



US Department
of Transportation
Federal Aviation
Administration

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

OMB No. 2120-0020
Exp: 04/30/2023

Electronic Tracking Number

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. §44701). Failure to report can result in a civil penalty for each such violation. (49 U.S.C. §46301(a))

1. Aircraft	Nationality and Registration Mark N71VK	Serial No. TE-387	
	Make BEECHCRAFT	Model 95-C55	Series BARON
2. Owner	Name (As shown on registration certificate) Brandon Cottingham		Address (As shown on registration certificate)
			Address 601 N 20th St
			City Lamesa State TX
		Zip 79331 Country USA	

3. For FAA Use Only

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial No.
<input checked="" type="checkbox"/>	<input type="checkbox"/>	AIRFRAME	_____	(As described in Item 1 above)	_____
<input type="checkbox"/>	<input type="checkbox"/>	POWERPLANT			
<input type="checkbox"/>	<input type="checkbox"/>	PROPELLER			
<input type="checkbox"/>	<input type="checkbox"/>	APPLIANCE	Type		
			Manufacturer		

6. Conformity Statement

A. Agency's Name and Address		B. Kind of Agency	
Name HAROLD HOLLADAY	<input checked="" type="checkbox"/> U. S. Certified Mechanic <input type="checkbox"/> Foreign Certified Mechanic <input type="checkbox"/> Certified Repair Station <input type="checkbox"/> Certified Maintenance Organization	Manufacturer	
Address 2103 S. Highway 349		C. Certificate No.	
City LAMESA State TX			
Zip 79331 Country USA		AEP 3173828 IA	

D: I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B <input type="checkbox"/>	Signature/Date of Authorized Individual Harold Holladay 04-21-2023
--	---

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ Approved ☐ Rejected

BY	FAA Fit. Standards Inspector	Manufacturer	Maintenance Organization	Persons Approved by Canadian Department of Transport
	FAA Designee	Repair Station <input checked="" type="checkbox"/>	Inspection Authorization	Other (Specify)

Certificate or Designation No. AEP 3173828 IA	Signature/Date of Authorized Individual Harold Holladay 04-21-2023
---	---

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

N7LVK
04-21-2023


Nationality and Registration Mark
Date

Inspected Airplane to comply with AD 90-08-14 and Mandatory Service Bulletin SB53-2269 RA. Found crack from rivet to rivet, about 3/8 inch in length, on the aluminum cover. Installed Front Carry Thru Spar Doubler Kit, P/N 58-4008-1-5, installed in accordance with Installation Instructions included in Doubler Kit. Right stringer was damaged. Repaired the right stringer in accordance with Engineering Order # 2023-B55-TE-387-01, revision 1A, dated 29 March 2023, signed by DER Raul Echevarria, DER 632537688 <copy attached>. Substantiation Report for Stringer Repair (static Substantiation), Report # J5V-2023-003 by Engineer S. Javadoff <issued by Echevarria, Inc., Lighthouse Point, Florida> <copy attached>. FAA Form 8110-3 Filed with FAA, Aircraft Registration Branch, Oklahoma City, OK, signed by DER Raul Echevarria, DER 632537688. <copy attached>.


Attached Sheets Are:

Engineering Order 2023-B55-TE-387-01, Revision 1A Substantiation Report for Stringer Repair # J5V-2023-003
FAA Form 8110-3, signed by Raul Echevarria, DER 632537688

☒ Additional Sheets Are Attached

U.S. DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION DETERMINATION OF COMPLIANCE WITH AIRWORTHINESS STANDARDS			1. PROJECT NO. (if applicable)	
AIRCRAFT OR AIRCRAFT COMPONENT IDENTIFICATION				
2. MAKE Textron Aviation Inc.		3. MODEL NO. 95-C55		4. TYPE (Aircraft, Engine, Propeller, Actuator etc.) Aircraft
5. NAME OF APPLICANT SOUTH PLAINS AVIATION				
PURPOSE OF DATA				
6. IN SUPPORT OF: ___ TC/ATC ___ STC ___ PMA <input checked="" type="checkbox"/> Major Repair ___ Major Alteration ___ Other (Explain)				
PROJECT SPECIFIC INFORMATION: Stringer Repair				
PURPOSE OF SUBMITTAL: In support of FAA form 337 for Major Repair for aircraft S/N TE-387				
LIST OF DATA				
List the data for this submittal including applicable drawings, material specifications, and process specifications and any other data that shows or contributes to a showing of compliance with the applicable requirements listed in block 9. A reference to a drawing list, including revision level, may be used.				
7. IDENTIFICATION EO 2023-B55-TE-387-01 Rev IR Date: Mar 29, 2023 JSV-2023-003 Rev IR Date: Mar 30, 2023		8. TITLE OF DATA Stringer Repair BEECHCRAFT BARON 95-C55 STRINGER REPAIR STATIC SUBSTANTIATION Notes: This approval is for engineering design data only. It indicates the data listed above demonstrates compliance only with the regulations specified by paragraph and subparagraph listed below as 'Applicable Requirements'. 1. This is not a Damage Tolerance Approval. 2. This approval covers structural details only.		
9. APPLICABLE REQUIREMENTS (List specific sections and amendment levels) 14 CFR 23.601 [Amdt 23-0]; 23.603 (all) [Amdt 23-23]; 23.605 (a) [Amdt 23-23]; 23.611 [Amdt 23-7]; 23.613 (all) [Amdt 23-23]				
10. FAA DESIGNEE APPROVAL - As directed by the Administrator and in accordance with the conditions and limitations of authorization under 14 CFR, Part 183, data listed above, and on attached sheets numbered _____, have been examined in accordance with established procedures. I therefore <input checked="" type="checkbox"/> APPROVE the data above <input type="checkbox"/> RECOMMEND APPROVAL of the data above FOR MAJOR REPAIR OR MAJOR ALTERATION ONLY - Other data approvals <input checked="" type="checkbox"/> ARE REQUIRED <input type="checkbox"/> ARE NOT REQUIRED EXPLAIN : This form does not constitute FAA approval of all the engineering data necessary for substantiation of compliance with the requirements needed for the complete repair. The needs of 14 CFR 23.573 are not included in this approval and require separate approval. <input type="checkbox"/> MANAGING OFFICE WAS CONTACTED (required when approval was made outside the U.S and/or involved critical or life limited parts)				
11. DER/ODA NUMBER 632537688		12. PRINTED NAME Raul Echevarria		
13. TECHNICAL DISCIPLINE DER-T (Structural Engineering)		14. SIGNATURE <i>Raul Echevarria</i>  Digitally Signed 03/31/2023		16. DATE 03/31/2023
FAA APPROVAL (For FAA use when designee recommends approval above, or when approval is reserved for the FAA)				
17. PRINTED NAME/FAA OFFICE		18. TECHNICAL DISCIPLINE		
19. SIGNATURE		20. DATE		

ENGINEERING ORDER

	EO No.	DATE	REVISION	PAGE
	2023-B55-TE-387-01	Mar 29, 2023	IR	1

Prepared by:
J. Bolivar

Date:
Mar 29, 2023

Approved by:
R. Echevarria

Date:
Mar 29, 2023

TITLE:

Stringer Repair

AIRCRAFT EFFECTIVITY:

CUSTOMER	MODEL	MSN	REGISTRY	LINE NUMBER	VARIABLE NUMBER	TAT	TAC
South Plains Aviation	Beechcraft Baron 95-C55	TE-387	N71VK	N/A	N/A	4,108.6	N/A

DESCRIPTION:

Per OEM design, the lower stringer runs along the fuselage and ends on Carry Thru Spar horizontal flange.

During repair of Carry Thru Spar on above aircraft in accordance with Beech SB 53-2669 R2, the lower stringer was accidentally trimmed out before the Carry thru Spar. Thus, in order to restore the installation, a repair angle made of .080" 2024-T3 aluminum sheet, is going to be used to splice the stringer with Carry thru Spar.

Affected stringer has a thickness of .075" and the fuselage skin .032".

REFERENCES:


1. Beechcraft Baron 55 SRM latest version.
2. Beechcraft Baron 55 CCMH latest version.

TYPE OF ENGINEERING ORDER:

- ☒ Modification or Alteration
- ☐ Repair
- ☐ Inspection
- ☐ Other:

- ☒ Major
- ☐ Minor

ENGINEERING ORDER

 SOUTH PLAINS AVIATION	EO No.	DATE	REVISION	PAGE
	2023-B55-TE-387-01	Mar 29, 2023	IR	2

NOTES:

1. This Engineering Order is applicable to the referenced aircraft only.
2. No deviation from this procedure must be performed, unless is approved by DER.

WEIGHT AND BALANCE:

Not affected for this modification.

WORKINSTRUCTIONS:

No deviations from the following work instructions, are authorized without approval from Engineering

DISPOSITION	Mech	Insp
1. Remove enough system/ structure to get access to the Carry thru Spar, per applicable SRM/ AMM reference.	H H	H H
2. Perform a detailed visual inspection around the installation area, looking for any discrepancy.	H H	H H
3. Form -1 Repair Angle using a .080" 2024-T3 aluminum sheet.	H H	H H
4. Maintain a minimum .22R internal bend radius.	H H	H H
5. Brake sharp edges.	H H	H H
6. Temporarily install -1 Repair Angle over the stringer and Carry thru Spar and drill back from existing holes per Baron 55 SRM applicable reference. Ensure 3 fasteners common to the stringer horizontal flange.	H H	H H

ENGINEERING ORDER



EO No.

DATE

REVISION

PAGE

2023-B55-TE-387-01

Mar 29, 2023

IR

3

7. Drill Qty: 2EA new holes to install MS20470D5-X rivets between -1 Repair Angle vertical flange and stringer vertical flange per Baron 55 SRM applicable references.

H H

H H

8. Remove parts and deburr holes.

H H

H H

9. Protect all bare metal with a layer of Alodine and two layers of primer per standard practices.

H H

H H

10. Apply P/S 870 sealant to all mating surfaces. Ok to use any approved alternative.

H H

H H

11. Install -1 Repair Angle, using MS20470D5-X rivets on vertical flange and horizontal flanges of the repair angle as per figure 1.

H H

H H

12. Perform a general visual inspection on reinforced areas and ensure all tools were removed.

H H

H H

13. Close area as required.

H H

H H

ENGINEERING ORDER



EO No.

DATE

REVISION

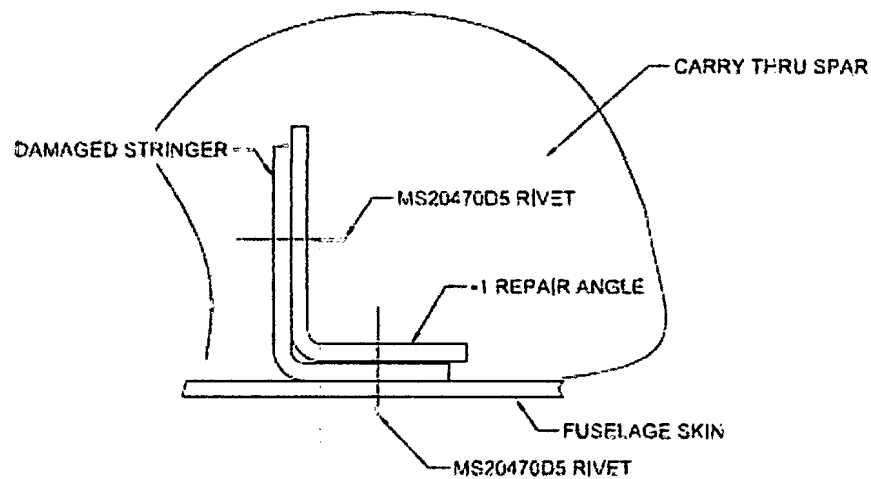
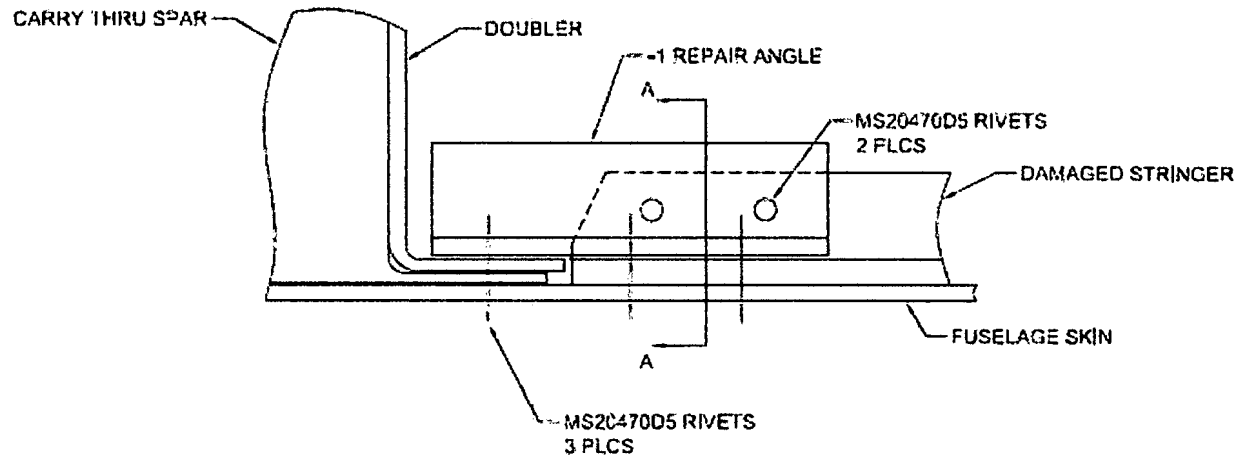
PAGE

2023-B55-TE-387-01

Mar 29, 2023

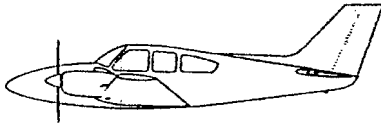
IR

4

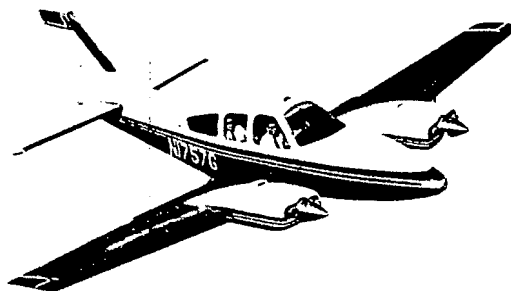


SECTION A-A

Figure 1: Repair Sketch

DATE: 3/30/2023	JSV Aviation 	TYPE: <input checked="" type="checkbox"/> STR <input type="checkbox"/> SYS <input type="checkbox"/> HYD
REPORT: JSV-2023-003		ENGINEER: S. Javadoff
MODEL: Beechcraft Baron 95-C55	SUBSTANTIATION REPORT	CLASSIFICATION: <input checked="" type="checkbox"/> MAJOR <input type="checkbox"/> MINOR

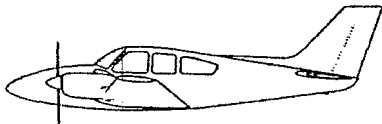
JSV AVIATION



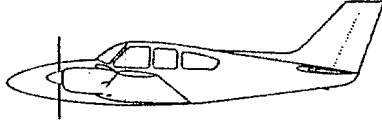
BEECHCRAFT BARON 95-C55
STRINGER REPAIR
STATIC SUBSTANTIATION

REPORT: JSV-2023-003

Rev. IR

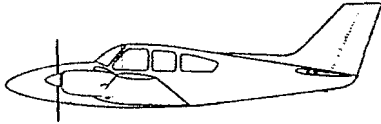
DATE: 3/30/2023	<div style="text-align: center;"> JSV Aviation  </div>	TYPE: <input checked="" type="checkbox"/> STR <input type="checkbox"/> SYS <input type="checkbox"/> HYD
REPORT: JSV-2023-003		ENGINEER: S. Javadoff
MODEL: Beechcraft Baron 95-C55	SUBSTANTIATION REPORT	CLASSIFICATION: <input checked="" type="checkbox"/> MAJOR <input type="checkbox"/> MINOR

REVISION HISTORY	3
ABSTRACT	4
REFERENCES	5
ASSUMPTIONS & LIMITATIONS	6
GEOMETRY and DESCRIPTION	7
STRINGER REPAIR - STATIC ANALYSIS	10
LOAD PATH	10
ALLOWABLES	10
LOADS	11
ANALYSIS – STATIC	12

DATE: 3/30/2023	<div style="text-align: center;"> JSV Aviation  </div>	TYPE: <input checked="" type="checkbox"/> STR <input type="checkbox"/> SYS <input type="checkbox"/> HYD
REPORT: JSV-2023-003		ENGINEER: S. Javadoff
MODEL: Beechcraft Baron 95-C55	SUBSTANTIATION REPORT	CLASSIFICATION: <input checked="" type="checkbox"/> MAJOR <input type="checkbox"/> MINOR

REVISION HISTORY

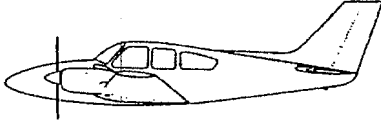
REPORT REVISION	REPORT DATE	ENGINEER	DEFINITION	CHANGES	APPROVAL
IR	03/30/2023	S. Javadoff	2023-B55-TE-387-01	Initial Release	R. Echevarria

DATE: 3/30/2023	<div style="text-align: center;"> JSV Aviation  </div>	TYPE: <input checked="" type="checkbox"/> STR <input type="checkbox"/> SYS <input type="checkbox"/> HYD
REPORT: JSV-2023-003		ENGINEER: S. Javadoff
MODEL: Beechcraft Baron 95-C55	SUBSTANTIATION REPORT	CLASSIFICATION: <input checked="" type="checkbox"/> MAJOR <input type="checkbox"/> MINOR

ABSTRACT

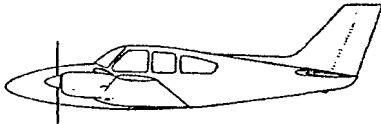
This document provides static substantiation for repair to lower stringer of Beechcraft Baron 95-C55 a/c.

The 2023-B55-TE-387-01 Rev IRI Engineering Order provides dimensions, instructions and hardware requirements for this repair.

DATE: 3/30/2023	<div style="text-align: center;"> JSV Aviation  </div>	TYPE: <input checked="" type="checkbox"/> STR <input type="checkbox"/> SYS <input type="checkbox"/> HYD
REPORT: JSV-2023-003		ENGINEER: S. Javadoff
MODEL: Beechcraft Baron 95-C55	SUBSTANTIATION REPORT	CLASSIFICATION: <input checked="" type="checkbox"/> MAJOR <input type="checkbox"/> MINOR

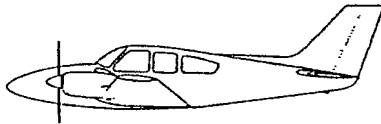
REFERENCES

1. Beechcraft Baron 55 SRM latest version.
2. Beechcraft Baron 55 CCMH latest version.
3. MMPDS-06, Metallic Materials Properties Development and Standardization, April 2011

DATE: 3/30/2023	<div style="text-align: center;"> JSV Aviation  </div>	TYPE: <input checked="" type="checkbox"/> STR <input type="checkbox"/> SYS <input type="checkbox"/> HYD
REPORT: JSV-2023-003		ENGINEER: S. Javadoff
MODEL: Beechcraft Baron 95-C55	SUBSTANTIATION REPORT	CLASSIFICATION: <input checked="" type="checkbox"/> MAJOR <input type="checkbox"/> MINOR

ASSUMPTIONS & LIMITATIONS

1. Repair is done in accordance with a/c SRM
2. Material used for repair is the same as for original stringer.

DATE: 3/30/2023	<div style="text-align: center;"> JSV Aviation  </div>	TYPE: <input checked="" type="checkbox"/> STR <input type="checkbox"/> SYS <input type="checkbox"/> HYD
REPORT: JSV-2023-003		ENGINEER: S. Javadoff
MODEL: Beechcraft Baron 95-C55		<div style="text-align: center; font-size: 1.5em; font-weight: bold;">SUBSTANTIATION REPORT</div>

GEOMETRY and DESCRIPTION

Per OEM design, the lower stringer runs along the fuselage and ends on Carry Thru Spar horizontal flange.

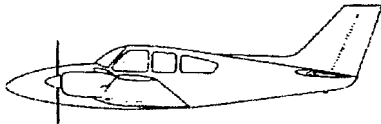
AIRCRAFT EFFECTIVITY:							
CUSTOMER	MODEL	MSN	REGISTRY	LINE NUMBER	VARIABLE NUMBER	TAT	TAC
South Plains Aviation	Beechcraft Baron 95-C55	TE-387	N71VK	N/A	N/A	4,108.6	N/A

During repair of Carry Thru Spar on above aircraft in accordance with Beech SB 53-2669 R2, the lower stringer was accidentally trimmed out before the Carry thru Spar. Thus, in order to restore the installation, a repair angle made of .080" 2024-T3 aluminum sheet, is going to be used to splice the stringer with Carry thru Spar.

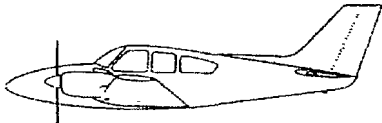
Affected stringer has a thickness of .075" and the fuselage skin .032".

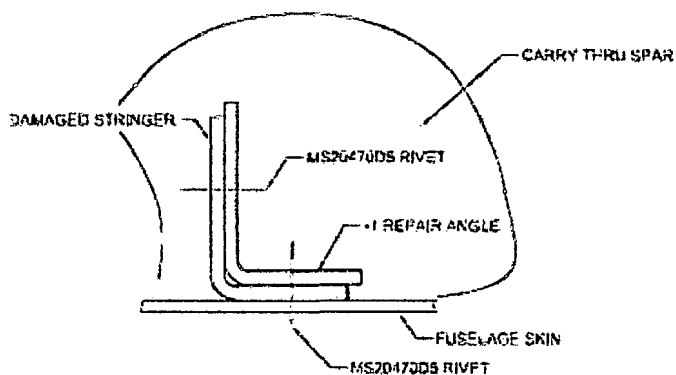
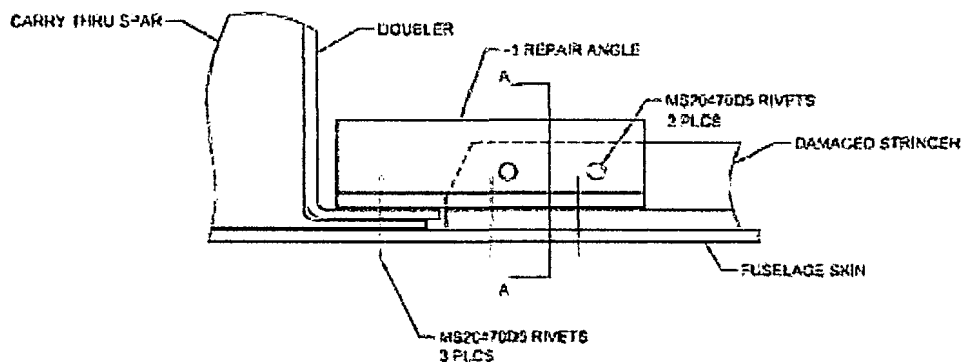
DISPOSITION

1. Remove enough system/ structure to get access to the Carry thru Spar, per applicable SRM/ AMM reference.
2. Perform a detailed visual inspection around the installation area, looking for any discrepancy.
3. Form -1 Repair Angle using a .080" 2024-T3 aluminum sheet.
4. Maintain a minimum .22R internal bend radius.
5. Brake sharp edges.
6. Temporarily install -1 Repair Angle over the stringer and Carry thru Spar and drill back from existing holes per Baron 55 SRM applicable reference. Ensure 3 fasteners common to the stringer horizontal flange.

DATE: 3/30/2023	<div style="text-align: center;"> JSV Aviation  </div>	TYPE: <input checked="" type="checkbox"/> STR <input type="checkbox"/> SYS <input type="checkbox"/> HYD
REPORT: JSV-2023-003		ENGINEER: S. Javadoff
MODEL: Beechcraft Baron 95-C55	SUBSTANTIATION REPORT	CLASSIFICATION: <input checked="" type="checkbox"/> MAJOR <input type="checkbox"/> MINOR

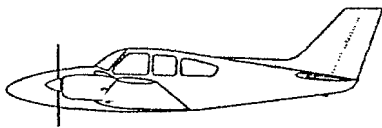
7. Drill Qty: 2EA new holes to install MS20470D5-X rivets between -1 Repair Angle vertical flange and stringer vertical flange per Baron 55 SRM applicable references.
8. Remove parts and deburr holes.
9. Protect all bare metal with a layer of Alodine and two layers of primer per standard practices.
10. Apply P/S 870 sealant to all mating surfaces. Ok to use any approved alternative.
11. Install -1 Repair Angle, using MS20470D5-X rivets on vertical flange and horizontal flanges of the repair angle as per figure 1.
12. Perform a general visual inspection on reinforced areas and ensure all tools were removed.
13. Close area as required.

DATE: 3/30/2023	JSV Aviation 	TYPE: <input checked="" type="checkbox"/> STR <input type="checkbox"/> SYS <input type="checkbox"/> HYD
REPORT: JSV-2023-003		ENGINEER: S. Javadoff
MODEL: Beechcraft Baron 95-C55	SUBSTANTIATION REPORT	CLASSIFICATION: <input checked="" type="checkbox"/> MAJOR <input type="checkbox"/> MINOR



SECTION A-A

Figure 1: Repair Sketch

DATE: 3/30/2023	JSV Aviation 	TYPE: <input checked="" type="checkbox"/> STR <input type="checkbox"/> SYS <input type="checkbox"/> HYD
REPORT: JSV-2023-003		ENGINEER: S. Javadoff
MODEL: Beechcraft Baron 95-C55	SUBSTANTIATION REPORT	CLASSIFICATION: <input checked="" type="checkbox"/> MAJOR <input type="checkbox"/> MINOR

STRINGER REPAIR - STATIC ANALYSIS

LOAD PATH

The critical load in the substantiation of this repair remains intact. Area lost is compensated by adding 0.080" thick -1 repair angle made from 2024-T3 Al sheet.

ALLOWABLES

AL SHEET:

2024-T3 AMS-QQ-A-250/4 per MMPDS-06 Table 3.2.4.0(b1)

FASTENERS:

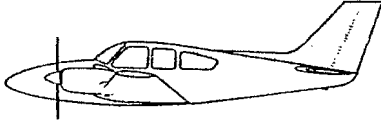
Per MMPDS-06 Table 8.1.2 (b)

MS20470AD5-X

0.125 in, Solid Rivet

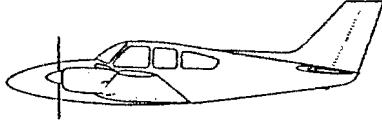
Table 8.1.2(b). Single Shear Strength of Solid Rivets*

Undriven			Driven		Rivet Designation	Rivet Size							
Rivet Material	F _u (ksi)		Rivet Material	F _u ^b (ksi)		1/16	3/32	1/8	5/32	3/16	1/4	5/16	3/8
	Mm	Max				Driven Single Shear Strength, lbs ^c							
5056-H32	24	n/a	5056-H321 ^d	28 ^e	B ^f	99	203	363	556	802	1450	2290	3275
2117-T4	26	n/a	2117-T3	30 ^e	AD	106	217	389	596	860	1555	2455	3510
2017-T4	35	42	2017-T3	38	D	134	275	493	755	1085	1970	3115	4445
2024-T4	37	n/a	2024-T31	41 ^e	DD	145	297	532	814	1175	2125	3360	4795
7050-T73	41	46	7050-T731 ^g	43 ^e	E ^h	152	311	558	854	1230	2230	3520	5030
Monel	49	59	Monel	52 ^e	M	183	376	674	1030	1490	2695	4260	6085
Ti-45Nb	50	59	Ti-45Nb	53 ^e	T	187	384	687	1050	1515	2745	4340	6200
A-286	85	95	A-286	90 ^e	-	317	651	1165	1785	2575	4065	7375	10500

DATE: 3/30/2023	<div style="text-align: center;"> JSV Aviation  </div>	TYPE: <input checked="" type="checkbox"/> STR <input type="checkbox"/> SYS <input type="checkbox"/> HYD
REPORT: JSV-2023-003		ENGINEER: S. Javadoff
MODEL: Beechcraft Baron 95-C55	SUBSTANTIATION REPORT	CLASSIFICATION: <input checked="" type="checkbox"/> MAJOR <input type="checkbox"/> MINOR

LOADS

Loads are not altered by this repair.

DATE: 3/30/2023	<div style="text-align: center;"> JSV Aviation  </div>	TYPE: <input checked="" type="checkbox"/> STR <input type="checkbox"/> SYS <input type="checkbox"/> HYD
REPORT: JSV-2023-003		ENGINEER: S. Javadoff
MODEL: Beechcraft Baron 95-C55	SUBSTANTIATION REPORT	CLASSIFICATION: <input checked="" type="checkbox"/> MAJOR <input type="checkbox"/> MINOR

ANALYSIS – STATIC

Repair performed per 2023-B55-TE-387-01 restores original structure capability by adding in more cross sectional area than is removed by trim out operation.

Trimmed out stringer is 0.075" thick 2024-T3 AL SH

-1 repair angle is 0.080" thick 2024-T3 AL SH

Table 3.2.4.0(b). Design Mechanical and Physical Properties of 2024 Aluminum Alloy Sheet and Plate

Specification	AMS 4037 and AMS-QQ-A-250/4 ^a					AMS-QQ-A-250/4 ^a		
Form	Sheet					Sheet	Plate	
Temper	T3					T361		
Thickness, in.	0.008-0.009	0.010-0.128		0.129 - 0.249		0.020-0.062	0.063-0.249	0.250-0.500
Basis	S	A	B	A	B	S	S	S
Mechanical Properties:								
F_{σ} , ksi:								
L	64	64	65	64	66	68	69	67
LT	63	63	64	63	65	67	68	66
ST
F_{σ} , ksi:								
L	47	47	48	47	48	56	56	54
LT	42	42	43	42	43	50	51	49
ST
F_{σ} , ksi:								
L	39	39	40	39	40	47	48	46
LT	45	45	46	45	46	53	54	52
ST
F_{σ}^b , ksi	39	39	40	40	41	42	42	41
$F_{\sigma}^{b,c}$, ksi:								
(e/D = 1.5)	104	104	106	106	107	111	112	109
(e/D = 2.0)	129	129	131	131	133	137	139	135
$F_{\sigma}^{b,c}$, ksi:								
(e/D = 1.5)	73	73	75	73	75	82	84	81
(e/D = 2.0)	88	88	90	88	90	97	99	96
e, percent:								
LT	10	11	...	11	...	8	9	9 ^e
E , 10 ³ ksi	10.5					10.7		
E , 10 ³ ksi	10.7					10.9		
G, 10 ³ ksi	4.0					4.0		
μ	0.33					0.33		
Physical Properties:								
ω , lb/in. ³	0.100							
C, K, and α	See Figure 3.2.4.0							

Revised: Apr 2008, MMPDS-04, Item 05-14.

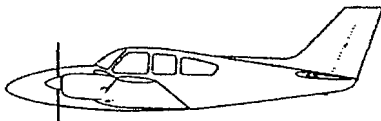
a Mechanical properties were established under MIL-QQ-A-250/4.

b Grain direction unknown.

c Bearing values are "dry pin" values per Section 1.4.7.1. See Table 3.1.2.1.1

d See Table 3.2.4.0(c)

e 10% for 0.500 inch.

DATE: 3/30/2023	<div style="text-align: center;"> JSV Aviation  </div>	TYPE: <input checked="" type="checkbox"/> STR <input type="checkbox"/> SYS <input type="checkbox"/> HYD
REPORT: JSV-2023-003		ENGINEER: S. Javadoff
MODEL: Beechcraft Baron 95-C55	SUBSTANTIATION REPORT	CLASSIFICATION: <input checked="" type="checkbox"/> MAJOR <input type="checkbox"/> MINOR

Assuming that both t capping and repair angle are 1 x 1 inch angle the increase in area after repair is 6.7%

$$A_{orig} = (1 \text{ in} + 1 \text{ in}) \cdot 0.075 \text{ in} = 0.15 \text{ in}^2$$

$$A_{repair} = (1 \text{ in} + 1 \text{ in}) \cdot 0.080 \text{ in} = 0.16 \text{ in}^2$$

Original load carrying capability is:

$$P_{orig} = A_{orig} \cdot F_{tu} = 0.15 \text{ in}^2 \cdot 64,000 \text{ psi} = 9600 \text{ lbf}$$

Repair load carrying capability is:

$$P_{repair} = A_{repair} \cdot F_{tu} = 0.16 \text{ in}^2 \cdot 64,000 \text{ psi} = 10240 \text{ lbf}$$

Original stringer was attached to carry through spar with 1 MS20470D5 rivet. The -1 repair angle as attached with four MS20470D5 rivets to the stringer and with 1 MS20470D5 rivet to the carry through spar thus load carrying capability is not affected by repair.

Repair to stringer on Beechcraft Baron 95-C55 a/c is structurally acceptable. No further analysis is required.



US Department
of Transportation
Federal Aviation
Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

OMB No. 2120-0020
Exp: 5/31/2018

Electronic Tracking Number

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. §44701). Failure to report can result in a civil penalty for each such violation. (49 U.S.C. §46301(a))

1. Aircraft	Nationality and Registration Mark N71VK	Serial No. TE-387		
	Make BEECH	Model 95-C55	Series BARON	
2. Owner	Name (As shown on registration certificate) Cottingham, Brandon		Address (As shown on registration certificate)	
			Address 601 N 20th St.	
			City LAMESA	State TX
			Zip 79331	Country USA

3. For FAA Use Only

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial No.
<input type="checkbox"/>	<input type="checkbox"/>	AIRFRAME		(As described in Item 1 above)	
<input type="checkbox"/>	<input checked="" type="checkbox"/>	POWERPLANT	Left: Continental Right: Continental	IO-520-C IO-520-C	1258167C 231709R
<input type="checkbox"/>	<input type="checkbox"/>	PROPELLER			
<input type="checkbox"/>	<input type="checkbox"/>	APPLIANCE	Type		
			Manufacturer		

6. Conformity Statement

A. Agency's Name and Address		B. Kind of Agency	
Name HAROLD HOLBADA	Address 2103 S Highway 349 City LAMESA State TX Zip 79331 Country USA	<input checked="" type="checkbox"/> U. S. Certified Mechanic	Manufacturer
		Foreign Certified Mechanic	C. Certificate No.
		Certified Repair Station	
		Certified Maintenance Organization	AEP 3173828 IA

D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B <input type="checkbox"/>	Signature/Date of Authorized Individual Harold Holbada 12-05-2022
--	---

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ Approved ☐ Rejected

BY	FAA Flt. Standards Inspector	Manufacturer	Maintenance Organization	Persons Approved by Canadian Department of Transport
	FAA Designee	Repair Station	<input checked="" type="checkbox"/> Inspection Authorization	Other (Specify)

Certificate or Designation No. AEP 3173828 IA	Signature/Date of Authorized Individual Harold Holbada 12-05-2022
---	---

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

NOTICE

8. Description of Work Accomplished (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Installation of D'Shannon Products Cooling Air Baffles Kits in reference to STC Number SA092A05C, Certificate issued to:
D'Shannon Products, LTD.
1309 County Road 134
Buffalo, MN 55313.

Air Baffles Kits were installed on both engines of Beech Baron
N71VK: Left Engine - Continental IO-520-C, Serial #: 1258167C
And Right Engine - Continental IO-520-C, Serial #: 231709R

Air Baffles Kits were installed in reference to Drawing List
DL87002, Revision 11 (Even), dated 02/20/2020 and Installation
Manual DSP-IM95-4 REV B, dated 04-20-19-Dwg DSP-IM95-4-1C
Option C, (Item 3 Installation Manual, listed on front page of
STC # SA092A05C), using parts for two kits (one for each engine)
as listed by D'Shannon Products, P/N DL87002 OPTC, Serial #:
190415-05, pages 1, 2 + 3 of Dwg-3 Backing List Engine Baron 55
Baffle OPTC Complete.

Note: Copy of STC Number SA092A05C, Parts List for Kits, P/N
DL87002, and Instructions for Continued Airworthiness, DP-ICA-200,
Revision B for STC SA092A05C for Beech 55 have been included
in Airplane logs for N71VK.

☐ Additional Sheets Are Attached



US Department
of Transportation
Federal Aviation
Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020
11/30/2007

Electronic Tracking Number

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. §44701). Failure to report can result in a civil penalty for each such violation. (49 U.S.C. §46301(a))

1. Aircraft	Nationality and Registration Mark USA N71VK	Serial No. TE-387	
	Make BEECH	Model 95-C55	Series
2. Owner	Name (As shown on registration certificate) SOUTH PLAINS AVIATION TRAININGF CENT		Address (As shown on registration certificate) PO BOX 179
			City Lamesa State TX
			Zip 79331 Country USA

3. For FAA Use Only

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial No.
<input type="checkbox"/>	<input checked="" type="checkbox"/>	AIRFRAME		(As described in Item 1 above)	
<input type="checkbox"/>	<input type="checkbox"/>	POWERPLANT			
<input type="checkbox"/>	<input type="checkbox"/>	PROPELLER			
<input type="checkbox"/>	<input type="checkbox"/>	APPLIANCE	Type		
			Manufacturer		

6. Conformity Statement

A. Agency's Name and Address		B. Kind of Agency	
Name Abilene Aero, Inc.	Address 2850 Airport Blvd. City Abilene State TX Zip 79602 Country USA	<input type="checkbox"/> U. S. Certificated Mechanic	<input type="checkbox"/> Manufacturer
		<input type="checkbox"/> Foreign Certificated Mechanic	C. Certificate No.
		<input checked="" type="checkbox"/> Certificated Repair Station	GSTR469E
		<input type="checkbox"/> Certificated Maintenance Organization	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B <input type="checkbox"/>	Signature/Date of Authorized Individual October 2, 2020 Stuart Douglas
--	---

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ Approved ☐ Rejected

BY	FAA Fit. Standards Inspector	Manufacturer	Maintenance Organization	Persons Approved by Canadian Department of Transport
	FAA Designee <input checked="" type="checkbox"/>	Repair Station	Inspection Authorization	

Certificate or Designation No. GSTR469E	Signature/Date of Authorized Individual October 2, 2020 Stuart Douglas
---	---

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

USA N71VK

10/02/2020

Nationality and Registration Mark

Date

Installed GTN-750, GA-35 WAAS Antenna IAW Garmin STC# SA02019SE-D and Install Manual 190-01007-C1 Rev:1.
Installed Garmin GTX-345 ADSB Transponder IAW Garmin STC# SA01714WI and Install Manual 190-00734-10 Rev:14.
The Installed ADS-B out system has shown to meet the equipment performance requirements of 14 CFR section 91.227.

Installed Dual Garmin G5 HSI / ADI, Dual G5 Batterys, Single GAD-29B, Single GMU-11, and associated mounts, hardware and wiring IAW Garmin STC # SA01818WI and Install Manual 190-01112-10 Rev:21.

Installed Garmin EIS IAW Garmin Installation Manual P/N190-01717-00 Rev 13 Dated May 31, 2020.

Installed Genesys 3100 Autopilot IAW Genesys Installation Manual P/N Dated and STC SA09755DS dated 9-11-19.

☐ Additional Sheets Are Attached



US Department
of Transportation
Federal Aviation
Administration

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

OMB No. 2120-0020
Exp: 9/31/2018

Electronic Tracking Number

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. §44701). Failure to report can result in a civil penalty for each such violation. (49 U.S.C. §46301(a))

1. Aircraft	Nationality and Registration Mark N71VK	Serial No. TE-387	
	Make Beechcraft	Model 95-C55	Series
2. Owner	Name (As shown on registration certificate) South Plains Aviation Training Center		Address (As shown on registration certificate)
			Address P.O. Box 179 City Lamesa State TX Zip 79331 Country USA

3. For FAA Use Only

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial No.
<input type="checkbox"/>	<input checked="" type="checkbox"/>	AIRFRAME	_____	(As described in Item 1 above)	_____
<input type="checkbox"/>	<input type="checkbox"/>	POWERPLANT			
<input type="checkbox"/>	<input type="checkbox"/>	PROPELLER			
<input type="checkbox"/>	<input type="checkbox"/>	APPLIANCE	Type		
			Manufacturer		

6. Conformity Statement

A. Agency's Name and Address		B. Kind of Agency		
Name	Harold Hobbsday	<input checked="" type="checkbox"/> U. S. Certificated Mechanic	Manufacturer	
Address	2103 S. Hwy. 349	<input type="checkbox"/> Foreign Certificated Mechanic	C. Certificate No.	
City	Lamesa State TX	<input type="checkbox"/> Certificated Repair Station	A+P 3173828	
Zip	79331 Country USA	<input type="checkbox"/> Certificated Maintenance Organization		

D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B <input type="checkbox"/>	Signature/Date of Authorized Individual <div style="display: flex; justify-content: space-between; align-items: center;"> <div>Harold Hobbsday</div> <div>15 June 2020</div> </div>
--	--

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☐ Approved ☐ Rejected

BY	FAA Flt. Standards Inspector	Manufacturer	Maintenance Organization	Persons Approved by Canadian Department of Transport
	FAA Designee	Repair Station	<input checked="" type="checkbox"/> Inspection Authorization	Other (Specify)
Certificate or Designation No. IA 3173828		Signature/Date of Authorized Individual <div style="display: flex; justify-content: space-between; align-items: center;"> <div>Harold Hobbsday</div> <div>15 June 2020</div> </div>		

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

N71VK

Nationality and Registration Mark

15 June 2020

Date

Action:

Installed cockpit Sunvisor Assemblies in reference to
STC 3599 NM, Rosen Sunvisor Systems, using Rosen
Sunvisors, RBB-300-1A, SERIAL #: M234734

☐ Additional Sheets Are Attached

1372-1069 37 APR 24 '68

APR 1 01968

☆ SPO : 1965 OF-760-962

FORM APPROVED
BUDGET BUR. NO. 04-R063

FEDERAL AVIATION AGENCY

AIRCRAFT USE AND INSPECTION REPORTPRINT OR TYPE ALL ENTRIES EXCEPT SIGNATURES.
SHADED BOXES ARE FOR FAA USE**SECTION I—IDENTIFICATION**

1. AIRCRAFT REGISTRATION MARK N- 2786T		AIRCRAFT MFR. SERIAL NO. TE-387	
2. AIRCRAFT MAKE Beech		AIRCRAFT MODEL 95-C-55	
3. NAME OF OWNER (As shown on Certificate of Registration) Combs Aircraft, Inc. - Dealer Certif. 68-37			
4. ADDRESS OF OWNER (As shown on Certificate of Registration) NUMBER AND STREET: Stapleton Intl. Airport CITY: Denver, STATE: Colorado			
5. AIRCRAFT IS BASED AT: AIRPORT: Stapleton Intl. Airport CITY: Denver		COUNTY: Denver STATE OF: Colorado	03576

SECTION II—ACTIVITY DATA

6. DATE OF PREVIOUS PERIODIC INSPECTION 5/3/67		AIRCRAFT TIME IN SERVICE PREVIOUS 12 MONTHS 42.3 HRS.	
7. PRIMARY USE OF AIRCRAFT (Check one box only)			
1	EXECUTIVE TRANSPORTATION	6	AIR TAXI
2	BUSINESS TRANSPORTATION	7	INDUSTRIAL, SPECIAL
3	<input checked="" type="checkbox"/> PERSONAL	8	OTHER (Specify)
4	AERIAL APPLICATION		
5	INSTRUCTION		

SECTION III—INSPECTION REPORT for #100

8. NAME OF INSPECTING AGENCY Combs Aircraft Inc.																			
9. TYPE INSPECTING AGENCY 3 <input type="checkbox"/> INSPECTION AUTHORIZATION 5 <input type="checkbox"/> MANUFACTURER 6 <input checked="" type="checkbox"/> REPAIR STATION	CERTIFICATE NO. AND TYPE Repair Sta. 4152 Class 3																		
10. TYPE OF INSPECTION 1 <input checked="" type="checkbox"/> PERIODIC 2 <input type="checkbox"/> PROGRESSIVE	DATE OF INSPECTION 4/8/68																		
CONDITION OF AIRCRAFT 1 <input checked="" type="checkbox"/> AIRWORTHY 2 <input type="checkbox"/> UNAIRWORTHY																			
11. SIGNATURE OF CONDUCTOR OF THIS INSPECTION <i>Errol H. Fritinger</i>																			
<table border="1"> <tr> <td>EA</td> <td>SW</td> <td>GE</td> <td>WE</td> <td>AL</td> <td>PC</td> <td>SO</td> <td>EU</td> <td>FAA DIST. OFF NO.</td> </tr> <tr> <td>1</td> <td>2</td> <td>3</td> <td>4</td> <td>5</td> <td>6</td> <td>7</td> <td>8</td> <td>9</td> </tr> </table>		EA	SW	GE	WE	AL	PC	SO	EU	FAA DIST. OFF NO.	1	2	3	4	5	6	7	8	9
EA	SW	GE	WE	AL	PC	SO	EU	FAA DIST. OFF NO.											
1	2	3	4	5	6	7	8	9											

FEDERAL AVIATION AGENCY

POSTAGE AND FEES
PAID
FEDERAL AVIATION AGENCY

OFFICIAL BUSINESS

General Aviation District Office #3
Jefferson County Airport
Broomfield, Colorado 80020

10:

APR 17
OKLAHOMA

General Aviation District Office #3
Jefferson County Airport
Broomfield, Colorado 80020

ANCE 11-1-58 J WITH
AIRCRAFT REGIS

0326-1364 17 JAN 4 '68

DEC 15 1967

FEDERAL AVIATION AGENCY

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)Form Approved
Budget Bureau No. 04-R060.1

FOR FAA USE ONLY

OFFICE IDENTIFICATION DEN-GADO
4-5-03

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.

1. AIRCRAFT	MAKE BEECHCRAFT	MODEL 95-C55
	SERIAL NO. TE-387	NATIONALITY AND REGISTRATION MARK N-2786T
2. OWNER	NAME (As shown on registration certificate) COMBS AIRCRAFT, INC.	ADDRESS (As shown on registration certificate) HANGAR #7 - STAPLETON INT'L. AIRPORT DENVER, COLORADO 80207

3. FOR FAA USE ONLY

4. UNIT IDENTIFICATION

5. TYPE

UNIT	MAKE	MODEL	SERIAL NO.	REPAIR	ALTERATION
AIRFRAME	***** (As described in item 1 above) *****				X
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				

Verified by Operator #100

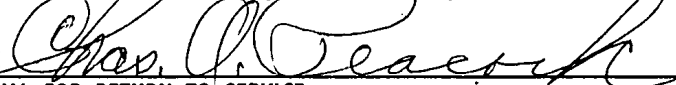
6. CONFORMITY STATEMENT

A. AGENCY'S NAME AND ADDRESS	B. KIND OF AGENCY	C. CERTIFICATE NO.
AIRCRAFT RADIO & ACCESSORY CO., INC. HANGAR #4 STAPLETON INTERNATIONAL AIRPORT DENVER, COLORADO 80207	U.S. CERTIFICATED MECHANIC	3932
	FOREIGN CERTIFICATED MECHANIC	
	X CERTIFICATED REPAIR STATION	RADIO CLASS 1,2&3
	MANUFACTURER	INST. CLASS 1,2&3

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

DATE
DECEMBER 12, 1967

SIGNATURE OF AUTHORIZED INDIVIDUAL



7. APPROVAL FOR RETURN TO SERVICE

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Agency and is ☒ APPROVED ☐ REJECTED

BY	FAA FLT. STANDARDS INSPECTOR	X	MANUFACTURER	INSPECTION AUTHORIZATION	OTHER (Specify)
	FAA DESIGNEE		REPAIR STATION	CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT	
DATE OF APPROVAL OR REJECTION 12-12-67		CERTIFICATE OR DESIGNATION NO. 3932		SIGNATURE OF AUTHORIZED INDIVIDUAL	



Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

NOTICE

8. DESCRIPTION OF WORK ACCOMPLISHED (if more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

THE FOLLOWING LISTED ELECTRONICS EQUIPMENT HAS BEEN INSTALLED. KING KXP-750 ATC TRANSPONDER AND KS-505 POWER SUPPLY MOUNTED ON BEECHCRAFT INSTALLED RADIO SHELF AND SECURED WITH 8.32 AIRCRAFT MACHINE SCREWS AND RIVNUTS. ALL RADIOS INSTALLED IN RADIO PANEL. INSTRUMENTS INSTALLED IN INSTRUMENT PANEL. RADAIR STROBE LIGHT SYSTEM INSTALLED WITH RADAIR INSTALLATION KIT AND PER RADAIR INSTRUCTIONS. APPLICABLE PORTIONS OF AC 43.13-1 AND 43.13-2 HAVE BEEN COMPLIED WITH. AIRCRAFT FILE AND LOG CURRENTLY AMENDED.

STATION INSTALLED AT

63.0	KING KX-160 VHF NAV/COM.
9.0	KING KS-505 POWER SUPPLY
6.0	KING KXP-750 ATC TRANSPONDER
-	KING KA-22 ANTENNA
11.0	KING KMA-128 AMPLIFIER
63.0	KING KI-211A INDICATOR
63.0	KING TRANSPONDER CONTROL
151.0	ARC A-15 ANTENNA (COMM)
190.0	MARKER BEACON ANTENNA
35.0	KING TRANSPONDER ANTENNA
195.0	RADAIR POWER SUPPLY
110.0	RADAIR STROBE LIGHT
63.0	HOBBS HOUR METER
60.0	WIRE, PLUGS, ETC.

-----END-----

REPLACEMENT SHEET WITH
FAA AIRCRAFT REGISTRY

JAN 3 4 01 PM '68
OKLAHOMA CITY, OKLA.

1378-1537

MAY 5 '68

FEDERAL AVIATION AGENCY

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
 Budget Bureau No. 04-R060.1

FOR FAA USE ONLY

OFFICE IDENTIFICATION DEN-GADJ

4-5-03

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.

1. AIRCRAFT	MAKE Beech	MODEL 95-C55
	SERIAL NO. TE 387	NATIONALITY AND REGISTRATION MARK N 2786T
2. OWNER	NAME (As shown on registration certificate) Combs Aircraft, Inc.	ADDRESS (As shown on registration certificate) Stapleton International Airport Denver, Colorado

3. FOR FAA USE ONLY

4. UNIT IDENTIFICATION

5. TYPE

UNIT	MAKE	MODEL	SERIAL NO.	REPAIR	ALTER- ATION
AIRFRAME	***** (As described in item 1 above) *****				X
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				

6. CONFORMITY STATEMENT

A. AGENCY'S NAME AND ADDRESS	B. KIND OF AGENCY	C. CERTIFICATE NO.
Kensair Corporation Jefferson County Airport Broomfield, Colorado	<input type="checkbox"/> U.S. CERTIFICATED MECHANIC	Repair Station #3986
	<input checked="" type="checkbox"/> FOREIGN CERTIFICATED MECHANIC	
	<input checked="" type="checkbox"/> CERTIFICATED REPAIR STATION	
	<input type="checkbox"/> MANUFACTURER	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

DATE March 5, 1968	SIGNATURE OF AUTHORIZED INDIVIDUAL <i>[Signature]</i>
------------------------------	--

7. APPROVAL FOR RETURN TO SERVICE

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Agency and is ☒ APPROVED ☐ REJECTED

BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	INSPECTION AUTHORIZATION	OTHER (Specify) Verified by Operator #49
	FAA DESIGNEE	X REPAIR STATION	CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT	

DATE OF APPROVAL OR REJECTION 3/5/68	CERTIFICATE OR DESIGNATION NO. 3986	SIGNATURE OF AUTHORIZED INDIVIDUAL <i>[Signature]</i>
--	---	--

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

OKLAHOMA

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Installed the following electronic equipment: I/A/W AC 43.13-2, Figures 2.1 and 3.1:

Old Empty Weight
KN-60, King DME

Weight
3310.0
7.5
3317.5

Arm
79.0
63.0
78.97

Moment
261515
473
261988

Operational and calibration checks performed by Air Radio, Inc., C.R.S. #4473, Radio, Class 1, 2 & 3. Instrument Limited.

Equipment list amended.

-----THE END-----

☐ ADDITIONAL SHEETS ARE ATTACHED

378-1986

APR 10 1968

FEDERAL AVIATION AGENCY

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved
Budget Bureau No. 04-R060.1

FOR FAA USE ONLY

OFFICE IDENTIFICATION
DEN-3403
4-5-03

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.

1. AIRCRAFT	MAKE Beech	MODEL 95-C-55
	SERIAL NO. TE-387	NATIONALITY AND REGISTRATION MARK N 2786T
2. OWNER	NAME (As shown on registration certificate) Combe Aircraft Inc	ADDRESS (As shown on registration certificate) Stapleton International Airport Denver Colorado

3. FOR FAA USE ONLY

4. UNIT IDENTIFICATION

5. TYPE

UNIT	MAKE	MODEL	SERIAL NO.	REPAIR	ALTERATION
AIRFRAME	***** (As described in item 1 above) *****				X
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				

6. CONFORMITY STATEMENT

A. AGENCY'S NAME AND ADDRESS Combs Aircraft Inc Stapleton International Airport Denver Colorado	B. KIND OF AGENCY U.S. CERTIFICATED MECHANIC FOREIGN CERTIFICATED MECHANIC <input checked="" type="checkbox"/> CERTIFICATED REPAIR STATION MANUFACTURER	C. CERTIFICATE NO. Repair Station 4152 Class 3
--	---	---

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

DATE 4/8/68	SIGNATURE OF AUTHORIZED INDIVIDUAL <i>Ronald Frutiger</i>
----------------	--

7. APPROVAL FOR RETURN TO SERVICE

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Agency and is ☒ APPROVED ☐ REJECTED

BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	INSPECTION AUTHORIZATION	OTHER (Specify)
	FAA DESIGNEE <input checked="" type="checkbox"/>	REPAIR STATION	CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT	Verified by Operator #41

DATE OF APPROVAL OR REJECTION 4/8/68	CERTIFICATE OR DESIGNATION NO. 4152	SIGNATURE OF AUTHORIZED INDIVIDUAL <i>Ronald Frutiger</i>
---	--	--

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

1. Install Dual Control Arm as per Beech drawing No 35-524662

Weight increase 5 lbs @ 72"

Aircraft New E.W. 3321.5 @ 78.65 Moment of 262267.

END

APR 17 4 19 PM '68

OKLAHOMA

COPIES FILED WITH
FAA AIRCRAFT REGISTRY
ADDITIONAL SHEETS ARE ATTACHED ☐

UNITED STATES OF AMERICA — FEDERAL AVIATION AGENCY

STANDARD AIRWORTHINESS CERTIFICATE

1. NATIONALITY AND REGISTRATION MARKS	2. MANUFACTURER AND MODEL	3. AIRCRAFT SERIAL NUMBER	4. CATEGORY
N2786T	BEECHCRAFT - 95-055	TE-387	Normal

5. AUTHORITY AND BASIS FOR ISSUANCE

This airworthiness certificate is issued pursuant to the Federal Aviation Act of 1958 and certifies that, as of the date of issuance, the aircraft to which issued has been inspected and found to conform to the type certificate therefor, to be in condition for safe operation, and has been shown to meet the requirements of the applicable comprehensive and detailed airworthiness code as provided by Annex 8 to the Convention on International Civil Aviation, except as noted herein.

Exceptions:

None.

6. TERMS AND CONDITIONS

Unless sooner surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator, this airworthiness certificate is effective as long as the maintenance, preventative maintenance, and alterations are performed in accordance with Parts 21, 43, and 71 of the Federal Aviation Regulations, as appropriate, and the aircraft is registered in the United States.

DATE OF ISSUANCE	EAA REPRESENTATIVE <i>Chester A. Rembleske</i> FOR BEECH AIRCRAFT CORPORATION	DESIGNATION NUMBER
May 3, 1967		DOA CE-2

Any alteration, reproduction, or misuse of this certificate may be punishable by a fine not exceeding \$1,000, or imprisonment not exceeding 3 years, or both. THIS CERTIFICATE MUST BE DISPLAYED IN THE AIRCRAFT IN ACCORDANCE WITH APPLICABLE FEDERAL AVIATION REGULATIONS.



FEDERAL AVIATION AGENCY APPLICATION FOR AIRWORTHINESS CERTIFICATE (FAR 21)				INSTRUCTIONS: Print or type. Do not write in shaded areas; these are for FAA use only. Submit original only to an authorized FAA Representative. If additional space is required, use reverse side.		
I. AIRCRAFT DESCRIPTION	1. REGISTRATION MARK N2786T	2. AIRCRAFT MAKE BEECHCRAFT	3. AIRCRAFT MODEL DESIGNATION 95-C55		FAA CODING 1152708	
	4. AIRCRAFT SERIAL NO. TE-387	5. ENGINE MAKE Continental	6. ENGINE MODEL DESIGNATION IO-520-C			
	7. NO. OF ENGINES 2	8. PROPELLER MAKE McCauley	9. PROPELLER MODEL DESIGNATION 2AF34C55		10. YR. AIRC. MANUFACTURED 1967	
II. CERTIFICATION REQUESTED	A. APPLICATION IS HEREBY MADE FOR:					
	<input checked="" type="checkbox"/> 3	ORIGINAL CERTIFICATE—AIRCRAFT IS		<input checked="" type="checkbox"/> NEW	USED	
	<input type="checkbox"/> 4	AMENDMENT OR MODIFICATION OF CURRENT CERTIFICATE				
	<input type="checkbox"/> 5	OTHER (Specify)				
	B. AIRWORTHINESS CLASSIFICATION (For multiple certification, check appropriate items)					
	<input checked="" type="checkbox"/> 1	STANDARD (Indicate category)		<input checked="" type="checkbox"/> NORMAL	UTILITY	
	<input type="checkbox"/> 2	LIMITED				
	<input type="checkbox"/> 3	RESTRICTED (Indicate operation(s) to be conducted)	<input type="checkbox"/> 1	AGRICULTURE & PEST CONTROL	<input type="checkbox"/> 2	AERIAL SURVEYING
			<input type="checkbox"/> 4	FOREST (Wild life conservation)	<input type="checkbox"/> 5	PATROLLING
			<input type="checkbox"/> 0	OTHER (Specify)	<input type="checkbox"/> 6	WEATHER CONTROL
<input type="checkbox"/> 4	EXPERIMENTAL (Indicate operation(s) to be conducted)	<input type="checkbox"/> 1	RESEARCH AND DEVELOPMENT	<input type="checkbox"/> 2	AMATEUR BUILT	
		<input type="checkbox"/> 4	RACING	<input type="checkbox"/> 0	TO SHOW COMPLIANCE WITH FAR	
<input type="checkbox"/> 5	PROVISIONAL (Indicate class)	<input type="checkbox"/> 1	CLASS I	<input type="checkbox"/> 2	CLASS II	
III. OWNER'S CERTIFICATION	A. REGISTERED OWNER (As shown on Certificate of Aircraft Registration)					
	NAME BEECH AIRCRAFT CORPORATION		ADDRESS (Number, street, city, State and ZIP code) 9709 E. Central Wichita, Kansas 67201			
	B. AIRCRAFT CERTIFICATION BASIS					
	<input checked="" type="checkbox"/> X	AIRCRAFT SPECIFICATION OR TYPE CERTIFICATION DATA SHEET (Give No. and Rev. No.) 3A16, Rev. 27		AIRWORTHINESS DIRECTIVE SUMMARY (Give year and last card No.)		
		AIRCRAFT LISTING (Give page No(s.))		SUPPLEMENTAL TYPE CERTIFICATE (Give No(s.))		
	C. AIRCRAFT OPERATION AND MAINTENANCE RECORDS					
	<input checked="" type="checkbox"/> X	AIRCRAFT NEW—NO PREVIOUS HISTORY		TOTAL AIRFRAME HOURS RECORDED	FAA CODING	
		RECORDS MAINTAINED IN COMPLIANCE WITH FAR 91.173				
	I hereby certify that I am the registered owner (or his agent) of the aircraft described above; that the aircraft is registered with the Federal Aviation Agency in accordance with section 501 of the Federal Aviation Act of 1958, and applicable Federal Aviation Regulations; and that the aircraft has been inspected and is airworthy and eligible for the classification(s) requested.					
	DATE OF APPLICATION May 3, 1967		NAME AND TITLE (Print or type) C. W. Myers - Leadman Final & Flight Inspection.		SIGNATURE <i>C. W. Myers</i>	
IV. INSPECTION AGENCY VERIFICATION	IN ACCORDANCE WITH FAR 21.183, THE AIRCRAFT DESCRIBED ABOVE HAS BEEN INSPECTED AND FOUND AIRWORTHY BY:					
	<input type="checkbox"/> 2	CERTIFICATED AIR CARRIER (Give Certificate No.)	<input type="checkbox"/> 3	CERTIFICATED MECHANIC (IA) (Give Certificate No.)	<input type="checkbox"/> 6	CERTIFICATED DOMESTIC REPAIR STATION (Give Certificate No.)
	<input type="checkbox"/> 5	AIRCRAFT MANUFACTURER (Give name of firm)				
	DATE				TITLE	
	SIGNATURE verified by Operator #79					
V. AIRWORTHINESS DOCUMENTATION	A. CURRENT OPERATING LIMITATIONS AVAILABLE IN AIRCRAFT		<input checked="" type="checkbox"/> X	G. THIS INSPECTION RECORDED IN AIRCRAFT RECORDS		
	B. CURRENT OPERATING LIMITATIONS ATTACHED			H. STATEMENT OF CONFORMITY, FAA FORM 317 (Attach when required)		
	<input checked="" type="checkbox"/> X	C. CURRENT APPROVED FLIGHT MANUAL AVAILABLE IN AIRCRAFT		I. PREVIOUS AIRWORTHINESS CERTIFICATE ISSUED IN ACCORDANCE WITH FAR _____ CAR _____ (Orig. attached)		
	D. DATA, DRAWINGS, PHOTOGRAPHS, ETC. (Attach when required)			J. AIRWORTHINESS CERTIFICATE ISSUED IN ACCORDANCE WITH FAR 21.183 (a) (Copy attached)		
	<input checked="" type="checkbox"/> X	E. CURRENT WEIGHT & BALANCE INFORMATION AVAILABLE IN AIRCRAFT				
	F. MAJOR REPAIR AND ALTERATION; FAA FORM-327 (Attach when req.)		<input checked="" type="checkbox"/> X			
VI. FAA REPRESENTATIVE CERTIFICATION	I have inspected the aircraft described above, find it conforms to its type certificate, and is in condition for safe operation.					
	B. I have inspected the aircraft described above, find it is in condition for safe experimental operation.					
	G. Current Airworthiness Certificate amended.					
DATE May 3, 1967		DESIGNATION NO. DOA CE-2		OFFICE NO. 1CT EMDO 3-0-43	SIGNATURE <i>Chester A. Rembleske</i> FOR BEECH AIRCRAFT CORPORATION	

MICRO

UNITED STATES OF AMERICA — FEDERAL AVIATION AGENCY

STANDARD AIRWORTHINESS CERTIFICATE

1. NATIONALITY AND REGISTRATION MARKS	2. MANUFACTURER AND MODEL	3. AIRCRAFT SERIAL NUMBER	4. CATEGORY
N2786T	BEECHCRAFT - 95-055	TE-387	Normal

5. AUTHORITY AND BASIS FOR ISSUANCE

This airworthiness certificate is issued pursuant to the Federal Aviation Act of 1958 and certifies that, as of the date of issuance, the aircraft to which issued has been inspected and found to conform to the type certificate therefor, to be in condition for safe operation, and has been shown to meet the requirements of the applicable comprehensive and detailed airworthiness code as provided by Annex 8 to the Convention on International Civil Aviation, except as noted herein.

Exceptions:

None.

6. TERMS AND CONDITIONS

Unless sooner surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator, this airworthiness certificate is effective as long as the maintenance, preventative maintenance, and alterations are performed in accordance with Parts 21, 43, and 91 of the Federal Aviation Regulations, as appropriate, and the aircraft is registered in the United States.

DATE OF ISSUANCE

May 3, 1967

FAA REPRESENTATIVE

Chester A. Hemblesie
FOR BEECH AIRCRAFT CORPORATION

DESIGNATION NUMBER

DOA CE-2

Any alteration, reproduction, or misuse of this certificate may be punishable by a fine not exceeding \$1,000, or imprisonment not exceeding 3 years, or both. THIS CERTIFICATE MUST BE DISPLAYED IN THE AIRCRAFT IN ACCORDANCE WITH APPLICABLE FEDERAL AVIATION REGULATIONS.

MICRO

UNITED STATES OF AMERICA
FEDERAL AVIATION AGENCY

Form approved.
Budget Bureau No. 04-RO59.

APPLICATION FOR EXPORT CERTIFICATE OF AIRWORTHINESS

EXPORT CERTIFICATE NO.

INSTRUCTIONS.—This application is to be submitted to an authorized FAA representative (one copy) when the product(s) to be exported is (are) presented for inspection. Use Part I for Class I products and Part II for Class II. For complete aircraft execute items 1 through 11 as applicable. For engines and propellers, omit items 5A and 6. Part III is for FAA use only.

PART I

1. APPLICATION IS MADE FOR AN EXPORT CERTIFICATE OF AIRWORTHINESS TO COVER THE PRODUCT(S) DESCRIBED BELOW, WHICH IS (ARE):

☐ NEW ☒ USED (Aircraft) ☐ NEWLY OVERHAULED

2. NAME AND ADDRESS OF EXPORTER

Combs Aircraft Inc.
Stapleton Intl Airport
Denver, Colorado

3. NAME AND ADDRESS OF FOREIGN PURCHASER

Archie Newall, Jr.
Calgary, Alberta Canada

4. COUNTRY OF DESTINATION

Canada

5. DESCRIPTION OF PRODUCT(S)

TYPE	MAKE AND MODEL	IDENTIFICATION NO.	SERIAL NUMBERS	FAA SPEC. NO.	OPERATING TIME (Hours)	
					SINCE OVERHAUL	TOTAL
a. AIRCRAFT	Beech 95C55	CF-DYN	TE-387	3A16		42.3
b. ENGINES	Continental 10-520-C		R. 125816-7-C	E5CE		42.3
	Continental 10-520-C		L-125793-7-C			
c. PROPELLERS	McCauley - 2AF-34C55-H		R-670103 L-671042	P5EA		42.3

6. THE AIRCRAFT WAS GIVEN A SATISFACTORY FLIGHT TEST ON _____
(Date)

7. DOES THE PRODUCT COMPLY WITH ALL APPLICABLE CIVIL AIR REGULATIONS, AIRWORTHINESS DIRECTIVES, AND OTHER FAA REQUIREMENTS?

☒ YES ☐ NO (Explain in "Remarks")

8. HAVE APPLICABLE SPECIAL REQUIREMENTS OF THE IMPORTING COUNTRY BEEN COMPLIED WITH?

☐ YES ☐ NO (Explain in "Remarks")

9. HAS PROPER PREVENTATIVE TREATMENT BEEN APPLIED TO PRODUCTS SUSCEPTIBLE TO RAPID CORROSION WHEN BEING SHIPPED?

☐ YES ☐ NO

10. REMARKS

11. EXPORTER'S CERTIFICATION

The undersigned certifies that the above statements are true and that the product(s) described herein is (are) airworthy and in condition for safe operation except as may be noted under item 10 "Remarks," above.

4/8/68

(Date)

Eurold Krizinger
(Signature of Applicant or Authorized Representative)

Chief Inspector

(Title)

PART II

12. APPLICATION IS MADE FOR APPROVAL OF AERONAUTICAL PARTS FOR EXPORT AS INDICATED BELOW:

13. NAME AND ADDRESS OF EXPORTER

14. NAME AND ADDRESS OF FOREIGN PURCHASER

15. COUNTRY OF DESTINATION

16. PARTS ARE ELIGIBLE FOR INSTALLATION ON _____

(Make and Model Class I Product)

(FAA Spec. No.)

17. THE PARTS ARE (check one):

☐ NEW

☐ NEWLY OVERHAULED

18. THE PARTS ARE DESCRIBED (check one):

☐ BY NAME, PART NUMBER, AND QUANTITY ON THE ATTACHED INVOICE OR PACKING SHEET NO. _____

☐ BELOW BY NAME, PART NUMBER, AND QUANTITY.

19. EXPORTER'S CERTIFICATION

I certify that the foregoing statements are true and that the parts described herein are airworthy and conform to FAA approved design data.

(Date)

(Signature of Applicant or Authorized Representative)

(Title)

PART III—APPROVAL (FOR FAA USE ONLY)

20. IT IS CONSIDERED THAT THE PRODUCT(S) DESCRIBED IN _____ IS (ARE) AIRWORTHY AND CONFORM(S) TO PERTINENT REQUIREMENTS EXCEPT AS NOTED IN ITEM 10.
(Part I or II)

(Signature)

(Number)

(Date)

Check One:

☐ DMIR

☐ DMCB

☐ DMR

☐ FAA INSPECTOR

21. _____ APPROVAL TAGS, FORMS FAA-186, WERE ISSUED FOR THE PARTS DESCRIBED IN PART II.
(Quantity)

22. EXPORT FILE SPOT-CHECKED BY:


Robert H. Lewis

(FAA Supervising Inspector)

3
(D. O. No.)

4/10/68
(Date)

The United States of America
Federal Aviation Agency
Washington, D.C.

No. E-76710

Export Certificate of Airworthiness

This certifies *that the product identified below and more particularly described in Specification (s)¹ of the Federal Aviation Agency, Numbered* **A-3A16, E-5CE-8, P-5EA-11** *has been examined and as of the date of this certificate, is considered airworthy in accordance with a comprehensive and detailed airworthiness code of the United States Government, and is in compliance with those special requirements of the importing country filed with the United States Government, except as noted below. This certificate in no way attests to compliance with any agreements or contracts between the vendor and purchaser, nor does it constitute authority to operate an aircraft.*

Product **Airplane**

Manufacturer **Beech Aircraft Corp.**

Model **95-C55**

Serial No. **TE-387**

New ☐ Newly Overhauled ☐

Used Aircraft ☒

Country to which exported **Canada**

Exceptions **None**

Engine Installed:

Continental IO-520-C

**Serial No. R-125816-7-C
L-125793-7-C**

Propeller Installed:

McCauley 2AF-34C-55H

**Serial No. R-670103
L-671042**

William A. Stewart
William A. Stewart

4/10/68

Date

Federal Aviation Administration

Agency Represented and Number
DEN GADO 4-5-03

District Office or Designee Number

¹ For complete aircraft, list applicable specification or Type Certificate Data Sheet numbers for the aircraft, engine, and propeller. Applicable specifications or Type Certificate Data Sheet, if not attached to this export certificate, will have been forwarded to the appropriate governmental office of the importing country.



US Department
of Transportation
Federal Aviation
Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020
11/30/2007

Electronic Tracking Number

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. §44701). Failure to report can result in a civil penalty for each such violation. (49 U.S.C. §46301(a))

1. Aircraft	Nationality and Registration Mark United States N71VK	Serial No. TE-387	
	Make Beechcraft	Model 95-C55	Series
2. Owner	Name (As shown on registration certificate) Phillips, Joe E	Address (As shown on registration certificate) Address P O Box 4554	
		City Ruidoso State NM	Zip 88345 Country

3. For FAA Use Only

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial No.
<input checked="" type="checkbox"/>	<input type="checkbox"/>	AIRFRAME		(As described in Item 1 above)	
<input type="checkbox"/>	<input type="checkbox"/>	POWERPLANT			
<input type="checkbox"/>	<input type="checkbox"/>	PROPELLER			
<input type="checkbox"/>	<input type="checkbox"/>	APPLIANCE	Type		
			Manufacturer		

6. Conformity Statement

A. Agency's Name and Address		B. Kind of Agency	
Name Melvern L Nelson		<input checked="" type="checkbox"/> U. S. Certificated Mechanic	Manufacturer
Address 208 Polk 87		<input type="checkbox"/> Foreign Certificated Mechanic	C. Certificate No.
City Mabelvale State AR		<input type="checkbox"/> Certificated Repair Station	1769599IA
Zip 72103 Country		<input type="checkbox"/> Certificated Maintenance Organization	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B <input type="checkbox"/>	Signature/Date of Authorized Individual September 30, 2010 Melvern L. Nelson
--	---

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ Approved ☐ Rejected

BY	FAA Fit. Standards Inspector	Manufacturer	Maintenance Organization	Persons Approved by Canadian Department of Transport
	FAA Designee	Repair Station	<input checked="" type="checkbox"/> Inspection Authorization	
Certificate or Designation No. 1769599IA		Signature/Date of Authorized Individual September 30, 2010 Melvern L. Nelson		

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

United States N71VK

09/30/2010

Nationality and Registration Mark

Date

Serial Number: TE-387

Repaired left hand elevator by installation of the following parts:

P/N

96-610006-9U Upper skin, new from manufacturer

96-610006-11U Lower skin, new from manufacturer

96-610006-3 Nose skin, new from manufacturer

45-610005-94 Rib, new from manufacturer

Symmetry and alignment were maintained during repairs with the use of fixtures. General procedures accomplished referencing Beech 95-C55 Service Manual and Parts Catalog. All parts installed with original type fasteners in accordance with AC 43.13-1B, Chapter 4, Section 4-57.

No change to empty weight and balance.

*****Nothing Follows*****

☐ Additional Sheets Are Attached



U.S. Department
of Transportation
Federal Aviation
Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020

For FAA Use Only

Office Identification

MRD SW11617FSA

INSTRUCTIONS: Print or type all entries, See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000.00 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make Beechcraft	Model 95-C55
	Serial No. TE-387	Nationality and Registration Mark United States N71VK
2. Owner	Name (As shown on registration certificate) Phillips, Joe E	Address (As shown on registration certificate) P O Box 4554 Ruidoso, NM 88345

3. For FAA Use Only

4. Unit Identification

5. Type

Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	~~~~~ (As described in Item 1 above) ~~~~~				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency	C. Certificate No.
Vernon Hampton 1307 Ransom Road Mena, AR 71953	<input checked="" type="checkbox"/> U.S. Certificated Mechanic	431417955
	<input type="checkbox"/> Foreign Certificated Mechanic	
	<input type="checkbox"/> Certificated Repair Station	
	<input type="checkbox"/> Manufacturer	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date 05/06/2005	Signature of Authorized Individual Vernon Hampton <i>[Signature]</i>
---------------------------	--

7. Approval For Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is

☒ APPROVED ☐ REJECTED

BY	FAA Fit Standards Inspector	Manufacturer	<input checked="" type="checkbox"/>	Inspection Authorization	Other (Specify)
	FAA Designee	Repair Station	<input type="checkbox"/>	Person Approved by Transport Canada Airworthiness Group	
Date of Approval or Rejection 05/06/2005		Certificate or Designation No. 431417955	Signature of Authorized Individual Vernon Hampton <i>[Signature]</i>		

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Date: May 6, 2005 Reg. # N71VK S/N TE-387

Installed new pilot's side window P/N WT-2237 in accordance with Great Lakes Aero Products drawing # W-2237 dated October 17, 1996 and installation instructions #2AF dated 10/17/96 approved by Supplemental Type Certificate # SA00690CH, amended Feb. 18, 1999.

Installed new entry door side window P/N WT-2049 in accordance with Great Lakes aero Products drawing # WT-2049, Rev. C, dated Oct. 6, 1998 and installation instructions #A2 dated 10/17/96 approved by Supplemental Type Certificate # SA00690CH, amended Feb. 18, 1999.

Revised aircraft weight and balance:—

*****END*****

RECEIVED

MAY n 9 2005

LIT-FSDO-11

☐ Additional Sheets are Attached



U.S. Department
of Transportation
Federal Aviation
Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020

For FAA Use Only

Office Identification

2011 11/15/05

INSTRUCTIONS: Print or type all entries, See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000.00 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make Beechcraft	Model 95-C55
	Serial No. TE-387	Nationality and Registration Mark United States N71VK
2. Owner	Name (As shown on registration certificate) Phillips, Joe E	Address (As shown on registration certificate) P O Box 4554 Ruidoso, NM 88345

3. For FAA Use Only

4. Unit Identification				5. Type	
Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	~~~~~ (As described in Item 1 above) ~~~~~			X	
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency	C. Certificate No.
Vernon Hampton 1307 Ransom Road Mena, AR 71953	<input checked="" type="checkbox"/> U.S. Certificated Mechanic	431417955
	<input type="checkbox"/> Foreign Certificated Mechanic	
	<input type="checkbox"/> Certificated Repair Station	
	<input type="checkbox"/> Manufacturer	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date 05/06/2005	Signature of Authorized Individual Vernon Hampton <i>Ce</i>
--------------------	--

7. Approval For Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is

☒ APPROVED ☐ REJECTED

BY	FAA Fit Standards Inspector	Manufacturer	X	Inspection Authorization	Other (Specify)
	FAA Designee	Repair Station		Person Approved by Transport Canada Airworthiness Group	
Date of Approval or Rejection 05/06/2005		Certificate or Designation No. 431417955	Signature of Authorized Individual Vernon Hampton <i>Ce</i>		

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Date: May 6, 2005

Registration: N71VK

S/N: TE-387

Repaired belly section of airframe as follows:

Installed the following new parts:

P/N 002-410066-29	Keel, LH fwd - new from manufacturer
002-410066-33	Keel, LH aft - new from manufacturer
96-420001-121	Skin, lower fwd - new from manufacturer
96-410031-37	Skin, lower LH fwd - new from manufacturer
*95-420015-55	Skin, lower center - locally fabricated
*96-420001-67	Skin, lower LH mid - locally fabricated
*96-420001-68	Skin, lower RH mid - locally fabricated.

Repaired lower flanges of belly bulkheads at station 68.00, station 82.96 and station 83.09 by installation of repair angles fabricated from 2024-T3 .032 alloy using procedures outlined in AC 43.13-1B, Chapter 4, Sections 4-50, 4-53 4-57 and 4-58.

General procedures were followed referencing Beechcraft 55 Shop Manual and Parts Catalog. All new parts installed using original type rivets in accordance with AC 43.13-1B, Chapter 4, Section 4-57. Some skins identified above with an (*) are locally fabricated from original type and thickness material using original skins as templates and procedures outlined in AC 43.13-1B, Chapter 4, Section 4-50 and 4-53.

No change to aircraft weight and balance.

*****NOTHING FOLLOWS*****

RECEIVED

MAY 19 2005

LIT-FSDO-11



U.S. Department
of Transportation
Federal Aviation
Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020

For FAA Use Only

Office Identification

21810 8/11/05

INSTRUCTIONS: Print or type all entries, See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000.00 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make Beechcraft	Model 95-C55
	Serial No. TE-387	Nationality and Registration Mark United States N71VK
2. Owner	Name (As shown on registration certificate) Phillips, Joe E	Address (As shown on registration certificate) P O Box 4554 Ruidoso, NM 88345

3. For FAA Use Only

--	--	--	--	--	--

4. Unit Identification

5. Type

Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	----- (As described in Item 1 above) -----				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency	C. Certificate No.
Vernon Hampton	<input checked="" type="checkbox"/> U.S. Certificated Mechanic	431417955
1307 Ransom Road	<input type="checkbox"/> Foreign Certificated Mechanic	
Mena, AR 71953	<input type="checkbox"/> Certificated Repair Station	
	<input type="checkbox"/> Manufacturer	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date 05/06/2005	Signature of Authorized Individual Vernon Hampton <i>[Signature]</i>
--------------------	---

7. Approval For Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is

☒ APPROVED ☐ REJECTED

BY	FAA Fit Standards Inspector	Manufacturer	<input checked="" type="checkbox"/>	Inspection Authorization	Other (Specify)
	FAA Designee	Repair Station		Person Approved by Transport Canada Airworthiness Group	

Date of Approval or Rejection 05/06/2005	Certificate or Designation No. 431417955	Signature of Authorized Individual Vernon Hampton <i>[Signature]</i>
---	---	---

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Date: May 6, 2005 Reg. # N71VK S/N TE-387

Installed battery vent fairing in accordance with South Seas Ventures, Inc. Drawing, P/N 15627A, Rev. 7, dated 5/17/01 and approved by Supplemental Type Certificate #SA01151CH, dated April 14, 2004.

Revised pilot's operating handbook with Supplement #P/M002-40001-33.

Revised aircraft weight and balance.

*****END*****

RECEIVED

MAY 09 2005

LIT-FSDO-11

☐ Additional Sheets are Attached



U.S. Department
of Transportation
Federal Aviation
Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020

For FAA Use Only

Office Identification

CW-FSDO-19

LB2 FSDO
NOV 27 2001

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make RATHEON	Model 95-C55
	Serial No. TE387	Nationality and Registration Mark N71VK
2. Owner	Name (As shown on registration certificate) EDWARD JOE PHILLIPS	Address (As shown on registration certificate) PO BOX 4554 RUIDODO NM 88345

3. For FAA Use Only

4. Unit Identification				5. Type	
Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	(As described in item 1 above)				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency	C. Certificate No.
GENE SEITZ 2313-58TH LUBBOCK, TEXAS 79412	<input checked="" type="checkbox"/> U.S. Certificated Mechanic	2148062
	<input type="checkbox"/> Foreign Certificated Mechanic	
	<input type="checkbox"/> Certificated Repair Station	
	<input type="checkbox"/> Manufacturer	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date 11-9-2001	Signature of Authorized Individual <i>Gene Seitz</i>
-------------------	---

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA Fit Standards Inspector	Manufacturer	Inspection Authorization		
	FAA Designee	Repair Station	Person Approved by Transport Canada Airworthiness Group		
Date of Approval or Rejection 11-26-2001		Certificate or Designation No. 2148062	Signature of Authorized Individual <i>Gene Seitz</i>		

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

71VK

INSTALLED AIR/OIL SEPARATOR MANUFACTURED BY M-20 OIL SEPARATOR, LLC UNDER STC SA02033AT ON THE LEFT SIDE OF ENGINE AS PER DRAWING #304. INSTALLED DRAIN FROM SEPARATOR TO LOW PRESSURE PORT ON LEFT SIDE OF ENGINE.

***** NOTHING FOLLOWS *****



Additional Sheets Are Attached

United States of America
Department of Transportation - Federal Aviation Administration
Supplemental Type Certificate
(Continuation Sheet)

Number SA02033AT
Date of Issuance: November 19, 1999
Date Amended: December 21, 2000
Date Reissued: January 18, 2001

Model 300 Eligibility Continued:

Piper Aircraft Corporation (cont'd):

1A4	PA-20, -20S, -20-115", -20S-115", -20-135", -20S-135"	CAR 3
1A6	PA-22, -22-108, -22-135, -22S-135, -22-150, -22S-150, -22-160, -22S-160	CAR 3
1A10	PA-23, PA-23-160, PA-23-235, PA-23-250 (Navy UO-1), PA-E23-250	CAR 3
A1EA	PA-30, PA-39, PA-40	CAR 3, FAR 23
A20SO	PA-31, PA-31-300, PA-31-325, PA-31-350	CAR 3, FAR 23
A7SO	PA34-200, -200T, -220T (Seneca III), -220T (Seneca IV), -220T (Seneca V)	FAR 23
A9SO	PA-36-285, PA-36-300	FAR 23
A18SO	PA-38-112	FAR 23, 36
A19SO	PA-44-180, -44-180T	FAR 23
A25SO	PA-46-310P, PA-46-350P	FAR 23, 25, 36

Raytheon Aircraft Company (Formerly Beech Aircraft Corporation):

A777	35, A35, B35, C35, D35, E35, G35 F35, 35R	CAR 3
3A15	F35, S35, V35A, 36, A36	CAR 3
3A16	95-C35, 95-C35A, D55, D55A, E55, E55A, 58, 58A	CAR 3
AICE	19A, B19, M19A, 23, A23, A23A, A23-19, A23-24, B23, C23, A24, A24R, B24R, C24R	CAR 3
3A15	35-33, 35-A33, 35-B33, 35-C33, 35-C33A, E33, E33A, E33C, F33, F33A, F33C, G33, H35, J35, K35, M35, N35, P35, S35, V35, V35A, V35B, 36, A36, A36TC, B36TC	CAR 3, FAR 23, FAR 36



US Department
of Transportation
Federal Aviation
Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020

For FAA Use Only

Office Identification

507SD005 MA

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make BEEHCRAFT	Model 95C55
	Serial No. TE-387	Nationality and Registration Mark N71VK
2. Owner	Name (As shown on registration certificate) Chris J. Price	Address (As shown on registration certificate) 6100 O'Briant Court Greensboro, NC 27410

3. For FAA Use Only

4. Unit Identification

5. Type

Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	~~~~~ (As described in Item 1 above) ~~~~~				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency	C. Certificate No.
Chris J. Price 6100 O'Briant Court Greensboro, NC 27410	<input checked="" type="checkbox"/> U.S. Certificated Mechanic	305569359
	<input type="checkbox"/> Foreign Certificated Mechanic	
	<input type="checkbox"/> Certificated Repair Station	
	<input type="checkbox"/> Manufacturer	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date 15 September 1999	Signature of Authorized Individual
---------------------------	--

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ APPROVED ☐ REJECTED

BY	FAA Fit. Standards Inspector	Manufacturer	<input checked="" type="checkbox"/>	Inspection Authorization	Other (Specify)
	FAA Designee	Repair Station		Person Approved by Transport Canada Airworthiness Group	
Date of Approval or Rejection 15 September 1999		Certificate or Designation No. 305569359IA		Signature of Authorized Individual 	

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

REMOVED BRITIAN B-S AUTOPILOT SYSTEMS AND ALL COMPONENTS FROM AIRCRAFT.

INSTALLED S-TEC 50 TWO AXIS AUTOMATIC FLIGHT GUIDANCE SYSTEM MODEL ST-223. THE INSTALLATION WAS PER SUPPLEMENTAL TYPE CERTIFICATE NUMBER SA5290SW-D. ALSO USED WAS BULLETIN NO. 323, REVISION 3, DATED 1-23-96 AND MASTER DRAWING LIST NO. 92251, REVISION C, DATED 1-23-96. FAA APPROVED SUPPLEMENT TO PILOT'S OPERATING HANDBOOK OR FAA APPROVED FLIGHT MANUAL P/N 89349, DATED 3-27-84 WAS INSTALLED IN AIRCRAFT.

THE ABOVE INSTALLATION MEETS OR EXCEEDS THE APPLICABLE SECTIONS OF AC 43.13-1B AND AC 43.13-2A AS A REFERENCE GUIDE. THE AIRCRAFT EQUIPMENT LIST WAS REVISED AND WEIGHT AND BALANCE DATA UPDATED. AN ENTRY WAS MADE IN THE PERMANENT LOGS CONCERNING THE INSTALLATION.

-----END-----

☐ Additional Sheets Are Attached



US Department
of Transportation
Federal Aviation
Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020

For FAA Use Only

Office Identification

50750005 MS

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make BEEHCRAFT	Model 95C55
	Serial No. TE-387	Nationality and Registration Mark N71VK
2. Owner	Name (As shown on registration certificate) Chris J. Price	Address (As shown on registration certificate) 6100 O'Briant Court Greensboro, NC 27410

3. For FAA Use Only

4. Unit Identification

5. Type

Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	~~~~~ (As described in Item 1 above) ~~~~~				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency	C. Certificate No.
Chris J. Price 6100 O'Briant Court Greensboro, NC 27410	<input checked="" type="checkbox"/> U.S. Certificated Mechanic	305569359
	<input type="checkbox"/> Foreign Certificated Mechanic	
	<input type="checkbox"/> Certificated Repair Station	
	<input type="checkbox"/> Manufacturer	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date 6/1/99	Signature of Authorized Individual Chris J. Price
----------------	--

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ APPROVED ☐ REJECTED

BY	FAA Flt. Standards Inspector	Manufacturer	<input checked="" type="checkbox"/>	Inspection Authorization	Other (Specify)
	FAA Designee	Repair Station		Person Approved by Transport Canada Airworthiness Group	
Date of Approval or Rejection 6/1/99		Certificate or Designation No. 305569359	Signature of Authorized Individual Chris J. Price		

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

REMOVED ORIGINAL ROTATING BEACON AND INSTALLED WHELEN DIRECT REPLACEMENT ANTI-COLLISION STROBE LIGHT. INSTALLATION IS PER STC SA615EA AND WHELEN INSTALLATION AND SERVICE MANUAL DATED AUGUST 1, 1996. THE WORK MEETS OR EXCEEDS ALL APPLICABLE SECTIONS OF AC 43.13-1B AND AC 43.13-2A. THE EQUIPMENT LIST AND WEIGHT AND BALANCE WERE UPDATED. AN ENTRY WAS MADE IN THE PERMANENT LOGS CONCERNING THIS INSTALLATION.

-----END-----

☐ Additional Sheets Are Attached



US Department
of Transportation
Federal Aviation
Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020

For FAA Use Only

Office Identification

50750005 MJ

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make BEECHCRAFT	Model 95C55
	Serial No. TE-387	Nationality and Registration Mark N71VK
2. Owner	Name (As shown on registration certificate) CHRIS J. PRICE	Address (As shown on registration certificate) 6100 O'Briant Court Greensboro, NC 27410

3. For FAA Use Only

4. Unit Identification

5. Type

Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	~~~~~ (As described in Item 1 above) ~~~~~				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

6. Conformity Statement

A. Agency's Name and Address Chris J. Price 6100 O'Briant Court Greensboro, NC 27410	B. Kind of Agency <input checked="" type="checkbox"/> U.S. Certificated Mechanic <input type="checkbox"/> Foreign Certificated Mechanic <input type="checkbox"/> Certificated Repair Station <input type="checkbox"/> Manufacturer	C. Certificate No. 305569359
---	--	---------------------------------

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date 6/1/99	Signature of Authorized Individual Chris J. Price
----------------	--

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ APPROVED ☐ REJECTED

BY	FAA Flt. Standards Inspector	Manufacturer	<input checked="" type="checkbox"/> Inspection Authorization	Other (Specify)
	FAA Designee	Repair Station	Person Approved by Transport Canada Airworthiness Group	
Date of Approval or Rejection 6/1/99		Certificate or Designation No. 305569359	Signature of Authorized Individual Chris J. Price	

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

REMOVED THE FOLLOWING RADIOS: TELEX INTERCOM, BENDIX T12D ADF SYSTEMS (2EA) AND FOSTER LRN 500 LORAN. INSTALLED THE FOLLOWING RADIOS: KR87 ADF RECEIVER, KI227 ADF INDICATOR, KA44B ADF LOOP ANTENNA, KN64 DME RECEIVER, AND PS-3000 INTERCOM SYSTEM. THE RADIOS WERE INSTALLED IN ACCORDANCE WITH MANUFACTURER'S INSTALLATION MANUAL AND USING APPLICABLE SECTIONS OF AC 43.13-1B AND AC 43.13-2A AS A REFERENCE GUIDE TO APPROVED INSTALLATION PRACTICES. THE EQUIPMENT INSTALLED WAS RAMP TESTED AND FOUND TO FUNCTION PROPERLY WITH NO INTERFERENCE TO ANY OTHER SYSTEM OR FLIGHT CONTROL. THE EQUIPMENT LIST WAS AMENDED AND WEIGHT AND BALANCE DATA UPDATED. AN ENTRY WAS MADE IN THE PERMANENT LOGS CONCERNING THIS INSTALLATION.

-----END-----

☐ Additional Sheets Are Attached



US Department
of Transportation
Federal Aviation
Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved

OMB No. 2120-0020

For FAA Use Only

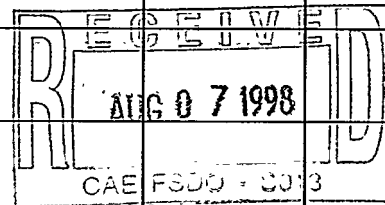
Office Identification
50-FSDO-13

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each violation (Section 901 of Federal Aviation Act of 1958).

1. Aircraft	Make BEECHCRAFT	Model 95-C55
	Serial No. TE-387	Nationality and Registration Mark N71VK
2. Owner	Name (As shown on registration certificate) KEENER & ASSOC. Inc	Address (As shown on registration certificate) PO BOX 1545 2800 Terminal Blvd N. MYRTLE BEACH, SC 29598

3. For FAA Use Only

4. Unit Identification				5. Type	
Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	~~~~~ (As described in Item 1 above) ~~~~~				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				



6. Conformity Statement

A. Agency's Name and Address CARL K KNUTH PO BOX 1114 N MYRTLE BEACH, SC 29598	B. Kind of Agency <input checked="" type="checkbox"/> U.S. Certified Mechanic <input type="checkbox"/> Foreign Certified Mechanic <input type="checkbox"/> Certified Repair Station <input type="checkbox"/> Manufacturer	C. Certificate No. 5432603728 JUL 1998
--	---	--

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date 28 JUL 1998	Signature of Authorized Individual
----------------------------	--

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ APPROVED ☐ REJECTED

BY	FAA Fit. Standards Inspector	Manufacturer	<input checked="" type="checkbox"/>	Inspection Authorization	Other (Specify)
	FAA Designee	Repair Station	<input type="checkbox"/>	Person Approved by Transport Canada Airworthiness Group	
Date of Approval or Rejection 28 JUL 1998		Certificate or Designation No. 54326037	Signature of Authorized Individual 		

NOTICE

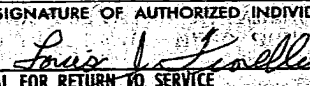
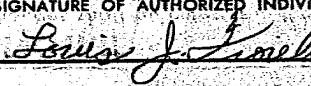
Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

REMOVED 2ea PAPER ELEMENT AIR FILTERS AND REPLACED WITH 2ea BRACKETT AIR
FILTERS IN ACCORDANCE WITH STC-SA71GL LIST No.1 AND FAA-PMA SUPPLEMENT No.1.
NO CHANGE IN WEIGHT AND BALANCE. END

☐ Additional Sheets are Attached

DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)				Form Approved Budget Bureau No. 04-R060.1 FOR FAA USE ONLY OFFICE IDENTIFICATION S0-15	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.					
1. AIRCRAFT	MAKE	BEECH		MODEL	95-C55
	SERIAL NO.	S/N TE 387		NATIONALITY AND REGISTRATION MARK	N71VK
2. OWNER	NAME (As shown on registration certificate)			ADDRESS (As shown on registration certificate)	
	AGRICULTURAL LAND DEVELOPMENT INC.			6249 SE 58th AVE. OCALA, FL. 34480	
3. FOR FAA USE ONLY					
4. UNIT IDENTIFICATION					
UNIT	MAKE	MODEL		SERIAL NO.	
AIRFRAME	(As described in item 1 above)				
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
5. TYPE					
REPAIR ALTERATION					
X					
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS		B. KIND OF AGENCY		C. CERTIFICATE NO.	
LOUIS J. FIORELLI		X U.S. CERTIFICATED MECHANIC		A&P1046134	
222 MARION OAKS LN		FOREIGN CERTIFICATED MECHANIC			
OCALA, FL. 34473		CERTIFICATED REPAIR STATION			
		MANUFACTURER			
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE		SIGNATURE OF AUTHORIZED INDIVIDUAL			
7/5/93					
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is: <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	X INSPECTION AUTHORIZATION	OTHER (Specify)	
	FAA DESIGNEE	REPAIR STATION	CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT		
DATE OF APPROVAL OR REJECTION		CERTIFICATE OR DESIGNATION NO.	SIGNATURE OF AUTHORIZED INDIVIDUAL		
7/5/93		IA1046134			

NOTICE

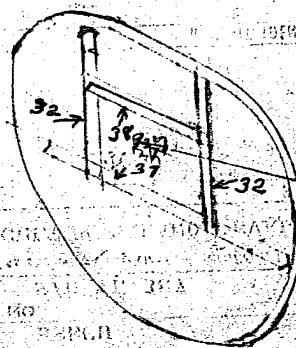
Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

337 REPAIR

1. REPAIRED BUCKLED SKIN WEB OF FORWARD BULKHEAD STA. 0.00 IN ACCORDANCE WITH ARMY AVIATION MAINTENANCE ENGINEERING MANUAL AIRCRAFT STRUCTURE. REPAIRED PER TM 55-405-4 AND APPLICABLE PORTIONS OF AC 42.13-1A. MATERIAL SAME GAUGE AND ALLOY AS ORIGINAL 2024T3. RIVETS WERE SOLID 2117-T4 AN470-4 RIVET PATTERN SAME AS ORIGINAL.
2. REPLACED DAMAGED OUTBOARD LEFT HAND TRAILING EDGE WING SKIN P/N 95-110014-199 AND LOWER OUTBOARD TRAILING EDGE SKIN P/N 95-110014-211 FACTORY PARTS WERE USED. ALL RIVETS FOR REPAIR SAME AS ORIGINAL ALSO USING PORTIONS OF AC 43.13-1A.

SEE DRAWING BELOW ON BULKHEAD REPAIR. CHAPTER 2 AC 43.13-1A METAL STRUCTURES SECTION 3 METAL REPAIR PROCEDURES PAR. 96 PAR 97.



STA. 0.00 FUSELAGE BULKHEAD

SLIGHT BUCKLE IN BULKHEAD
INSTALLED DOUBLER UNDER 32, 37, 38
EXISTING ANGLES

☐ ADDITIONAL SHEETS ARE ATTACHED



MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020

For FAA Use Only

Office Identification
50-FSB-15

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make	BEECH	Model	95-C55
	Serial No.	S/N TE 387	Nationality and Registration Mark	N71VK
2. Owner	Name (As shown on registration certificate)		Address (As shown on registration certificate)	
	AGRICULTURAL LAND DEVELOPEMENT INC		6249 SE 58th AVE OCALA, FL. 32671	

3. For FAA Use Only

4. Unit Identification				5. Type	
Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	(As described in Item 1 above)				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

A. Agency's Name and Address

LOUIS J. FIORELLI
3700 SW 157th ST. RD.
OCALA, FL. 32673

B. Kind of Agency

- ☒ U.S. Certified Mechanic
☐ Foreign Certified Mechanic
☐ Certified Repair Station
☐ Manufacturer

C. Certificate No.

A&P1046134

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date

7/5/93

Signature of Authorized Individual

Louis J. Fiorelli

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ APPROVED ☐ REJECTED

BY	FAA Ftl. Standards Inspector	Manufacturer	<input checked="" type="checkbox"/> Inspection Authorization	Other (Specify)
	FAA Designee	Repair Station	Person Approved by Transport Canada Airworthiness Group	
Date of Approval or Rejection		Certificate or Designation No.	Signature of Authorized Individual	
7/5/93		IA1046134	<i>Louis J. Fiorelli</i>	

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.


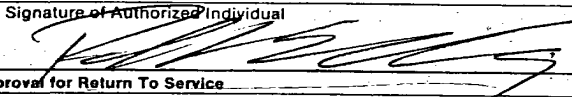
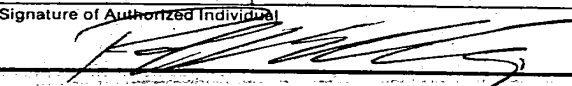
8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

337 Alteration

1. Removed damaged nose cone P/N 96-410021-607 and installed new Beechcraft nose cone P/N 96-410021-609 optional part per equipment list.
2. Installed new Beechcraft factory taxi lite assembly P/N 96-367045. Installed Beechcraft kit no. 58-3004-1P Equipment list was up dated to reflect change in aircraft log book.

☐ Additional Sheets Are Attached

 US Department of Transportation Federal Aviation Administration		MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)		Form Approved OMB No. 2120-0020 For FAA Use Only Office Identification SD-FSD00-15	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).					
1. Aircraft	Make BEECH		Model 95-C55		
	Serial No. TE-387		Nationality and Registration Mark N71VK		
2. Owner	Name (As shown on registration certificate) AGRICULTURAL LAND DEVELOPMENT, INC.		Address (As shown on registration certificate) 6249 SE. 58 AVE. OCALA, FL. 32671		
3. For FAA Use Only					
4. Unit Identification					
Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	(As described in Item 1 above)				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				
6. Conformity Statement					
A. Agency's Name and Address		B. Kind of Agency		C. Certificate No.	
OCALA AVIONICS, INC. P.O. BOX 2853 OCALA, FL. 32678		U.S. Certified Mechanic		VH4R596M	
		Foreign Certified Mechanic			
		X Certified Repair Station			
		Manufacturer			
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
Date 11-4-91		Signature of Authorized Individual 			
7. Approval for Return To Service					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA Fit. Standards Inspector	Manufacturer	Inspection Authorization	Other (Specify)	
	FAA Designee	Repair Station	Person Approved by Transport Canada Airworthiness Group		
Date of Approval or Rejection 11-4-91		Certificate or Designation No. VH4R596M	Signature of Authorized Individual 		

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

REMOVED

- 1ea. KING KMA-12B AUDIO/MARKER SYSTEM
- 1ea. KING KX-160 NAV/COM SYSTEM
- 1ea. SPERRY C-2 RMI/COMPASS SYSTEM

INSTALLED

- 1ea. KING KMA-24 AUDIO/MARKER SYSTEM
- 1ea. KING KX-155 COM/NAV/GS SYSTEM WITH KI-209 VOR/ILS INDICATOR
- 1ea. BFG WX-900 STORMSCOPE SYSTEM
- 1ea. TELEX TC-200 INTERCOM SYSTEM
- 1ea. DAVTRON 655-2 DENSITY ALTITUDE INDICATOR
- 1ea. TROLL FN-200 FAN

SYSTEMS INSTALLED IN ACCORDANCE WITH MAUFACTURERS SPECIFICATIONS AND AC 43.13-2A.

PANEL UNITS MOUNTED IN RADIO STACK AT STATION 63.0 AS PER PARAGRAPH 23 OF AC 43.13-2A. USED EXISTING ANTENNA SYSTEMS ON AIRCRAFT.

STORMSCOPE ANTENNA MOUNTED ON BELLY AT STATION 238.0 AS PER PARAGRAPH 43 OF AC43.13-2A.

COOLING FAN MOUNTED IN BAGGAGE COMPARTMENT AT STATION 40.0 AS PER PARAGRAPH 25 OF AC 43.13-2A.

ALL WORK PERFORMED USING ACCEPTABLE METHODS, PRACTICES AND TECHNIQUES OUTLINED IN CHAPTER 11 OF AC 43.13-1A AND CHAPTERS 1, 2 & 3 OF AC 43.13-2A

EQUIPMENT LIST AND WEIGHT AND BALANCE WERE REVISED TO REFLECT CHANGES
TOTAL ELECTRICAL LOAD DOES NOT EXCEED 80% OF TOTAL AIRCRAFT LOAD.

SYSTEMS FUNCTION TESTED AND FOUND TO PERFORM SATISFACTORILY.

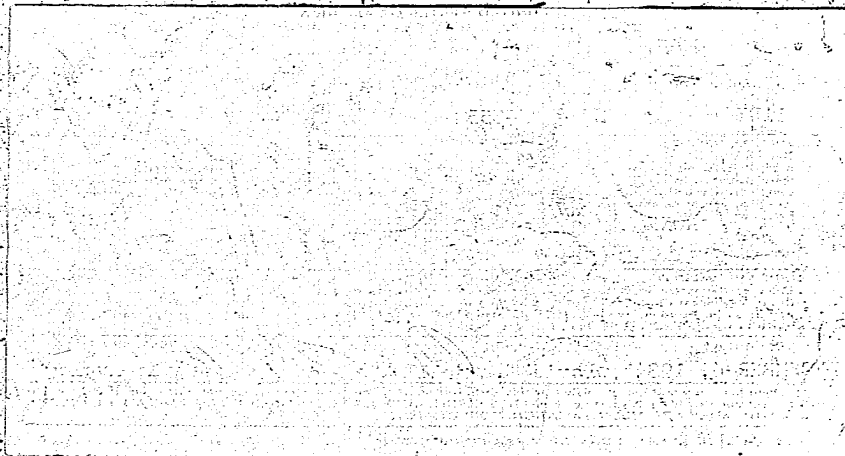
-----END-----

☐ Additional Sheets Are Attached

UNITED STATES OF AMERICA DEPARTMENT OF TRANSPORTATION—FEDERAL AVIATION ADMINISTRATION			
STANDARD AIRWORTHINESS CERTIFICATE			
1. NATIONALITY AND REGISTRATION MARKS	2. MANUFACTURER AND MODEL	3. AIRCRAFT SERIAL NUMBER	4. CATEGORY
N11DY	BEECH 99-C55	TE-387	NORMAL
5. AUTHORITY AND BASIS FOR ISSUANCE This airworthiness certificate was issued pursuant to the Federal Aviation Act of 1958 and certifies that, as of the date of issuance, the aircraft to which issued herein is inspected and found to conform to the type certificate therefor, to be in condition for safe operation, and has been shown to meet the requirements of the applicable comprehensive and detailed airworthiness code as provided by Annex 8 to the Convention on International Civil Aviation, except as noted herein. Exceptions: NONE			
6. TERMS AND CONDITIONS Unless sooner surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator, this airworthiness certificate is effective as long as the maintenance, preventative maintenance, and alterations are performed in accordance with Parts 43, and 91 of the Federal Aviation Regulations, as appropriate, and the aircraft is registered in the United States.			
DATE OF ISSUANCE	FAA REPRESENTATIVE	DESIGNATION NUMBER	
DECEMBER 6, 1984	David T. [Signature]	DAR-10-FS-SW	
Any alteration, reproduction, or misuse of this certificate may be punished by a fine not exceeding \$1,000, or imprisonment not exceeding 3 years, or both. THIS CERTIFICATE MUST BE DISPLAYED IN THE AIRCRAFT IN ACCORDANCE WITH APPLICABLE FEDERAL AVIATION REGULATIONS.			
FAA Form 8100-2 (7-67) FORMERLY FAA FORM 1362			

FAA AIRCRAFT REGISTRY
CAMERA NO. 3

DATE: 10-4-91



UNITED STATES OF AMERICA
DEPARTMENT OF TRANSPORTATION—FEDERAL AVIATION ADMINISTRATION
STANDARD AIRWORTHINESS CERTIFICATE

1. NATIONALITY AND REGISTRATION MARKS N71VK	2. MANUFACTURER AND MODEL BEECH 95-C55	3. AIRCRAFT SERIAL NUMBER TE-387	4. CATEGORY NORMAL
--	--	---	------------------------------

5. AUTHORITY AND BASIS FOR ISSUANCE
This airworthiness certificate is issued pursuant to the Federal Aviation Act of 1958 and certifies that, as of the date of issuance, the aircraft to which issued has been inspected and found to conform to the type certificate therefor, to be in condition for safe operation, and has been shown to meet the requirements of the applicable comprehensive and detailed airworthiness code as provided by Annex 8 to the Convention on International Civil Aviation, except as noted herein.
Exceptions:

NONE

6. TERMS AND CONDITIONS
Unless sooner surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator, this airworthiness certificate is effective as long as the maintenance, preventative maintenance, and alterations are performed in accordance with Parts 21, 43, and 91 of the Federal Aviation Regulations, as appropriate, and the aircraft is registered in the United States.

(Replacement) December 6, 1984	FAA REPRESENTATIVE Emanuel Sylvia	DESIGNATION NUMBER SO-FSDO-15
-----------------------------------	---	---

Any alteration, reproduction, or misuse of this certificate may be punishable by a fine not exceeding \$1,000, or imprisonment not exceeding 3 years, or both. THIS CERTIFICATE MUST BE DISPLAYED IN THE AIRCRAFT IN ACCORDANCE WITH APPLICABLE FEDERAL AVIATION REGULATIONS.

FAA Form 8100-2 (8-82)

GPO 892-604

FAA AIRCRAFT REGISTRY
CAMERA NO. 3

DATE: 10-4-91



DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				RECEIVED	
MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)				Form Approved FAA Form No. 04-R060.1 FOR FAA USE ONLY	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.				OFFICE IDENTIFICATION	
1. AIRCRAFT		MAKE Beechcraft	MODEL B995-C55	NATIONALITY AND REGISTRATION MARK NLDY	
2. OWNER		NAME (As shown on registration certificate) Danny White Inc.	ADDRESS (As shown on registration certificate) 6510 LBJ Freeway #200 Dallas, Tx. 75420		
3. FOR FAA USE ONLY					
4. UNIT IDENTIFICATION					
UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE	
AIRFRAME	***** (As described in item 1 above) *****			REPAIR	ALTERATION
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS		B. KIND OF AGENCY		C. CERTIFICATE NO.	
MARK ARDIZZONE Rt. 1 Rockhill Rd. McKinney Tx. 75069		<input checked="" type="checkbox"/> U.S. CERTIFICATED MECHANIC <input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC <input type="checkbox"/> CERTIFICATED REPAIR STATION <input type="checkbox"/> MANUFACTURER		452157668	
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE 9-30-87		SIGNATURE OF AUTHORIZED INDIVIDUAL Mark Ardizzone			
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED:					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	<input checked="" type="checkbox"/> INSPECTION AUTHORIZATION	OTHER (Specify)	
	FAA DESIGNEE	REPAIR STATION	CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT		
DATE OF APPROVAL OR REJECTION 9-30-87		CERTIFICATE OR DESIGNATION NO. 452157668	SIGNATURE OF AUTHORIZED INDIVIDUAL Mark Ardizzone		

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

N11DY

9-30-87

1. Installed Foster Air data LRN 500 Loran in R. H. radio rack with antenna and preamp in overhead area of cabin.
2. Installation completed using manufacturers instructions and A.C. 43.13-1A Chapter 2 + 11 and A.C. 43.13-2A Chapter 2 para 23, 27, Chapter 3 paragraph 30.
3. Installation placarded: "for VFR use only".
4. Weight and balance computed and entered in aircraft records.

END

☐ ADDITIONAL SHEETS ARE ATTACHED

DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				Form Approved Budget Bureau No. 04-R060.1	
MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)				FOR FAA USE ONLY	
				OFFICE IDENTIFICATION	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.					
1. AIRCRAFT	MAKE	Beech		MODEL	95-C55
	SERIAL NO.	TE-387		NATIONALITY AND REGISTRATION MARK	N11DY
2. OWNER	NAME (As shown on registration certificate)			ADDRESS (As shown on registration certificate)	
	Danny White, Inc.			6510 LBJ Freeway #200 Dallas, Tx 75240	
3. FOR FAA USE ONLY					
4. UNIT IDENTIFICATION					
UNIT	MAKE	MODEL	SERIAL NO.	REPAIR	ALTERATION
AIRFRAME	***** (As described in item 1 above) *****				XX
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
5. TYPE					
RECEIVED APR 18 1987 FAA - DALLAS FSDO-63					
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS			B. KIND OF AGENCY		C. CERTIFICATE NO.
Mark Ardizzone Rt. 1 Rockhill Rd. McKinney Tx 75069			<input checked="" type="checkbox"/> U.S. CERTIFICATED MECHANIC <input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC <input type="checkbox"/> CERTIFICATED REPAIR STATION <input type="checkbox"/> MANUFACTURER		452157668
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE			SIGNATURE OF AUTHORIZED INDIVIDUAL		
3-18-87			Mark Ardizzone		
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	XX	INSPECTION AUTHORIZATION	OTHER (Specify)
	FAA DESIGNEE	REPAIR STATION		CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT	
DATE OF APPROVAL OR REJECTION		CERTIFICATE OR DESIGNATION NO.	SIGNATURE OF AUTHORIZED INDIVIDUAL		
3-18-87		452157668	Mark Ardizzone		

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

N11DY

3-18-87

1. Installed Whelen strobe light modle SA, DF 28
STC SA 800 EA on belly of aircraft using
A440 mounting adapter.
2. Installation accomplished in accordance with
AC 43.13-1A Chapter.11 section.2 paragraph.424a,
AC 43.13-2A Chapter 4 paragraph 51a(1), 55a, and
manufacturers supplied instructions.
3. Weight change computed and entered in
aircraft records.

----- END -----

☐ ADDITIONAL SHEETS ARE ATTACHED

FAA AIR-FAST REGISTRY		APPLICATION FOR AIRWORTHINESS CERTIFICATE		INSTRUCTIONS — Print or type. Do not write in shaded areas. These are for FAA use only. Submit original only to an authorized FAA Representative. If additional space is required, use an attachment. For special flight permits complete Sections II and VI or VII as applicable.	
I. AIRCRAFT DESCRIPTION	1. REGISTRATION MARK	2. AIRCRAFT BUILDER'S NAME (Make)	3. AIRCRAFT MODEL DESIGNATION	FAA CODING	
	N11DY	BEECHCRAFT	95-C55	1752748	
	4. AIRCRAFT SERIAL NO.	5. ENGINE BUILDER'S NAME (Make)	6. ENGINE MODEL DESIGNATION	17432	
	TE 387	CONTINENTAL	IO 520-C		
7. NUMBER OF ENGINES	8. PROPELLER BUILDER'S NAME (Make)	9. PROPELLER MODEL DESIGNATION	10. AIRCRAFT IS CHECKED FOR:		
TWO	McCAULEY	2AF34C55-LMO	IMPORT		
APPLICATION IS HEREBY MADE FOR: (Check applicable items)					
A. 1. <input checked="" type="checkbox"/> STANDARD AIRWORTHINESS CERTIFICATE (Indicate Class below) <input checked="" type="checkbox"/> NORMAL <input type="checkbox"/> LITLITY <input type="checkbox"/> ACROBATIC <input type="checkbox"/> TRANSPORT <input type="checkbox"/> GLIDER <input type="checkbox"/> BALLOON					
B. <input type="checkbox"/> SPECIAL AIRWORTHINESS CERTIFICATE (Check applicable items)					
2. <input type="checkbox"/> LIMITED					
3. <input type="checkbox"/> PREVISIONAL (Indicate class)					
4. <input type="checkbox"/> RESTRICTED (Indicate operation(s) to be conducted)					
5. <input type="checkbox"/> EXPERIMENTAL (Indicate operation(s) to be conducted)					
6. <input type="checkbox"/> SPECIAL FLIGHT PERMIT (Indicate operation to be conducted then complete Section VI or VII as applicable or revise side)					
7. <input type="checkbox"/> TO SHOW COMPLIANCE WITH FAR					
8. <input type="checkbox"/> FERRY FLIGHT FOR REPAIRS, ALTERATIONS, MAINTENANCE OR STORAGE					
9. <input type="checkbox"/> EVACUATE FROM AREA OF IMPENDING DANGER					
10. <input type="checkbox"/> OPERATION IN EXCESS OF MAXIMUM CERTIFICATED TAKE-OFF WEIGHT					
11. <input type="checkbox"/> DELIVERING OR EXPORT					
12. <input type="checkbox"/> CUSTOMER DEMONSTRATION FLIGHTS					
C. <input type="checkbox"/> MULTIPLE AIRWORTHINESS CERTIFICATE (Check ABOVE Restricted, Experimental, and Special Flight Permits as applicable)					
A. REGISTERED OWNER: (As shown on certificate of aircraft registration)					
NAME: DANNY WHITE, INC. ADDRESS: 6510 LBJ FREEWAY, SUITE 200 DALLAS, TX 75240					
B. AIRCRAFT CERTIFICATION BASIS: (Check applicable blocks and complete items as indicated)					
1. AIRCRAFT SPECIFICATION OR TYPE CERTIFICATE DATA SHEET (Give No. and Revision No.)					
X 3A16 REVISION 62					
2. AIRCRAFT LISTING (Give page number(s))					
X B-84-19					
C. AIRCRAFT OPERATION AND MAINTENANCE RECORDS					
1. CHECK IF RECORDS IN COMPLIANCE WITH FAR 91.173					
X 10-15-84					
2. TOTAL AIRFRAME HOURS					
2082					
3. EXPERIMENTAL ONLY (Enter date of last inspection as required by FAR 91.173)					
D. CERTIFICATION: I hereby certify that I am the registered owner (or his agent) of the aircraft described above, that the aircraft is registered with the FAA and is in compliance with Section 501 of the Federal Aviation Act of 1958 and applicable Federal Aviation Regulations, and that the aircraft has been inspected and found eligible for the airworthiness certificate requested.					
DATE OF APPLICATION: 11-01-84					
NAME AND TITLE (Print or type): DANNY WHITE, PRESIDENT					
SIGNATURE: [Signature]					
IV. INSPECTION AGENCY VERIFICATION					
A. THE AIRCRAFT DESCRIBED ABOVE HAS BEEN INSPECTED AND FOUND AIRWORTHY BY (Complete this section only if FAR 21.153(d) applies)					
2. FAR PART 121 OR 127 CERTIFICATE HOLDER (Give Certificate No.)					
3. CERTIFICATED MECHANIC (Give Certificate No.)					
X A&P 452157668					
4. CERTIFICATED REPAIR STATION (Give Certificate No.)					
5. AIRCRAFT MANUFACTURER (Give name of firm)					
DATE: 10-15-84					
TITLE: MECHANIC					
SIGNATURE: [Signature]					
V. FAA REPRESENTATIVE CERTIFICATION					
1. (Check ALL applicable blocks in items 4 and 5)					
A. I find that the aircraft described in Section I or VII meets requirements for:					
4. <input checked="" type="checkbox"/> THE CERTIFICATE REQUESTED					
B. Inspection for a special flight permit under Section VII was conducted by:					
FAA INSPECTOR					
FAA DESIGNEE					
DATE: 12-6-84					
DISTRICT OFFICE: FS-100-63					
DESIGNER'S SIGNATURE AND NO. [Signature] DHE-10-FS-5W					
FAA INSPECTOR'S SIGNATURE					

Typed by # 10

FAA AIRCRAFT REGISTRY
CAMERA NO. 3 DATE: 10-24-89

10-24-89

I. AIRCRAFT DESCRIPTION		APPLICATION FOR AIRWORTHINESS CERTIFICATE		INSTRUCTIONS — Print or type. Do not write in shaded areas; these are for FAA use only. Submit original only to an authorized FAA Representative. If additional space is required, use an attachment. For special flight permits complete Sections II and VI or VII as applicable.																																																																																									
1. REGISTRATION MARK	2. AIRCRAFT BUILDER'S NAME (Make)	3. AIRCRAFT MODEL DESIGNATION	FAA CODING																																																																																										
N11DY	BEECHCRAFT	95-C55	115 2708																																																																																										
4. AIRCRAFT SERIAL NO	5. ENGINE BUILDER'S NAME (Make)	6. ENGINE MODEL DESIGNATION	17032																																																																																										
TE 387	CONTINENTAL	10 520-C																																																																																											
7. NUMBER OF ENGINES	8. PROPELLER BUILDER'S NAME (Make)	9. PROPELLER MODEL DESIGNATION	10. AIRCRAFT IS (Check if applicable)																																																																																										
TWO	MCCAULEY	2AF34C55-LMO	IMPORT																																																																																										
APPLICATION IS HEREBY MADE FOR: (Check applicable items)																																																																																													
<input checked="" type="checkbox"/> STANDARD AIRWORTHINESS CERTIFICATE (Indicate category) <input checked="" type="checkbox"/> AERIAL <input type="checkbox"/> UTILITY <input type="checkbox"/> ACROBATIC <input type="checkbox"/> TRANSPORT <input type="checkbox"/> GLIDER <input type="checkbox"/> BALLOON																																																																																													
<input type="checkbox"/> SPECIAL AIRWORTHINESS CERTIFICATE (Check appropriate items)																																																																																													
<table border="1"> <tr> <td>2</td> <td>LIMITED</td> <td>1</td> <td>CLASS I</td> </tr> <tr> <td>5</td> <td>PROVISIONAL (Indicate class)</td> <td>2</td> <td>CLASS II</td> </tr> <tr> <td>3</td> <td>RESTRICTED (Indicate operation(s) to be conducted)</td> <td>1</td> <td>AGRICULTURE AND PEST CONTROL</td> </tr> <tr> <td>4</td> <td>EXPERIMENTAL (Indicate operation(s) to be conducted)</td> <td>2</td> <td>AERIAL SURVEYING</td> </tr> <tr> <td></td> <td></td> <td>3</td> <td>AERIAL ADVERTISING</td> </tr> <tr> <td></td> <td></td> <td>4</td> <td>FOREST (Wildlife conservation)</td> </tr> <tr> <td></td> <td></td> <td>5</td> <td>PATROLLING</td> </tr> <tr> <td></td> <td></td> <td>6</td> <td>WEATHER CONTROL</td> </tr> <tr> <td></td> <td></td> <td>7</td> <td>CARRIAGE OF CARGO</td> </tr> <tr> <td></td> <td></td> <td>8</td> <td>OTHER (Specify)</td> </tr> <tr> <td></td> <td></td> <td>1</td> <td>RESEARCH AND DEVELOPMENT</td> </tr> <tr> <td></td> <td></td> <td>2</td> <td>AMATEUR BUILT</td> </tr> <tr> <td></td> <td></td> <td>3</td> <td>EXHIBITION</td> </tr> <tr> <td></td> <td></td> <td>4</td> <td>CREW TRAINING</td> </tr> <tr> <td></td> <td></td> <td>5</td> <td>MKT SURVEY</td> </tr> <tr> <td></td> <td></td> <td>6</td> <td>TO SHOW COMPLIANCE WITH FAR</td> </tr> <tr> <td></td> <td></td> <td>7</td> <td>FERRY FLIGHT FOR REPAIRS, ALTERATIONS, MAINTENANCE OR STORAGE</td> </tr> <tr> <td></td> <td></td> <td>8</td> <td>EVACUATE FROM AREA OF IMPENDING DANGER</td> </tr> <tr> <td></td> <td></td> <td>9</td> <td>OPERATION IN EXCESS OF MAXIMUM CERTIFICATED TAKE OFF WEIGHT</td> </tr> <tr> <td></td> <td></td> <td>10</td> <td>DELIVERING OR EXPORT</td> </tr> <tr> <td></td> <td></td> <td>11</td> <td>PRODUCTION FLIGHT TESTING</td> </tr> <tr> <td></td> <td></td> <td>12</td> <td>CUSTOMER DEMONSTRATION FLIGHTS</td> </tr> </table>						2	LIMITED	1	CLASS I	5	PROVISIONAL (Indicate class)	2	CLASS II	3	RESTRICTED (Indicate operation(s) to be conducted)	1	AGRICULTURE AND PEST CONTROL	4	EXPERIMENTAL (Indicate operation(s) to be conducted)	2	AERIAL SURVEYING			3	AERIAL ADVERTISING			4	FOREST (Wildlife conservation)			5	PATROLLING			6	WEATHER CONTROL			7	CARRIAGE OF CARGO			8	OTHER (Specify)			1	RESEARCH AND DEVELOPMENT			2	AMATEUR BUILT			3	EXHIBITION			4	CREW TRAINING			5	MKT SURVEY			6	TO SHOW COMPLIANCE WITH FAR			7	FERRY FLIGHT FOR REPAIRS, ALTERATIONS, MAINTENANCE OR STORAGE			8	EVACUATE FROM AREA OF IMPENDING DANGER			9	OPERATION IN EXCESS OF MAXIMUM CERTIFICATED TAKE OFF WEIGHT			10	DELIVERING OR EXPORT			11	PRODUCTION FLIGHT TESTING			12	CUSTOMER DEMONSTRATION FLIGHTS
2	LIMITED	1	CLASS I																																																																																										
5	PROVISIONAL (Indicate class)	2	CLASS II																																																																																										
3	RESTRICTED (Indicate operation(s) to be conducted)	1	AGRICULTURE AND PEST CONTROL																																																																																										
4	EXPERIMENTAL (Indicate operation(s) to be conducted)	2	AERIAL SURVEYING																																																																																										
		3	AERIAL ADVERTISING																																																																																										
		4	FOREST (Wildlife conservation)																																																																																										
		5	PATROLLING																																																																																										
		6	WEATHER CONTROL																																																																																										
		7	CARRIAGE OF CARGO																																																																																										
		8	OTHER (Specify)																																																																																										
		1	RESEARCH AND DEVELOPMENT																																																																																										
		2	AMATEUR BUILT																																																																																										
		3	EXHIBITION																																																																																										
		4	CREW TRAINING																																																																																										
		5	MKT SURVEY																																																																																										
		6	TO SHOW COMPLIANCE WITH FAR																																																																																										
		7	FERRY FLIGHT FOR REPAIRS, ALTERATIONS, MAINTENANCE OR STORAGE																																																																																										
		8	EVACUATE FROM AREA OF IMPENDING DANGER																																																																																										
		9	OPERATION IN EXCESS OF MAXIMUM CERTIFICATED TAKE OFF WEIGHT																																																																																										
		10	DELIVERING OR EXPORT																																																																																										
		11	PRODUCTION FLIGHT TESTING																																																																																										
		12	CUSTOMER DEMONSTRATION FLIGHTS																																																																																										
<input type="checkbox"/> MULTIPLE AIRWORTHINESS CERTIFICATE (Check ABOVE Restricted Operation and Standard or Limited as applicable)																																																																																													
III. OWNER'S CERTIFICATION																																																																																													
A. REGISTERED OWNER (As shown on certificate of aircraft registration)																																																																																													
NAME			ADDRESS																																																																																										
DANNY WHITE, INC.			6510 LBJ FREEWAY, SUITE 200 DALLAS, TX 75240																																																																																										
B. AIRCRAFT CERTIFICATION BASIS (Check applicable blocks and complete items as indicated)																																																																																													
AIRCRAFT SPECIFICATION OR TYPE CERTIFICATE DATA SHEET (Give No. and Revision No.)			AIRWORTHINESS DIRECTIVES (Check if all applicable ADs complied with and give latest AD No.)																																																																																										
X 3A16 REVISION 62			X B 84-19																																																																																										
AIRCRAFT LISTING (Give page number(s))			SUPPLEMENTAL TYPE CERTIFICATE (List number of each STC incorporated)																																																																																										
C. AIRCRAFT OPERATION AND MAINTENANCE RECORDS																																																																																													
CHECK IF RECORDS IN COMPLIANCE WITH FAR 91.173			TOTAL AIRFRAME HOURS		EXPERIMENTAL ONLY (Enter hours flown since last certificate issued or renewed)																																																																																								
X 10-15-84			2082																																																																																										
D. CERTIFICATION — I hereby certify that I am the registered owner (or his agent) of the aircraft described above, that the aircraft is registered with the Federal Aviation Administration in accordance with Section 501 of the Federal Aviation Act of 1958, and applicable Federal Aviation Regulations, and that the aircraft has been inspected and is airworthy and eligible for the airworthiness certificate requested.																																																																																													
DATE OF APPLICATION			NAME AND TITLE (Print or type)		SIGNATURE																																																																																								
10-01-84			DANNY WHITE, PRESIDENT		<i>Danny White</i>																																																																																								
IV. INSPECTION AGENCY VERIFICATION																																																																																													
A. THE AIRCRAFT DESCRIBED ABOVE HAS BEEN INSPECTED AND FOUND AIRWORTHY BY (Complete this section only if FAR 21.183(d) applies)																																																																																													
FAR PART 121 OR 135 CERTIFICATE HOLDER (Give Certificate No.)		CERTIFICATED MECHANIC (Give Certificate No.)		CERTIFICATED REPAIR STATION (Give Certificate No.)																																																																																									
		A&P 452157668																																																																																											
B. AIRCRAFT MANUFACTURER (Give name of firm)																																																																																													
typed by # 20																																																																																													
DATE		TITLE		SIGNATURE																																																																																									
10-15-84		MECHANIC		<i>Mark Ordizzone</i>																																																																																									
V. FAA REPRESENTATIVE CERTIFICATION																																																																																													
(Check ALL applicable blocks in items A and B)																																																																																													
A. I find that the aircraft described in Section I or VI meets requirements for:																																																																																													
<input checked="" type="checkbox"/> THE CERTIFICATE REQUESTED																																																																																													
<input type="checkbox"/> AMENDMENT OR MODIFICATION OF CURRENT AIRWORTHINESS CERTIFICATE																																																																																													
B. Inspection for a special flight permit under Section VII was conducted by:			FAA INSPECTOR																																																																																										
			FAA DESIGNEE																																																																																										
			CERTIFICATE HOLDER UNDER																																																																																										
			FAR 65 FAR 121 127 or 135 FAR 145																																																																																										
DATE		DISTRICT OFFICE		DESIGNEE'S SIGNATURE AND NO.																																																																																									
12-6-1984		FSDO-63		<i>Donald R. Kent</i> D&P-10-FS-SW																																																																																									
FAA INSPECTOR'S SIGNATURE																																																																																													

VI. PRODUCTION FLIGHT TESTING	A. MANUFACTURER	
	NAME	ADDRESS
	B. PRODUCTION BASIS (Check applicable item)	
	<input type="checkbox"/> PRODUCTION CERTIFICATE (Give production certificate number) <input type="checkbox"/> TYPE CERTIFICATE ONLY <input type="checkbox"/> APPROVED PRODUCTION INSPECTION SYSTEM	
VII. SPECIAL FLIGHT PERMIT - APPLIES OTHER THAN PRODUCTION FLIGHT TEST	C. GIVE QUANTITY OF CERTIFICATES REQUIRED FOR OPERATING NEEDS	
	DATE OF APPLICATION	SIGNATURE
	NAME AND TITLE (Print or type)	
	A. DESCRIPTION OF AIRCRAFT	
	REGISTERED OWNER	ADDRESS
	BUILDER (Make)	MODEL
	SERIAL NUMBER	REGISTRATION MARK
	B. DESCRIPTION OF FLIGHT	
	FROM	TO
	VIA	DEPARTURE DATE
DURATION		
C. CREW REQUIRED TO OPERATE THE AIRCRAFT AND ITS EQUIPMENT		
<input type="checkbox"/> PILOT <input type="checkbox"/> CO-PILOT <input type="checkbox"/> NAVIGATOR <input type="checkbox"/> OTHER (Specify)		
D. THE AIRCRAFT DOES NOT MEET THE APPLICABLE AIRWORTHINESS REQUIREMENTS AS FOLLOWS		
E. THE FOLLOWING RESTRICTIONS ARE CONSIDERED NECESSARY FOR SAFE OPERATION (Use attachment if necessary)		
F. CERTIFICATION — I hereby certify that I am the registered owner (or his agent) of the aircraft described above; that the aircraft is registered with the Federal Aviation Administration in accordance with Section 501 of the Federal Aviation Act of 1958, and applicable Federal Aviation Regulations; and that the aircraft has been inspected and is airworthy for the flight described.		
DATE		
NAME AND TITLE (Print or type)		
SIGNATURE		
VIII. AIRWORTHINESS DOCUMENTATION (FAA use only)	<input checked="" type="checkbox"/> A. Operating Limitations and Markings in Compliance with FAR 91.31 as Applicable	G. Statement of Conformity, FAA Form 8130-9 (Attach when required)
	<input type="checkbox"/> B. Current Operating Limitations Attached	H. Foreign Airworthiness Certification for Import Aircraft (Attach when required)
	<input type="checkbox"/> C. Data, Drawings, Photographs, etc. (Attach when required)	I. Previous Airworthiness Certificate Issued in Accordance with FAR _____ CAR _____ (Original Attached)
	<input checked="" type="checkbox"/> D. Current Weight and Balance Information Available in Aircraft	J. Current Airworthiness Certificate Issued in Accordance with FAR _____ CAR _____ (Copy attached)
	<input checked="" type="checkbox"/> E. Major Repair and Alteration, FAA Form 337 (Attach when required)	
	<input checked="" type="checkbox"/> F. This Inspection Recorded in Aircraft Records	

UNITED STATES OF AMERICA
DEPARTMENT OF TRANSPORTATION—FEDERAL AVIATION ADMINISTRATION
STANDARD AIRWORTHINESS CERTIFICATE

1. NATIONALITY AND REGISTRATION MARKS N11DY	2. MANUFACTURER AND MODEL BEECH 8R-55	3. AIRCRAFT SERIAL NUMBER 1-387	4. CATEGORY NORMAL
<p>5. AUTHORITY AND BASIS FOR ISSUANCE This airworthiness certificate is issued pursuant to the Federal Aviation Act of 1958 and certifies that, as of the date of issuance, the aircraft to which issued has been inspected and found to conform to the type certificate therefor, to be in condition for safe operation, and has been shown to meet the requirements of the applicable comprehensive and detailed airworthiness standards provided by Annex 8 to the Convention on International Civil Aviation, except as noted herein. Exception: NONE</p>			
<p>6. TERMS AND CONDITIONS Unless sooner surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator, this airworthiness certificate is effective as long as the maintenance, preventive maintenance, and alterations are performed in accordance with Parts 43, and 91 of the Federal Aviation Regulations, as appropriate, and the aircraft is registered in the United States.</p>			
DATE OF ISSUANCE DECEMBER 6, 1984	FAA REPRESENTATIVE David E. STRAIN	DESIGNATION NUMBER DAR-10-FS-SW	

FAA AIRCRAFT REGISTRY

CAMERA NO. 3 DATE: 10-24-89

[illegible]

FAA AIRCRAFT REGISTRY

CAMERA NO. 3 DATE: 10-24-89



U.S. Department
of Transportation
Federal Aviation
Administration

Memorandum

Wichita Aircraft Certification Office
1801 Airport Road, Room 100
Mid-Continent Airport.
Wichita, KS 67209

Subject: Airplane Flight Manual/Pilot's Operating
Handbook for a Beech Model 95-C55

Date: November 8, 1984

From: *Lawrence A. Herron*
Robert A. Gambrill, Jr., Manager
Wichita Aircraft Certification Office, ACE-115W

Reply to:
Attn. of: Herron:4402

To: Gene Berrier, Manager
Manufacturing Inspection Branch, ASW-180

This is to verify our conversation concerning Beech POH P/N 96-590010-29
being added to Aircraft Specification No. 3A16. This POH is applicable to
Beechcraft Barons with S/N TE-1 through TE-942 except TE-938. The afore-
mentioned POH can be used as an alternate to P/N 55-590006-3.

NOV 19 1984

FAA AIRCRAFT REGISTRY

FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY

FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY

FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY

FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY

FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY

FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY

FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY

FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY

FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY

FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY
FAA AIRCRAFT REGISTRY

U.S. Department of Transportation Federal Aviation Administration		APPLICATION FOR AIRWORTHINESS CERTIFICATE		INSTRUCTIONS — Print or type. Do not write in shaded areas, these are for FAA use only. Submit original only to an authorized FAA Representative. If additional space is required, use an attachment. For special flight permits complete Sections II and VI or VII as applicable.	
I. AIRCRAFT DESCRIPTION	1. REGISTRATION MARK	2. AIRCRAFT BUILDER'S NAME (Make)	3. AIRCRAFT MODEL DESIGNATION	FAA CODING	
	4. AIRCRAFT SERIAL NO.	5. ENGINE BUILDER'S NAME (Make)	6. ENGINE MODEL DESIGNATION		
	7. NUMBER OF ENGINES	8. PROPELLER BUILDER'S NAME (Make)	9. PROPELLER MODEL DESIGNATION	10. AIRCRAFT IS (Check if applicable) IMPORT	
	N11DY	BEECHCRAFT	95-C55		
	TE 387	CONTINENTAL	IO 520-C		
	TWO	McCAULEY	2AF34C55-LMO		
II. CERTIFICATION REQUESTED	APPLICATION IS HEREBY MADE FOR: (Check applicable items)				
	A. 1. STANDARD AIRWORTHINESS CERTIFICATE (Indicate category) <input checked="" type="checkbox"/> NORMAL <input type="checkbox"/> UTILITY <input type="checkbox"/> ACROBATIC <input type="checkbox"/> TRANSPORT <input type="checkbox"/> GLIDER <input type="checkbox"/> BALLOON				
	B. SPECIAL AIRWORTHINESS CERTIFICATE (Check appropriate items)				
	2. LIMITED				
	3. PROVISIONAL (Indicate class)				
	4. RESTRICTED (Indicate operation(s) to be conducted)				
	5. EXPERIMENTAL (Indicate operation(s) to be conducted)				
	6. SPECIAL FLIGHT PERMIT (Indicate operation to be conducted, then complete Section VI or VII as applicable on reverse side)				
	7. TO SHOW COMPLIANCE WITH FAR				
	8. FERRY FLIGHT FOR REPAIRS, ALTERATIONS, MAINTENANCE OR STORAGE				
III. OWNER'S CERTIFICATION	C. MULTIPLE AIRWORTHINESS CERTIFICATE (Check ABOVE "Restricted Operation" and "Standard or Limited" as applicable)				
	A. REGISTERED OWNER (As shown on certificate of aircraft registration)				
	NAME		ADDRESS		
	DANNY WHITE, INC.		6510 LBJ FREEWAY, SUITE 200 DALLAS, TX 75240		
	B. AIRCRAFT CERTIFICATION BASIS (Check applicable blocks and complete items as indicated)				
	AIRCRAFT SPECIFICATION OR TYPE CERTIFICATE DATA SHEET (Give No. and Revision No.)		AIRWORTHINESS DIRECTIVES (Check if all applicable AD's complied with and give latest AD No.)		
	X 3A16 REVISION 62		X B 84-19		
	AIRCRAFT LISTING (Give page number(s))		SUPPLEMENTAL TYPE CERTIFICATE (List number of each STC incorporated)		
	C. AIRCRAFT OPERATION AND MAINTENANCE RECORDS				
	CHECK IF RECORDS IN COMPLIANCE WITH FAR 91.173		TOTAL AIRFRAME HOURS		EXPERIMENTAL ONLY (Enter hours flown since last certificate issued or renewed)
X 10-15-84		2082			
D. CERTIFICATION — I hereby certify that I am the registered owner (or his agent) of the aircraft described above that the aircraft is registered with the Federal Aviation Administration in accordance with Section 501 of the Federal Aviation Act of 1958, and applicable Federal Aviation Regulations, and that the aircraft has been inspected and is airworthy and eligible for the airworthiness certificate requested.					
DATE OF APPLICATION		NAME AND TITLE (Print or type)		SIGNATURE	
11-01-84		DANNY WHITE, PRESIDENT		<i>[Signature]</i>	
IV. INSPECTION AGENCY VERIFICATION	A. THE AIRCRAFT DESCRIBED ABOVE HAS BEEN INSPECTED AND FOUND AIRWORTHY BY (Complete this section only if FAR 21.183(d) applies)				
	FAR PART 121 OR 127 CERTIFICATE HOLDER (Give Certificate No.)		CERTIFICATED MECHANIC (Give Certificate No.)		CERTIFICATED REPAIR STATION (Give Certificate No.)
			X A&P 452157668		
	AIRCRAFT MANUFACTURER (Give name of firm)				
V. FAA REPRESENTATIVE CERTIFICATION	DATE		TITLE		SIGNATURE
	10-15-84		MECHANIC		<i>[Signature]</i>
	(Check ALL applicable blocks in items A and B)				
	A. I find that the aircraft described in Section I or VII meets requirements for:				
B. Inspection for a special flight permit under Section VII was conducted by:					
DATE		DISTRICT OFFICE		DESIGNEE'S SIGNATURE AND NO.	FAA INSPECTOR'S SIGNATURE
12-6-MBA		FSDO-63		<i>[Signature]</i> DAB-10-FS-SW	

VI. PRODUCTION FLIGHT TESTING	A. MANUFACTURER	
	NAME	ADDRESS
	B. PRODUCTION BASIS (Check applicable item)	
	<input type="checkbox"/> PRODUCTION CERTIFICATE (Give production certificate number) <input type="checkbox"/> TYPE CERTIFICATE ONLY <input type="checkbox"/> APPROVED PRODUCTION INSPECTION SYSTEM	
VII. SPECIAL FLIGHT PERMIT PURPOSES OTHER THAN PRODUCTION FLIGHT TEST	C. GIVE QUANTITY OF CERTIFICATES REQUIRED FOR OPERATING NEEDS	
	DATE OF APPLICATION	NAME AND TITLE (Print or type) SIGNATURE
	A. DESCRIPTION OF AIRCRAFT	
	REGISTERED OWNER	ADDRESS
	BUILDER (Make)	MODEL
	SERIAL NUMBER	REGISTRATION MARK
	B. DESCRIPTION OF FLIGHT	
	FROM TO	
	VIA DEPARTURE DATE DURATION	
	C. CREW REQUIRED TO OPERATE THE AIRCRAFT AND ITS EQUIPMENT	
<input type="checkbox"/> PILOT <input type="checkbox"/> CO-PILOT <input type="checkbox"/> NAVIGATOR <input type="checkbox"/> OTHER (Specify)		
D. THE AIRCRAFT DOES NOT MEET THE APPLICABLE AIRWORTHINESS REQUIREMENTS AS FOLLOWS:		
E. THE FOLLOWING RESTRICTIONS ARE CONSIDERED NECESSARY FOR SAFE OPERATION (Use attachment if necessary)		
F. CERTIFICATION — I hereby certify that I am the registered owner (or his agent) of the aircraft described above; that the aircraft is registered with the Federal Aviation Administration in accordance with Section 501 of the Federal Aviation Act of 1958, and applicable Federal Aviation Regulations; and that the aircraft has been inspected and is airworthy for the flight described.		
DATE	NAME AND TITLE (Print or type) SIGNATURE	
VIII. AIRWORTHINESS DOCUMENTATION (FAA use only)	<input checked="" type="checkbox"/> A. Operating Limitations and Markings in Compliance with FAR 91.31 as Applicable	
	<input checked="" type="checkbox"/> B. Current Operating Limitations Attached	
	<input checked="" type="checkbox"/> C. Data, Drawings, Photographs, etc. (Attach when required)	
	<input checked="" type="checkbox"/> D. Current Weight and Balance Information Available in Aircraft	
	<input checked="" type="checkbox"/> E. Major Repair and Alteration, FAA Form 337 (Attach when required)	
	<input checked="" type="checkbox"/> F. This Inspection Recorded in Aircraft Records	
<input checked="" type="checkbox"/> G. Statement of Conformity, FAA Form 8130-9 (Attach when required)		
<input checked="" type="checkbox"/> H. Foreign Airworthiness Certification for Import Aircraft (Attach when required)		
<input checked="" type="checkbox"/> I. Previous Airworthiness Certificate Issued in Accordance with FAR _____ CAR _____ (Original Attached)		
<input checked="" type="checkbox"/> J. Current Airworthiness Certificate Issued in Accordance with FAR 21.183(d) _____ (Copy attached)		

DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				Form approved Budget Bureau No. 04-R060.1 FOR FAA USE ONLY	
MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)				DALF900-63	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.					
1. AIRCRAFT	MAKE	BEECHCRAFT		MODEL	B95C55
	SERIAL NO.	TE387		NATIONALITY AND REGISTRATION MARK	NIDDY
2. OWNER	NAME (As shown on registration certificate)		ADDRESS (As shown on registration certificate)		
	DANNY WHITE, INC.		6510 LBJ #200 DALLAS, TEXAS 75240		
3. FOR FAA USE ONLY					
4. UNIT IDENTIFICATION					
UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE	
AIRFRAME	***** (As described in item 1 above) *****			REPAIR	ALTERATION
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS		B. KIND OF AGENCY		C. CERTIFICATE NO.	
TEXOMA AVIONICS, INC. P.O. BOX 888 MCKINNEY AIRPORT MCKINNEY, TEXAS 75069		<input checked="" type="checkbox"/> U.S. CERTIFICATED MECHANIC <input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC <input checked="" type="checkbox"/> CERTIFICATED REPAIR STATION <input type="checkbox"/> MANUFACTURER		CRS-202-104	
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE		SIGNATURE OF AUTHORIZED INDIVIDUAL			
3/26/85		James P. Bush			
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	INSPECTION AUTHORIZATION		
	FAA DESIGNEE	REPAIR STATION	CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT		
DATE OF APPROVAL OR REJECTION		CERTIFICATE OR DESIGNATION NO.	SIGNATURE OF AUTHORIZED INDIVIDUAL		
3/26/85		202-104	James P. Bush		

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

B. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Removed, KX-160, KI-201, KES-505 Power Supply, KXP-750, KES-570 CONTROL.

INSTALLED KX-155 NAV/COMM, KI-208 INDICATOR, KF-76A ATC, D120 BLIND ENCODER.

INSTALLATION COMPLETED USING MFG. DATA AND A.C. 43.13-1, 2A CHAPTERS 2, 3 AS GUIDELINES.

WT & BALANCE INFORMATION ENTERED IN WT. & BAL. SECTION.

END

☐ ADDITIONAL SHEETS ARE ATTACHED

DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)				Form Approved Budget Bureau No. 04-R060.1 FOR FAA USE ONLY DAITSD0063	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.					
1. AIRCRAFT	MAKE Beechcraft		MODEL 95-055		
	SERIAL NO. TE-387		NATIONALITY AND REGISTRATION MARK N11DY		
2. OWNER	NAME (As shown on registration certificate) Danny White, Inc.		ADDRESS (As shown on registration certificate) 6510 LBJ Fwy. #200 Dallas, Texas 75240		
3. FOR FAA USE ONLY					
The alteration identified herein complied with applicable airworthiness requirements and is approved only for the above described aircraft subject to conformity inspection by a person authorized in FAR 43. Date <u>12-4-84</u> Signature <u>Mark Archizzone</u> Date _____ Signature _____ FAA Inspector, SW-FSDO-63 (OWL)					
4. UNIT IDENTIFICATION					5. TYPE
UNIT	MAKE	MODEL	SERIAL NO.	REPAIR	ALTERATION
AIRFRAME	As described in item 1 above				XX
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS			B. KIND OF AGENCY		C. CERTIFICATE NO.
Aeromark, Inc. Rt. 1, Rockhill Rd. McKinney, Texas 75069			<input checked="" type="checkbox"/> U.S. CERTIFICATED MECHANIC		452157668
			<input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC		
			<input type="checkbox"/> CERTIFICATED REPAIR STATION		
			<input type="checkbox"/> MANUFACTURER		
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE 12-4-84			SIGNATURE OF AUTHORIZED INDIVIDUAL <u>Mark Archizzone</u>		
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	<input checked="" type="checkbox"/> INSPECTION AUTHORIZATION		OTHER (Specify)
	FAA DESIGNEE	REPAIR STATION	CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT		
DATE OF APPROVAL OR REJECTION 12-4-84		CERTIFICATE OR DESIGNATION NO. 452157668		SIGNATURE OF AUTHORIZED INDIVIDUAL <u>Mark Archizzone</u>	

FAA AIRCRAFT REGISTRY
CAMERA NO. 4

DATE: 3-25-85

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

B. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

N11DY

12-4-84

1. The following installation was made by an unknown person or persons at an unknown time.
2. Two 120 Volt 750 Watt AC heater units, one located under each engine cowling outboard side just forward of firewall.
3. Mounting of these heaters is accomplished by brackets and clamps securing units to engine mount.
4. The wiring harness consist of two 14 gauge wires from each heater routed with existing aircraft wiring harness to two 120 Volt AC male plug receptacles located beneath a small hinged door on the left outboard nacelle.
5. Small hinged door placarded, "120 Volt A.C. Ground power operation only".
6. Weight and balance previously figured by reweighing of aircraft.

----- END -----

☐ ADDITIONAL SHEETS ARE ATTACHED

FAA AIRCRAFT REGISTRY
CAMERA NO. 4

DATE: 3-25-85

DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				Form Approved Budget Bureau No. 04-R060.1	
MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)				FOR FAA USE ONLY	
				OFFICE IDENTIFICATION DAL FSDO-63	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.					
1. AIRCRAFT	MAKE	Beechcraft		MODEL	95-C55
	SERIAL NO.	TE-387		NATIONALITY AND REGISTRATION MARK	N11DY
2. OWNER	NAME (As shown on registration certificate)	Danny White, Inc.		ADDRESS (As shown on registration certificate)	6510 LEJ Fwy. #200 Dallas, Texas 75240
3. FOR FAA USE ONLY					
<p>The alteration identified herein complied with applicable airworthiness requirements and is approved for the above described aircraft subject to conformity therewith by a person authorized in FAR 43.9.</p> <p>12-4-84 <i>Mark Ardizzone</i> FSDO-63</p>					
4. UNIT IDENTIFICATION					5. TYPE
UNIT	MAKE	MODEL	SERIAL NO.	REPAIR	ALTERATION
AIRFRAME	***** (As described in item 1 above) *****				XX
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS			B. KIND OF AGENCY		C. CERTIFICATE NO.
Aeromark, Inc. Rt. 1, Rockhill Rd. McKinney, Texas 75069			<input checked="" type="checkbox"/> U.S. CERTIFICATED MECHANIC <input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC <input type="checkbox"/> CERTIFICATED REPAIR STATION <input type="checkbox"/> MANUFACTURER		452157668
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE 12-4-84			SIGNATURE OF AUTHORIZED INDIVIDUAL <i>Mark Ardizzone</i>		
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	XX	INSPECTION AUTHORIZATION	OTHER (Specify)
	FAA DESIGNEE	REPAIR STATION		CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT	
DATE OF APPROVAL OR REJECTION 12-4-84		CERTIFICATE OR DESIGNATION NO. 452157668		SIGNATURE OF AUTHORIZED INDIVIDUAL <i>Mark Ardizzone</i>	

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

N11DY-

12-4-84

1. The following installation was made by an unknown person or persons at an unknown time.
2. Two 120 Volt 750 Watt AC heater units, one located under each engine cowling outboard side just forward of firewall.
3. Mounting of these heaters is accomplished by brackets and clamps securing units to engine mount.
4. The wiring harness consist of two 14 gauge wires from each heater routed with existing aircraft wiring harness to two 120 Volt AC male plug receptacles located beneath a small hinged door on the left outboard nacelle.
5. Small hinged door placarded, "120 Volt A.C. Ground power operation only".
6. Weight and balance previously figured by reweighing of aircraft.

END

☐ ADDITIONAL SHEETS ARE ATTACHED