined in accor	limitations of authorization under dance with established procedures. I					
APPROVE the dat	a above	ID APPROVAL of the data above							
	R MAJOR ALTERATION OF	NLY Other data approvals 🖲 AR	E REQUIRED AF	RE NOT REQUIRED					
EXPLAIN: This form does not cons requirements needed for	titute FAA approval of a	II the engineering data necessar	ry for substantiation included in this a	on of compliance with the					
approval.	requirements needed for the complete repair. The needs of 14 CPR 23.573 are not included in this approval and require separate approval. MANAGING OFFICE WAS CONTACTED (required when approval was made outside the U.S and/or invloved critical or life limited parts)								
11. DER/ODA NUMBER	2 11110 00111110 125 (100	12. PRINTED NAME	10.00 110 0.0 11001	Trioved emical of the armica parts)					
632537688		Raul Echevarria							
13. TECHNICAL DISCIPLINE DER-T (Structural Engineering) 14. SIGNATURE Digitally Signed 03/31/2023									
Solver (Condition of the Condition of th		Raul Echevarria 📴 👸	/31/2023	03/31/2023					
		esignee recommends approval abo	ove, or when approva	l is reserved for the FAA)					
17. PRINTED NAME/FAA	OFFICE	18. TECHNICAL DISCIPLINE							
19. SIGNATURE		20. DATE							
		and the state of t							

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ENGINEERING ORDER DATE EO No. **REVISION PAGE** 2023-B55-TE-387-01 Mar 29, 2023 IR OUTH PLAINS 1 AVIATION Prepared by: Date: Approved by: Date: J. Bolivar Mar 29, 2023 R. Echevarria Mar 29, 2023 TITLE: **Stringer Repair AIRCRAFT EFFECTIVITY:** LINE VARIABLE **CUSTOMER** MODEL MSN REGISTRY TAT TAC NUMBER **NUMBER** South Beechcraft **Plains** Baron TE-387 N71VK N/A N/A 4,108.6 N/A Aviation 95-C55 **DESCRIPTION:** Per OEM design, the lower stringer runs along the fuselage and ends on Carry Thru Spar horizontal flange. During repair of Carry Thru Spar on above aircraft in accordance with Beech SB 53-2669 R2, the lower stringer was accidentally trimmed out before the Carry thru Spar. Thus, in order to restore the installation, a repair angle made of .080" 2024-T3 aluminum sheet, is going to be used to splice the stringer with Carry thru Spar. Affected stringer has a thickness of .075" and the fuselage skin .032". **REFERENCES:** Beechcraft Baron 55 SRM latest version. 2. Beechcraft Baron 55 CCMH latest version. **TYPE OF ENGINEERING ORDER:** ☑ Modification or Alteration ☑ Major □ Repair ☐ Minor ☐ Inspection ☐ Other:

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ENGINEERING ORDER									
	EO No.	DATE	REVISION	PAGE					
South Plains AVIATION	2023-B55-TE-387-01	Mar 29, 2023	IR	2					

NOTES:

- 1. This Engineering Order is applicable to the referenced aircraft only.
- 2. No deviation from this procedure must be performed, unless is approved by DER.

WEIGHT AND BALANCE:

Not affected for this modification.

WORKINSTRUCTIONS:

No deviations from the following work instructions, are authorized without approval from Engineering

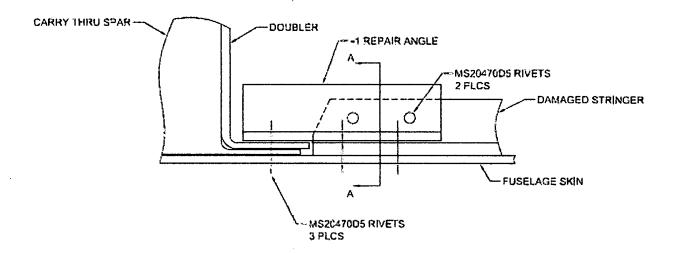
DISPOSITION	Mech	Insp
Remove enough system/ structure to get access to the Carry thru Spar, per applicable SRM/ AMM reference.	HH	HH
Perform a detailed visual inspection around the installation area, looking for any discrepancy.	H H	нн
3. Form -1 Repair Angle using a .080" 2024-T3 aluminum sheet.	HH	HH
4. Maintain a minimum .22R internal bend radius.	H H	H H
5. Brake sharp edges.	HH	HH
6. Temporarily install -1 Repair Angle over the stringer and Carry thru Spar and drill back from existing holes per Baron 55 SRM applicable reference. Ensure 3 fasteners common to the stringer horizontal flange.	A H	H H

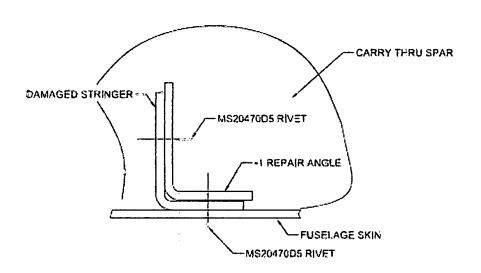
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South Plains AVIATION	2023-B55-TE-387-01	Mar	29, 2023	I	R	3
7. Drill Qty: 2EA new ho	oles to install MS20470D5-X ri	vets				
·	ngle vertical flange and string eron 55 SRM applicable refere		HH		Н	/4
8. Remove parts and de	eburr holes.		HH		H H	
Protect all bare meta layers of primer per s	I with a layer of Alodine and tandstandard practices.	:wo	AH		HH	
10. Apply P/S 870 sealan	nt to all mating surfaces. Ok to	o use	H H		H}	1
	le, using MS20470D5-X rivets prizontal flanges of the repair		H)*		HH	
12. Perform a general vis	sual inspection on reinforced a	areas	H H		μН	
13. Close area as require	ed.		H A		H F	\mathbf{f}

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ENGINEERING ORDER EO No. DATE REVISION PAGE OUTH PLAINS AVIATION. 2023-B55-TE-387-01 Mar 29, 2023 IR 4





SECTION A-A

Figure 1: Repair Sketch

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DATE:
3/30/2023

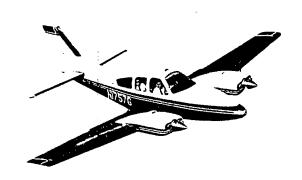
REPORT:
JSV-2023-003

MODEL:
Beechcraft Baron 95-C55

REPORT:
SUBSTANTIATION REPORT

CLASSIFICATION:
■ MAJOR □ MINOR

JSV AVIATION



BEECHCRAFT BARON 95-C55 STRINGER REPAIR STATIC SUBSTANTIATION

REPORT: JSV-2023-003

Rev. IR

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DATE: 3/30/2023	JSV Aviation	TYPE: ☑ STR □ SYS □ HYD
REPORT: JSV-2023-003		ENGINEER: S. Javadoff
MODEL: Beechcraft Baron 95-C55	SUBSTANTIATION REPORT	CLASSIFICATION: ■ MAJOR □MINOR

REVISION HISTORY	3
ABSTRACT	4
REFERENCES	5
ASSUMPTIONS & LIMITATIONS	6
GEOMETRY and DESCRIPTION	7
STRINGER REPAIR - STATIC ANALYSIS	10
LOAD PATH	10
ALLOWABLES	10
LOADS	11
ANALYSIS - STATIC	12

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DATE: 3/30/2023	JSV Aviation	TYPE: ⊠ STR □ SYS □ HYD
REPORT: JSV-2023-003		ENGINEER: S. Javadoff
MODEL: Beechcraft Baron 95-C55	SUBSTANTIATION REPORT	CLASSIFICATION: ■ MAJOR □MINOR

REVISION HISTORY

REPORT REVISION	REPORT DATE	ENGINEER	DEFINITION	CHANGES	APPROVAL
IR	03/30/2023	S. Javadoff	2023-B55-TE-387-01	Initial Release	R. Echevarria

DATE: 3/30/2023	JSV Aviation	TYPE: ⊠ STR □ SYS □ HYD
REPORT: JSV-2023-003		ENGINEER: S. Javadoff
MODEL: Beechcraft Baron 95-C55	SUBSTANTIATION REPORT	CLASSIFICATION:

ABSTRACT

This document provides static substantiation for repair to lower stringer of Beechcraft Baron 95-C55 a/c.

The 2023-B55-TE-387-01 Rev IRI Engineering Order provides dimensions, instructions and hardware requirements for this repair.

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DATE: 3/30/2023	JSV Aviation	TYPE: ⊠ STR □ SYS □ HYD
REPORT: JSV-2023-003		ENGINEER: S. Javadoff
MODEL: Beechcraft Baron 95-C55	SUBSTANTIATION REPORT	CLASSIFICATION: MAJOR DMINOR

REFERENCES

- 1. Beechcraft Baron 55 SRM latest version.
- 2. Beechcraft Baron 55 CCMH latest version.
- 3. MMPDS-06, Metallic Materials Properties Development and Standardization, April 2011

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DATE: 3/30/2023	JSV Aviation	TYPE: ⊠ STR □ SYS □ HYD
REPORT: JSV-2023-003		ENGINEER: S. Javadoff
MODEL: Beechcraft Baron 95-C55	SUBSTANTIATION REPOR	CLASSIFICATION: MAJOR DMINOR

ASSUMPTIONS & LIMITATIONS

- 1. Repair is done in accordance with a/c SRM
- 2. Material used for repair is the same as for original stringer.

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DATE: 3/30/2023	JSV Aviation	TYPE: ☑ STR ☐ SYS ☐ HYD
REPORT: JSV-2023-003		ENGINEER: S. Javadoff
MODEL: Beechcraft Baron 95-C55	SUBSTANTIATION REPORT	CLASSIFICATION: ■ MAJOR □MINOR

GEOMETRY and DESCRIPTION

Per OEM design, the lower stringer runs along the fuselage and ends on Carry Thru Spar horizontal flange.

AIRCRAFT EFFECTIVITY:									
CUSTOMER	MODEL	MSN	REGISTRY	LINE NUMBER	VARIABLE NUMBER	TAT	TAC		
South Plains Aviation	Beechcraft Baron 95-C55	TE-387	N71VK	N/A	N/A	4,108.6	N/A		

During repair of Carry Thru Spar on above aircraft in accordance with Beech SB 53-2669 R2, the lower stringer was accidentally trimmed out before the Carry thru Spar. Thus, in order to restore the installation, a repair angle made of .080" 2024-T3 aluminum sheet, is going to be used to splice the stringer with Carry thru Spar.

Affected stringer has a thickness of .075" and the fuselage skin .032".

DISPOSITION

- Remove enough system/ structure to get access to the Carry thru Spar, per applicable SRM/ AMM reference.
- Perform a detailed visual inspection around the installation area, looking for any discrepancy.
- Form -1 Repair Angle using a .080" 2024-T3 aluminum sheet.
- 4. Maintain a minimum .22R internal bend radius.
- 5. Brake sharp edges.
- Temporarily Install -1 Repair Angle over the stringer and Carry thru Spar and drill back from existing holes per Baron 55 SRM applicable reference. Ensure 3 (asteners common to the stringer horizontal flange.

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DATE: 3/30/2023	JSV Aviation	TYPE: ⊠ STR □ SYS □ HYD
REPORT: JSV-2023-003		ENGINEER: S. Javadoff
MODEL: Beechcraft Baron 95-C55	SUBSTANTIATION REPORT	CLASSIFICATION:

- Drill Qty: 2EA new holes to install MS20470D5-X rivets between -1 Repair Angle vertical flange and stringer vertical flange per Baron 55 SRM applicable references.
- 8. Remove parts and deburr holes.
- Protect all bare metal with a layer of Alodine and two layers of primer per standard practices.
- Apply P/S 870 sealant to all mating surfaces. Ok to use any approved alternative.
- 11. Install -1 Repair Angle, using MS20470D5-X rivets on vertical flange and horizontal flanges of the repair angle as per figure 1.
- 12. Perform a general visual inspection on reinforced areas and ensure all tools were removed.
- 13. Close area as required.

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DATE:
3/30/2023

REPORT:
JSV-2023-003

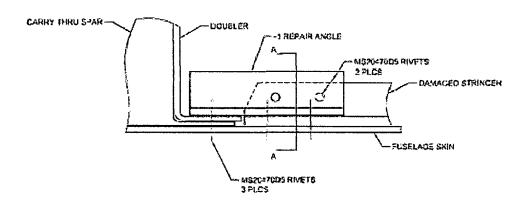
MODEL:
Beechcraft Baron 95-C55

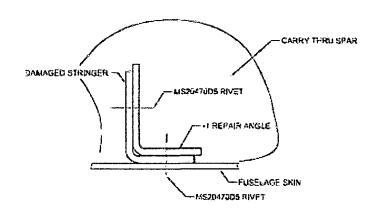
SUBSTANTIATION REPORT

TYPE:
STR □ SYS □ HYD

ENGINEER:
S. Javadoff

CLASSIFICATION:
MAJOR □ MINOR





SECTION A-A

Figure 1: Repair Sketch

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DATE: 3/30/2023	JSV Aviation	TYPE: ⊠ STR □ SYS □ HYD
REPORT: JSV-2023-003		ENGINEER: S. Javadoff
MODEL: Beechcraft Baron 95-C55	SUBSTANTIATION REPORT	CLASSIFICATION: ■ MAJOR □MINOR

STRINGER REPAIR - STATIC ANALYSIS

LOAD PATH

The critical load in the substantiation of this repair remains intact. Area lost is compensated by adding 0.080" thick -1 repair angle made from 2024-T3 Al sheet.

ALLOWABLES

AL SHEET:

2024-T3 AMS-QQ-A-250/4 per MMPDS-06 Table 3.2.4.0(b1)

FASTENERS:

Per MMPDS-06 Table 8.1.2 (b)

MS20470AD5-X

0.125 in, Solid Rivet

Table 8.1.2(b). Single Shear Strength of Solid Rivets°

U	Undriven Driven			River Size									
Rivet	Fa	(ksi)	Rivet	Fah (ksi)	Rivet	1/16	3/32	1/8	5/32	3/16	1/4	5/16	3/8
Material	Mm	Max	Material		Designation			Drive	n Single S	ear Streng	th, fbs		
5056-H32	2.4	n/a	5056-H321 ⁴	28°	B ^r	99	203	363	556	802	1450	2290	3275
2117-T4	26	n/a	2117-73	317	AD	106	217	389	596	560	1555	2455	3510
2017-T4	35	42	2017-773	38	Ð	1,34	275	493	755	1085	1970	3115	4445
2024-T4	37	n/a	2024-731	412	DD	145	297	532	814	1175	2125	3360	4795
7050-173	11	46	7050-17314	43°	E*	152	311	558	854	1230	2230	3520	5030
Monel	49	59	Monel	52"	M	183	376	674	1030	1490	2695	4260	6085
Ti-45Nb	50	59	Ti-45Nb	53"	'r	187	384	687	1050	1515	2745	4340	6200
A-286	85	95	A-286	90°		317	651	1165	1785	2575	4665	7375	10500

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DATE: 3/30/2023	JSV Aviation	TYPE: ⊠ STR □ SYS □ HYD
REPORT: JSV-2023-003		ENGINEER: S. Javadoff
MODEL: Beechcraft Baron 95-C55	SUBSTANTIATION REPORT	CLASSIFICATION: ■ MAJOR □MINOR

LOADS

Loads are not altered by this repair.

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DATE: 3/30/2023	JSV Aviation	TYPE: ⊠ STR □ SYS □ HYD		
REPORT: JSV-2023-003		ENGINEER: S. Javadoff		
MODEL: Beechcraft Baron 95-C55	SUBSTANTIATION REPORT	CLASSIFICATION: ■ MAJOR □MINOR		

ANALYSIS - STATIC

Repair performed per 2023-B55-TE-387-01 restores original structure capability by adding in more cross sectional area than is removed by trim out operation.

Trimmed out stringer is 0.075" thick 2024-T3 AL SH

-1 repair angle is 0.080" thick 2024-T3 AL SH

Table 3.2.4.0(b_1). Design Mechanical and Physical Properties of 2024 Aluminum Alloy Sheet and Plate

Specification	AM	S 4037	and AM:	AMS	AMS-QQ-A-250/4*			
Form	Sheet					Sheet		Plate
Temper			T:3	T361				
Thickness, in	0.008- 0.009 0.010-)-0.128	0.129 - 0.249		0.020- 0.062	0.063- 0.249	0.250+ 0.500
Basis	S	Α	В	A	В	S	S	S
Mechanical Properties:						T		
F_{α} , ksi;	1		į		l	1	Į.	Į.
Ü	64	64	65	64	66	68	69	67
LT	63	63	64	63	65	67	68	66
st								
$F_{\rm re}$, ksi:					!	1	1	1
Ĺ	47	47	48	47	48	56	56	54
LT	42	42	43	42	43	50	51	49
ST								
$F_{\rm eff}$ ksi:						1	ŀ	1
<u>Ľ</u>	39	39	40	39	40)	47	48	46
LT	45	45	46	45	46	53	54	52
ST					-:	1 :::	1	
$F_{\mathfrak{m}}^{\mathfrak{d}}$, kši	39	39	40	40	41	42	42	41
F_{tri}^{mba} , ksi:						1		
(e/D = 1.5)	104	104	106	106	107	111	112	109
(e/D = 2.0)	129	129	131	131	133	137	139	135
F_{hay} , ksi:				.		1	١.,	l
$(e/D = 1.5) \dots$	73	73	75 90	73	75	82	84	81
(e/D = 2.0)	88	88	90	88	90	1 9/	99	96
e, percent;	10	a		1		8	9	gr.
E. 10 ¹ ksi	—	لسسا		10.5		1 "	L	10.7
E_i , 10^3 ksi	1			10.7				10.7
G. 10' ksi	4.0							4.0
μ	0.33							0.33
	 		·-·					1 4.00
Physical Properties:	ļ				100			
ω, lb/in.	0,100							
C. K. and a	See Figure 3.2.4.0							

Revised: Apr 2008, MMPDS-04, Item 05-14.

a Mechanical properties were established under MIL-QQ-A-250/4.

b Gruin direction unknown.

c Bearing values are "dry pin" values per Section 1.4.7.1. See Tuble 3.1.2.1.1

d See Table 3.2.4.thc)

e 10/3 for 0.500 inch.

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DATE: 3/30/2023	JSV Aviation	TYPE: ⊠ STR □ SYS □ HYD
REPORT: JSV-2023-003		ENGINEER: S. Javadoff
MODEL: Beechcraft Baron 95-C55	SUBSTANTIATION REPORT	CLASSIFICATION:

Assuming that both t capping and repair angle are 1 x 1 inch angle the increase in area after repair is 6.7%

$$A_{orig} = (1 in + 1 in) \cdot 0.075 in = 0.15 in^2$$

 $A_{revair} = (1 in + 1 in) \cdot 0.080 in = 0.16 in^2$

Original load carrying capability is:

$$P_{orig} = A_{orig} \cdot F_{tu} = 0.15 \ in^2 \cdot 64,000 \ psi = 9600 \ lbf$$

Repair load carrying capability is:

$$P_{repair} = A_{repair} \cdot F_{tu} = 0.16 in^2 \cdot 64,000 psi = 10240 lbf$$

Original stringer was attached to carry through spar with 1 MS20470D5 rivet. The -1 repair angle as attached with four MS20470D5 rivets to the stringer and with 1 MS20470D5 rivet to the carry through spar thus load carrying capability is not affected by repair.

Repair to stringer on Beechcraft Baron 95-C55 a/c is structurally acceptable. No further analysis is required.

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US Department
of Transportation
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Administration

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

OMB No. 2120-0020 Exp: 5/31/2018	Electronic Tracking Number
	For FAA Use Only

INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for

Suc	tructions ch violat	s and dispos ion. (49 U.S	ition of this .C. §46301	form. (a))	This report is n	equir	ed b	y law (49 U.:	S.C. §44701).	. Fail	lure to rep	ort can re	sult in a civil penalty for each
			y and Regis	` ''	n Mark				Serial No.	•			
			'IVK	,	, man				i i	~~~	_ 307	,	
1. Air	rcraft	Make	1 7 1					TE-387					Series
		BEE	-ah						a	5-0	Baaru		
Name (As shown on registration certificate)								, <u></u>	Address (A			ristration (BARON certificate)
		,	Cottingham, Brandon							001	N at		St.
2. Ov	vner	Cott							city LAMESA				State TX
			J	,					Zip	79.		Count	
							3. F	or FAA Use	Only				
	4. Ty	pe					5. U	nit Identifica	ation				
Re	epair	Alteration	Unit			Mal	ke			ı	Model		Serial No.
			AIRFRAM	E					(As desci	ribed	in Item 1 a	above)	
		⊠	DOWEDD	POWERPLANT LEFT: Contine				۵۱	IO-520-C			1258167C	
		\square	FOVERF	LAINI	Right: Continental				IO-520-C			231 709 R	
			PROPELL	.ER	<u> </u>								
	APPLIANCE Manufacturer												
						6.		nformity Sta				·	
	· · · · ·	Name and A					B. k	Kind of Agend					
Name		Borg"	HOPF	rdD	AU		X		ated Mechanic			Man	ufacturer
City			ighwai	1 3.	Siale TX		\Box		oreign Certificated Mechanic C. Certificate No.				cate No.
Zip	•	nesa 331 <u>°</u>	untry <u>U</u>	SA	2.816	_			Repair Station Maintenance Or	rganiza	ation	AEP 3	3173828 IA
D.	have b	een made in	accordance	e with	on made to the u the requirement to the best of my	s of F	Part ₄	43 of the U.S	i 5 above and . Federal Avia	d desc ation	cribed on the Regulation	he reverse is and tha	e or attachments hereto at the information
per 1	nded rar 4 CFR I			Signa	ature/Date of Aut	horiz	ed In	ndividual					
App.	D				Maral	//	gu	soulse,	2				12-05-2022
								al for Return					
Pui Adi	rsuant t ministrat	to the authorion of the Fed	ority given deral Aviatio	perso on Adr	ns specified be ministration and i	elow, is	the	unit identif	ied in item :	5 wa	s inspecte		manner prescribed by the
BY -		AA Flt. Stand spector	It. Standards tor Manufacturer			Maintenance Org		rganization Persons Approve Department of Tr					
	F	AA Designee		Repa	air Station	Χ	Ins	pection Auth	orization	C	ther (Spec	cify)	
	icate or	. In		Signa	ture/Date of Aut	Horiz	ed Jr	ndividual					
-	nation N		TO !		3/ 1.1	(9)		11 1					
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Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

And Right Engine-Continental IO-520-C. SERICL*; 231709R. Air Battles Kits Were installed in reference to Drawing bist Du 87002, Revision 11 (Eleven) dated 03/20/2020 And Installation Oftion C. (Item 3 Installation Manual, listed on tront page of STC # SA 09220, using parts for Two Kits Cone for each engine, As listed by D'Shannon Products, PW Du 87002 OPTC, Serial #: 190 415-05, pages 1, 2+3 of DAC-3 Backing bist Engine Barron 55 Baffle OPTC Complete. Note: Copy of STC Number SA 092205C, Parts bist for Kits, PN Note: Copy of STC Mumber SA 092305C, Parts bist for Kits, PN Revision B for STC SA 092305C for Desen 55 have been included in Airplane bogs for NTIVK.
Air Baffles Kits WERE installed on both Engines of Beech Baron NTIVK: Left Engine - Continental IO-530-C. Serial #: 1858167C
Thennon troducts, bTD. 4309 County Road 134 Buffalo, MM 55313.
Trotallation of D'Shannon Products Cooling Air Baffles Kits in reference to STC Number SA09220 Certificate issued to:
12-05-2032 12-05-205-2032 12-05-205-2032 12-05-205-205-2052 12-05-205-205-205-205-205-205-205-205-205
8. Description of Work Accomplished (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Additional Sheets Are Attached

6
US Department
of Transportation
Federal Aviation
Administration

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form App OMB No. 11/30/200	2120-0020	Electronic Tracking Number
	F	or FAA Use Only

INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. §44701). Failure to report can result in a civil penalty for each such violation. (49 U.S.C. §46301(a)) Nationality and Registration Mark Serial No. USA N71VK TE-387 1. Aircraft Make Model Series **BEECH** 95-C55 Address (As shown on registration certificate)
Address PO BOX 179 Name (As shown on registration certificate) SOUTH PLAINS AVIATION TRAININGF CENT 2. Owner State TX Lamesa Country USA 79331 3. For FAA Use Only 5. Unit Identification 4. Type Repair Alteration Unit Make Model Serial No. (As described in Item 1 above) X AIRFRAME **POWERPLANT** PROPELLER Туре APPLIANCE Manufacturer 6. Conformity Statement A. Agency's Name and Address B. Kind of Agency Abilene Aero, Inc. U. S. Certificated Mechanic Manufacturer Address 2850 Airport Blvd. Foreign Certificated Mechanic C. Certificate No City Abilene TX Certificated Repair Station GSTR469E 79602 Country Certificated Maintenance Organization D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge. Signature/Date of Authorized Individual Extended range fuel per 14 CFR Part 43 Арр. В October 2, 2020 Stuart Douglas 7. Approval for Return to Service Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is X Approved Rejected FAA Flt. Standards Persons Approved by Canadian Manufacturer Maintenance Organization Department of Transport Inspector BY X FAA Designee Inspection Authorization Repair Station Certificate or Signature/Date of Authorized Individual Designation No.

Stuart Douglas

October 2, 2020

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

	USA N71VK	10/02/2020
	Nationality and Registration	Mark Date
talled GTN-750, GA-35 WAAS Antenna IAW Garmin STC# talled Garmin GTX-345 ADSB Transponder IAW Garmin ST is Installed ADS-B out system has shown to meet the equipm 227. talled Dual Garmin G5 HSI / ADI, Dual G5 Batterys, Single G	C# SA01714WI and Install ment performance requiren GAD-29B, Single GMU-11,	Manual 190-00734-10 Rev:14 nents of 14 CFR section and associated mounts,
rdware and wiring IAW Garmin STC # SA01818WI and Insta stalled Garmin EIS IAW Garmin Installation Manual P/N190-0 stalled Genesys 3100 Autopilot IAW Genesys Installation Ma	01717-00 Rev 13 Dated Ma	ay 31, 2020.
		·

US Department
of Transportation
Federal Aviation
Administration

ОМВ No. 2120-0020 Exp: 5/31/2018	Electronic Tracking Number
	For FAA Use Only

US Departme of Transporta Federal Avia Administrati	ation ation		ne, Powerplant, I			ance)			For FAA Use Only
instructio	CTIONS: Print ns and dispos ation. (49 U.S	ition of this	form. This report is	14 CF requir	FR §43.9, Part 43 red by law (49 U.S.	Appendix B, a C. §44701). F	and AC 43.9 ailure to rep	-1 (or sub ort can res	sequent revision thereof) for sult in a civil penalty for each
	Nationalit	y and Regi	stration Mark		Serial No.			7	
1. Aircraft	Make		CI			Model			Series
	Name (As	cher shown on	registration certificate	e)		Address (As	C55 shown on re	gistration d	certificate)
2. Owner	South	Plair	15 Aviatio	n 7		Address P.C.	Box 1	State TX	
					Center'	1	331	Countr	11 - 4
					3. For FAA Use O	nly			
l									
4. T	уре		-		5. Unit Identificat	ion			
Repair	Alteration	Uni	t	Ma	ke		Model		Serial No.
	×	AIRFRAN	IE			(As described in Item 1 above)			
		POWERP	LANT						
		PROPELI	_ER						
			Туре						
		APPLIAN	CE Manufacturer						
A Agonovida	Name and A	ddrana	· · · · · · · · · · · · · · · · · · ·	6	. Conformity State				
		hhhAD	Au		B. Kind of Agency U. S. Certificat			Manu	ufacturer
Address 21		Hwy.	3449		Foreign Certific	cated Mechanic		C. Certific	
	<u>Amesa</u> !33[_∝	untry US	State		Certificated Re	pair Station aintenance Organ	nization	A+P	3173818
D. I certi have	ify that the rep been made in	air and/or a accordanc		its of I	i) identified in item (above and de	escribed on t	he reverse	or attachments hereto
Extended raper 14 CFR App. B			Signature/Date of Au	sthoriz	zed Individual Kallylly	Ý			15 June 2020
					proval for Return()				
Pursuant Administr	to the authorator of the Feo	ority given deral Aviati	persons specified b on Administration and	elow, is	the unit identifie	d in item 5 v Approved	was inspect Rejecte		manner prescribed by the
1 1.	FAA Flt. Stand Inspector	lards	Manufacturer		Maintenance Org	anization		ns Approved tment of Tra	d by Canadian ansport
	FAA Designee		Repair Station	X	Inspection Author	ization	Other (Spe	cify)	
Certificate of Designation		3_	Signature/Date of A	thoriz	red Individual				15 June 2020

N.	O	T1/	~ =
13		I۱	JE.

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)
N71VK 15June 2020
Nationality and Registration Mark Date
Action:
Installed cockpit Sunvisor Assemblies in reference to
STC 3599 NM, Rosen Sunvisor Systems, using Rosen
Sunvisors, RBB-300-1A, SERIAL #: M234734
[Additional Shoots Are Attached

1372-1069 37 Apr 24'58

		APR 1	0 1	968	}		
⊋ \$PO	: 1965	i OF760962				FORM APPROVED BUDGET BUR. NO.	04-R06
		FEDERAL AVIA	TION	AGENC'	Υ		
		AIRCRAFT USE AND	INS	SPEC.	TION	REPORT	
		PRINT OR TYPE ALL ENT \$HADED BOXES	ARE I	OR FA	AA USE	TURES.	
		SECTION I—I	DEN	TIFICA	TION		
1. A	IRCR	AFT REGISTRATION MARK	AIRC	RAFT M	FR. SERIAI	L NO.	
١	1—	2786T		TE-	387		
2. A	IRCR.	AFT MAKE	AIRC	RAFT M	ODEL		
		Beech 🤞 🛣		95-	C-55		
3. 1	YAME	OF OWNER (As-shown on Certificate	of Re	gistratio	on)		
l co	mb	s Aircraft, Inc De	eale	er Co	ertif	. 68-37	
4. A	DDRI	ESS OF OWNER (As shown on Certifica	ate of	Registra	ation)	<u> </u>	
,	NUMB	ER AND STREET: Stapleton	[nt]	Ι Δ	irnor	٠	
	CITY:	Denver	STAT	E: /	<u>Color</u>		
	IRCR	Denver, AFT IS BASED AT: CO	UNTY:	Deni	COLOI		. وسراس
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·	CITY:	Denver STA	ATE OI	Co	lorad	ο	
		SECTION II—					
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7. P	RIMA	RY USE OF AIRCRAFT (Check one box of	nly)				
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2		BUSINESS TRANSPORTATION	7	11	NDUSTRIAL	., SPECIAL	
3	X	PERSONAL	8	0	THER (Sp	ecify)	
4		AERIAL APPLICATION					
5		INSTRUCTION	 		<u> </u>		-
	1	SECTION III—IN	SPEC	ном.	VREPORT	rator #100	
8. 1	AME	OF INSPECTING AGENCY				12.01 // 100	
	Cor	mbs Aircraft Inc.				:	
9. 1		INSPECTING AGENCY		CERTIFI	CATE NO	. AND TYPE	
3	· 🗆	INSPECTION 5 MANUFACTURER		Rei	pair	Sta. 4152	
	_	AUTHORIZATION 6 T REPAIR STATION	ı	•	lass		
10.	TYPE	OF INSPECTION DATE OF IN				N OF AIRCRAFT	
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				3 1 2		No.	

OFFICIAL BUSINESS

General Aviation District Office #3 Jefferson County Airport Broomfield, Colorado 80020 General Aviation District Office Services Broomfield, Colorado 80020

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FEDERAL VIATION AGENCY

MAJOR REPAIR AND ALTERATION

Form Approved Budget Bureau No. 04-R060.1 FOR FAA USE ONLY OFFICE IDENTIFICATION

		(Airrram	e, r	owerpiant, Pr	open	er,	or Applian	ice)			-	N-GADO -5-03
INST	rRUCT nstruction	IONS: Print o	or typ	oe all entries. See	FAR -	43.9,	FAR 43 Appe	endix B, and	AC 43.9-1 (or	subsequent	revision t	nereof)
	45.	MAKE		BEECHCRAFT				MODEL	95-C5	55		
1. AIRO	CKAFI	SERIAL NO.		TE-387				NATIONALI	Y AND REGIS		ARK	
2. OWI	NED	NAME (As sho		on registration certifi	cate)			ADDRESS (A	As shown on regi	stration certif	icate) T * L . A I	RPORT
2. 011	NLK	Сом	88	AIRCRAFT, I	1C -			Denver,	COLORADO	802	07	
					3	. F0	R FAA USE ON	LY	,			
				4. UN	IT IDE	NTIF	ICATION				5.	TYPE
UNIT MAKE							MODEL		SERIAL	NO.	REPAIR	ALTER- ATION
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PROPEL	LLER											
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APPLIA	INCE	MANUFACTURER						Verified	by Operato	or #100		
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at	tachme	nts hereto ĥav	e be	d/or alteration ma en made in accorda urnished herein is t	ance v	vith	the requirement	nts of Part 4	3 of the U.S. I			
DATE	C	DECEMBER	12,	1967		SIC	CHATURE OF	AUTHORIZED	INDIVIDUAL	-//		
				7	APPRO	VAL	FOR RETURN TO	OSERVICE		<i>—</i>		
Pursu the A	ant to	the authority trator of the F	giver eder	n persons specified al Aviation Agency	below and	, the	e unit identifie X APPROVE	ed in item 4 D RI	was inspected EJECTED	in the mann	er prescri	bed by
ВУ		FLT. STANDARDS		MANUFACTURER		INSI	PECTION AUTHORI	ZATION	OTHER (Specify)			
01	FAA	DESIGNEE	×	REPAIR STATION		OF	NADIAN DEPARTM TRANSPORT INSP AIRCRAFT				Λ	
DATE (REJECTI	ION	ROVAL OR 2-12-67		CERTIFICATE OR DESIGNATION NO 3932	Э.	-		AUTHORIZE	INDIVIDUAL	//		

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with air-craft nationality and registration mark and date work completed.)

THE FOLLOWING LISTED ELECTRONICS EQUIPMENT HAS BEEN INSTALLED. KING KXP-750 ATC TRANSPONDER AND KS-505 POWER SUPPLY MOUNTED ON BEECHCRAFT INSTALLED RADIO SHELF AND SECURED WITH 8.32 AIRCRAFT MACHINE SCREWS AND RIVNUTS. ALL RADIOS INSTALLED IN INSTALLED IN INSTALLED IN INSTALLED STROBE LIGHT SYSTEM INSTALLED WITH RADAIR INSTALLED IN INSTALLED WITH RADAIR INSTALLED WITH RADAIR STROBE LIGHT APPLICABLE PORTIONS OF AC 43.13-1 AND 43.13-2 HAVE BEEN COMPLIED WITH. AIRCRAFT FILE AND LOG CURRENTLY AMENDED.

0.09	Wire, plues, etc.
0.63	HOBBS HOUR METER
0.011	RADAIR STROBE LIGHT
0.261	RADAIR POWER SUPPLY
32.0	KING TRANSPONDER ANTENNA
0.061	МАРКЕЯ ВЕРСОИ АИТЕИИА
0.131	(MMOD) ANTENNA SI-A DAA
0.69	KING TRANSPONDER CONTROL
0.68	KING KI-2114 INDICATOR
0.69	KING KMA-128 AMPLIFIER
0.11 -	KING KA-22 ANTENNA
0.9 -	KING KXP-750 ATC TRANSPONDER
0 ° 6 -	KING KS-505 POWER SUPPLY
0.69	KING KK-160 VHF NAV/COM.
STATION INSTALLED AT	ITEMS INSTALLED

JAN 3 4 01 PM 68 OKLAHOMA GITY: OKLA

FEDERAL AVIATION AGENCY

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved Budget Bureau No. 04-R060.1

FOR FAA USE ONLY

OFFICE IDENTIFICATION DEN-GADS

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INSTRUC'	FIONS: Print or ions and disposi	typ	e all entries. See	FAR 4	43.9, FAR 43 App	endix B, and	AC 43.9-1 (or	subsequent i	evision t	hereof)
	MAKE Bee	ch				MODEL	95-C5	5		
1. AIRCRAFT	SERIAL NO.	38	7			NATIONALI	TY AND REGIS N 278	TRATION MA	\RK	
2 OWNED	· ·		n registration certific			SPESS (ton Inte	stration certification	cote) A	irpor
2. OWNER	Combs	Ai —	ecraft, In			Denver	, Colora			
				3	. FOR FAA USE ON	ILY				
			4. UN	IT IDE	NTIFICATION				5.	TYPE
UNIT		MA	KE		MODEL	_	SERIAL	NO.	REPAIR	ALTER- ATION
AIRFRAME	****	***	•••••••••••••••••••••••••••••••••••••••	As des	scribed in item 1 al	oove) ******	***********	••		x
POWERPLAN	r	•								
										
PROPELLER										
	TYPE				,					
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				6.	CONFORMITY STATE			·		<u> </u>
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	ir Corpor			•		FICATED MECHA		#3986	Jiai	LON
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DI-OOM.	Teru, co.				MANUFACTURE	R				
attachm	ents hereto have	bee	d/or alteration ma en made in accorda rnished herein is t	nce v	vith the requireme	ents of Part 4	i3 of the U.S. I	described or ederal Aviat	the revion Regu	erse or lations
DATE					SIGNATURE OF					
March	5, 1968		·		,		nuit			
D					VAL FOR RETURN T					had h
			persons specified al Aviation Agency		is APPROVE	D RI	EJECTED		er prescri	bed by
1 1	A FLT. STANDARDS		MANUFACTURER		INSPECTION AUTHOR		fied by Ope	rator #49		
	A DESIGNEE	X	REPAIR STATION		OF TRANSPORT INSE	PECTOR				
REJECTION,			CERTIFICATE OR DESIGNATION NO) .	SIGNATURE OF	AUTHORIZED	MAINTOUAL			

Meight and balance or operating limitation changes shall be entered in the appropriate aircraft record.

An alteration must be compatible, with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

craft nationality and registration mark and date work completed.) 8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with air-

Figures 2.1 and 3.1: Installed the following electronic equipment: I/A/W AC 43.13-2,

880196	<u> </u>	3 4166
674	63.0	3.7
<u> </u>	0:64	3370°0
Moment	Arma	Meight

KN-60, King DME Old Empty Weight

#4473, Radio, Class 1, 2 & 3. Instrument Limited. Operational and calibration checks performed by Air Radio, Inc., C.R.S.

Equipment list amended.

APR 1 0 1968

FEDERAL AVIATION AGENCY

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved Budget Bureau No. 04-R060.1

FOR FAA USE ONLY

OFFICE IDENTIFICATION DETAINED

~5**.03**

IN for	STRUCT instructi	CIONS: Print of ons and dispo	or typ	pe all entries. See	FAR 4	£3.9 ,	FAR 43 Appe	ndix B, and	AC 43.9-1	(or subsequent	revision t	hereof)		
		MAKE						MODEL						
, ,	DCDAET	F	eec	h					95-	C_55	•			
1. A	IRCRAFT	SERIAL NO.						NATIONALI	TY AND RE	C-55 GISTRATION M	ARK			
			T	E-387				N S	786T					
		NAME (As sho	own c	on registration certific	cate)					registration certif	îcate)			
2. 0	WNER	Combo A:	W ~ ~	- 61 T		Stapleton International Airport								
		Combe Ai	rcr	urt inc			Denver Colorado							
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				4. UN	IT IDE	NTIF	ICATION				5.	TYPE		
	UNIT MAKE						MODEL		SE	RIAL NO.	REPAIR	ALTER- ATION		
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∣ ⊅er.	verCo]	Lorado			_		MANUFACTURER			Clrs	5 3			
D.	attachme	nts hereto hav	e be	d/or alteration ma en made in accorda urnished herein is t	ince w	vith	the requiremen	its of Part 4	3 of the U					
DATE						SIG	NATURE OF	AUTHORIZED	V ÜNDIVIDU.	AL /				
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				7. 1	APPRO'	VAL	FOR RETURN TO	SERVICE	_O_	_//				
Pur the	suant to Adminis	the authority trator of the I	giver Feder	n persons specified al Aviation Agency	below and	the is x	APPROVEI	d in item 4	was inspec	ted in the mann	er prescri	bed by		
BY		FLT. STANDARDS PECTOR		MANUFACTURER			PECTION AUTHORIZ		OTHER (Special					
	FAA	DESIGNEE	x	REPAIR STATION		OF OF	IADIAN DEPARTME TRANSPORT INSPE AIRCRAFT			Operator #4	1	7		
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FAA	Form	337 (1-65) (OBSC	LETE PREVIOUS EL	OITIC	4		·	0	0052-025-	-8000 (8320)		

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

1. Install Dual Control Arm as per Beech drawing No 35-524662 Weight increase 5 lbs @ 72"

Aircrait New E.W. 3321.5 3 78.65 Woment of S62867.

END

APR 17 4 19 PM '68 OKLAHOMA : ...

■ ADDITIONAL SHEETS ARREST ATTACHED ALL CHED ALL CHEST
UNITED STATES OF AMERICA — FEDERAL AVIATION AGENCY

STANDARD AIRWORTHINESS CERTIFICATE

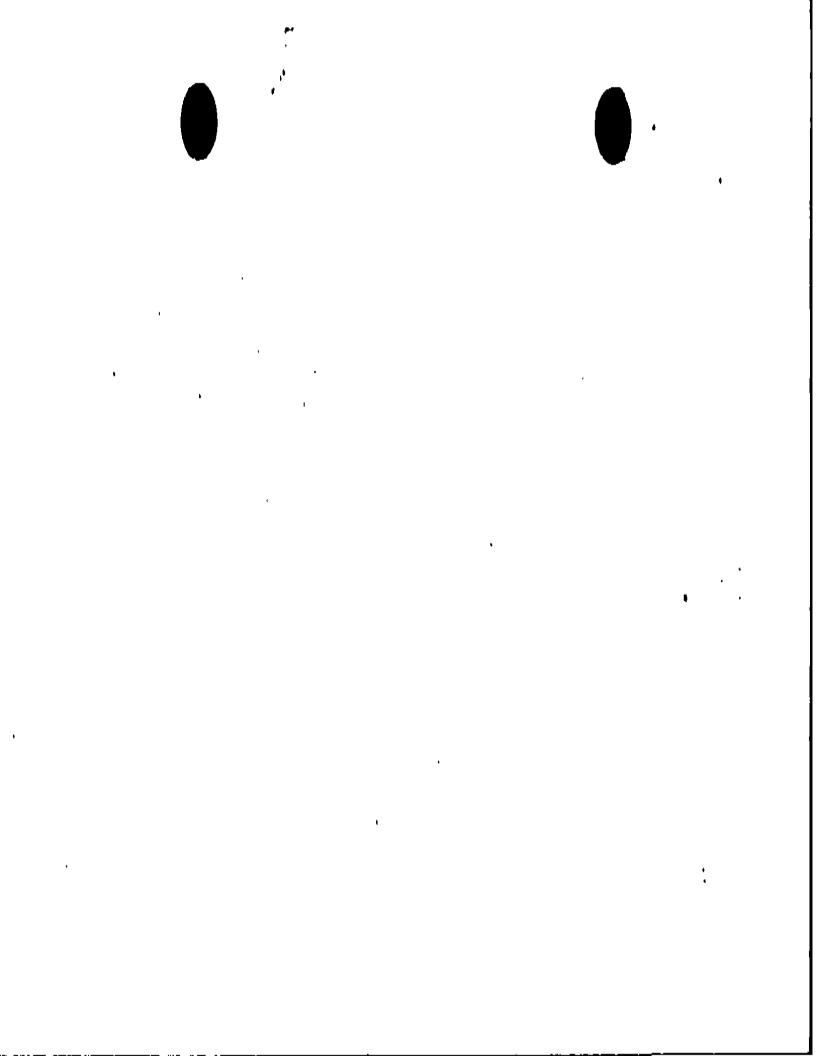
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1. NATIONALITY AND REGISTRATION MARKS	MANUFACTURER AI	ND MÓDEL JARIO	3. AIRCRAFT SER NUMBER	AL 4. CATEGORY
N2786T	BEECHORAF	r - 95-C35	TE-887	Normal
date of issuance, the therefor, to be in co	ertificate /is issued pu e aircraft/ to which i ondition/for/safe ope detailed/airworthines	ssued has been inspected the been straight and has been	ted and found to cont shown to meet the req	and certifies that, as of the form to the type certificate juirements of the applicable intion on International Civi
this airworthiness cer	dered, suspended, rev rtificate is effective a ance with Parts 21/14	s long as the maintena 3 and 3 of Feder	nce, preventative/maini	olished by the Administrator tenance, and alterations are s, as appropriate, and the
DATE OF ISSUANCE	EAA REPRESEN		Mary	DESIGNATION NUMBER
May 3, 1967	The ster FOR BEE		PORATION	DOA_CE-2
Any alteration, reproduc	tion, or misuse of th	is certificate may be p	unishable by a fine no	ot exceeding \$1,000, or im

prisonment not exceeding 3 years, or both. THIS CERTIFICATE MUST BE DISPLAYED IN THE AIRCRAFT IN ACCORDANCE WITH APPLICABLE FEDERAL AVIATION REGULATIONS.

FAA Form 1362 (7-65)

☆ GPO: 1966 O - 798-353

0052-040-8000 (8100)



1248 Com to the 14.67

		Al	PPLICATION F				S CER	TIFIC	CAI	Γ Ε		the tho	iTRUCTIONS: Print or to se are for FAA use or rized FAA Representat reverse side.	ly.	Submit o	original or	haded areas; aly to an au- e is required,
NO.			tration mark 786T	2. AIRCRAFT MA BEEC	KE CHCRA	FT			3. A		MODEL -055		IGNATION		FAA CI // (ODING	08
AFT IPTI			AFT SERIAL NO.	5. ENGINE MAK	Œ				6. E	NGINE A			GNATION		·•		·········
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- A			OF ENGINES	8. PROPELLER M					9. F	PROPELLER	MODE	L DE	SIGNATION		10. YR.	<i>-</i>	NUFACTURED
		2		McCar	ıley					2AF3	4C55	;				(19)67	
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8	4	_	AMENDMENT OR MODE	FICATION OF CL	IRRENT C	ERTIFICAT	r E			 						-1	<u> </u>
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ERT	3		(Indicate operation(s) to be conducted)		4	 	(Wild life	e conse	rvali	on)	5	<u>.</u>	PATROLLING	6	_ w	EATHER C	ONTROL
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INSPECTION AGENCY VERIFICATION						l											
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Q X		D. C	DATA, DRAWINGS, PHOT	OGRAPHS, ETC.	(Attach	when re	equired)		1	FAR _			CAR		_	(Orig	. attached)
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UNITED STATES OF AMERICA — FEDERAL AVIATION AGENCY

STANDARD AIRWORTHINESS CERTIFICATE

1. NATIONALITY AND REGISTRATION MARKS	2. MANUFACTURER	AND MODEL AT	3. AIRCRANUMBI	'	CATEGORY
N2786T	BEECHERAV	PT - 95-055		187	Vorma 1
date of issuance, the	ertificate/is issued e aircraft/to which ondition/for/safe o detailed/airworthin	pursuant to the Federal in issued has been in peration; and has been in ess/code as provided	spected and found een shown to meet	to conform to the requirements	he type certificat of the applicabl
6. TERMS AND CONDITION Unless sooner surrent this airworthiness center performed in accordance aircraft is registered.	dered, suspended, r rtificate is effective ance with Parts 2];	as long as the-main	tenance, preventativ	se established by	na faiterations of
May 3, 1967	FAA REPRES Choste FOR BL	MACH CHIBLE	Slic Lecrporat		TION NUMBER
Any alteration, reproduc	tion, or misuse of	this certificate may	be punishable by a	fine not exceedi	ng \$1,000, or im

Any alteration, reproduction, or misuse of this certificate may be punishable by a fine not exceeding \$1,000, or imprisonment not exceeding 3 years, or both. THIS CERTIFICATE MUST BE DISPLAYED IN THE AIRCRAFT IN ACCORDANCE WITH APPLICABLE FEDERAL AVIATION REGULATIONS.

FAA Form 1362 (7.65)

☆ GPO: 1986 O - 798-353

0052-040-8000 (8100)

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UNITED STATES OF AMERICA FEDERAL AVIATION AGENCY

EXPORT CERTIFICATE NO.

Form approved. Budget Bureau No. 04–RO59.

APPLICATION FOR EXPORT CERTIFICATE OF AIRWORTHINESS

INSTRUCTIONS.—This application is to be submitted to an authorized FAA representative (one copy) when the product(s) to be exported is (are) presented for inspection. Use Part I for Class I products and Part II for Class II. For complete aircraft execute items 1 through 11 as applicable. For engines and propellers, omit items 5A and 6. Part III is for FAA use only.

			ADTI					
			ART I					
1. APPLICATION IS NEW	MADE FOR AN EXPORT CERTIFICATE OF AI	RWORTHINESS TO	COVER THE PRODUCT(S) DESCRIBE	ED BELOW, WHICH	IS (ARE):			
2. NAME AND ADE	DRESS OF EXPORTER	3. NAME AN	ID ADDRESS OF FOREIGN PURCHA	ASER	4. COUNTRY OF DES	MOITANIT		
=	rcraft Inc. n Intl Airport Colorado	I	e Newall, Jr. ry, Alberta Canad	a	Canada			
5. DESCRIPTION OF	PRODUCT(S)							
		IDENTIFI-		FAA	OPERATING 1	IME (Hours)		
TYPE	MAKE AND MODEL	CATION NO.	SERIAL NUMBERS	SPEC. NO.	SINCE OVERHAUL	TOTAL		
a. AIRCRAFT	Beech 95C55	CF-DYN	TE-387	3A16		4243		
b. ENGINES	Continental 10-520-0	C	R. 125816-7-C	E5CE		42.3		
	Continental IO-520-0	C	L-125793-7-C					
	McCauley - 2AF-340	С55-Н	R-670103	P5 EA		42.3		
c. PROPELLERS			L-671042					
	VAS GIVEN A SATISFACTORY FLIGHT TEST	(D	ale)					
7. DOES THE PROD	UCT COMPLY WITH ALL APPLICABLE CIVIL NO (Explain in "Remarks")	AIR REGULATIONS,	AIRWORTHINESS DIRECTIVES, AN	ID OTHER FAA REG	DUIREMENTS?			
<u> </u>	E SPECIAL REQUIREMENTS OF THE IMPORT	ING COUNTRY BEEF	N COMPLIED WITH?					
☐ YES	NO (Explain in "Remarks")							
	VENTATIVE TREATMENT BEEN APPLIED TO	PRODUCTS SUSCEP	TIBLE TO RAPID CORROSION WHE	N REING SHIPPED	2			
YES	Пио				•			
10. REMARKS								
	ersigned certifies that the above sta operation except as may be noted u			scribed herein i	s (are) airworthy	and in con-		
	Ω	//						

4/8/68

(Date)

(Signature of Applicant or Authorized Representative)

Chief Inspector

(Title)

	PART II		
12. APPLICATION IS MADE FOR APPROVAL OF AERONAUTICA	L PARTS FOR EXPORT AS INDICATED BELOV	٧:	
13. NAME AND ADDRESS OF EXPORTER	14. NAME AND ADDRESS OF FOREIGN	PURCHASER	15. COUNTRY OF DESTINATION
16. PARTS ARE ELIGIBLE FOR INSTALLATION ON			
	(Make and Model Class I Prode	uct)	(FAA Spec. No.)
17. THE PARTS ARE (check one):	NEWLY OVERHAULED		
18. THE PARTS ARE DESCRIBED (check one):			7
_	• •	•	
BY NAME, PART NUMBER, AND QUANTITY ON THE ATTAC	HED INVOICE OR PACKING SHEET NO.		•
BELOW BY NAME, PART NUMBER, AND QUANTITY.			
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19. EXPORTER'S CERTIFICATION I certify that the foregoing statements are to	cue and that the parts described by	rain are airmorthy ar	nd conform to FAA approved
design data.	de and that the parts described he	stem are an worthy at	a comorni to PAA approved
(Date) (Signat	ture of Applicant or Authorized Representative)	(Title)
	PART III—APPROV	AL (FOR FAA USE	ONLY)
		-	
20. IT IS CONSIDERED THAT THE PRODUCT(S) DESCRIBED IN	(Part I or II) IS (ARE) AIR	WORTHY AND CONFORM(S) TO PERTINENT REQUIREMENTS
EXCEPT AS NOTED IN ITEM 10.			
(Signature)		(Number)	(Date)
	DMR FAA INSPECTOR		, -
Check One:	DMR FAA INSPECTOR		
21 APPROVAL TAGS, FORMS FAA-186, V	VERE ISSUED FOR THE PARTS DESCRIBED IN	I PART II.	
(Quantity) 22. EXPORT FILE SPOT-CHECKED BY:			
Se III			
Town Land			
Robert H. Lewis (FAA Supervising Inspector)		(D. O. No.)	4/10/68 (Date)
(FAA Supervising Inspector)		(D. O. 140.)	(Date)

The United States of America Jederal Aviation Agency

Washington, B.C.

No.	E-76710
•	

Export Certificate of Airworthiness

This contifies were the second	
the Tederal Aviation Agency, Numbered A-3A16,	undunvera particularly).describediin Geoification (si) ^l eg E-5CE-8, P-5EA-11
has been examined and as of the date of this cort	
.comprehensive.and.detailed.airworthiness.code.of.the	United States Government, and is in compliance with
.thase special requirements of the importing count	try filed with the United States Government, except
.asr.moted.below. This cortificate in mo way attests to	r.compliance.with.amy.agreements.or.contracts.between
the wender, and purchaser, mer does it constitute autho	ority.to.operate.am.aircraft.
Broduct. Airplane	Engine Installed: Continental IO-520-C
Manufacturer: Beech Aircraft Corp.	Serial No. R-125816-7-C L-125793-7-C
Model. 95-C55	
Lorial No. TB-387	Propeller Installed: McCauley 2AF-34C-55H
New 🗆 Newby Overhauled 🗆	Serial No. R-670103 L-671042
Used Livorafle 💍	
Country to which exported.	
Exceptions. None	
,	
William d. Stuart	
William A. Stewart	Federal Aviation Administration
4/10/68 Signature of Authorized Representative	Agency Represented and Number

Date

District Office or Designee Number

¹ For complete aircraft, list applicable specification or Type Certificate Data Sheet numbers for the aircraft, engine, and propeller. Applicable specifications or Type Certificate Data Sheet, if not attached to this export certificate, will have been forwarded to the appropriate governmental office of the importing country.

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Form Approved OMB No. 2120-0020 11/30/2007	Electronic Tracking Number
	or FAA Use Only

US Da			!	OLAM	R REPAIR AN	ID A	۱L	TERATION			11130120		For FAA Use Only
of Tra Feder	epartment nsportation ral Aviation nistration		(Airfra	ame, P	owerplant, Pr	ope	elle						
instr	uctions	ONS: Print and disposi n. (49 U.S.0	tion of th	nis form.	es. See Title 14 (This report is req	CFR Juire	¦ §₄ d b	43.9, Part 43 by law (49 U.S	Appendix E .C. §44701	3, and AC). Failure (43.9-1 to repo	(or sub	sequent revision thereof) for sult in a civil penalty for each
		Nationality United			n Mark N71V	K		,	Serial No TE-38				
1. Aire	craft	Make Beech	craft						Model 95-C5	 5			Series
Name (As shown on registration certificate) Phillips, Joe E									Address P	(As shown O Box 45 uidoso 3345	on reg 54		certificate) State NM
								or FAA Use		7343		Coun	try
							-						
	4. Typ	e			<u></u>		5. \	Unit Identifica	ition				
R€	pair	Alteration	U	Jnit		Mak	e e		*	Mod	lei		Serial No.
[X		AIRFRA	AME					(As described in Item 1 above)				
[POWE	RPLAN1	T								
[PROPE	ELLER									
[APPLIA	ANCE	Type Manufacturer								
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A. Ag		Name and A				\dashv	X X	Kind of Agend U. S. Certific	<u> </u>	ic		Mar	nufacturer
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Zip D.	72103	that the re	untry pair and/	or altera	ation made to the	unit(s) i	Certificated I	em 5 above	and desc	cribed o	n the rev	verse or attachments hereto
	have be	een made in	accorda	ance witl	h the requirements to the best of my k	of F	oar	t 43 of the U.S	i. Federal A	viation Re	gulatior	ns and the	at the information
	nded ran I4 CFR F B				nature/Date of Auth	1		Individual Liveu Melvern L.		Uls	<u></u>	-	
					<u> </u>		oro	val for Return)			
Pu Ad	rsuant t ministrat	o the auth lor of the Fe	ority giv deral Av	en pers		low,		ne unit identi		5 was	inspect Rejecte		e manner prescribed by th
		AA Flt. Stand	dards	Ма	nufacturer		м	laintenance O	rganization			ns Approve tment of T	ed by Canadian ransport
BY		AA Designe	e	Re	pair Station	х	lr	nspection Auth	orization	Othe	er (Spe	cify)	
	ficate or gnation t			Sigr	nature/Date of Aut	horiz	ed 2	Individual Lever	n d	Pela	· o	***	
1	176950	AIP		6.	entember 30, 201	<u>~</u>	t	Melvern		L			

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

3. Description of \((If more space i	Work Accomplished is required, attach additional sho	eets. Identify with aircraft n	ationality and registration mark and dat	e work completed.)
			United States N71VK	09/30/2010
			Nationality and Registration Mark	Date
Serial Number:	TE-387			
Repaired left ha	and elevator by installation of	the following parts:		
P/N 96-610006-9U 96-610006-11U 96-610006-3 45-610005-94	Upper skin, new from manuf Lower skin, new from manuf Nose skin, new from manuf Rib, new from manufacturer	acturer acturer		
Symmetry and a C55 Service Ma 4-57.	alignment were maintained du inual and Parts Catalog. All p	ring repairs with the use earts installed with origina	of fixtures. General procedures accordance with A	omplished referencing Beech 98 AC 43.13-1B, Chapter 4, Section
No change to er	mpty weight and balance.			
******	*********	***********Nothing Follow	S*****************************	**** *** *****
		•		
				•
			e e	
				•
		Additional Sh	aets Are Attached	

0 U.S. Department of Transportation

Federal Aviation Administration

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved

OMB No. 2120-0020

For FAA Use Only

Autimistratio	<i>J</i> 11									10011	UIII >M
			tries, See FAR 43.9,					or subsequent revisi	on thereof)	for instruction	ns
			ort is required by law ederal Aviation Act o			. 1421). Fallure u	o report can	1 result in a civii peri	alty not to e	xceea pi,uuu	1.00
	Make										
1. Aircraft	Beecho	raft	····					5-C55			
:	Serial No. TE-387	,					_	y and Registration N		174\ <i>I</i> Z	
<u>,</u>			on registration certific	nate)		j	1	nited States (As shown on reg		N71VK	
O Oumor	Phillips, Joe I		II I Ogioti daesi. e	Jaco,		1	P O Box	x 4554	listi duer.	unou.o,	
2. Owner						!	Ruidoso	o, NM 88345			
			3. For l	FAAI	lea!	A-1					
			J. 1011	FAA 、	JSe v	Only					
											}
			4. Unit	Ident	tifica	ation				5. Type	
Unit		Make	•			Model		Serial No		Repair	Alteration
VI						Piocoi		J	,. 	Пори	Alterane
AIRFRAME		~	~~~~~(As de	scrib	bed in Item 1 abov	/e) ~~~~	. ~ ~ ~ ~ ~			х
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POWERPLANT											
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PROPELLER											
	Туре										
APPLIANCE	Manufacturer										
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A Agency's Na	me and Address					Conformity State	ment		C. Certific	nN etc	
Vernon Ham			<u></u>		X	B. Kind of Agency X U.S. Certificated Mechanic					
1307 Ranso	m Road			İ	Foreign Certificated Mechanic			431417955			
Mena, AR 71	1953					Certificated Rep	pair_Station_			*****	
D. I certify that t	the renair and/or a	-Herat	tion made to the unit(/e) ide	ntifie	Manufacturer	and descri	ibed on the reverse (or attachme	nte hereto	
have been m	nade in accordance	e with	the requirements of	f Part	43 of						
furnished her	rein is true and cor	rrect	to the best of my kno	owled	ge. ——	<u> </u>					
Date OF 104	2/2005	_			Sign	nature of Authoriz	ed Individua	al/			
_ 05/06	6/2005				Ve	ernon Hampton	(1	The same of the sa			·
				7. /	Аррг	roval For Return	To Service	e			
			ons specified below, t Administration and i		it ide		-	=	escribed by	the	
		ation	Administration and i	15	ī	X APPROVE) [] F	REJECTED Specific			
1 }	Fit Standards ector		Manufacturer	х		pection Authorizat		Other (Specify	9		
FAA	A Designee		Repair Station		Car	rson Approved by nada Airworthines	s Group				
Date of Approval	=		Certificate or Designation No.	_	Sig	nature of Authoriz	ed Individu	 			
05/06	3/2005		421417055	,	14	Ht d					

Vernon Hampton (

431417955

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

B. Description of Work Accomplished (If more space is required, attach as Date: May 6, 2005	ed dditional sheets. Identify with Reg. # N71VK	aircraft nationality and registration mark and S/N TE-387	d date work completed.)
Installed new pilot's side wind dated October 17, 1996 and Certificate # SA00690CH, an	installation instructions:	cordance with Great Lakes Aero P #2AF dated 10/17/96 approved by	roducts drawing # W-2237 Supplemental Type
	6, 1998 and installation	in accordance with Great Lakes ae n instructions #A2 dated 10/17/96 a 999.	
Revised aircraft weight and b	alance		الم المستوالية المستوالية التي
*******		*****END*****************	*****
		RECEIVED	
		MAY N 9 2005	
		LIT-FSDO-11	
	•		
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Additional Sheets are Attached

U.S. Department of Transportation Federal Aviation Administration

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved OMB No. 2120-0020

For FAA Use Only

Office Identification

INSTRUCTIONS: Print or type all entries, See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000.00 for each such violation (Section 901 Federal Aviation Act of 1958).

for ea				ederal Aviation Act of			,	,. , u			- por	,		
		Make							Model					
4 A:	irome	Beecho	craft				95-C55							
1. A	ircraft	Serial No.							Nationality	y an	nd Registration I	Vlark		
		TE-387	<u>, </u>								ed States		N71VK	
2. O	wner	Name (As shown on registration certificate) Phillips, Joe E						Address (As shown on registration certificate) P O Box 4554 Ruidoso, NM 88345						
				3. For l	FAA U	se C	nly							
	· · · · · · · · · · · · · · · · · · ·			4 11-94	13				· Ladino, · · J				6 Type	
				4. Unit	Identi	ticat	ion			1			5. Type	T
Un	it		Make	•			٨	Model			Serial No	o. 	Repair	Alteration
AIRFF	RAME		~~~~~(,	As des	cribe	ed in i	ltem 1 abov	re) ~~~~	- ~ -	~ ~ ~ ~		х		
POWI	ERPLANT													
PROP	PELLER													
APPL	IANCE	Type Manufacturer												
		<u>. </u>				6 0	`onfo	rmity State	ement	l			-	1
A. Ad	ency's Na	me and Address						of Agency	Michie			C. Certifi	cate No.	
	on Ham					X U.S. Certificated Mechanic				A31A17055				
1307	7 Ranso	m Road				Foreign Certificated Mechanic					1 4314	431417955		
ivien	a, AR 7	1953				Certificated Repair Station				-				
D 1-	antific 44 - 4 4	ha ranair and/	alte1	ion made to the unit	(a) ida	otific		nufacturer	and dage=	ho	d on the reverse	or attachma	ante heroto	
ha	ve been m	ade in accordanc	e with	to the best of my known	Part	43 of								
Date	05/0	6/2005				_		of Authoriz	ed Individua	al				
					7. 4			For Return	To Service	e	- wye			
				ons specified below, Administration and	the un		ntified		vas inspecto	ed i	in the manner pr	rescribed by	/ the	
DV.		Fit Standards ector		Manufacturer	х	Inspection Authorization Other (Specify)						-		
BY	FAA	\ Designee		Repair Station				approved by Airworthines			,			
Date of Approval or Rejection 05/06/2005 Certificate or Designation No. 431417955						Signature of Authorized Individual Vernon Hampton								

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

the second secon
3. Description of Work Accomplished
(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)
Date: May 6, 2005 Registration: N71VK S/N: TE-387
Repaired belly section of airframe as follows:
Installed the following new parts:
P/N 002-410066-29 Keel, LH fwd - new from manufacturer
002-410066-33 Keel, LH aft - new from manufacturer
96-420001-121 Skin, lower fwd - new from manufacturer
96-410031-37 Skin, lower LH fwd - new from manufacturer
*95-420015-55 Skin, lower center - locally fabricated
*96-420001-67 Skin, lower LH mid - locally fabricated
*96-420001-68 Skin, lower RH mid - locally fabricated.
Repaired lower flanges of belly bulkheads at station 68.00, station 82.96 and station 83.09 by installation of repair angles fabricated from 2024-T3 .032 alloy using procedures outlined in AC 43.13-1B, Chapter 4, Sections 4-50, 4-53 4-57 and 4-58.
General procedures were followed referencing Beechcraft 55 Shop Manual and Parts Catalog. All new parts installed using original type rivets in accordance with AC 43.13-1B, Chapter 4, Section 4-57. Some skins identified above with an (*) are locally fabricated from original type and thickness material using original skins as templates and procedures outlined in AC 43.13-1B, Chapter 4, Section 4-50 and 4-53.
No change to aircraft weight and balance.

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Additional Sheets are Attached

U.S. Department of Transportation

Federal Aviation Administration

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020

For FAA Use Only

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INSTRUCTIONS: Print or type all entries, See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000.00 for each such violation (Section 901 Federal Aviation Act of 1958).

for each such violation (Section 901 Federal Aviation Act of 1958).										
Make				Model						
Beechcraf										
Serial No.								トリフィンル		
				<u> </u>						
Phillips, Joe E				P O Box 4554 Ruidoso, NM 88345						
3. For FAA Use Only										
4. Unit Ident				ification			5. Type			
Ma	Make		Model Serial N).	Repair	Alteration		
~~~~~~ ( As de:				scribed in Item 1 above)~~~~~~~~~					х	
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Type										
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					ement					
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Vernon Hampton								431417955		
1307 Ransom Road										
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made in accordance w	ith the requirements	of Part	43 of						·	
			Sigr	nature of Authoriz	ed Individu	ıal				
05/06/2005 Vernon Hampton										
					_			<del></del>		
			nit ide		_	REJECTED		ıne		
1	Manufacturer   Y			Inspection Authorization Other (Specify)						
AA Designee Repair Station			Person Approved by Transport Canada Airworthiness Group							
Designation No.										
	Make Beechcraft Serial No. TE-387 Name (As shown Phillips, Joe E  Type Manufacturer  Mame and Address mpton com Road 71953 at the repair and/or alter made in accordance wherein is true and correct to 6/2005  the authority given person of the Sepector AA Designee val or Rejection	violation (Section 901 Federal Aviation Act  Make Beechcraft Serial No. TE-387 Name (As shown on registration certify Phillips, Joe E  3. For Make  Make  A Un  Make  4. Un  Make  Type  Manufacturer   Make Beechcraft Serial No. TE-387 Name (As shown on registration certificate) Phillips, Joe E  3. For FAA  Type Manufacturer  Mame and Address mpton  oom Road 71953 It the repair and/or alteration made to the unit(s) idenade in accordance with the requirements of Partnerein is true and correct to the best of my knowled  06/2005  7. the authority given persons specified below, the unit or of the Federal Aviation Administration and is the Fit Standards spector  AA Designee  Repair Station  Read Carolifo Certificate or Designation No.  Certificate or Designation No.	Wiolation (Section 901 Federal Aviation Act of 1958).  Make Beechcraft Serial No. TE-387 Name (As shown on registration certificate) Phillips, Joe E  4. Unit Identificat  Make  A Unit Identificat  Make  Type Manufacturer  6. Clame and Address Mpton  Com Road 71953 It the repair and/or alteration made to the unit(s) identifier made in accordance with the requirements of Part 43 of the rein is true and correct to the best of my knowledge.  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For FAA Use Only  4. Unit Identification  Make Model	Make Beechcraft 99  Serial No. TE-387 National U  Name (As shown on registration certificate) Phillips, Joe E Ruidosi  4. Unit Identification  Make Model  4. Unit Identification  Make Model	Make   Beechcraft   95-C55	Make Beechcraft Serial No. TE-387 Name (As shown on registration certificate) Phillips, Joe E  4. Unit Identification Make Model PO Box 4554 Ruidoso, NM 88345  3. For FAA Use Only  4. Unit Identification Make Model Serial No.  (As described in Item 1 above)  Type Manufacturer  6. Conformity Statement Jame and Address B. Kind of Agency Foreign Certificated Mechanic Foreign Certificated Mechanic Certificated Repair Station Certificated Repair Station Manufacturer  7. Approval For Return To Service the authority given persons specified below, the unit identified in Item 4 was inspected in the manner prescribed by Vernon Hampton  7. Approval For Return To Service the authority given persons specified below, the unit identified in Item 4 was inspected in the manner prescribed by Vernon Hampton  7. Approval For Return To Service the authority given persons specified below, the unit identified in Item 4 was inspected in the manner prescribed by APROVED  REJECTED  Other (Specify) APROVED  REJECTED  Other (Specify) APPROVED  REJECTED  Other (Specify) APPROVED  Relation Authorization and a Signature of Authorization Individual Certificate or Designation No.  Signature of Authorization Certificate or Designation No.  Signature of Authorization Individual Certificate or Designation No.  Signature of Authorization Individual Certificate or Designation No.  Signature of Authorization Individual Certificate or Designation No.  Signature of Authorization Individual Certificate or Designation No.  Signature of Authorization Individual Certificate or Designation No.  Signature of Authorization Individual	Make   Beechcraft   Serial No.   TE-387   Nationality and Registration Mark United States   N71VK		

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished (If more space is required, attach additional sheets. Identify with aircraft nationality and registration Date: May 6, 2005 Reg. # N71VK S/N TE-387	mark and date work completed.)						
Installed battery vent fairing in accordance with South Seas Ventures, Inc. Draw 5/17/01 and approved by Supplemental Type Certificate #SA01151CH, dated A							
Revised pilot's operating handbook with Supplement #P/M002-40001-33.							
Revised aircraft weight and balance.							
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	MAY n 9 2005						
	LIT-FSDO-11						
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Additional Sheets are Attached							

U.S. Department of Transportation Federal Aviation Administration

### MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved OMB No.2120-0020

For FAA Use Only

Office Identification

ELOCELUS

mula INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958). Model Make 1. Aircraft **RATHEON** 95-C55 Serial No. Nationality and Registration Mark **TE387** N71VK Address (As shown on registration certificate) Name (As shown on registration certificate) Owner **EDWARD JOE PHILLIPS** PO BOX 4554 **RUIDODO NM 88345** 3. For FAA Use Only 4. Unit Identification 5. Type Serial No. Repair Alteration Unit Make Model X **AIRFRAME**  (As described in item 1 above)-**POWERPLANT PROPELLER** Type APPLIANCE Manufacturer 6. Conformity Statement C. Certificate No. B. Kind of Agency A. Agency's Name and Address U.S. Certificated Mechanic 2148062 **GENE SEITZ** 2313-58TH Foreign Certificated Mechanic LUBBOCK, TEXAS 79412 Certificated Repair Station Manufacturer D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge. Signature of Authorized Individual Date 11-9-2001 7. Approval for Return To Service Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is

APPROVED

REJECTED the Administrator of the Federal Aviation Administration and is Other (Specify) FAA Fit. Standards Manufacturer Inspection Authorization BY Inspector Person Appoved by Transport FAA Designee Repair Station Canada Airworthiness Group Date of Approval or Rejection Certificate or Designation No. Signature of Authorized Individual

2148062

11-26-2001



Fried Stries et America Bepartment of Transportation - Federal Abiation Administration

# Supplemental Type Certificate (Continuation Sheet)

Number SADZOJSAT

Date of Issuance: November 19, 1999 Date Amended: December 21, 2000 Date Reissued: January 18, 2001

#### Model 308 Eligibility Continued:

Piper Aircraft Corporation (cont'd):

1A4	PA-20, -20S, -20-"115", -20S-"115", -20-"135", -20S-"135"	CAR3
1A6	PA-22, -22-108, -22-135, -225-135, -22-150, -225-150, -22-160, -225-160	CAR3
1A10	PA-23, PA-23-160, PA-23-235, PA-23-250 (Narry UO-1), PA-E23-250	CAR3
AIEA	PA-30, PA-39, PA-40	CAR3, FAR23
A20SO	PA-31, PA-31-300, PA-31-325, PA-31-350	CAR 3, FAR 23
A7SO	PA34-200, -200T, -220T (Seneca III), -220T (Seneca IV), -220T (Seneca V)	FAR 23
A9SO	PA-36-285, PA-36-300	FAR 23
A18SO	PA-38-112	FAR 23, 36
A1980	PA-44-180,-44-180T	FAR 23
A2590	PA-46-310P, PA-46-350P	FAR 23, 25, 36
Raytheon Aire	craft Company (Formerly Beach Aistraft Corporation):	
A777	35, A35, B35, C35, D35, E35, G35 F35, 35R	CAR3
3A15	P35, S35, V35A, 36, A36	CAR 3
3A16	95-C35, 95-C35A, D55, D55A, E55, E55A, 5%, 5%A	CAR3
AICE	19A, B19, M19A, 23, A23, A23A, A23-19, A23-24, B23, C23, A24, A24R, B24R, C24R	CAR3
3A15	35-33, 35-A33, 35-B33, 35-C33, 35-C33A, E33, E33A, E33C, F33, F33A, F35C, G33, H35, J35, K35, M35, N35, P35, S35, V35, V35A, V35B, 36, A36, A36TC, B36TC	CAR 3, FAR 23, FAR 36

May alteration of this certificate as purished by a fine of met encouring \$1,000, or imprisonment met exceeding 3 years, or both.

TAX FORM \$119-2-1(10-43) FACE 9 45 13 FACES

This certificate may be transferred in accordance with fix 71.47.



## MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved OMB No. 2120-0020

For FAA Use Only

Office Identification
50 FSD005 MC)

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

for	each suc	h violation (Sec	ction 901 Federal A	viation A	ct of	1958).		cportcarresarrira	orvii peria	ity not to exc	cca ψ1,000
		Make BEECHCR	RAFT				Model 950	C55			
1. Aı	ircraft	Serial No. TE-387		· · · · · ·				lity and Registratio	n Mark		
		Name (As sho	own on registration	certifica	te)			s (As shown on regi		ertificate)	
2. Qi	wner	Chris J	J. Price					0 O'Briant C			
							Gree	ensboro, NC	27410		
			•		3.	For FAA Use	Only				
											:
		T			4.	Unit Identific	cation	T		5. Type	<del></del>
1	Unit		Make			Model		Serial No	).	Repair	Alteration
AIRFF	RAME			⊶(As de:	scribe	ed in Item 1 a	bove)				х
POWE	ERPLANT										
PROP	PELLER	,					,				
		Туре			_						
APPL	IANCE	Manufacturer		_							
					6. C	onformity Sta	atement				
A. A	gency's N	ame and Addre	ss			Kind of Age			C. Certi	ificate No.	
Ch	ris J.	Price			x		cated Mechanic		1		
		riant Cour			<u> </u>	<del></del>	rtificated Mech		4		
Gr	eensbor	ro, NC 274	-10		$\vdash$	Manufacture	Repair Station	<u>n</u>	3055	69359	
	1		- 1/ac alteration ma		-:4/5				<u> </u>		
	have beer	n made in acco		quirement	its of F y knov	Part 43 of the wledge.	U.S. Federa	e and described on t al Aviation Regulatio			
Date					Sig	gnature of Au	thorized Inc	lividual			
15	Septer	mber 1999				- 1h	is At				
				7. A	\ppro	val for Return	n To Service				
Pui	rsuant to ministrato	the authority g	given persons spec I Aviation Adminis	ified belo	ow, th	he unit identi	ified in item VED	4 was inspected in REJECTED		iner prescrib	ed by the
вү		Fit. Standards ector	Manufacturer	x	Ins	spection Autho	orization	Other (Specify)			
	ll	Designee	Repair Station		Car	rson Approved nada Airworth	ninéss Group				
Date o	of Approva Septer	alorRejection mber 1999	Certificate or Designation N	۱o.	Sig	gnature of Au	thorized Indi	ividual			
			130330333	/ 1.21		1/2	J4 ()	n.			

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished
(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

REMOVED BRITIAN B-S AUTOPILOT SYSTEMS AND ALL COMPONENTS FROM AIRCRAFT.

INSTALLED S-TEC 50 TWO AXIS AUTOMATIC FLIGHT GUIDANCE SYSTEM MODEL ST-223. THE INSTALLATION WAS PER SUPPLEMENTAL TYPE CERTIFICATE NUMBER SA5290SW-D. ALSO USED WAS BULLETIN NO. 323, REVISION 3, DATED 1-23-96 AND MASTER DRAWING LIST NO. 92251, REVISION C, DATED 1-23-96. FAA APPROVED SUPPLEMENT TO PILOT'S OPERATING HANDBOOK OR FAA APPROVED FLIGHT MANUAL P/N 89349, DATED 3-27-84 WAS INSTALLED IN AIRCRAFT.

THE ABOVE INSTALLATION MEETS OR EXCEEDS THE APPLICABLE SECTIONS OF AC 43.13-1B AND AC 43.13-2A AS A REFERENCE GUIDE. THE AIRCRAFT EQUIPMENT LIST WAS REVISED AND WEIGHT AND BALANCE DATA UPDATED. AN ENTRY WAS MADE IN THE PERMANENT LOGS CONCERNING THE INSTALLATION.

-----END------

## US Department of Transportation Federal Aviation Administration

**MAJOR REPAIR AND ALTERATION** (Airframe, Powerplant, Propeller, or Appliance) Form Approved OMB No. 2120-0020

For FAA Use Only

Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000

fo	r each su	ich violation (Se	ectio	n 901 Federal Aviat	ion	Act	t of 1	958).		•		• · · · · · · · · · · · · · · · · · · ·			
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1. A	ircraft	BEECHCR Serial No.	ALI	<u></u>				*****	95C5		- 14x 4:				
		TE-387							Nationality and Registration Mark N71VK						
			own	on registration cer	tific	cate	<u> </u>			(As shown on reg	istration o	ertificate)			
	\	Chris J					•			O'Briant C					
<b>2.</b> O	wner									nsboro, NC					
							3. F	For FAA Use Or	nly -						
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							4. l	Unit Identificati	on			5. Type			
	Unit	• •	Ма	ıke				Model		Serial No	D.	Repair	Alteration		
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		Туре													
APPI	IANCE														
•		Manufacture	r												
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A. A	gency's l	Name and Addre	ess				T	Kind of Agency			C. Certi	ficate No.			
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U.	nave be	en made in acco	ordar	r alteration made to nce with the require correct to the best	eme	ents	of Pa	art 43 of the U.S	n 4 above : S. Federal	and described on t Aviation Regulati	he reverse ons and th	or attachme at the inform	nts hereto nation		
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				·						vidual///-	#	,	ł		
	6/	1/99					Cl	hris J. Pr	ice	1 hos	フル	11	-		
								al for Return To							
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3Y		A Flt. Standards pector		Manufacturer		Х	Insp	ection Authoriza	ıtion	Other (Specify)			· · · · · · · · · · · · · · · · · · ·		
<i>-</i> 1	FA	A Designee		Repair Station				on Approved by ada Airworthines			_				
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	6/1/	99		305569359			Cl	nris J. Pr	ice	11	11/				

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished (If more space is required, attach additional action)	tional sheets. Identify with aircraft nationality	and registration mark and date work completed.)
REMOVED ORIGINAL ROTATING	G BEACON AND INSTALLED WHELEN INSTALLATION IS PER STC SA61	N DIRECT REPLACEMENT ANTI-
AND SERVICE MANUAL DATED	AUGUST 1, 1996. THE WORK MI	EETS OR EXCEEDS ALL APPLICABLE
SECTIONS OF AC 43.13-1B WERE UPDATED. AN ENTRY I	AND AC 43.13-2A. THE EQUIPM WAS MADE IN THE PERMANENT LO	MENT LIST AND WEIGHT AND BALANCE GS CONCERNING THIS INSTALLATION.
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	-   Additional Sheets Are Attach	ned

# US Department of Transportation

of Transportation
Federal Aviation
Administration

## MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved OMB No. 2120-0020

For FAA Use Only

Office Identification SO 4SDO 05

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

for	each suc	h violation (Se	ction	n 901 Federal Avial	tion Ac	w (49 tot of 1	958).	. 1421). га	nare to f	ιeħ.	on can rest	uitifia	civii pena	uty not to exc	eea \$ 1,000
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A	ail	Serial No.							Nationa	ality	y and Regis	stratio	n Mark		
		TE-387							N71	_					
	_	1		on registration cer	rtificat	e) ~					As shown c	_		ertificate)	
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									Gre	en	sboro,	NC 2	27410		
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Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8	Description of Work Accomplished  (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)  REMOVED THE FOLLOWING RADIOS: TELEX INTERCOM, BENDIX T12D ADF SYSTEMS (2EA) AND  FOSTER LRN 500 LORAN. INSTALLED THE FOLLOWING RADIOS: KR87 ADF RECEIVER, KI227 ADF  INDICATOR, KA44B ADF LOOP ANTENNA, KN64 DME RECEIVER, AND PS-3000 INTERCOM SYSTEM.  THE RADIOS WERE INSTALLED IN ACCORDANCE WITH MANUFACTURER'S INSTALLATION MANUAL  AND USING APPLICABLE SECTIONS OF AC 43.13-1B AND AC 43.13-2A AS A REFERENCE GUIDE  TO APPROVED INSTALLATION PRACTICES. THE EQUIPMENT INSTALLED WAS RAMP TESTED AND  FOUND TO FUNCTION PROPERLY WITH NO INTERFERENCE TO ANY OTHER SYSTEM OR FLIGHT  CONTROL. THE EQUIPMENT LIST WAS AMENDED AND WEIGHT AND BALANCE DATA UPDATED. AN  ENTRY WAS MADE IN THE PERMANENT LOGS CONCERNING THIS INSTALLATION.
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1	
	☐ Additional Sheets Are Attached

# US Department of Transportation

### MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved

OMB No. 2120-0020

For FAA Use Only Office Identification

	eral Aviation	**************************************	÷		Ť.	•			Identification - FSD0 - 13	
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and	disposition of	of this form. This n (Section 901 of	s repor	ort is required by law ( eral Aviation Act of 19	(49 U.S.	.C. 1421). Failure to r	report can resu	ult in a civil penalty not t	to exceed \$1,000	
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ľ	made in acc	cordance with the	e requ	ration made to the ur uirements of Part 43 ect to the best of my I	of the U	U.S. Federal Aviation I	e and describe Regulations ar	ed on the reverse or atta nd that the information	achments hereto h	nave been
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BY	<u> </u>	Designee		Repair Station		Person Approved by Canada Airworthines	ss Group		<u> </u>	
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	NOTICE
_	Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.
	3. Description of Work Accomplished If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)
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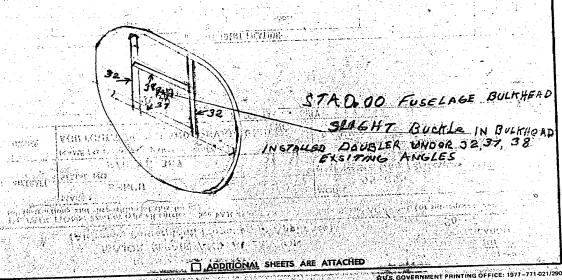
Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

#### 337 REPAIR

- REPAIRED BUCKLED SKIN WEB OF FORWARD BULKHEAD STA. 0.00 IN ACCORDANCE WITH ARMY AVIATION MAINTENCE ENGINEERING MANUAL AIRCRAFT STRUCTURE. REPAIRED PER TM 55-405-4 AND APPLICABLE PORTIONS OF AC 42.13-1A. MATERIAL SAME GAUGE AND ALLOY AS ORIGINAL 2024T3. RIVETS WERE SOLID 2117-T4 AN470-4 RIVET PATERN SAME AS ORIGINAL.
- REPLACED DAMAGED OUTBOARD LEFT HAND TRAILING EDGE WING SKIN P/N 95-110014-199 AND LOWER OUTBOARD TRAILING EDGE SKIN P/N 95-110014-211 FACTORY PARTS WERE USED. All 2. RIVETS FOR REPAIR SAME AS ORIGINAL ALSO USING PORTIONS OF AC 43.13-1A.

SEE DRAWING BELOW ON BULKHEAD REPAIR. CHAPTER 2 AC 43.13-1A METAL STRUCTURES SECTION 3 METAL REPAIR PROCEDURES PAR. 96 PAR 97.



US Departmen of Transportate	MAJOR F	REPAIR	ND ALTERAT	ION	OMB	opproved No. 2120-0020	
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Person Approved by Transport Canada Airworthiness Group Signature of Authorized Individual

Repair Station

Certificate or Designation No. IA1046134

Date of Approval or Rejection

7/5/93

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

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US Deportment . of Transportation
Federal Aviation Administration

### MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved OMB No. 2120-0020 For FAA Use Only
Office Identification

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FAA Form 337 (12-88)

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

lea. KING KMA-12B AUDIO/MARKER SYSTEM

lea. KING KX-160 NAV/COM SYSTEM

lea. SPERRY C-2 RMI/COMPASS SYSTEM

lea. KING KMA-24 AUDIO/MARKER SYSTEM
lea. KING KX-155 COM/NAV/GS SYSTEM WITH KI-209 VOR/ILS INDICATOR
lea. BFG WX-900 STORMSCOPE SYSTEM
lea. TELEX TC-200 INTERCOM SYSTEM
lea. DAVTRON 655-2 DENSITY ALTITUDE INDICATOR
lea. TROLL FN-200 FAN

SYSTEMS INSTALLED IN ACCORDANCE WITH MAUFACTURERS SPECIFICATIONS AND AC 43.13-2A.

PANEL UNITS MOUNTED IN RADIO STACK AT STATION 63.0 AS PER PARAGRAPH 23 OF AC 43.13-2A. USED EXISTING ANTENNA SYSTEMS ON AIRCRAFT.

STORMSCOPE ANTENNA MOUNTED ON BELLY AT STATION 238.0 AS PER PARAGRAPH 43 OF AC43.13-2A.

COOLING FAN MOUNTED IN BAGGAGE COMPARTMENT AT STATION 40.0 AS PER PARAGRAPH 25 OF AC 43.13-2A.

ALL WORK PERFORMED USING ACCEPTABLE METHODS, PRACTICES AND TECHNIQUES OUTLINED IN CHAPTER 11 OF AC 43.13-14 AND CHAPTERS 1.2 & 3 OF AC 43.13-2A

EQUIPMENT LIST AND WEIGHT AND BALANCE WERE REVISED TO REFLECT CHANGES TOTAL ELECTRICAL LOAD DOES NOT EXCEED 80% OF TOTAL AIRCRAFT LOAD.

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SYSTEMS FUNCTION TESTED AND FOUND TO PERFORM SATISFACTORLY. ----END----

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Additional Sheets Are Attached

# U.S.GPO:1990-0-568-012/40004

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DEPARTMENT OF TRANSPORTATION—FEDERAL AVIATION ADMINISTRATION

STANDARD AIR WORTHINESS CERTIFICATE

I. NATIONALITY AND AIR SECONDARY

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S. AUTHORITY AND BASIS FORTSSUIJES

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Aviation, except as noted better.

S. TERMS AND CONDITIONS

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AND CONDITIONS

DAR 10-FS-SW

APPLICABLE FEDERAL AVIATION REGULATIONS.

FAA Form 8100-2 (7-67) FORMERLY FAA FORM 1362

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		UNITED STATES OF AMERICA	
		ANICHORTATION EFFERAL AL	

IATION ADMINISTRATION

### STANDARD AIRWORTHINESS CERTIFICATE

1. NATIONALITY AND REGISTRATION MARKS 2. MANUFACTURER AND MODEL AIRCRAFT SE

4. CATEGORY

N71VK

BEECH 95-C55

TE-387

NORMAL

5. AUTHORITY AND BASIS FOR ISSUANCE
This airworthiness certificate is issued pursuant to the Federal Aviation Act of 1958 and certifies that, as of the date of issuance, the aircraft to which issued has been inspected and found to conform to the type certificate therefor, to be in condition for safe operation, and has been shown to meet the requirements of the applicable comprehensive and defailed airworthiness code as provided by Annex 8 to the Convention on International Civil Aviation, except as noted herein.

2. Exceptions:

#### NONE

5. TERMS AND CONDITIONS

Unless sooner surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator, this airworthiness certificate is effective as long as the maintenance, preventative maintenance, and alterations are performed in accordance with Parts 21, 43, and 91 of the Federal Aviation Regulations, as appropriate, and the aircraft is registered in the United States.

(Replacement)
December 6, 1984 Emanuel Sylvia

SO-FSD0-15 Any alteration, reproduction, or misuse of this certificate may be punishable by a fine not exceeding \$1,000, or imprisonment not exceeding \$1,000, or imprisonment not exceeding \$2,000, or imprisonment not exceed in \$2,000, or imprisonment not exceed in \$2,000, or imprisonment not exceed in \$2,000, or imprisonment not exceed in \$2,000, or imprisonment not exceed in \$2,000, or imprisonment not exceed in \$2,000, or imprisonment not exceed in \$2,000, or imprisonment not exceed in \$2,000, or imprisonment not exceed in \$2,000, or imprisonment not exceed in \$2,000, or imprisonment not exceed in \$2,000, or imprisonment not exceed in \$2,000, or imprisonment not exceed in \$2,000, or imprisonment not exceed in \$2,000, or imprisonment not exceed in \$2,000, or imprisonment not exceed in \$2,000, or imprisonment not exceed in \$2,000, or imprisonment not exceed in \$2,000, or imprisonment not exceed in \$2,000, or imprisonment not exceed in \$2,000, or imprisonment not exceed in \$2,000, or imprisonment not exceed in \$2,000, or imprisonment not exceed in \$2,000, or imprisonment not exceed in \$2,000, or imprisonment not exceed in \$2,000, or imprisonment not exceed in \$2,000, or imprisonment not exceed in \$2,000

FAA Form 8100-2 (8-82)

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Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

- 8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.) NILDY 9-30-87
- 1. Installed Foster Air data LRN 500 Lpran in R.H. radio rack with antenna and preamp in overhead area of cabin.
- 2. Installation completed using manufactorers instructions and A.C. 43.13-1A Chapter 2 +11 and A.C. 43.13.2A Chapter 2 para 23,27, Chapter 3 paragraph 30,
- 3. Installation placanded: "for VFR use only"
- 4. Weight and balance computed and envered in aircraft records.

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ADDITIONAL SHEETS ARE ATTACHED

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Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

- 8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

  NIIDY

  3-18-87
- 1. Installed Whelen strobe light modle SA, DF 28 ST.C SA 800 EA on belly of aircraft using A440 mounting adapter.
- 2-Installation accomplished in accordance with A.C. 43.13-1A Chapter 11 section. 2 paragraph 4240. AC 43.13-2A Chapter 4 paragraph 5[a(1), 550, and manufacturers, supplied instructions,
- 3- Weight change computed and entered in air Fratt records.

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ADDITIONAL SHEETS ARE ATTACHED

**★U.S. GOVERNMENT PRINTING OFFICE: 1977-771-021/29** 

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FAA Form 8130-6 (12-83) SUPERSEDES PREVIOUS EDITION

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U.S. Department of Transportation Federal Aviation Administration

### Memorandum

Wichita Aircraft Certification Office 1801 Airport Road, Room 100 Mid-Continent Airport. Wichita, KS 67209

Airplane Flight Manual/Pilot & Operating Handbook for a Beech Model 95-C55

Date: November 6, 1984

Xawumee A. Nemm.
Robert A. Gambrill, Jr., Manager
Wichita Aircraft Certification Office, ACE-115W

Reply to Alth. of: Herron: 4402

Gene Berrier, Manager Manufacturing Inspection Branch, ASW-180

This is to verify our conversation concerning Beech POH P/N 96-590010-29 being added to Aircraft Specification No. 3Al6. This POH is applicable to Beechcraft Barons with S/N TE-1 through TE-942 except TE-938. The aforementioned POH can be used as an alternate to P/N 55-590006-3.

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DEPARTMENT OF TRANSPORTATION
FEDERAL AVIATION ADMINISTRATION

Form Spproved
Budget Bureau No. 04-R060.1
FOR FAA USE ONLY

INSTRUCT	IONS: Print or typ	e all entries. See	FAR.	13.9, FAR 43 Appe	ndix B, and	A. 3.9-1 (or	subsequent	revision th	nereof)
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NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

. DESCRIP	TION OF WORK ACCOMPLISHED (If m	ore space is requi	ired, attach ac	lditional sheets.	Identify with air
craft n	ationality and registration m	ark and date wor	k completed.)	Walter State	
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FAA Form 337 (7-67)

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FAA AIRCRAFT REGISTRY CAMERA NO. 4 DATE: 3-25-85

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with air- craft nationality and registration mark and date work completed.)
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4. The waring harness consist of two 14 gauge wires from each heater routed with existing aircraft wiring harness to two 120 Volt AC male plug recepticles located beneath a small hinged door on the left outboard nacelle.
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5. Small hinged door placarded, "120 Volt A.C. Ground power operation only".
6. Weight and balance previously figured by reweighing of
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DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION Form Approved
Budget Bureau No. 04-R060.1

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END.

ADDITIONAL SHEETS ARE ATTACHED

C FAA AC 72-4908

± U.S. GFO: 1981 - 775-332/47