

Airworthiness Directive Compliance RecordFile ID: **18281088**USARL Research Date: **01/03/2002**

Manufacturer Cessna Aircraft Company		Model 182T			Part # : Serial # :	
AD Number Effective date	Description	Complied Date Time	Status Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility 3. Cert. Num. 2. Cert. Type 4. Author. by
71-21-01 01/01/1971 ©ATP	Superseded by 72-07-09 ©ATP	N/A	N/A S/S	Once ©ATP	 Signature:	1. CESSNA 2. REPAIR STATION 3. I97R796N 4. MAR 2 6 2002 SE 197R796N 16
71-22-02 R(1) 11/09/1971 ©ATP	To Decrease The Possibility Of Failure Of The Nose Gear Structure ©ATP	N/A	N/A PER S/N	Recur ©ATP	 Signature:	1. CESSNA 2. REPAIR STATION 3. I97R796N 4. MAR 2 6 2002 SE 197R796N 16
72-07-09 10/17/1974 ©ATP	To Detect Cracks And Bolt Looseness Which Could Lead To Inflight Separation Of ©ATP	N/A	N/A PER S/N	Recur ©ATP	 Signature:	1. CESSNA 2. REPAIR STATION 3. I97R796N 4. MAR 2 6 2002 SE 197R796N 16
73-17-01 08/16/1973 ©ATP	To Advise The Pilot Concerning Proper Fuel Transfer Pump Operation ©ATP	N/A	N/A PER MODEL	Once ©ATP	 Signature:	1. CESSNA 2. REPAIR STATION 3. I97R796N 4. MAR 2 6 2002 SE 197R796N 16
75-16-01 07/30/1975 ©ATP	To Preclude Inadvertent Fuel Exhaustion Due To Incorrect Fuel Placarded Capaci ©ATP	N/A	N/A PER S/N	Once ©ATP	 Signature:	1. CESSNA 2. REPAIR STATION 3. I97R796N 4. MAR 2 6 2002 SE 197R796N 16
77-04-05 03/01/1977 ©ATP	To Prevent Ingestion Of The Induction Air Box Seal Into The Carburetor ©ATP	N/A	N/A PER S/N	Once ©ATP	 Signature:	1. CESSNA 2. REPAIR STATION 3. I97R796N 4. MAR 2 6 2002 SE 197R796N 16

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Cessna Aircraft Company**Model**
182T**Part # :****Serial # :**

AD Number Effective date	Description	Complied Date Time	Status Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility 2. Cert. Type 3. Cert. Num. 4. Author. by
77-23-11 11/28/1977 ©ATP	To Preclude The Possibility Of An In-Flight Fire Due To A Loose Elt Antenna Co ©ATP	N/A	N/A PER S/N	Once ©ATP		1. CESSNA 2. REPAIR STATION 3. I97R796N 4. MAR 2 6 2002 SE 197R796N 16
78-01-14 01/23/1978 ©ATP	To Preclude The Possibility Of An In-Flight Fire Due To A Loose Elt Antenna Co ©ATP	N/A	N/A PER S/N	Once ©ATP		1. CESSNA 2. REPAIR STATION 3. I97R796N 4. MAR 2 6 2002 SE 197R796N 16
78-26-09 01/01/1978 ©ATP	Superseded by 79-10-14 ©ATP	N/A	N/A S/S	Once ©ATP		1. CESSNA 2. REPAIR STATION 3. I97R796N 4. MAR 2 6 2002 SE 197R796N 16
79-10-14 R1 05/30/1988 ©ATP	To Provide An Alternate Source Of Fuel Tank Venting In Case Of Fuel Tank Vent ©ATP	N/	N/A PER MODEL	Once ©ATP		1. CESSNA 2. REPAIR STATION 3. I97R796N 4. MAR 2 6 2002 SE 197R796N 16
79-25-07 12/13/1979 ©ATP	To Preclude The Possibility Of Electrical Or Electronic Component Damage Or An ©ATP	N/A	N/A PER S/N	Once ©ATP		1. CESSNA 2. REPAIR STATION 3. I97R796N 4. MAR 2 6 2002 SE 197R796N 16
						1. CESSNA 2. REPAIR STATION 3. I97R796N 4. Signature:

Airworthiness Directive Compliance RecordFile ID: **18281088**USARL Research Date: **01/03/2002**

Manufacturer Textron Lycoming		Model IO-540-AB1A5			Part # : Serial # :	
AD Number Effective date	Description	Complied Date Time	Status Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility 3. Cert. Num. 2. Cert. Type 4. Author. by
00-00-04 03/06/2001 ©ATP	Important Notice For Textron Lycoming Fuel Injected Engines Listed In Ad 93-02 ©ATP	PRE-A/W	NOTED INFORMATION	Once ©ATP		1. CESSNA 2. REPAIR STATION 3. I97R796N 4. MAR 26 2002
63-14-03 07/06/1963 ©ATP	Oil Pump Drive Shaft ©ATP	N/A	N/A PER S/N	Recur ©ATP		1. CESSNA 2. REPAIR STATION 3. I97R796N 4. MAR 26 2002
66-20-04 08/27/1966 ©ATP	To Prevent Further Failures Of Oil Filter Adapter Gasket, P/N 74904 ©ATP	N/A	N/A PER S/N	Once ©ATP		1. CESSNA 2. REPAIR STATION 3. I97R796N 4. MAR 26 2002
75-08-09 R(3) 08/18/1977 ©ATP	To Prevent Oil Pump Failures, Inspect, Replace And Assemble The Oil Pump Drive ©ATP	N/A	N/A PER S/N	Once ©ATP		1. CESSNA 2. REPAIR STATION 3. I97R796N 4. MAR 26 2002
75-09-15 04/30/1975 ©ATP	To Prevent Possible Fuel Starvation To The Engine ©ATP	N/A	N/A PER P/N	Once ©ATP		1. CESSNA 2. REPAIR STATION 3. I97R796N 4. MAR 26 2002
78-23-10 11/07/1978 ©ATP	To Prevent An In-Flight Power Loss Due To An Over Rich Condition, Contd. ©ATP	N/A	N/A PER S/N	Once ©ATP		1. CESSNA 2. REPAIR STATION 3. I97R796N 4. MAR 26 2002

Airworthiness Directive Compliance RecordFile ID: **18281088**USARL Research Date: **01/03/2002**

<u>Manufacturer</u>		<u>Model</u>		<u>Part # :</u>		<u>Serial # :</u>	
Textron Lycoming		IO-540-AB1A5					
AD Number Effective date	Description	Complied Date Time	Status Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility 2. Cert. Type 3. Cert. Num. 4. Author. by	
79-04-05 09/26/1979 ©ATP	To Prevent An In-Flight Power Loss Due To The Separation Of The P/N 2529192 Re ©ATP	N/A	N/A PER P/N	Once ©ATP		1. CESSNA 2. REPAIR STATION 3. I97R796N 4. MAR 26 2002	SE I97R796N 16
81-18-04 R2 06/07/1982 ©ATP	Superseded by 96-09-10 ©ATP	N/A	N/A S/S	Once ©ATP		1. CESSNA 2. REPAIR STATION 3. I97R796N 4. MAR 26 2002	SE I97R796N 16
91-08-07 C 05/06/1991 ©ATP	To Prevent Engine Power Loss And Possible Loss Of The Aircraft ©ATP	N/A	N/A PER ENGINE TYPE	Once ©ATP		1. CESSNA 2. REPAIR STATION 3. I97R796N 4. MAR 26 2002	SE I97R796N 16
91-14-22 08/19/1991 ©ATP	To Prevent Loosening Or Failure Of The Crankshaft Gear Retaining Bolt, Which M ©ATP	N/A	N/A NOT DUE	Recur ©ATP		1. CESSNA 2. REPAIR STATION 3. I97R796N 4. MAR 26 2002	SE I97R796N 16
92-12-05 07/10/1992 ©ATP	To Prevent Piston Pin Failure, Or Piston Release, And Engine Failure ©ATP	N/A	N/A PER S/N	Once ©ATP		1. CESSNA 2. REPAIR STATION 3. I97R796N 4. MAR 26 2002	SE I97R796N 16
93-02-05 06/14/1993 ©ATP	To Prevent Failure Of The Fuel Injector Fuel Lines Allowing Fuel To Spray Into ©ATP	Pre-A/W	Visual insp during flight test.	Recur ©ATP	100 hr	1. CESSNA 2. REPAIR STATION 3. I97R796N 4. MAR 26 2002	SE I97R796N 16

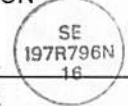
Airworthiness Directive Compliance Record

File ID: 18281088

USARL Research Date: 01/03/2002

Manufacturer Textron Lycoming		Model IO-540-AB1A5			Part # : Serial # :	
AD Number Effective date	Description	Complied Date Time	Status Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility 3. Cert. Num. 2. Cert. Type 4. Author. by
96-09-10 C 07/15/1996 ©ATP	To Prevent Oil Pump Failure Due To Impeller Failure, Which Could Result In An ©ATP	N/A	N/A per engine model	Once ©ATP	<div>Signature:</div>	1. CESSNA 2. REPAIR STATION 3. I97R796N 4. MAR 26 2002 <div>SE 197R796N 16</div>
96-23-03 12/17/1996 ©ATP	To Prevent An Inflight Engine Failure Due To Fuel Starvation, Which Could Resu ©ATP	N/A	N/A per P/N	Once ©ATP	<div>Signature:</div>	1. CESSNA 2. REPAIR STATION 3. I97R796N 4. MAR 26 2002 <div>SE 197R796N 16</div>
97-15-11 08/12/1997 ©ATP	To Prevent Piston Pin Failure, Which Could Result In Engine Failure ©ATP	N/A	N/A engine new.	Once ©ATP	<div>Signature:</div>	1. CESSNA 2. REPAIR STATION 3. I97R796N 4. MAR 26 2002 <div>SE 197R796N 16</div>
98-17-11 C 10/19/1998 ©ATP	To Prevent Crankshaft Failure Due To Cracking, Which Could Result In An Influg ©ATP	N/A	N/A new engine.	Once ©ATP	<div>Signature:</div>	1. CESSNA 2. REPAIR STATION 3. I97R796N 4. MAR 26 2002 <div>SE 197R796N 16</div>
98-18-12 09/28/1998 ©ATP	To Prevent Rotary Fuel Pump Leaks,Which Could Result In An Engine Failure,Engi ©ATP	N/A	N/A per fuel pump type	Recur ©ATP	<div>Signature:</div>	1. CESSNA 2. REPAIR STATION 3. I97R796N 4. MAR 26 2002 <div>SE 197R796N 16</div>
					<div>Signature:</div>	1. CESSNA 2. REPAIR STATION 3. I97R796N 4.

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AD Number Effective date	Description	Complied Date Time	Status Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility 2. Cert. Type 3. Cert. Num. 4. Author. by
<u>Manufacturer</u> Parker Hannifin	<u>Appliance Category</u> Vacuum Pumps		<u>Model</u> 216CW		<u>Part # :</u> <u>Serial # :</u>	
98-23-01 11/20/1998 ©ATP	To Prevent Failure Of The Primary Dry Air Pump Caused By Defective Flexible Co ©ATP	N/A	N/A per s/n	Once ©ATP		1. CESSNA 2. REPAIR STATION 3. I97R796N 4. Signature: MAR 28 2002 
						1. CESSNA 2. REPAIR STATION 3. I97R796N 4. Signature:
						1. CESSNA 2. REPAIR STATION 3. I97R796N 4. Signature:
						1. CESSNA 2. REPAIR STATION 3. I97R796N 4. Signature:
						1. CESSNA 2. REPAIR STATION 3. I97R796N 4. Signature:

Airworthiness Directive Compliance RecordFile ID: **51732ENG**

USARL Research Date: 11/08/2002

Manufacturer		Model		Part # :		
Textron Lycoming		IO-540-AB1A5		Serial # : L-28292-48A		
AD Number Effective date	Description	Complied Date Time	Status Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility 3. Cert. Num. 2. Cert. Type 4. Author. by
96-09-10 C 07/15/1996 ©ATP	TO PREVENT OIL PUMP FAILURE DUE TO IMPELLER FAILURE, WHICH COULD RESULT IN AN ©ATP		N/A PER ENGINE MODEL	Once ©ATP		1. 2. 3. 4. Signature:
96-23-03 12/17/1996 ©ATP	TO PREVENT AN INFLIGHT ENGINE FAILURE DUE TO FUEL STARVATION, WHICH COULD RESU ©ATP		N/A PER P/W	Once ©ATP		1. 2. 3. 4. Signature:
97-15-11 08/12/1997 ©ATP	TO PREVENT PISTON PIN FAILURE, WHICH COULD RESULT IN ENGINE FAILURE ©ATP		N/A NEW ENGINE	Once ©ATP		1. 2. 3. 4. Signature:
98-17-11 C 10/19/1998 ©ATP	TO PREVENT CRANKSHAFT FAILURE DUE TO CRACKING, WHICH COULD RESULT IN AN INFLIG ©ATP		N/A NEW ENGINE	Once ©ATP		1. 2. 3. 4. Signature:
98-18-12 09/28/1998 ©ATP	TO PREVENT ROTARY FUEL PUMP LEAKS,WHICH COULD RESULT IN AN ENGINE FAILURE,ENGI ©ATP		N/A PER FUEL PUMP TYPE. NOT RECURRING.	Recur ©ATP		1. 2. 3. 4. Signature:
2002-19-03 09/20/2002 ©ATP	To prevent crankshaft failure, which could result in total engine power loss, ©ATP		N/A NOT MODIFIED PER STC.	Once ©ATP		1. 2. 3. 4. Signature:

Airworthiness Directive Compliance RecordFile ID: **51732ENG**

USARL Research Date: 11/08/2002

ManufacturerModel

Textron Lycoming

IO-540-AB1A5

Part # :Serial # : L-28292-48A

AD Number Effective date	Description	Complied Date Time	Status Method of Compliance/Applicability	Once or Recur	Next Due Date Time	1. Facility 2. Cert. Type 3. Cert. Num. 4. Author. by
2002-20-51 E 10/01/2002 ©ATP	To prevent loss of all engine power and possible forced landing ©ATP		N/A DUE TO SERIAL NUMBER.	Once ©ATP		1. 2. 3. 4. Signature:
						1. 2. 3. 4. Signature:
						1. 2. 3. 4. Signature:
						1. 2. 3. 4. Signature:
						1. 2. 3. 4. Signature:
						1. 2. 3. 4. Signature:
						1. 2. 3. 4. Signature: