US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

OMB No. 2120-0020
Exp: 5/31/2018

Electronic Tracking Number

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. §44701). Failure to report can result in a civil penalty for each such violation. (49 U.S.C. §46301(a))

1. Aircraft	NATIONALITY AND REGISTRATION MARK: N1833D USA	SERIAL NO: CE-952
	MAKE: BEECH	MODEL: F33A / SERIES:
2. Owner	Name *(As shown on registration certificate)*: BUTLER JEROME KNOX	Address *(As shown on registration certificate)*: Address 1310 IMBODEN DR; City FRONT ROYAL; State VIRGINIA; Zip 22630-5341; Country USA

3. For FAA Use Only

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial No.
☐	☑	AIRFRAME	BEECH	*(As described in Item 1 above)*	CE-952
☐	☐	POWERPLANT			
☐	☐	PROPELLER			
☐	☐	APPLIANCE	Type / Manufacturer		

6. Conformity Statement

A. Agency's Name and Address		B. Kind of Agency		
NAME: On Wing LLC		☐ U. S. Certificated Mechanic	☐ Manufacturer	
ADDRESS: 511 Airport Rd		☐ Foreign Certificated Mechanic	C. Certificate No.	
CITY: Winchester	STATE Virginia	☑ Certificated Repair Station	2OWR084C	
ZIP: 22602	COUNTRY USA	☐ Certificated Maintenance Organization		

D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B ☐

Signature/Date of Authorized Individual: Robert Kline 12/27/22

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is: ☑ Approved ☐ Rejected

BY							
	☐	FAA Flt. Standards Inspector	☐	Manufacturer	☐	Maintenance Organization	☐ Persons Approved by Canadian Department of Transport
	☐	FAA Designee	☑	Repair Station	☐	Inspection Authorization	Other *(Specify)*

CERTIFICATE OR DESIGNATIOCN NO: 2OWR084C

Signature/Date of Authorized Individual: Robert Kline 12/27/22

FAA Form 337 (10/06)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished
(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

N1833D	USA	12/27/2022
Nationality and Registration Mark		Date:

1. Introduction: Installation of Garmin GTN 750Xi.
2: Description. Removed existing King KR 87 ADF Receiver PN: 066-1072-00 SN: 23472, Allied Signal KA92 GPS Antenna PN: 071-01553-0200 SN:19474, and King KLN 90B GPS PN: 066-04031-1121 SN: 26841. Installed Garmin GTN750Xi PN: 011-04634-00 SN: 5FR005839.
3: Control information.Reference Garmin GTN Xi Maintenance Manual PN:190-0107-C1 Rev.2 Section 3 for system operation and self test information.
4: Servicing information.The GTN Xi, GMA 35, and Flight Stream 210/510 do not require servicing. In the event of system failure, troubleshoot the GTN Xi, GMA 35, and Flight Stream 210/510 in accordance with Section 5 of the GTN M.M PN: 190-0107-C1 Rev.2
5: Maintenance instructions: Reference Garmin GTN Xi Maintenance Manual PN:190-0107-C1 Rev.2 Section 4.2- 4.6.
6: Troubleshooting information: Reference Garmin GTN Xi Maintenance Manual PN:190-0107-C1 Rev.2, Section 5.
7: Removal/Replace information: Reference Garmin GTN Xi Maintenance Manual PN:190-0107-C1 Rev.2, Section 6.
8: Diagram: Reference Garmin GTN Xi Part 23 AML STC Installation Manual PN:190-01007-C0 Rev.6 Section 4 and Appendix A&B.
9: Special inspection Requirements: None.
10: Application of Special Treatments: None.
11: Data: Reference Garmin GTN750Xi STC# SA02019SE-D, Garmin GTN Xi Part 23 AML STC Installation Manual PN: 190-01007-C0 Rev.6, Garmin GTN Xi Maintenance Manual PN:190-0107-C1 Rev.2, AC 43.13-2A and AC 43.13-1B with change 1.
12: List of Special tools: Calibrated milliohm meter with an accuracy of +0.1 milliohm or better is required to measure the electrical bonding between the GTN Xi/GMA system components and aircraft ground.
13: Commuter Aircraft: Not Applicable.
14: Recommended Overhaul Periods: None.
15: Approved Airworthiness limitations: There are no new airworthiness limitations associated with this equipment and/or installation.
16: Maintenance information ICAs. Revision instructions: Revisions to this installation shall submit a revised FAA Form 337 to the local FSDO.

Additional Sheets Are Attached

☐

US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

OMB No. 2120-0020
Exp: 5/31/2018

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1. Aircraft	NATIONALITY AND REGISTRATION MARK: N1833D USA	SERIAL NO: CE-952	
	MAKE: BEECH	MODEL: F33A	SERIES:
2. Owner	Name *(As shown on registration certificate)* BUTLER JEROME KNOX	Address *(As shown on registration certificate)* Address 1310 IMBODEN DR	
		City FRONT ROYAL	State VIRGINIA
		Zip 22630-5341	Country USA

3. For FAA Use Only

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial No.
☐	☑	AIRFRAME	BEECH	*(As described in Item 1 above)*	CE-952
☐	☐	POWERPLANT			
☐	☐	PROPELLER			
☐	☐	APPLIANCE	Type / Manufacturer		

6. Conformity Statement

A. Agency's Name and Address		B. Kind of Agency		
NAME: On Wing LLC		☐ U. S. Certificated Mechanic	☐ Manufacturer	
ADDRESS: 511 Airport Rd		☐ Foreign Certificated Mechanic	C. Certificate No.	
CITY: Winchester	STATE Virginia	☑ Certificated Repair Station	2OWR084C	
ZIP: 22602	COUNTRY USA	☐ Certificated Maintenance Organization		

D. I certify that the repair and/or alteration made to the unit(s) identified in Item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B ☐

Signature/Date of Authorized Individual: 12/27/22

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is: ☑ Approved ☐ Rejected

BY								
	☐	FAA Flt. Standards Inspector	☐	Manufacturer	☐	Maintenance Organization	☐	Persons Approved by Canadian Department of Transport
	☐	FAA Designee	☑	Repair Station	☐	Inspection Authorization		Other *(Specify)*

CERTIFICATE OR DESIGNATIOCN NO: 2OWR084C

Signature/Date of Authorized Individual: 12/27/22

FAA Form 337 (10/06)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished
(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

N1833D USA 12/27/2022

Nationality and Registration Mark Date:

1. Introduction: Installation of Garmin GTX 345R.
2: Description. Removed existing Ack Encoder PN:A30.02-00, King KT76A ATC Transponder PN: 066-1062-00 SN: 45015 and uAvionix Tail Beacon PN: UAV-1001362-001-L. Installed Garmin GTX345R PN: 010-01217-06 SN: 3EH040791 and Rami Transponder Antenna PN: AV-74-1.
3: Control information. Reference Garmin GTX 33X and GTX 3X5 ADS-B Maintenance Manual PN:190-00734-11 Rev.8 Section 3 for system operation and self test information.
4: Servicing information. GTX 33X and GTX 3X5 LRU maintenance is "on condition" only. Component-level overhaul is not required for the GTX 33X and GTX 3X5 with ADS-B installation. On Condition replacement and/or servicing should occur when an item exhibits conditions, symptoms, and/or abnormalities as defined in Section 5 of the GTX Maintenance Manual. Replacement and/or servicing should be made only after the technician troubleshoots the system by using the guidance in the GTX Maintenance Manual along with common avionics maintenance practices.
5: Maintenance instructions: Reference Garmin GTX 345 Maintenance Manual P/N 190-00734-11 Rev. 8, Section 4.4.
6: Troubleshooting information: Reference Garmin GTX 345 Maintenance Manual P/N 190-00734-11 Rev. 8, Section 5.
7: Removal/Replace information: Reference Garmin GTX 345 Maintenance Manual P/N 190-00734-11 Rev. 8, Section 6.
8: Diagram: Reference Garmin GTX 3XX Part 23 AML STC Installation Manual PN:190-00734-10 Rev.15 Appendix A&B.
9: Special inspection Requirements: None.
10: Application of Special Treatments: None.
11: Data: Reference Garmin STC# SA01714WI, Garmin GTX 345 Installation manual P/N 190-00734-10 Rev.15, Garmin GTX 33X and GTX 3X5 ADS-B Maintenance Manual P/N 190-00734-11 Rev. 8, AC 43.13-2A and AC 43.13-1B with change 1.
12: List of Special tools: Calibrated milliohm meter with an accuracy of +0.1 milliohm or better, Calibrated transponder ramp tester, Calibrated Pitot/static ramp tester, 50 Ω 5 watt antenna load.
13: Commuter Aircraft: Not Applicable.
14: Recommended Overhaul Periods: None.
15: Approved Airworthiness limitations: There are no new airworthiness limitations associated with this equipment and/or installation.
16: Maintenance information ICAs. Revision instructions: Revisions to this installation shall submit a revised FAA Form 337 to the local FSDO.

The installed ADS-B OUT system was shown to meet the equipment performance requirements of 14 CFR section 91.227.

Additional Sheets Are Attached ☐

US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

OMB No. 2120-0020
Exp: 5/31/2018

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1. Aircraft	NATIONALITY AND REGISTRATION MARK: N1833D USA	SERIAL NO: CE-952	
	MAKE: BEECH	MODEL: F33A	SERIES:
2. Owner	Name *(As shown on registration certificate)* BUTLER JEROME KNOX	Address *(As shown on registration certificate)* Address 1310 IMBODEN DR	
		City FRONT ROYAL	State VIRGINIA
		Zip 22630-5341	Country USA

3. For FAA Use Only

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial No.
☐	☑	AIRFRAME	BEECH	*(As described in Item 1 above)*	CE-952
☐	☐	POWERPLANT			
☐	☐	PROPELLER			
☐	☐	APPLIANCE	Type / Manufacturer		

6. Conformity Statement

A. Agency's Name and Address		B. Kind of Agency		
NAME: On Wing LLC		☐ U. S. Certificated Mechanic	☐ Manufacturer	
ADDRESS: 511 Airport Rd		☐ Foreign Certificated Mechanic	C. Certificate No.	
CITY: Winchester	STATE Virginia	☑ Certificated Repair Station	2OWR084C	
ZIP: 22602	COUNTRY USA	☐ Certificated Maintenance Organization		

D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B ☐

Signature/Date of Authorized Individual: 12/27/22

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is: ☑ Approved ☐ Rejected

BY								
	☐	FAA Flt. Standards Inspector	☐	Manufacturer	☐	Maintenance Organization	☐	Persons Approved by Canadian Department of Transport
	☐	FAA Designee	☑	Repair Station	☐	Inspection Authorization		Other *(Specify)*

CERTIFICATE OR DESIGNATIOCN NO: 2OWR084C

Signature/Date of Authorized Individual: 12/27/22

FAA Form 337 (10/06)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished
(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

N1833D USA 12/27/2022

Nationality and Registration Mark Date:

1. Introduction: Installation of Garmin G3X 10" display and asociated Garmin systems.
2. Discription: Removed Stormscope WX-10A PN: 78-8047-0984-4 SN:R0D04501091, Stormscope WX-10A Antenna PN: 78-8041-7950-1 SN: 2506, Stormscope WX-10A Processor PN: 78-8047-0985-1 SN: R0P04501091, Stormscope WX-10A Mounting Plate, Troll Technology Copr. Avionics Fan, United Instruments Airspeed PN: 8025 SN: 84824, United Instruments Altimeter PN: 5934-3 SN: 467965, United instruments Vertical Speed PN: 7040 SN: 11517, Aviation Instruments MFG. Turn Coordinator PN:507-0020-906/TC 120-2D SN: M2803, Beechcraft Com Antenna PN: 35-5006-1 SN: 1269, Beechcraft NAV Splitter PN:35-5024 SN: 5901, King KA 44 ADF Antenna PN: 071-1196-00 SN: 6922, King KI203 VOR/LOC Converter Indicator PN: 066-3034-00 SN: 13230, King KDI572 DME Indicator PN: 066-1069-00 SN: 3637, King KI227 ADF Indicator PN: 066-3063-00 SN: 5278, King KLN 90B GPS PN: 066-04031-1121 SN: 26841, King KMA 24 Marker Beacon Receiver PN: 066-1055-03 SN: 19359, King KN53 Navigation Receiver PN:066-1067-01 SN: 54144, King KNS 81 Navigational Computer PN: 066-4010-10 SN: 5346, King KR 87 ADF Receiver PN: 066-1072-00 SN: 23472. Installed Cobham VHF Blade Com Antenna PN:CI-138 SN: 695917, Garmin GA35 GPS Antenna PN: 013-00235-00 SN: 210524, Garmin GDU 460 PN: 011-02920-05 SN: 350011323, Garmin GMU11 PN: 011-04349-01 SN: 56J06878, Garmin GSU25 PN: 011-02929-51 SN: 5Q3003220, Garmin GTP59 PN: 011-00978-00 SN: 47971321, Garmin 6 Cylinder Txi Engine Sensor Kit PN:K00-01011-00 SN: 4TK008723, and Garmin GEA24 PN: 011-02848-01 SN: 2J2022403.
3. Control Information: Reference Garmin G3X Touch EFIS Maintenance Manual PN:190-02472-02 Rev.5 Section 3 for Controling and Operating information.
4. Servicing Information: G3X Touch EFIS LRUs maintenance is 'on condition' only. No component-level overhaul is required for the G3X Touch EFIS Part 23 AML STC installation. See section 6 of Garmin G3X M.M. PN:190-02472-02 Rev.5 for equipment removal and installation.
5. Maintenance Information: Reference Section 4.2-4.7 of the Garmin G3X M.M. PN:190-02472-02 Rev.5.
6. Troubleshooting Information: Reference Section 5 of the Garmin G3X M.M. PN:190-02472-02 Rev.5.
7. Removal/Replacement Information: Reference Section 6 of the Garmin G3X M.M. PN:190-02472-02 Rev.5.
8. Diagrams: Reference Appendix B and I of the Garmin G3X Installation Manual PN:190-02472-01 Rev.9. for installation diagrams and wiring diagram interconnections.
9. Special Inspection Requirements: Reference Section 4.7 of the Garmin G3X M.M. PN:190-02472-02 Rev.5.
10. Application of Special Treatments: There are no special treatment requirements for the Garmin G3X system or its components.
11. Data: Garmin G3X STC# SA01899WI, Garmin G3X Installation Manual PN:190-02472-01, Garmin G3X Maintenance Manual PN:190-02472-02 Rev.5, AC 43.13-1B with Change 1 and AC 43.13-2A.
12. List of Special Tools: The Garmin G3X system has no specials tools required to perform maintenance on its LRUs and components.
13. Commuter Aircraft?: Not Applicable.
14. Recommended Overhaul Periods: The Garmin G3X system does not require overhauling at a specific time period.
15. Approved Airworthiness Limitations: The G3X Touch EFIS is airworthy when installed, configured, and maintained in accordance with section 4 of the Garmin G3X Maintenance Manual PN:190-02472-02 Rev.5. There are no new (or additional) airworthiness limitations associated with this equipment and/or installation. The Airworthiness Limitations section is FAA-approved and specifies maintenance required under§§43.16 and 91.403 of Title 14 of the Code of Federal Regulations, unless an alternative program has been FAA-approved.
16. Maintenance Information ICAs. Revision Instructions: Revisions to this installation shall submit a revised FAA Form 337 to the local FSDO.

Additional Sheets Are Attached

☐

US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

OMB No. 2120-0020
Exp: 5/31/2018

Electronic Tracking Number

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. §44701). Failure to report can result in a civil penalty for each such violation. (49 U.S.C. §46301(a))

1. Aircraft	NATIONALITY AND REGISTRATION MARK: N1833D USA	SERIAL NO: CE-952	
	MAKE: BEECH	MODEL: F33A	SERIES:
2. Owner	Name *(As shown on registration certificate)* BUTLER JEROME KNOX	Address *(As shown on registration certificate)* Address 1310 IMBODEN DR	
		City FRONT ROYAL	State VIRGINIA
		Zip 22630-5341	Country USA

3. For FAA Use Only

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial No.
☐	☑	AIRFRAME	BEECH	*(As described in Item 1 above)*	CE-952
☐	☐	POWERPLANT			
☐	☐	PROPELLER			
☐	☐	APPLIANCE	Type / Manufacturer		

6. Conformity Statement

A. Agency's Name and Address		B. Kind of Agency		
NAME: Robert Kline		☑ U. S. Certificated Mechanic	☐ Manufacturer	
ADDRESS: 36181 Foxlore Farm Ln		☐ Foreign Certificated Mechanic	C. Certificate No.	
CITY: Round Hll	STATE VIRGINIA	☐ Certificated Repair Station	AP3807569IA	
ZIP: 20141	COUNTRY usa	☐ Certificated Maintenance Organization		

D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B ☐

Signature/Date of Authorized Individual: Robert R Kline 12/27/22

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is: ☑ Approved ☐ Rejected

BY								
	☐	FAA Flt. Standards Inspector	☐	Manufacturer	☐	Maintenance Organization	☐	Persons Approved by Canadian Department of Transport
	☐	FAA Designee	☐	Repair Station	☑	Inspection Authorization		Other *(Specify)*

CERTIFICATE OR DESIGNATION NO: AP3807569IA

Signature/Date of Authorized Individual: Robert R Kline 12/27/22

FAA Form 337 (10/06)

NOTICE

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8. Description of Work Accomplished
(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

N1833D USA 12/27/2022

Nationality and Registration Mark Date:

1. Introduction: Installation of the Garmin GFC500 2 Axis Autopilot with Optional Pitch Trim Control.
2. Description: Removed A.I.M Horizon Ref. ind. PN: 500DCF 28 SN:2883, A.I.M. Directional Gyro PN: 505-0010-902 SN: 20234, Airborne Instrument Air PN: 169-38 SN: 58-380017-3, Century 3 Autopilot Head PN: N/A SN: N/A, Century Flight System ADC PN: 1C535-1 SN: K11264-D, EDO-AIRE Mitchell CENT/ALT IIIC AMP PN: 1C515-1 CENT SN: 9291B, EDO-AIRE MITCHELL G/S COUPLER BOX PN: 1C493 SN: 10120E, EDO-AIRE MITCHELL GYRO SLAVING AMP BOX PN: 1C714 SN: 4005 A, EDO-AIRE MITCHELL Pitch Servo PN: 1C750-1-668P SN: 241, EDO-AIRE Mitchell Electronic Pitch Trim Amp. PN: 1C709-20 SN: 15224 F, EDO-AIRE MITCHELL Pitch Trim Servo PN: 1C469-6-662 SN: 21738A, EDO-AIRE Mitchell Radio Coupler PN: 1C388-2 SN: 3126, EDO-AIRE Mitchell Relay Box PN: 1A526-1 SN: 21719, EDO-AIRE MITCHELL Roll Servo PN: 1C465-1-662R SN: 10551 A, Mitchell Attitude Gyro PN:52D167M/IU367-102-2 SN: 3675E. Installed Garmin GMC507 PN: 011-04548-01 SN: 5H1100529, Garmin GSA 28 Pitch Trim Servo PN: 011-02927-21 SN:6N2003154, Garmin GSA28 Pitch Servo PN: 011-02927-21 SN: 6N2003159, Garmin GSA28 Roll Servo PN: 011-02927-21 SN: 6N2003076.
3. Control Information: Reference Section 3 Garmin GFC500 STC Maintenance Manual PN: 190-02291-01 Rev 10 for Controlling & Operating Information of the Garmin GFC500 Autopilot.
4. Servicing Information: The Garmin GFC500 Autopilot is ' On Condition ' Servicing only. In the event of a system failure, troubleshoot the Garmin GFC500 Autopilot in accordance with Section 5 of the Garmin GFC500 STC Maintenance Manual PN: 190-02291-01 Rev 10.
5. Maintenance Information: Reference Section 4.3, 4.4, & 4.5 of Garmin GFC500 STC Maintenance Manual PN: 190-02291-01 Rev 10 for Maintenance Instructions and Intervals for the Garmin GFC500 Components and LRUs.
6. Troubleshooting Information: Reference Section 5 of Garmin GFC500 STC Maintenance Manual PN: 190-02291-01 Rev 10 for Troubleshooting information and Instructions.
7. Removal/Replacement Information: Reference Section 6 of Garmin GFC500 STC Maintenance Manual PN: 190-02291-01 Rev 10 for information and instructions on removing and or replacing the Garmin GFC500 Components and LRUs.
8. Diagrams: Reference Appendix A of Garmin GFC500 STC Installation Manual PN: 190-02291-00 Rev 21 for wiring diagrams and wiring interconnections. Reference Garmin GFC500 STC Installation Manual Addendum PN: 190-02291-13 Rev 13 for Equipment Installation Diagrams and component routing.
9. Special Inspection Requirements: There are no special inspections required with the Garmin GFC500 Autopilot System and its LRUs.
10. Application of Special Treatments: There are no special treatment requirements for the Garmin GFC500 System and its components.
11. Data: STC# Garmin GFC500 STC# SA01866WI, Garmin GFC500 STC Installation Manual PN: 190-02291-00 Rev 21, Garmin GFC500 STC Installation Manual Addendum PN: 190-02291-13 Rev 13, Garmin GFC500 STC Maintenance Manual PN: 190-02291-01 Rev 10, & Garmin GFC500 STC Gain Addendum PN: 005-01470-21 Rev 3.
12. List of Special Tools: The Garmin GFC500 Autopilot has no specials tools required to perform maintenance on its LRUs and components.
13. Commuter Aircraft?: Not Applicable.
14. Recommended Overhaul Periods: The Garmin GFC500 Autopilot system does not require overhauling at a specific time period.
15. Approved Airworthiness Limitations: The Garmin GFC500 Autopilot is airworthy when installed, configured, and maintained in accordance with section 4 of Garmin GFC500 STC Maintenance Manual PN: 190-02291-01 Rev 10. There are no new (or additional) airworthiness limitations associated with this equipment and/or installation. The Airworthiness Limitations section is FAA approved and specifies maintenance required under§§ 43.16 and 91.403 of Title 14 of the Code of Federal Regulations unless an alternative program has been FAA approved.
16. Maintenance Information ICAs. Revision Instructions: Revisions to this installation shall submit a revised FAA Form 337 to the local FSDO.

Additional Sheets Are Attached

☐

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(Airframe, Powerplant, Propeller, or Appliance)

OMB No. 2120-0020
Exp: 5/31/2018

Electronic Tracking Number

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1. Aircraft	NATIONALITY AND REGISTRATION MARK: N1833D USA	SERIAL NO: CE-952	
	MAKE: BEECH	MODEL: F33A	SERIES:
2. Owner	Name *(As shown on registration certificate)* BUTLER JEROME KNOX	Address *(As shown on registration certificate)* Address 1310 IMBODEN DR	
		City FRONT ROYAL	State VIRGINIA
		Zip 22630-5341	Country USA

3. For FAA Use Only

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial No.
☐	☑	AIRFRAME	BEECH	*(As described in Item 1 above)*	CE-952
☐	☐	POWERPLANT			
☐	☐	PROPELLER			
☐	☐	APPLIANCE	Type / Manufacturer		

6. Conformity Statement

A. Agency's Name and Address		B. Kind of Agency		
NAME: On Wing LLC		☐ U. S. Certificated Mechanic	☐ Manufacturer	
ADDRESS: 511 Airport Rd		☐ Foreign Certificated Mechanic	C. Certificate No.	
CITY: Winchester	STATE Virginia	☑ Certificated Repair Station	2OWR084C	
ZIP: 22602	COUNTRY USA	☐ Certificated Maintenance Organization		

D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B ☐

Signature/Date of Authorized Individual: Robert Klein 12/27/22

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is: ☑ Approved ☐ Rejected

BY								
	☐	FAA Flt. Standards Inspector	☐	Manufacturer	☐	Maintenance Organization	☐	Persons Approved by Canadian Department of Transport
	☐	FAA Designee	☑	Repair Station	☐	Inspection Authorization		Other *(Specify)*

CERTIFICATE OR DESIGNATIOCN NO: 2OWR084C

Signature/Date of Authorized Individual: Robert Klein 12/27/22

FAA Form 337 (10/06)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished
(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

N1833D USA 12/27/2022

Nationality and Registration Mark Date:

1. Installation of Garmin G5 Electronic Attitude indicator.
2. Removed Mitchell Attitude gyro PN:IU367-102-2 SN:3675E, Slaved Navigation Situation Display PN:52D137-1333 SN: I-1050 and installed Garmin G5 PN: 011-03809-00 SN: 4JQ058025.
3. Control Information: Reference Garmin G5 Installation Manual PN: 190-01112-10 Rev 29, for system operation and self-test information.
4. Servicing Information: Refer to Section 4 of Garmin G5 Maintenance Manual PN: 190-01112-11 Rev 10 for servicing information.
5. Maintenance Instructions: Refer to Section 4 of Garmin G5 Maintenance Manual PN: 190-01112-11 Rev 10 for Maintenance instructions.
6. Troubleshooting Information: For troubleshooting the Dual G5 System, refer to Section 5 of Garmin G5 Maintenance Manual PN: 190-01112-11 Rev 10.
7. Removal/Replacement Information: Refer to section 6 of Garmin G5 Maintenance Manual PN:190-01112-11 Rev 10.
8. Diagrams: See figures in Section 4 of the Garmin G5 installation Manual PN: 190-01112-10 Rev. 29.
9. Special Inspection Requirements: Refer to Section 4.4 of Garmin G5 Maintenance Manual PN: 190-01112-11 Rev 10. After a suspected lightning strike the LPM shall be replaced. Reference Garmin G5 Electronic Flight Instrument Part 23 AML STC Installation Manual, 190-01112-10 Rev.29 section 4 and 5 for connector assembly and wiring requirements of LPM.
10. Application of Special Treatments: None.
11. Data: Reference Garmin G5 Installation Manual PN: 190-01112-10 Rev 29, Garmin G5 Maintenance Manual PN: 190-01112-11 Rev 10, Garmin STC#SA01818WI, AC43.13-1B with change 1 and AC43.13-2A.
12. Special Tools: None.
13. Commuter Aircraft: Not Applicable.
14. Recommended Overhaul Periods: None.
15. Approved Airworthiness Limitations: There are no new or additional airworthiness limitations associated with this equipment and/or installation. The Airworthiness Limitations section is FAA-approved and specifies maintenance required under§§43.16 and 91.403 of Title 14 of the Code of Federal Regulations, unless an alternative program has been FAA-approved.
16. Maintenance Information ICAs. Revision Instructions: Revisions to this installation shall submit a revised FAA Form 337 to the local FSDO.

Additional Sheets Are Attached

☐

US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

OMB No. 2120-0020
Exp: 5/31/2018

Electronic Tracking Number

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. §44701). Failure to report can result in a civil penalty for each such violation. (49 U.S.C. §46301(a))

1. Aircraft	NATIONALITY AND REGISTRATION MARK: N1833D USA	SERIAL NO: CE-952	
	MAKE: BEECH	MODEL: F33A	SERIES: N/A
2. Owner	Name *(As shown on registration certificate)* BUTLER JEROME KNOX	Address *(As shown on registration certificate)* Address 1310 IMBODEN DR	
		City FRONT ROYAL	State VIRGINIA
		Zip 22630-5341	Country USA

3. For FAA Use Only

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial No.
☐	☑	AIRFRAME	BEECH	*(As described in Item 1 above)*	CE-952
☐	☐	POWERPLANT			
☐	☐	PROPELLER			
☐	☐	APPLIANCE	Type / Manufacturer		

6. Conformity Statement

A. Agency's Name and Address		B. Kind of Agency		
NAME: Robert Kline		☑ U. S. Certificated Mechanic	☐ Manufacturer	
ADDRESS: 36181 Foxlore Farm Ln		☐ Foreign Certificated Mechanic	C. Certificate No.	
CITY: Round Hill	STATE Virginia	☐ Certificated Repair Station	AP3807659IA	
ZIP: 20141	COUNTRY USA	☐ Certificated Maintenance Organization		

D. I certify that the repair and/or alteration made to the unit(s) identified in Item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B ☐

Signature/Date of Authorized Individual: Robert Kline 12/27/22

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is: ☑ Approved ☐ Rejected

BY								
	☐	FAA Flt. Standards Inspector	☐	Manufacturer	☐	Maintenance Organization	☐	Persons Approved by Canadian Department of Transport
	☐	FAA Designee	☐	Repair Station	☑	Inspection Authorization		Other *(Specify)*

CERTIFICATE OR DESIGNATION NO: AP3807659IA

Signature/Date of Authorized Individual: Robert Kline 12/27/22

FAA Form 337 (10/06)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished
(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

N1833D USA 12/27/2022

Nationality and Registration Mark Date:

1. Introduction: Installation of CIES Fuel Quantity senders
2. Description: Removed existing Beechcraft Fuel Sensor PN: 96 38004101 SN: CB7740 2 A79, Beechcraft Fuel Sensor PN: 96 380042 1 SN: B7740 1 H79, Beechcraft Fuel Sensor PN: 96 380042 2 SN: CB7740-47/M79, Beechcraft Fuel Sensor PN: 96 380041 1 SN: CB7740. Installed CiES Outboard Fuel senders PN: CC284022-2218-101 SN: GT5177 & GT5178, CiES Inboard Fuel Senders PN: CC284022-2217-101 SN: GT5175 & GT5176,
3. Control information: Ref. CIES Installation Manual & ICA Doc. # CC-TR-2840-031 Rev. F
4. Servicing Information: Ref. Section 8 Part A(4) of CIES Installation Manual & ICA Doc. # CC-TR-2840-031 Rev. F
5. Maintenance instruction: Ref. Section 8 Part B of CIES Installation Manual & ICA Doc. # CC-TR-2840-031 Rev. F
6. Troubleshooting information: Ref. Section 8 Part B(2) of CIES Installation Manual & ICA Doc. # CC-TR-2840-031 Rev. F
7. Removal/Replacement information: Ref. Section 3.2 of CIES Installation Manual & ICA Doc. # CC-TR-2840-031 Rev. F
8. Diagrams: Ref. CIES Installation Manual & ICA Doc. # CC-TR-2840-031 Rev. F
9. Special inspection requirements: Not Applicable
10. Application of Special Treatments: Not Applicable
11.Data:STC# SA02511SE, CIES Installation Manual & ICA Doc. # CC-TR-2840-031 Rev. F and in Compliance with AC 43.13-1B with Change 1 and AC 43.13-2A.
12.List of Special Tools: Not Applicable
13. Commuter Aircraft?: Not Applicable.
14.Recommended Overhaul Periods: No additional overhaul time limitations. Removed and repair on-condition.
15.Approved Airworthiness Limitations: There are no new(or additional) Airworthiness Limitation associated with this installation.
16. Maintenance information, ICA's. Revision instructions: Revisions to this installation shall submit a revised FAA Form 337 to the local FSDO.

Additional Sheets Are Attached

☐

US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

OMB No. 2120-0020
Exp: 5/31/2018

Electronic Tracking Number

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. §44701). Failure to report can result in a civil penalty for each such violation. (49 U.S.C. §46301(a))

1. Aircraft	NATIONALITY AND REGISTRATION MARK: N1833D USA	SERIAL NO: CE-952
	MAKE: BEECH	MODEL: F33A / SERIES:
2. Owner	Name *(As shown on registration certificate)*: BUTLER JEROME KNOX	Address *(As shown on registration certificate)*: Address 1310 IMBODEN DR; City FRONT ROYAL; State VIRGINIA; Zip 22630-5341; Country USA

3. For FAA Use Only

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial No.
☐	☑	AIRFRAME	BEECH	*(As described in Item 1 above)*	CE-952
☐	☐	POWERPLANT			
☐	☐	PROPELLER			
☐	☐	APPLIANCE	Type / Manufacturer		

6. Conformity Statement

A. Agency's Name and Address		B. Kind of Agency		
NAME: Robert Kline		☑ U. S. Certificated Mechanic	☐ Manufacturer	
ADDRESS: 36181 Foxlore Farm Ln		☐ Foreign Certificated Mechanic	C. Certificate No.	
CITY: Round Hill	STATE Virginia	☐ Certificated Repair Station	AP3807569IA	
ZIP: 20141	COUNTRY USA	☐ Certificated Maintenance Organization		

D. I certify that the repair and/or alteration made to the unit(s) identified in Item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B ☐

Signature/Date of Authorized Individual: Robert R Kline 12/28/22

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is: ☑ Approved ☐ Rejected

BY								
	☐	FAA Flt. Standards Inspector	☐	Manufacturer	☐	Maintenance Organization	☐	Persons Approved by Canadian Department of Transport
	☐	FAA Designee	☐	Repair Station	☑	Inspection Authorization		Other *(Specify)*

CERTIFICATE OR DESIGNATIOCN NO: AP3807569IA

Signature/Date of Authorized Individual: Robert R Kline 12/28/22

FAA Form 337 (10/06)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished
(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

N1833D USA 12/28/2022

Nationality and Registration Mark Date:

1. Installation of Parmetheus PlusPAR36 Taxi Light.
2. Removed GE Taxi light PN:4587 and installed Whelen LED Parmetheus Plus Par 36 taxi light PN:01-0771833-25 SN: 541413.
3. Control Information: Reference Whelen Installation Manual PN: 14793D.
4. Servicing Information: Reference Whelen Installation Manual PN: 14793D, Page 3.
5. Maintenance Instructions: Reference Whelen Installation Manual PN: 14793D, Page 3.
6. Troubleshooting Information: Reference Whelen Installation Manual PN: 14793D, Page 4.
7. Removal/Replacement Information: Reference Whelen Installation Manual PN: 14793D, Page 1.
8. Diagrams: Reference Whelen Installation Manual PN: 14793D, Pages 2, 4, and 5.
9. Special Inspection Requirements: None
10. Application of Special Treatments: None.
11. Data: Reference Whelen Installation Manual PN: 14793D, Whelen STC#SA00344BO, AC43.13-1B with change 1 and AC43.13-2A.
12. Special Tools: None.
13. Commuter Aircraft: Not Applicable.
14. Recommended Overhaul Periods: None.
15. Approved Airworthiness Limitations: There are no new or additional airworthiness limitations associated with this equipment and/or installation.
16. Maintenance Information ICAs. Revision Instructions: Revisions to this installation shall submit a revised FAA Form 337 to the local FSDO.

Additional Sheets Are Attached

☐

US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

OMB No. 2120-0020
Exp: 5/31/2018

Electronic Tracking Number

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. §44701). Failure to report can result in a civil penalty for each such violation. (49 U.S.C. §46301(a))

1. Aircraft	NATIONALITY AND REGISTRATION MARK: N1833D USA	SERIAL NO: CE-952	
	MAKE: BEECH	MODEL: F33A	SERIES:
2. Owner	Name *(As shown on registration certificate)* BUTLER JEROME KNOX	Address *(As shown on registration certificate)* Address 1310 IMBODEN DR	
		City FRONT ROYAL	State VIRGINIA
		Zip 22630-5341	Country USA

3. For FAA Use Only

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial No.
☐	☑	AIRFRAME	BEECH	*(As described in Item 1 above)*	CE-952
☐	☐	POWERPLANT			
☐	☐	PROPELLER			
☐	☐	APPLIANCE	Type / Manufacturer		

6. Conformity Statement

A. Agency's Name and Address		B. Kind of Agency		
NAME: On Wing LLC		☐ U. S. Certificated Mechanic	☐ Manufacturer	
ADDRESS: 511 Airport Rd		☐ Foreign Certificated Mechanic	C. Certificate No.	
CITY: Winchester	STATE Virginia	☑ Certificated Repair Station	2OWR084C	
ZIP: 22602	COUNTRY USA	☐ Certificated Maintenance Organization		

D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B ☐

Signature/Date of Authorized Individual: Robert Kline 12/28/22

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is: ☑ Approved ☐ Rejected

BY								
	☐	FAA Flt. Standards Inspector	☐	Manufacturer	☐	Maintenance Organization	☐	Persons Approved by Canadian Department of Transport
	☐	FAA Designee	☑	Repair Station	☐	Inspection Authorization		Other *(Specify)*

CERTIFICATE OR DESIGNATIOCN NO: 2OWR084C

Signature/Date of Authorized Individual: Robert Kline 12/28/22

FAA Form 337 (10/06)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished
(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

N1333D	USA	12/28/2022
Nationality and Registration Mark		Date:

1. Introduction: Installation of RadioRax Rail Tray mounting kit.
2: Description: Installed RadioRax Rail Tray mounting kit PN:RK1020.
3: Control information: Reference RadioRax Installation Manial PN:RKD1020-2 Rev.B.
4: Servicing information: Reference RadioRax Instructions for Continued Airworthiness PN: RKD1020-4 Rev.A.
5: Maintenance instructions: Reference RadioRax Instructions for Continued Airworthiness PN: RKD1020-4 Rev.A.
6: Troubleshooting information: Reference RadioRax Instructions for Continued Airworthiness PN: RKD1020-4 Rev.A, and RadioRax Installation Manial PN:RKD1020-2 Rev.B.
7: Removal/Replace information: Reference RadioRax Installation Manial PN:RKD1020-2 Rev.B.
8: Diagram: Reference RadioRax Installation Manial PN:RKD1020-2 Rev.B.
9: Special inspection Requirements: None.
10: Application of Special Treatments: None.
11: Data: Reference RadioRax STC# SA01330LA, RadioRax Installation Manial PN:RKD1020-2 Rev.B, RadioRax Instructions for Continued Airworthiness PN: RKD1020-4 Rev.A, and AC 43.13-1B with change 1.
12: List of Special tools: None
13: Commuter Aircraft: Not Applicable.
14: Recommended Overhaul Periods: None.
15: Approved Airworthiness limitations: There are no new airworthiness limitations associated with this equipment and/or installation.
16: Maintenance information ICAs. Revision instructions: Revisions to this installation shall submit a revised FAA Form 337 to the local FSDO.

Additional Sheets Are Attached

☐

US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

OMB No. 2120-0020
Exp: 5/31/2018

Electronic Tracking Number

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. §44701). Failure to report can result in a civil penalty for each such violation. (49 U.S.C. §46301(a))

1. Aircraft	NATIONALITY AND REGISTRATION MARK: N1833D	SERIAL NO: CE-952	
	MAKE: Beechcraft	MODEL: F33A	SERIES: N/A
2. Owner	Name *(As shown on registration certificate)* BUTLER JEROME KNOX	Address *(As shown on registration certificate)* Address 1310 IMBODEN DR City FRONT ROYAL State Virginia Zip 22630-5341 Country USA	

3. For FAA Use Only

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial No.
☐	☑	AIRFRAME	Beechcraft	*(As described in Item 1 above)*	CE-952
☐	☐	POWERPLANT			
☐	☐	PROPELLER			
☐	☐	APPLIANCE	Type Manufacturer		

6. Conformity Statement

A. Agency's Name and Address		B. Kind of Agency		
NAME: David Edelman		☑ U. S. Certificated Mechanic	☐ Manufacturer	
ADDRESS: 411 Old Bethel Rd		☐ Foreign Certificated Mechanic	C. Certificate No.	
CITY: Edinburg	STATE VA	☐ Certificated Repair Station	4432100	
ZIP: 22824	COUNTRY USA	☐ Certificated Maintenance Organization		

D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B ☐	Signature/Date of Authorized Individual David Edelman 12/14/22

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is: ☑ Approved ☐ Rejected

BY								
	☐	FAA Flt. Standards Inspector	☐	Manufacturer	☐	Maintenance Organization	☐	Persons Approved by Canadian Department of Transport
	☐	FAA Designee	☐	Repair Station	☑	Inspection Authorization		Other *(Specify)*

CERTIFICATE OR DESIGNATION NO: 3807659 IA	Signature/Date of Authorized Individual 12/14/22

FAA Form 337 (10/06)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

N1833D	12/13/2022
Nationality and Registration Mark	Date:

1. Installation of STC SA01120SE Door Steward kit B40C10-1 in reference to Installation Instructions and ICA PN: MVA-B40C10M&O Rev.F, and AC 43.14-1B with change 1.
2. Removed main door stop PN:35-400379 and baggage door stop PN169-380032-5. Installed Door Steward kit B40C10-1 on Rear baggage door and on Cabin door.
3. Control Information: Reference Beechcraft Bonanza F33A POH PN:33-590009-13A9
4. Servicing Information: Refer Section B of MVA-B40C10M&O
5. Maintenance Instructions: Refer Section B of MVA-B40C10M&O
6. Troubleshooting Information: Refer Section B of MVA-B40C10M&O
7. Removal/Replacement Information: Refer Section B of MVA-B40C10M&O
8. Diagrams: See Section A1 of MVA-B40C10M&O
9. Special Inspection Requirements: None
10. Application of Special Treatments: None.
11. Data: Reference STC SA01120SE, Installation Instructions & ICA PN: MVA-B40C10M&O Rev.F, and AC 43.14-1B with change 1.
12. Special Tools: None.
13. Commuter Aircraft: Not Applicable.
14. Recommended Overhaul Periods: None.
15. Approved Airworthiness Limitations: There are no new or additional airworthiness limitations associated with this equipment and/or installation.
16. Maintenance Information ICAs. Revision Instructions: Revisions to this installation shall submit a revised FAA Form 337 to the local FSDO.

Additional Sheets Are Attached

☐

U.S. Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No.2120-0020

For FAA Use Only

Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed $1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make BEECH	Model F33A
	Serial No. *CE-952*	Nationality and Registration Mark N1833D
2. Owner	Name (As shown on registration certificate) HARR JOSEPH L. HARR JOANN R.	Address (As shown on registration certificate) P.O. BOX 3370 CAMP VERDE, AZ. 86322-3370

3. For FAA Use Only

4. Unit Identification				**5. Type**	
Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	*(As described in item 1 above)*				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type Manufacturer				

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency		C. Certificate No.
John Atterholt Legend Aviation LLC DBA Airzona Air-Craftsman 2020 Clubhouse Drive Prescott, AZ 86301	X	U.S. Certificated Mechanic	A&P 3287177
		Foreign Certificated Mechanic	
		Certificated Repair Station	
		Manufacturer	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date	Signature of Authorized Individual
12-17-2008	John Atterholt

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☑ APPROVED ☐ REJECTED

BY							
		FAA Flt. Standards Inspector		Manufacturer	X	Inspection Authorization	Other *(Specify)*
		FAA Designee		Repair Station		Person Appoved by Transport Canada Airworthiness Group	

Date of Approval or Rejection	Certificate or Designation No.	Signature of Authorized Individual
12-17-2008	A&P 3287177 IA	JOHN ATTERHOLT

FAA Form 337 (12-88)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Replaced cabin door window with Great Lakes Aero Products window, PN: W-2224 IAW STC #SA00690CH and Installation Instructions #A2D, Revision A, dated December 9, 1998.

ICA, if any, are included in the Installation Instructions and are a permanent part of the aircraft records. Weight change negligible.

********** **NOTHING FOLLOWS** **********

☐ Additional Sheets Are Attatched

FAA APPROVED MODEL LIST (AML) NO. SA00690CH
GREAT LAKES AERO PRODUCTS
INSTALLATION OF GLAP "PILOT SIDE and DOOR" WINDOWS
Original Issue Date: October 2, 1996

ITEM	AIRCRAFT MAKE	AIRCRAFT MODEL	ORIGINAL TYPE CERTIFICATE NUMBER	CERTIFICATION BASIS FOR ALTERATION	INSTALLATION INSTRUCTIONS NUMBER	INSTALLATION INSTRUCTIONS REVISION NO. & DATE	APM SUPPLEMENT NUMBER/DATE	AML AMENDMENT DATE
1	Raytheon	35, A35, B35, C35, D35, E35, F35, G35	A-777	CAR 3	INSTL. A2	Revised 10-17-96	N/A	2/18/99
2	Raytheon	E33, E33A, E33C, F33, F33A, F33C, G33, H35, J35, K35, M35, N35, P35, S35, V35, V35A, V35B, 35-33, 35-A33, 35-B33, 35-C33, 35-C33A, 36, A36TC	[illegible]15	CAR 3	INSTL. A2	Revised 10-17-96	N/A	2/18/99
3	Raytheon	D55, D55A, E55, E55A, 56TC, A56TC, 58, 58A, 95, B95, B95A, D95A, E95, 95-55, 95-A55, 95-B55, 95-B55A, 95-B55B(T-42A), 95-C55, 95-C55A	3A16	CAR 3	INSTL. A2	Revised 10-17-96	N/A	2/18/99
4	Raytheon	35, A35, B35, C35, D35, E35, F35, G35	A-777	CAR 3	INSTL. A2A, or A2B	Revised 10-5-96	N/A	2/18/99

United States of America
Department of Transportation — Federal Aviation Administration

Supplemental Type Certificate

Number SA00690CH

This certificate, issued to Great Lakes Aero Products
915 Kearsley Park Blvd.
Flint, MI 48503

certifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon meets the airworthiness requirements of Part • of the • Regulations. •

Original Product — Type Certificate Number •
Make •
Model •

•See attached FAA Approved Model List (AML) No. SA00690CH for list of approved airplane models and applicable airworthiness regulations.

Description of Type Design Change:
Installation of Great Lakes Aero Products, Pilot Side and Door Windows in accordance with Great Lakes Aero Products Drawing Numbers W-2048, Rev. F, dtd October 17, 1996, W-2049, Rev. C, dtd October 6, 1998, W-2221, Rev. B dtd. February 18, 1999, W-2222, Rev. B, dtd., February 18, 1999, W-2223, Rev. A, dtd., January 22, 1999, W-2224, Rev. B, dtd., January 26, 1999, W-2236, Rev. A, dtd., January 23, 1999, W-2237, Rev. None, dtd. October 17, 1996, W-2238, Rev. none, dtd October 17, 1996, and W-2239, Rev. A, dtd. November 18, 1998, and Great Lakes Aero Products Installation Instructions as listed on AML No. SA00690CH, or later FAA approved revisions.

Limitations and Conditions:
1. Compatibility of this design change with previously approved modifications must be determined by the installer. A copy of this Certificate and FAA Approved Model List (AML) No. SA00690CH, or later FAA approved revision must be maintained as part of the permanent records for the modified aircraft.
2. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission.

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of application October 2, 1996
Date of issuance July 25, 1997
Date reissued
Date amended February 18, 1999

FEDERAL AVIATION ADMINISTRATION

By direction of the Administrator
Mary Ellen A. Schutt (Signature)
Mary Ellen A. Schutt
Manager, Airframe & Administrative Branch
Chicago Aircraft Certification Office
(Title)

Any alteration of this certificate is punishable by a fine of not exceeding $1,000, or imprisonment not exceeding 3 years, or both.

FAA Form 8110-2 (10-68)

This certificate may be transferred in accordance with FAR 21.47.

FAA APPROVED MODEL LIST (AML) NO. SA00690CH
GREAT LAKES AERO PRODUCTS
INSTALLATION OF GLAP "PILOT SIDE and DOOR" WINDOWS

Original Issue Date: October 2, 1996

ITEM	AIRCRAFT MAKE	AIRCRAFT MODEL	ORIGINAL TYPE CERTIFICATE NUMBER	CERTIFICATION BASIS FOR ALTERATION	INSTALLATION INSTRUCTIONS		AFM SUPPLEMENT NUMBER/DATE	AML AMENDMENT DATE
					NUMBER	REVISION NO. & DATE		
5	Raytheon	E33, E33A, E33C, F33, F33A, F33C, G33, H35, J35, K35, M35, N35, P35, S35, V35, V35A, V35B, 35-33, 35-A33, 35-B33, 35-C33, 35-C33A, 36, A36	3A15	CAR 3	INSTL. A2A, or A2B	Revised 10-5-96	N/A	2/18/99
6	Raytheon	D55, D55A, E55, E55A, 56TC, A56TC, 58, 58A, 95, B95, B95A, D95A, E95, 95-55, 95-A55, 95-B55, 95-B55A, 95-C55, 95-C55A	3A16	CAR 3	INSTL. A2A, or A2B	Revised 10-5-96	N/A	2/18/99
7	Raytheon	F33A, V35B, A36, A36TC	3A15	CAR 3	INSTL. A2C	Rev. A, dtd., 11/22/98	N/A	2/18/99
8	Raytheon	E55, E555A, 58, 58A, 95-B55, 95-B55A	3A16	CAR 3	INSTL. A2C	Rev. A, dtd., 11/22/98	N/A	2/18/99
9	Raytheon	F33A, V35B, A36, A36TC	3A15	CAR 3	INSTL A2D	Rev. A, dtd., 12/9/98	N/A	2/18/99
10	Raytheon	E55, E55A, 58, 58A, 95-B55, 95-B55A	3A16	CAR 3	INSTL. A2D	Rev. A, dtd., 12/9/98	N/A	2/18/99
11	Raytheon	F33A, V35B, A36, A36TC	3A15	CAR 3	INSTL. A2E	Rev. A, dtd., 1/23/99	N/A	2/18/99

FAA APPROVED MODEL LIST (AML) NO. SA00690CH
GREAT LAKES AERO PRODUCTS
INSTALLATION OF GLAP "PILOT SIDE and DOOR" WINDOWS

Original Issue Date: October 2, 1996

ITEM	AIRCRAFT MAKE	AIRCRAFT MODEL	ORIGINAL TYPE CERTIFICATE NUMBER	CERTIFICATION BASIS FOR ALTERATION	INSTALLATION INSTRUCTIONS		AFM SUPPLEMENT NUMBER/DATE	AML AMENDMENT/ DATE
					NUMBER	REVISION NO. & DATE		
12	Raytheon	35, A35, B35, C35, D35, E35, F35, G35	A-777	CAR 3	INSTL. A2F	Rev. None, dtd., 10/17/96	N/A	2/18/99
13	Raytheon	E33, E33A, E33C, F33, F33A, F33C, G33, H35, J35, K35, M35, N35, P35, S35, V35, V35A, V35B, 35-33, 35-A33, 35-B33, 35-C33, 35-C33A, 36, A36, A36TC	3A15	CAR 3	INSTL. A2F	Rev. None, dtd., 10/17/96	N/A	2/18/99
14	Raytheon	D55, D55A, E55, E55A, 56TC, A56TC, 58, 58A, 95, B95, B95A, D95A, E95, 95-55, 95-A55, 95-B55, 95-B55A, 95-B55B(T-42A), 95-C55, 95-C55A	3A16	CAR 3	INSTL. A2F	Rev. None, dtd., 10/17/96	N/A	2/18/99
15	Raytheon	F33, F33A, F33C, G33, V35B, A36, A36TC	3A15	CAR 3	INSTL. B-1	Retyped 6/10/93	N/A	2/18/99
16	Raytheon	E55, E55A, A56TC, 58, 58A, 95-B55, 95-B55A	3A16	CAR 3	INSTL. B-1	Retyped 6/10/93	N/A	2/18/99

FAA Approved: Mary Ellen A. Schutt
Mary Ellen A. Schutt, Manager,
Airframe & Administrative Branch
Chicago Aircraft Certification Office

U.S. Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No.2120-0020

For FAA Use Only

Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed $1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make BEECH	Model F33A
	Serial No. CE952	Nationality and Registration Mark N1833D
2. Owner	Name (As shown on registration certificate) HARR JOSEPH L. HARR JOANN R.	Address (As shown on registration certificate) P.O. BOX 3370 CAMP VERDE, AZ. 86322-3370

3. For FAA Use Only

4. Unit Identification				**5. Type**	
Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	*(As described in item 1 above)*				X
POWERPLANT	TCM	IO-520-BB	274580-R		X
PROPELLER					
APPLIANCE	Type Manufacturer				

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency		C. Certificate No.
JOHN ATTERHOLT ARIZONA AIR-CRAFTSMAN, INC. dba MILE HIGH AVIONICS 2020 CLUBHOUSE DRIVE PRESCOTT, ARIZONA 86301	X	U.S. Certificated Mechanic	526045567 AP
		Foreign Certificated Mechanic	
		Certificated Repair Station	
		Manufacturer	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date	Signature of Authorized Individual
12-04-06	JOHN ATTERHOLT

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☑ APPROVED ☐ REJECTED

BY		FAA Flt. Standards Inspector		Manufacturer	X	Inspection Authorization	Other *(Specify)*
		FAA Designee		Repair Station		Person Appoved by Transport Canada Airworthiness Group	

Date of Approval or Rejection	Certificate or Designation No.	Signature of Authorized Individual
12-04-06	526045567 IA	JOHN ATTERHOLT

FAA Form 337 (12-88)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Installed single piece sloped windshield IAW STC #SA997SO and Beryl D'Shannon Aviation Specialties, Inc. Speed sloped windshield kit Installation Instructions, revision C, dated 03-01-1992.

Installed U.S. Airparts roller rocker arm assemblies, PN: US92E exhaust and US92I intake IAW STC #SE09559SC and U.S. Airparts Installation Instructions, Form US92in, dated 04-03-99.

ICA, if any, are contained in the Installation Instructions and are a permanent part of the aircraft records. Revised weight and balance.

********** **NOTHING FOLLOWS** **********

☐ Additional Sheets Are Attatched

United States Of America

Department of Transportation - Federal Aviation Administration

Supplemental Type Certificate

Number SE09559SC

This Certificate issued to

U.S. Airparts, Inc.
an Arizona LLC
31 Industry Dr.
Mt. Ida, AR 71957

certifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon meets the airworthiness requirements of Part * of the * Regulations.

Original Product Type Certificate Number: •

Make: [See attached FAA approved model list (AML) for list of approved models and applicable airworthiness regulations]

Model:

Description of Type Design Change: Installation of U.S. Airparts roller rocker arm assemblies P/N US92E exhaust and US92I intake, in accordance with U.S Airparts installation instructions, Form US92in, dated April 3, 1999, or later FAA approved revisions.

Limitations and Conditions: Compatibility of this design change with previously approved modifications must be determined by the installer. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission.

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of application: March 17, 1999

Date of issuance: September 22, 1999

Date reissued:

Date amended:

By direction of the Administrator

(Signature)

S. Frances Cox
Manager, Special Certification Office
Southwest Region

(Title)

FAA APPROVED MODEL LIST (AML)

SE09559SC

U.S. Airparts, Inc
An Arizona LLC
31 Industrial Drive
Mt. Ida, AR 71957

Date of application: March 17, 1999
Date of issuance: September 22, 1999

Item	Aircraft Make	Aircraft Model	Original Type Certificate Number	Regulation/ Part
1	Teledyne Continental Motors	IO-470-D, -E, -F, -G, -H, -L, -M, -N, -P, -R, -S, -T, -U, -V, -LO, -VO L/IO-470-A	3E1	CAR 13
2	Teledyne Continental Motors	TSIO-470-B, -C, -D	3E3	CAR 13
3	Teledyne Continental Motors	IO-520-A, -B, -BA, -BB, -C, -CB, -D, -E, -F, -J, -K, -L, -M, -MB L/IO-520-P	E5CE	CAR 13
4	Teledyne Continental Motors	TSIO-520-A, -AF, -B, -BB, -BE, -C, -CE, -D, -DB, -E, -EB, -G, -H, -J, -JB, -K, -KB, -L, -LB, -M –N, -NB, -P, -R, -T, -U, -UB, -VB, -WB	E8CE	CAR 13
5	Teledyne Continental Motors	IO-550-A, -B, -C, -D, -E, -F, -L	E3SO	CAR 13
6	Teledyne Continental Motors	TSIOL-550-A, -B, -C	E4SO	CAR 13
7	Teledyne Continental Motors	TSIO-550-A, -B, -C, -E	E5SO	CAR 13
8	Teledyne Continental Motors	O-470-B, - G, -K, -L, -M, -R, -S, -U IO-470-A, -C	E-273	CAR 13

FAA Approved: S. Gray Cox

Date: September 22, 1999

United States of America

Department of Transportation — Federal Aviation Administration

Supplemental Type Certificate

Number SA997SO

This certificate, issued to Beryl D'Shannon Aviation Specialties, Inc.

certifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon meets the airworthiness requirements of Part 3/03 *of the* Civil Air *Regulations.*

Original Product — Type Certificate Number: 3A15, A-777

Make Beech

Model: All models on 3A15 (Part 3)
All models on A-777 (Part 03)

Description of Type Design Change.

Installation of a single-piece sloped windshield in accordance with Beryl D'Shannon Aviation Specialties, Inc., Drawing B19000, Revision D, dated November 9, 1979, (or later FAA approved revision).

Limitations and Conditions This approval should not be extended to other aircraft of the same model on which other approved modifications have been made unless it has first been determined that the interrelationship between these various modifications will have no adverse affect on the airworthiness of the aircraft. This determination should include consideration of significant changes in weight distribution such as an increase in the fixed disposable weight in the fuselage.

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of application: April 24, 1976

Date reissued.

Date of issuance: June 7, 1976

Date amended: May 7, 1979, Nov. 21, 1979

By direction of the Administrator

Francis Rock

FRANCIS C. ROCK, *(Signature)* Assistant Chief,
Engineering and Manufacturing Branch
Flight Standards Division
(Title)

Any alteration of this certificate is punishable by a fine of not exceeding $1,000, or imprisonment not exceeding 3 years, or both.

FAA Form 8110-2 (10-68)

This certificate may be transferred in accordance with FAR 21.47

HCD

U.S. Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No.2120-0020

For FAA Use Only

Office Identification HCD WP07

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed $1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make Beechcraft	Model F-33A
	Serial No. CE-952	Nationality and Registration Mark USA N1833D
2. Owner	Name (As shown on registration certificate) Joseph L. Harr	Address (As shown on registration certificate) P.O. Box 3370 Camp Verde, AZ 86322

3. For FAA Use Only

4. Unit Identification / **5. Type**

Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	(As described in item 1 above)				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type Manufacturer				

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency	C. Certificate No.
Robert A. Luna P.O. 1120 Clarkdale, AZ 86324	X U.S. Certificated Mechanic Foreign Certificated Mechanic Certificated Repair Station Manufacturer	564198630A&P

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date	Signature of Authorized Individual
January 11, 2006	

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☑ APPROVED ☐ REJECTED

BY							
		FAA Flt. Standards Inspector		Manufacturer	X	Inspection Authorization	Other *(Specify)*
		FAA Designee		Repair Station		Person Appoved by Transport Canada Airworthiness Group	

Date of Approval or Rejection	Certificate or Designation No.	Signature of Authorized Individual
January 11, 2006	564198630IA	

FAA Form 337 (12-88)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

January 11, 2006 N1833D Tach 2700

Removed Hartzell PHC-C3YF-IBF, installed Hartzell PHC-C3YF-1RF/F8068 I/A/W Hartzell Drawing No. 999005, Revision A, dated March 28, 2002, Revision B dated June 24, 2003, and STC SA10135SC

No change to weight and balance.

********** **NOTHING FOLLOWS** **********

☐ Additional Sheets Are Attatched

*U.S.GPO:1994-568-012/00019

United States Of America

Department of Transportation - Federal Aviation Administration

Supplemental Type Certificate

Number SA10135SC

This Certificate issued to Hartzell Propeller, Inc.
One Propeller Place
Piqua, OH 45356

certifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon meets the airworthiness requirements of Part 3 *of the* Civil Air *Regulations.*

Original Product Type Certificate Number : 3A15

Make : Beech

Model : 35-C33A; 36; A36; E33A; E33C; F33A; F33C; S35; V35; V35A; V35B

Description of Type Design Change:
Installation of Hartzell model PHC-C3YF-1RF/F8068 propeller in accordance with Hartzell Drawing No. 999005, Revision A, dated March 28, 2002, or later FAA approved revision. FAA approved Airplane Flight Manual Supplement AFMS999005, dated April 3, 2003, or later FAA approved revision is required.

Limitations and Conditions: This installation includes the above model aircraft equipped with Teledyne Continental Motors IO-520-B, IO-520-BA, IO-520-BB, or IO-550-B model engine. Compatibility of this design change with previously approved modifications must be determined by the installer. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission.

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of application : November 12, 2002

Date of issuance : April 03, 2003

Date reissued :

Date amended :

By direction of the Administrator

(Signature)

S. Frances Cox
Manager, Special Certification Office
Southwest Region

(Title)

Any alteration of this certificate is punishable by a fine of not exceeding $1,000, or imprisonment not exceeding 3 years, or both.

 This certificate may be transferred in accordance with FAR 21.47.

HARTZELL PROPELLER INC.

TOP PROP PERFORMANCE CONVERSIONS

One Propeller Place
Piqua, Ohio 45356-2634
U.S.A.
Phone: 937-778-4200

HARTZELL

Authorization For Use of STC Document

This Supplemental Type Certificate is the property of Hartzell Propeller, Inc. Hartzell Propeller, Inc. authorizes the use of the STC data only for the aircraft specified below. Any use of this STC beyond the aircraft specified below is a violation of proprietary laws and is subject to legal action.

Please keep the STC, installation instructions, Flight Manual Supplement, and this authorization sheet, in a secure location. There will be a charge for issuing replacement documents.

STC Number:	SA10135SC
Aircraft Model:	BEECH F33A BONANZA
Serial Number:	CE-952
Aircraft Registration:	N1833D
Date Issued:	12/08/05
Issued by:	Casey Verzi
Casey Verzi	Customer Service Representative, Hartzell Top Prop Performance Conversions

Hartzell Customer Reference: CO 79197 ROCKY MOUNTAIN PROPELLER 11/5/97

U.S. Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No.2120-0020

For FAA Use Only

Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed $1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make BEECH	Model F33A
	Serial No. CE-952	Nationality and Registration Mark N1833D
2. Owner	Name (As shown on registration certificate) HARR JOSEPH L. HARR JOANN R.	Address (As shown on registration certificate) P.O. BOX 3370 CAMP VERDE, AZ. 86322-3370

3. For FAA Use Only

4. Unit Identification				**5. Type**	
Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	*(As described in item 1 above)*				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type Manufacturer				

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency		C. Certificate No.
JOHN ATTERHOLT ARIZONA AIR-CRAFTSMAN, INC. DBA MILE HIGH AVIONICS 2020 CLUBHOUSE DRIVE PRESCOTT, ARIZONA 86301	X	U.S. Certificated Mechanic	A&P 526045567
		Foreign Certificated Mechanic	
		Certificated Repair Station	
		Manufacturer	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date	Signature of Authorized Individual	
11-15-05		JOHN ATTERHOLT

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☑ APPROVED ☐ REJECTED

BY		FAA Flt. Standards Inspector		Manufacturer	X	Inspection Authorization	Other *(Specify)*
		FAA Designee		Repair Station		Person Appoved by Transport Canada Airworthiness Group	

Date of Approval or Rejection	Certificate or Designation No.	Signature of Authorized Individual	
11-15-05	IA 526045567		JOHN ATTERHOLT

FAA Form 337 (12-88)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Replaced Insight Instrument Corporation Graphic Engine Monitor GEM602 with Insight Instrument Corporation Graphic Engine Monitor Model GEM-610 IAW STC #SA01860NY and Gem Series Installation Manual, drawing 8258, version 3.0 dated May 1996.

ICA, if any, are included in the Installation Manual and are a permanent part of the aircraft records. No weight change.

********** **NOTHING FOLLOWS** **********

☐ Additional Sheets Are Attatched

United States of America

Department of Transportation -- Federal Aviation Administration

Supplemental Type Certificate

IMPORT

Number SA01860NY

This certificate issued to Insight Instrument Corporation
Box 122
Fort Erie, ON L2A 5M6
Canada

*certifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon meets the airworthiness requirements of Part * of the * Regulations.*

Original Product -- Type Certificate Number : *

Make : *

Model : *

* See attached FAA Approved Model List (AML) No. SA01860NY for list of approved airplane models and applicable airworthiness regulations

Description of Type Design Change:
Installation of Graphic Engine Monitors (G.E.M.), Models GEM-610 and GEMINI 1200, must be in accordance with GEM Series Installation Manual, Drawing 8258, Version 3.0, dated May 1996.

Limitations and Conditions :

1. Operation must be in accordance with Insight Instrument Corporation Flight Manual Supplement FMS610/1200, Revision -, Transport Canada approved April 24, 2003.
2. Maintenance must be completed in accordance with Insight Instrument Corporation Instructions for Continued Airworthiness Graphic Engine Monitor, Amendment N/C, dated December 14, 1999, Transport Canada accepted April 24, 2003.
3. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written permission of that permission.
4. Compatibility of this design change with previously approved modifications must be determined by the installer.
5. A copy of this certificate and the FAA Approved Model List (AML) No. SA01860NY, issued February 3, 2004, or later FAA approved revisions, must be maintained as part of the permanent records for the modified aircraft.

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of application : August 22, 2003

Date of issuance : February 3, 2004

Date reissued :

Date amended :

By direction of the Administrator

(Signature)

Vito A. Pulera
Manager
New York Aircraft Certification Office

(Title)

Any alteration of this certificate is punishable by a fine of not exceeding $1,000, or imprisonment not exceeding 3 years, or both.

This certificate may be transferred in accordance with FAR 21.47.

FAA APPROVED MODEL LIST (AML) NO. SA01860NY
Insight Instrument Corporation
For
Installation of Graphic Engine Monitors

Issue Date: February 3, 2004

ITEM	AIRCRAFT MAKE	AIRCRAFT MODEL	ORIGINAL TYPE CERTIFICATION NUMBER	CERTIFICATION BASIS FOR ALTERATION	AML AMENDMENT DATE
1	Beechcraft	23- all, 24-all	A1CE	CAR 3	NA
2	Beechcraft	33-all, 36-all	3A15	CAR 3	NA
3	Beechcraft	35-all	A-777/3A15	CAR 3	NA
4	Beechcraft	45-all	5A3	CAR 3	NA
5	Bellanca	17-30/31-all	1A3/A18CE	CAR 3	NA
6	Bellanca	14-19-all	1A3	FAR 23	NA
7	Bellanca	14-13-all	A-773	CAR 4a	NA
8	Cessna	150, 152-all	3A19	CAR 3	NA
9	Cessna	170-all	A-799	CAR 3	NA
10	Cessna	172-all	3A12	CAR 3	NA
11	Cessna	175-all	3A17	CAR 3	NA
12	Cessna	177-all	A13CE	FAR 23	NA
13	Cessna	177RG	A20CE	FAR 23	NA
14	Cessna	180-all	5A6	CAR 3	NA
15	Cessna	182-all	3A13	CAR 3	NA
16	Cessna	185-all	3A24	CAR 3	NA
17	Cessna	205-all, 210-all	3A21	CAR 3	NA
18	Cessna	206-all	A4CE	CAR 3	NA
19	Cessna	207-all	A16CE	FAR 23	NA
20	Composite Aircraft Corp (Windecker)	AC-7	A7SW	FAR23	NA
21	Consolidated Aeronautics (Lake)	LA-4-all	1A13	CAR 3	NA
22	Found Aircraft Development Inc.	FBA-2C, FBA-2C1	A7EA	CAR 3	NA
23	Globe Swift	GC-1A/1B	A-766	CAR 4a	NA
24	Gulfstream Aerospace	AA-5B	A16EA	FAR 23	NA
25	Helio Enterprises Inc.	H-295	1A8	CAR 3	NA

U.S. Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No.2120-0020

For FAA Use Only

Office Identification
WP07

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed $1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make BEECH	Model F-33A
	Serial No. CE 952	Nationality and Registration Mark N1833D
2. Owner	Name (As shown on registration certificate) HARR JOSEPH L	Address (As shown on registration certificate) PO BOX 3370 CAMP VERDE AZ. 86322-3370

3. For FAA Use Only

4. Unit Identification				**5. Type**	
Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	*(As described in item 1 above)*				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type Manufacturer				

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency	C. Certificate No.
ARIZONA AIR-CRAFTSMAN 2020 CLUBHOUSE DR. PRESCOTT, AZ. 86301	U.S. Certificated Mechanic Foreign Certificated Mechanic X Certificated Repair Station Manufacturer	ZM3R029M AIRFRAME CLASS I AIRFRAME CLASS III

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date	Signature of Authorized Individual
FEB 10,2004	RONALD P WEAVER

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☑ APPROVED ☐ REJECTED

BY							Other *(Specify)*
		FAA Flt. Standards Inspector		Manufacturer		Inspection Authorization	
		FAA Designee	X	Repair Station		Person Appoved by Transport Canada Airworthiness Group	

Date of Approval or Rejection	Certificate or Designation No.	Signature of Authorized Individual
FEB 10,2004	ZM3R029M	RONALD P WEAVER

FAA Form 337 (12-88)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

INSTALLED HORIZON INSTRUMENTS, INC. MODEL P-1000 PER STC SA5491NM AND MANUFACTURES INSTALLATION INSTRUCTION MANUAL P103050, REV E. INSTRUCTIONS FOR CONTINUED AIRWORTHINESS, IF ANY, ARE CONTAINED IN THE ABOVE MENTIONED INSTALLATION MANUAL.

********** **NOTHING FOLLOWS** **********

☐ Additional Sheets Are Attatched

*U.S.GPO:1994-568-012/00019

U.S. Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No.2120-0020

For FAA Use Only

Office Identification
WP07

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed $1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make BEECH	Model F33A
	Serial No. CE-952	Nationality and Registration Mark N1833D
2. Owner	Name (As shown on registration certificate) HARR JOSEPH	Address (As shown on registration certificate) PO BOX 3370 CAMP VERDE AZ. 86322-3370

3. For FAA Use Only

4. Unit Identification				**5. Type**	
Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	*(As described in item 1 above)*				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type Manufacturer				

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency	C. Certificate No.
ARIZONA AIR-CRAFTSMAN 2020 CLUBHOUSE DR PRESCOTT, AZ. 86301	U.S. Certificated Mechanic Foreign Certificated Mechanic X Certificated Repair Station Manufacturer	ZM3R029M AIRFRAME CLASS I AIRFRAME CLASS III

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date	Signature of Authorized Individual	
JAN 7, 2004	Donald P Weaver	RONALD P WEAVER

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☑ APPROVED ☐ REJECTED

BY		FAA Flt. Standards Inspector		Manufacturer		Inspection Authorization	Other *(Specify)*
		FAA Designee	X	Repair Station		Person Appoved by Transport Canada Airworthiness Group	

Date of Approval or Rejection	Certificate or Designation No.	Signature of Authorized Individual	
JAN 7, 2004	ZM3R029M	Donald P Weaver	RONALD P WEAVER

FAA Form 337 (12-88)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

INSTALLED AEROTECH COMPONENTS, INC. CV1J4 (CLEAR-VIEW) INLINE PHEUMATIC AIR FILTER PER STC SA289CH AND MANUFACTURES INSTALLATION INSTRUCTIONS REV-3-2/13/95.
CONTINUED AIRWORTHIENESS IS TO CHANGE FILTER ON AN "AS NEEDED BASIS"

********** **NOTHING FOLLOWS** **********

☐ Additional Sheets Are Attatched

*U.S.GPO:1994-568-012/00019

US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020

For FAA Use Only

Office Identification
WP-21

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This form is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed $1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make: **Beechcraft**	Model: **F-33-A**
	Serial No.: **CE-952**	Nationality and Registration Mark: **N1833D**
2. Owner	Name *(As shown on registration certificate)*: **Joe Harr**	Address *(As shown on registration certificate)*: **2125 N. Mooney Lane, Camp Verde, AZ 86322**

3. For FAA Use Only

4. Unit Identification / **5. Type**

Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	*(As described in item 1 above)*				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type / Manufacturer				

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency	C. Certificate No.
G&D Aero Products Inc. **6951 Flight Road** **Riverside, CA 92504** **Hanger #7**	U.S. Certificated Mechanic Foreign Certificated Mechanic X Certificated Repair Station Manufacturer	**LIMITED** **G7DR870J**

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date	Signature of Authorized Individual
10-30-2002	Gale R. Calhoon

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is [X] APPROVED [ ] REJECTED

BY						
		FAA Flt. Standards Inspector		Manufacturer		Inspection Authorization	Other (Specify)
		FAA Designee	X	Repair Station		Person Approved by Transport Canadian Airworthiness Group	

Date of Approval or Rejection	Certificate or Designation No.	Signature of Authorized Individual
10/30/02	**G7DR870J**	Gale R. Calhoon

FAA Form 337 (12-88)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

On October 30, 2002, Installed G&D Aero Product Inc. Thermal Pane Window inserts part #GD-F-33A according to STC# SA3759NM. Weight and balance, and equipment list revised.

----------END----------

☐ Additional Sheets Are Attached

U.S. Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No.2120-0020

For FAA Use Only

Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed $1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make BEECH	Model F33A
	Serial No. CE-952	Nationality and Registration Mark N1833D
2. Owner	Name (As shown on registration certificate) HARR JOSEPH L HARR JOANN R	Address (As shown on registration certificate) P.O. BOX 3370 CAMP VERDE, AZ. 86322-3370

3. For FAA Use Only

4. Unit Identification / **5. Type**

Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	*(As described in item 1 above)*				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type Manufacturer				

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency	C. Certificate No.
ARIZONA AIR-CRAFTSMAN, INC. DBA MILE HIGH AVIONICS 2020 CLUBHOUSE DRIVE PRESCOTT, ARIZONA 86305	U.S. Certificated Mechanic Foreign Certificated Mechanic X Certificated Repair Station Manufacturer	CRS# ZM3R029M

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date	Signature of Authorized Individual
11-19-01	JOHN ATTERHOLT

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☑ APPROVED ☐ REJECTED

BY							Other *(Specify)*
		FAA Flt. Standards Inspector		Manufacturer		Inspection Authorization	
		FAA Designee	X	Repair Station		Person Appoved by Transport Canada Airworthiness Group	

Date of Approval or Rejection	Certificate or Designation No.	Signature of Authorized Individual
11-19-01	ZM3R029M	JOHN ATTERHOLT

FAA Form 337 (12-88)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Installed Eagle Technologies Co. Wing Fuel Tank quick Drain Valves, Kit PN BS-2, I.A.W. STC #SA0111OCH and Eagle Technologies Co., Installation Instructions for Quick Drain Valve Assy. QDV-3, 730B, Rev. G. No weight change.

********** **NOTHING FOLLOWS** **********

☐ Additional Sheets Are Attatched

*U.S.GPO:1994-568-012/00019

U.S. Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No.2120-0020

For FAA Use Only

Office Identification WPQ7 RRC

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C.1421). Failure to report can result in a civil penalty not to exceed $1,000 for each such violation (Section 901 Federal Aviation Act 1958)

1. Aircraft	Make: BEECHCRAFT	Model: F33A
	Serial No.: CE952	Nationality and Registration Mark: N1833D
2. Owner	Name *(As shown on registration certificate)*: HARR, JOSEPH L. HARR, JOANN R.	Address *(As shown on registration certificate)*: BOX 3370 MONTEZUMA HEIGHTS APT 20B, CAMP VERDE, AZ. 86322

3. For FAA Use Only

4. Unit Identification / **5. Type**

Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	*(As described in item 1 above)*				
POWERPLANT	TELEDYNE CONT.	IO-520-BB	274580R		X
PROPELLER					
APPLIANCE	Type Manufacturer				

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency	C. Certificate No.
AIRZONA AIR-CRAFTSMAN, INC. 2020 CLUBHOUSE DRIVE PRESCOTT, AZ. 86305	U.S. Certificated Mechanic Foreign Certificated Mechanic X Certificated Repair Station Manufacturer	ZM3R029M Radio Class I & II

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date	Signature of Authorized Individual
11-04-98	JOHN ATTERHOLT John Atterholt

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is [X] APPROVED [ ] REJECTED

BY							Other *(Specify)*
		FAA Flt. Standards Inspector		Manufacturer		Inspection Authorization	
		FAA Designee	X	Repair Station		Person Approved by Transport Canada Airworthiness Group	

Date of Approval or Rejection	Certificate or Designation No.	Signature of Authorized Individual
11-04-98	ZM3R029M	JOHN ATTERHOLT John Atterholt

FAA Form 337 (12-88)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished
(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

INSPECTED PREVIOUSLY INSTALLED GAMIJECTORS KIT NO. GAD13AA, INSTALLATION CONFORMS TO S.T.C. #SE09217SC AND GAMIJECTOR INSTALLATION PROCEDURE # IP-96-001, REV. 004, DATED SEPTEMBER 25, 1996. NO CHANGE IN WEIGHT & BALANCE.

**********End**********

☐ Additional Sheets Are Attached

U.S.GPO:1989-0-663-171

FAA AIRCRAFT REGISTRY
CAMERA NO. 3N DATE: 7-29-94

FAA COPY

FEDERAL AVIATION AGENCY

MAJOR REPAIR AND ALTERATION

(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
Budget Bureau No. 04-R060.1

FOR FAA USE ONLY

OFFICE IDENTIFICATION
SDL-ESOD

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.

1. AIRCRAFT	MAKE: BEECHCRAFT	MODEL: F33A
	SERIAL NO.: CE952	NATIONALITY AND REGISTRATION MARK: U.S.A. N1833D
2. OWNER	NAME (As shown on registration certificate): JOSEPH L. HARR & JOANN R. HARR	ADDRESS (As shown on registration certificate): P.O. BOX 3370 CAMP VERDE, ARIZONA 86322

3. FOR FAA USE ONLY

4. UNIT IDENTIFICATION				5. TYPE	
UNIT	MAKE	MODEL	SERIAL NO.	REPAIR	ALTERATION
AIRFRAME	(As described in item 1 above)				XX
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE / MANUFACTURER				

6. CONFORMITY STATEMENT

A. AGENCY'S NAME AND ADDRESS	B. KIND OF AGENCY	C. CERTIFICATE NO.
Arizona Air-Craftsman 290 Shrine Rd, #5 Sedona, AZ 86336	U.S. CERTIFICATED MECHANIC FOREIGN CERTIFICATED MECHANIC X CERTIFICATED REPAIR STATION MANUFACTURER	ZM3R029M Powerplant-Class 1 Airframe-Limited

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

DATE	SIGNATURE OF AUTHORIZED INDIVIDUAL
5/16/94	TONY R. RILEY

7. APPROVAL FOR RETURN TO SERVICE

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Agency and is ☒ APPROVED ☐ REJECTED

BY							OTHER (Specify)
		FAA FLT. STANDARDS INSPECTOR		MANUFACTURER		INSPECTION AUTHORIZATION	
		FAA DESIGNEE	XX	REPAIR STATION		CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT	

DATE OF APPROVAL OR REJECTION	CERTIFICATE OR DESIGNATION NO.	SIGNATURE OF AUTHORIZED INDIVIDUAL
5/16/94	ZM3R029M	TONY R. RILEY

FAA Form 337 (1-65) OBSOLETE PREVIOUS EDITION 0052-025-8000 (8320)

FAA AIRCRAFT REGISTRY
CAMERA NO. 3N DATE: 7-29-94

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED *(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)*

AIRCRAFT NUMBER N1833D SERIAL NUMBER CE952

Verified previous installation of Shadin Company Inc. Digital fuel management system, P/N-90281 in accordance with STC-SA449GL and installation drawings dated 5-20-94 code # 4017-26.

☐ ADDITIONAL SHEETS ARE ATTACHED

U.S. GOVERNMENT PRINTING OFFICE : 1985 OP—781-948

FAA AIRCRAFT REGISTRY
CAMERA NO. 3N DATE: 7-29-94

FEDERAL AVIATION AGENCY

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
Budget Bureau No. 04-R060.1

FOR FAA USE ONLY

OFFICE IDENTIFICATION
WA-FSDO-07

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.

1. AIRCRAFT	MAKE: Beechcraft	MODEL: F33A
	SERIAL NO.: CE952	NATIONALITY AND REGISTRATION MARK: N1833D
2. OWNER	NAME (As shown on registration certificate): Harr, Joseph L. Harr, Joann R.	ADDRESS (As shown on registration certificate): Box 3370 Montezuma Heights Apt 20B, Camp Verde, AZ 86322

3. FOR FAA USE ONLY

4. UNIT IDENTIFICATION

UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE: REPAIR	5. TYPE: ALTERATION
AIRFRAME	(As described in item 1 above)				X
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE MANUFACTURER				

6. CONFORMITY STATEMENT

A. AGENCY'S NAME AND ADDRESS	B. KIND OF AGENCY	C. CERTIFICATE NO.
Jay A. Kerger dba Arizona Air-Craftsman 290 Shrine Rd, #5 Sedona, AZ 86336	U.S. CERTIFICATED MECHANIC FOREIGN CERTIFICATED MECHANIC X CERTIFICATED REPAIR STATION MANUFACTURER	ZM3R029M Powerplant-Class 1 Airframe-Limited

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

DATE	SIGNATURE OF AUTHORIZED INDIVIDUAL
4-29-94	Dennis E. O'Leary Dennis E. O'Leary

7. APPROVAL FOR RETURN TO SERVICE

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Agency and is ☒ APPROVED ☐ REJECTED

BY						
	FAA FLT. STANDARDS INSPECTOR		MANUFACTURER		INSPECTION AUTHORIZATION	OTHER (Specify)
	FAA DESIGNEE	XX	REPAIR STATION		CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT	

DATE OF APPROVAL OR REJECTION	CERTIFICATE OR DESIGNATION NO.	SIGNATURE OF AUTHORIZED INDIVIDUAL
4-29-94	ZM3R029M	Dennis E. O'Leary Dennis E. O'Leary

FAA Form 337 (1-65) OBSOLETE PREVIOUS EDITION

0052-025-8000 (8320)

FAA AIRCRAFT REGISTRY
CAMERA NO. 3N DATE: 7-29-94

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED *(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)*

Install Space Machine products STC SA5870NM, middle rudder hinge bracket IAW drawing SMP-11 dated 12/1/92 and SMP Installation Instructions dated 1/21/94. Weight change negligible.

N1833D
SN: CE952

☐ ADDITIONAL SHEETS ARE ATTACHED

U.S. GOVERNMENT PRINTING OFFICE : 1985 OF—761-748

FAA AIRCRAFT REGISTRY
CAMERA NO. 2N DATE: 12.14.93

FEDERAL AVIATION AGENCY

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
Budget Bureau No. 04-R060.1

FOR FAA USE ONLY

OFFICE IDENTIFICATION
SOL-E300

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.

1. AIRCRAFT	MAKE: BEECHCRAFT	MODEL: F-33A
	SERIAL NO.: CE-952	NATIONALITY AND REGISTRATION MARK: N1833d
2. OWNER	NAME (As shown on registration certificate): JOESPH HARR	ADDRESS (As shown on registration certificate): P.O. BOX 3370 CAMP VERDE, ARIZ. 86322

3. FOR FAA USE ONLY

4. UNIT IDENTIFICATION				5. TYPE	
UNIT	MAKE	MODEL	SERIAL NO.	REPAIR	ALTERATION
AIRFRAME	(As described in item 1 above)				XX
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE / MANUFACTURER				

6. CONFORMITY STATEMENT

A. AGENCY'S NAME AND ADDRESS	B. KIND OF AGENCY	C. CERTIFICATE NO.
Jay A. Kerger dba Arizona Air-Craftsman 290 Shrine Rd, #5 Sedona, AZ 86336	U.S. CERTIFICATED MECHANIC FOREIGN CERTIFICATED MECHANIC X CERTIFICATED REPAIR STATION MANUFACTURER	ZM3R029M Powerplant-Class 1 Airframe-Limited

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

DATE	SIGNATURE OF AUTHORIZED INDIVIDUAL
10/13/93	TONY R. RILEY

7. APPROVAL FOR RETURN TO SERVICE

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Agency and is ☒ APPROVED ☐ REJECTED

BY						
	FAA FLT. STANDARDS INSPECTOR		MANUFACTURER		INSPECTION AUTHORIZATION	OTHER (Specify)
	FAA DESIGNEE	XX	REPAIR STATION		CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT	

DATE OF APPROVAL OR REJECTION	CERTIFICATE OR DESIGNATION NO.	SIGNATURE OF AUTHORIZED INDIVIDUAL
10/13/93	ZM3R029M	TONY R. RILEY

FAA Form 337 (1-65) OBSOLETE PREVIOUS EDITION 0052-025-8000 (8320)

FAA AIRCRAFT REGISTRY
CAMERA NO. 2N DATE: 12.14.93

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED *(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)*

INSTALLED INSIGHT INSTRUMENT CORP. GRAPHIC ENGINE MONITOR SYSTEM
I.A.W. INSIGHT INSTRUCTION INSTALLATION MANUAL AND DRAWINGS.
FOR MODEL 602 LIST 8200, 5/0493 AND INSTALLATION DATED SEPT. 1, 1982.
REMOVED OPTIONAL SPECIAL EQUIPMENT ALCOR SINGLE POINT EGT AND INSTALLED-
G.E.M. IN PLACE AT INSTRUMENT PANEL. EQUIPMENT LIST REVISED.
UNIT P/N-1261, SOFTWARE LEVEL-602D, SERIAL NO.-007342, MANUF.DATE-9/16/93.
FLIGHT SUPPLEMENT PLACED IN AIRCRAFT FLIGHT MANUAL.

☐ ADDITIONAL SHEETS ARE ATTACHED

U.S. GOVERNMENT PRINTING OFFICE : 1985 OF—761-748

FAA AIRCRAFT REGISTRY
CAMERA NO. 3 DATE: 7-22-91

US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

RECEIVED MAY 22 1991

Form Approved
OMB No. 2120-0020

For FAA Use Only
Office Identification WP-05 R

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed $1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make: Beechcraft	Model: F 33A
	Serial No.: CE-952	Nationality and Registration Mark: N1833D
2. Owner	Name (As shown on registration certificate): Harr Joseph L. Harr Joann R.	Address (As shown on registration certificate): Box 3370 Monteluma Heights Apt 20B Camp Verdie, AZ 86322

3. For FAA Use Only

4. Unit Identification / 5. Type

Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	(As described in Item 1 above)				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency	C. Certificate No.
Flite-Comm Electronics, Inc. 3271 Airflite Way Long Beach, CA 90807	U.S. Certificated Mechanic Foreign Certificated Mechanic X Certificated Repair Station Manufacturer	Radio Class I,II, III(XU3R984L)Limited Instrument, Limited Airframe, Specialize Services

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date	Signature of Authorized Individual
5/16/91	Todd H. Foster

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is [X] APPROVED [ ] REJECTED

BY						
	FAA Flt. Standards Inspector		Manufacturer		Inspection Authorization	Other (Specify)
	FAA Designee	X	Repair Station		Person Approved by Transport Canada Airworthiness Group	

Date of Approval or Rejection	Certificate or Designation No.	Signature of Authorized Individual
5/16/91	XU3R984L	Todd H. Foster

FAA Form 337 (12-88)

FAA AIRCRAFT REGISTRY
CAMERA NO. 3 DATE: 7-22-91

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished
(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Installed KLN-88, Bent Whip KA-84, FN-100B Cooling, RSU-21, and ATS-80.

Installation made in accordance with AC 43.13-IA, Chapters 11 and 15, AC 43.13-2A, Chapters 2 and 3, AC 20-121A, Paragraph 3, Appendix I and manufacturer's instruction manual.

Loran left/right, to/from and nav flag information are switched to pilot's HSI/Flight Director and Autopilot and copilot's repeater indicator. Also installed ILS energize interlock. Loran switching is accomplished through an annunciator switch.

Instrument panel placarded as follows: "Approved for VFR only".

Ground and flight check made and Loran exhibits no interference to other aircraft communications or navigations systems.

Electrical load does not exceed 80% of charging system.

EMI check made and no interference noted.
Weight and balance revised.

LAST ITEM

☐ Additional Sheets Are Attached

FAA AIRCRAFT REGISTRY
CAMERA NO. /N DATE: 5-15-91

U.S. DEPARTMENT OF TRANSPORTATION
FEDERAL AVIATION ADMINISTRATION

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
Budget Bureau No. 04-R060.1

FOR FAA USE ONLY

OFFICE IDENTIFICATION WP8200-7

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.

1. AIRCRAFT	MAKE: Beechcraft	MODEL: F33A
	SERIAL NO.: CE952	NATIONALITY AND REGISTRATION MARK: N1833D
2. OWNER	NAME (As shown on registration certificate): Harr Joseph L Harr Joann R	ADDRESS (As shown on registration certificate): Box3370 Montezuma Heights Apt 20B, Camp Verde, AZ 86322

3. FOR FAA USE ONLY

4. UNIT IDENTIFICATION / 5. TYPE

UNIT	MAKE	MODEL	SERIAL NO.	REPAIR	ALTERATION
AIRFRAME	(As described in item 1 above)				X
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE MANUFACTURER				

6. CONFORMITY STATEMENT

A. AGENCY'S NAME AND ADDRESS	B. KIND OF AGENCY	C. CERTIFICATE NO.
Arizona Air-Craftsman 290 Shrine Rd. #5 Sedona, AZ 86336	U.S. CERTIFICATED MECHANIC FOREIGN CERTIFICATED MECHANIC XX CERTIFICATED REPAIR STATION MANUFACTURER	ZM3R029M

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

DATE	SIGNATURE OF AUTHORIZED INDIVIDUAL
03-11-91	Dennis E O'Leary

7. APPROVAL FOR RETURN TO SERVICE

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ APPROVED ☐ REJECTED

BY							OTHER (Specify)
		FAA FLT. STANDARDS INSPECTOR		MANUFACTURER		INSPECTION AUTHORIZATION	
		FAA DESIGNEE	X	REPAIR STATION		CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT	

DATE OF APPROVAL OR REJECTION	CERTIFICATE OR DESIGNATION NO.	SIGNATURE OF AUTHORIZED INDIVIDUAL
03-11-91	ZM3R029M	Dennis E O'Leary

FAA Form 337 (7-67) (8320)

FAA AIRCRAFT REGISTRY
CAMERA NO. 1N DATE: 5-15-91

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED *(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)*

Install Upper Rudder Hinge Brackett in accordance with STC SA4899NM, dated 08-28-90 and Spacecraft Machine Products Master Drawing List SMP-100, Rev. A, dated 07-17-90. Weight added is negligible.
****

☐ ADDITIONAL SHEETS ARE ATTACHED

☆U.S. GOVERNMENT PRINTING OFFICE: 1986-659-121/40260

FAA AIRCRAFT REGISTRY
CAERA NO. 2N DATE: 9-27-89

U.S. DEPARTMENT OF TRANSPORTATION
FEDERAL AVIATION ADMINISTRATION

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
Budget Bureau No. 04-R060.1

FOR FAA USE ONLY

OFFICE IDENTIFICATION
ACE-61

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.

1. AIRCRAFT	MAKE: Beechcraft	MODEL: F33A
	SERIAL NO.: CE-952	NATIONALITY AND REGISTRATION MARK: N1833D
2. OWNER	NAME (As shown on registration certificate): Joseph L. Harr	ADDRESS (As shown on registration certificate): 10192 Cary Circle, Cypress, Ca 90630

3. FOR FAA USE ONLY

RECEIVED
JUN 17 1988
ACE-FSDO-61
DES MOINES, IOWA

4. UNIT IDENTIFICATION

UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE: REPAIR	5. TYPE: ALTERATION
AIRFRAME	(As described in item 1 above)				X
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE / MANUFACTURER				

6. CONFORMITY STATEMENT

A. AGENCY'S NAME AND ADDRESS	B. KIND OF AGENCY	C. CERTIFICATE NO.
GENE SIMPSON AIRCRAFT SALES, INC. P.O. Box 13, Municipal Airport Onawa, Iowa 51040	U.S. CERTIFICATED MECHANIC FOREIGN CERTIFICATED MECHANIC X CERTIFICATED REPAIR STATION MANUFACTURER	CRS 361-3 Limited Airframe Beechcraft

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

DATE	SIGNATURE OF AUTHORIZED INDIVIDUAL
6-15-88	R. Rudnick

7. APPROVAL FOR RETURN TO SERVICE

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is [X] APPROVED [ ] REJECTED

BY							
		FAA FLT. STANDARDS INSPECTOR		MANUFACTURER		INSPECTION AUTHORIZATION	OTHER (Specify)
		FAA DESIGNEE	X	REPAIR STATION		CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT	

DATE OF APPROVAL OR REJECTION	CERTIFICATE OR DESIGNATION NO.	SIGNATURE OF AUTHORIZED INDIVIDUAL
6-15-88	CRS 361-3	R. Rudnick

FAA Form 337 (7-67)

(8320)

FAA AIRCRAFT REGISTRY
CAMERA NO. 2N DATE: 9-29-89

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED *(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)*

Installed Beryl D'Shannon Aviation Specialties, Inc., Fiberglas Tip Tanks, and fuel transfer system in accordance with STC SA153EA, Drawing No. B1404, Rev. E, dated July 25, 1977. See New Weight and Balance This Date.

RECEIVED

☐ ADDITIONAL SHEETS ARE ATTACHED

U S GPO 1981 775-332 47

FAA AIRCRAFT REGISTRY
CAMERA NO. 2N DATE: 9-29-89

RECEIVED
MAR 23 1988
LGB FSDO 5

DEPARTMENT OF TRANSPORTATION
FEDERAL AVIATION ADMINISTRATION

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
Budget Bureau No. 04-R060.1

FOR FAA USE ONLY
OFFICE IDENTIFICATION
WP25

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.

1. AIRCRAFT	MAKE: BEECH	MODEL: F33A
	SERIAL NO.: CE-952	NATIONALITY AND REGISTRATION MARK: N-1833D
2. OWNER	NAME (As shown on registration certificate): JOSEPH L. HARR	ADDRESS (As shown on registration certificate): 10192 CARY CIRCLE CYPRESS, CA. 90630

3. FOR FAA USE ONLY

4. UNIT IDENTIFICATION

UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE REPAIR	5. TYPE ALTERATION
AIRFRAME	(As described in item 1 above)				X
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				

6. CONFORMITY STATEMENT

A. AGENCY'S NAME AND ADDRESS	B. KIND OF AGENCY	C. CERTIFICATE NO.
WEST END AVIATION 2837 E. SPRING ST. LONG BEACH, CA. 90806	X U.S. CERTIFICATED MECHANIC FOREIGN CERTIFICATED MECHANIC CERTIFICATED REPAIR STATION MANUFACTURER	DONALD P. TOWNSEND A&P 550989302

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

DATE: 3-16-88

SIGNATURE OF AUTHORIZED INDIVIDUAL: Donald P. Townsend
DONALD P. TOWNSEND

7. APPROVAL FOR RETURN TO SERVICE

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is [X] APPROVED [ ] REJECTED

BY				
FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	X	INSPECTION AUTHORIZATION	OTHER (Specify)
FAA DESIGNEE	REPAIR STATION		CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT	

DATE OF APPROVAL OR REJECTION	CERTIFICATE OR DESIGNATION NO.	SIGNATURE OF AUTHORIZED INDIVIDUAL
3-16-88	618006	R.T. Cook R.T. COOK

FAA Form 337 (7-67)

(8320)

FAA AIRCRAFT REGISTRY
CAMERA NO. N DATE: 9-27-89

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED *(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)*

3-16-88 N-1833D

INSTALLED AILERON AND WING FLAP SEALS AS PER STC SA1176SO. WORK ACCOMPLISHED IN ACCORDANCE WITH D'SHANNON PRODUCTS DRAWING B-6300 REV A. DATED OCTOBER 8, 1980, INSTALLATION INSTRUCTIONS, AND FAR-AC-43-13-1A. WEIGHT CHANGE NEGLIGIBLE.

NOTHING FOLLOWS

☐ ADDITIONAL SHEETS ARE ATTACHED

☆U.S. GOVERNMENT PRINTING OFFICE: 1977—771-021/34

FAA AIRCRAFT REGISTRY
CAMERA NO. 2 DATE: 7-20-87

DEPARTMENT OF TRANSPORTATION
FEDERAL AVIATION ADMINISTRATION

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
Budget Bureau No. 04-R060.1

FOR FAA USE ONLY
OFFICE IDENTIFICATION
WP-FSDO-9

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.

1. AIRCRAFT	MAKE: BEECHCRAFT	MODEL: F33A
	SERIAL NO.: CE-952	NATIONALITY AND REGISTRATION MARK: USA N1833D
2. OWNER	NAME (As shown on registration certificate): KARAHADIAN RANCHES	ADDRESS (As shown on registration certificate): P.O. BOX 756 THERMAL, CALIF 92274

3. FOR FAA USE ONLY

4. UNIT IDENTIFICATION / 5. TYPE

UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE: REPAIR	5. TYPE: ALTERATION
AIRFRAME	(As described in item 1 above)				X
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE / MANUFACTURER				

6. CONFORMITY STATEMENT

A. AGENCY'S NAME AND ADDRESS	B. KIND OF AGENCY	C. CERTIFICATE NO.
AVIONICS AT PALOMAR 2188 B PALOMAR AIRPORT RD CARLSBAD, CALIF 92008	U.S. CERTIFICATED MECHANIC FOREIGN CERTIFICATED MECHANIC X CERTIFICATED REPAIR STATION MANUFACTURER	403-37 RADIO 1,2,3 LIMITED AIRFRAME

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

DATE: 8/20/85
SIGNATURE OF AUTHORIZED INDIVIDUAL: Bill [illegible]

7. APPROVAL FOR RETURN TO SERVICE

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is [X] APPROVED [ ] REJECTED

BY						
	FAA FLT. STANDARDS INSPECTOR		MANUFACTURER		INSPECTION AUTHORIZATION	OTHER (Specify)
	FAA DESIGNEE	X	REPAIR STATION		CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT	

DATE OF APPROVAL OR REJECTION	CERTIFICATE OR DESIGNATION NO.	SIGNATURE OF AUTHORIZED INDIVIDUAL
8/20/85	403-37	Bill [illegible]

FAA Form 337 (7-67) ☆ U.S. Government Printing Office 1977—772-646/141 (8320)

FAA AIRCRAFT REGISTRY
CAMERA NO. 8 DATE: 7-20-87

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED *(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)*

1. Installed 3 M Ryan Stormscope model WX10A in accordance with manufacturers installation instructions.

2. Installed Stormscope Antenna on belly of aircraft at STA. 129 using Supplied doubler plate.

3. Stormscope processor and display unit installed in space provided by aircraft manufacturer.

4. An electrical load check was performed and it was found that the maximum electrical load does not exceed 80% of the generating capacity of the electrical charging system.

5. All materials used and all the above work performed was in accordance with acceptable methods and techniques set forth in AC 43 .13-1A and 43.13-2A.

6. Revised aircraft weight and balance records.

----------END----------

☐ ADDITIONAL SHEETS ARE ATTACHED

AUG 2 1985

FAA AIRCRAFT REGISTRY
CAMERA NO. 3 DATE: 8-16-85

5

DEPARTMENT OF TRANSPORTATION
FEDERAL AVIATION ADMINISTRATION

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
Budget Bureau No. 04-R060.1

FOR FAA USE ONLY

OFFICE IDENTIFICATION

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.

1. AIRCRAFT	MAKE: BEECHCRAFT	MODEL: E33A
	SERIAL NO.: CE-952	NATIONALITY AND REGISTRATION MARK: USA N1833D
2. OWNER	NAME (As shown on registration certificate): KARAHADIAN RANCHES INC.	ADDRESS (As shown on registration certificate): P.O. Box 756 THERMAL, CALIF. 92274

3. FOR FAA USE ONLY

4. UNIT IDENTIFICATION / 5. TYPE

UNIT	MAKE	MODEL	SERIAL NO.	REPAIR	ALTERATION
AIRFRAME	(As described in item 1 above)				X
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE / MANUFACTURER				

6. CONFORMITY STATEMENT

A. AGENCY'S NAME AND ADDRESS	B. KIND OF AGENCY	C. CERTIFICATE NO.
AVIONICS AT PALOMAR 2188B PALOMAR AIRPORT RD CARLSBAD, CALIF 92008	U.S. CERTIFICATED MECHANIC FOREIGN CERTIFICATED MECHANIC X CERTIFICATED REPAIR STATION MANUFACTURER	403-37 RADIO 1,2,3, LIMITED AIRFRAME

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

DATE	SIGNATURE OF AUTHORIZED INDIVIDUAL
3/23/84	Bill Fry

7. APPROVAL FOR RETURN TO SERVICE

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is X APPROVED ☐ REJECTED

BY							
		FAA FLT. STANDARDS INSPECTOR		MANUFACTURER		INSPECTION AUTHORIZATION	OTHER (Specify)
		FAA DESIGNEE	X	REPAIR STATION		CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT	

DATE OF APPROVAL OR REJECTION	CERTIFICATE OR DESIGNATION NO.	SIGNATURE OF AUTHORIZED INDIVIDUAL
3/23/84	403-37	Bill Fry

FAA Form 337 (7-67)

(8320)

FAA AIRCRAFT REGISTRY
CAMERA NO. 3 DATE: 8-16-85

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED *(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)*

1. Relocated No. 2 nav indicator to co-pilots instrument panel.

2. Installed aviation instrument manufacturing corporation 200DCL directional gyro and 500 DCFL attitude gyro in pilots instrument subpanel.

3. An electrical load check was performed and it was found that the maximum continious electrical load does not exceed 80% of the generating capacity of the electrical charging system.

4. All materials used and all the above work accomplished was in accordance with acceptable methods and techniques set forth in AC 43.13-1A and 43.13-2A.

5. Revised aircraft weight and balance records.

----------END----------

☐ ADDITIONAL SHEETS ARE ATTACHED

☆U.S. GOVERNMENT PRINTING OFFICE: 1977—771-021/344

FAA AIRCRAFT REGISTRY
CAMERA NO. 3 DATE: 8-16-85

⑧ 4/25/81

Form Approved
Budget Bureau No. 04-R0058

DEPARTMENT OF TRANSPORTATION
FEDERAL AVIATION ADMINISTRATION

APPLICATION FOR AIRWORTHINESS CERTIFICATE

INSTRUCTIONS—Print or type. Do not write in shaded areas; these are for FAA use only. Submit original only to an authorized FAA Representative. If additional space is required, use an attachment. For special flight permits complete Sections II and VI or VII as applicable.

I. AIRCRAFT DESCRIPTION

1. REGISTRATION MARK	2. AIRCRAFT BUILDER'S NAME (make)	3. AIRCRAFT MODEL DESIGNATION	4. YR. MFG.	FAA CODING
N1833D	BEECH AIRCRAFT CORP.	F33A	1981	1151425
5. AIRCRAFT SERIAL NO.	6. ENGINE BUILDER'S NAME (make)	7. ENGINE MODEL DESIGNATION		170-32
CE-952	Continental	IO-520-BB		
8. NUMBER OF ENGINES	9. PROPELLER BUILDER'S NAME (make)	10. PROPELLER MODEL DESIGNATION		11. AIRCRAFT IS:
1	Hartzell	PHC-C3YF-1RF		~~NEW~~ USED IMPORT

II. CERTIFICATION REQUESTED

APPLICATION IS HEREBY MADE FOR: (Check applicable items)

A 1 X STANDARD AIRWORTHINESS CERT. (Indicate category) NORMAL X UTILITY ACROBATIC TRANSPORT GLIDER BALLOON

B SPECIAL AIRWORTHINESS CERTIFICATE (Check appropriate items)

2	LIMITED									
5	PROVISIONAL (Indicate class)	1	CLASS I							
		2	CLASS II							
3	RESTRICTED (Indicate operation(s) to be conducted)	1	AGRICULTURE & PEST CONTROL	2	AERIAL SURVEYING	3	AERIAL ADVERTISING			
		4	FOREST (Wild life conservation)	5	PATROLLING	6	WEATHER CONTROL			
		0	OTHER (Specify)							
4	EXPERIMENTAL (Indicate operation(s) to be conducted)	1	RESEARCH AND DEVELOPMENT	2	AMATEUR BUILT	3	EXHIBITION			
		4	RACING	5	CREW TRAINING	6	MKT. SURVEY			
		0	TO SHOW COMPLIANCE WITH FAR							
8	SPECIAL FLIGHT PERMIT (Indicate operation to be conducted then complete Section VI or VII as applicable on reverse side)	1	FERRY FLIGHT FOR REPAIRS, ALTERATIONS, MAINTENANCE OR STORAGE							
		2	EVACUATE FROM AREA OF IMPENDING DANGER							
		3	OPERATION IN EXCESS OF MAX. CERTIFICATED TAKE-OFF WEIGHT							
		4	DELIVERING OR EXPORT	5	PRODUCTION FLIGHT TESTING					

C 6 MULTIPLE AIRWORTHINESS CERTIFICATE (Check appropriate Restricted Operation and Standard or Limited as applicable above)

III. OWNER'S CERTIFICATION

A. REGISTERED OWNER (As shown on Certificate of Aircraft Registration) IF DEALER, CHECK HERE → X

NAME: BEECH AIRCRAFT CORP.

ADDRESS: 9709 E. Central
Wichita, Kansas 67201

B. AIRCRAFT CERTIFICATION BASIS (Check applicable blocks and complete items as indicated)

X AIRCRAFT SPECIFICATION OR TYPE CERTIFICATION DATA SHEET (Give No. and Revision No.): 3A15, Rev. 65

X AIRWORTHINESS DIRECTIVES (Check if all applicable AD's complied with and give latest AD No.): Issue 81-07

AIRCRAFT LISTING (Give page No(s).): N/A

X SUPPLEMENTAL TYPE CERTIFICATE (List number of each STC incorporated): SA785CE, SA3231SW-D

C. AIRCRAFT OPERATION AND MAINTENANCE RECORDS

X CHECK IF RECORDS IN COMPLIANCE WITH FAR 91.173

TOTAL AIRFRAME HOURS—~~Enter for used aircraft only~~: 2:25

3 EXPERIMENTAL ONLY—Enter hours flown since last certificate issued or renewed: N/A

D. CERTIFICATION—I hereby certify that I am the owner (or his agent) of the aircraft described above; that the aircraft is registered with the Federal Aviation Administration in accordance with Section 501 of the Federal Aviation Act of 1958, and applicable Federal Aviation Regulations; and that the aircraft has been inspected and is airworthy and eligible for the airworthiness certificate requested.

DATE OF APPLICATION: 4-23-81

NAME AND TITLE (Print or type): Ronn J. Nelson - Foreman Flight Inspection

SIGNATURE: Ronn J. Nelson

IV. INSPECTION AGENCY VERIFICATION

A. THE AIRCRAFT DESCRIBED ABOVE HAS BEEN INSPECTED AND FOUND AIRWORTHY BY: (Complete this section only if FAR 21.183 (d) applies)

2 FAR PART 121 OR 127 CERTIFICATE HOLDER (Give Certificate No.)

3 CERTIFICATED MECHANIC (Give Certificate No.)

6 CERTIFICATED REPAIR STATION (Give Certificate No.)

5 AIRCRAFT MANUFACTURER (Give Name of Firm)

DATE | TITLE | SIGNATURE

V. FAA REPRESENTATIVE CERTIFICATION

(Check ALL applicable blocks) I find that the aircraft described in Section I or VII meets the requirements for: [X] The certification requested, or [4] Amendment or modification of its current airworthiness certificate. Inspection for a special flight permit under Section VII was conducted by: [ ] FAA Inspector; certificate holder under [ ] FAR 65, [ ] FAR 121 or 127, or [ ] FAR 145.

DATE	DISTRICT OFFICE	DESIGNEE'S SIGNATURE AND NO.	FAA INSPECTOR'S SIGNATURE
4-23-81	ACE EMDO 3-0-43	D. M. Porteous, DOA, PC#8	

FAA Form 8130-6 (7-70)

FAA AIRCRAFT REGISTRY
CAMERA NO. 3 DATE: 8-16-85

VI. PRODUCTION FLIGHT TESTING

A. MANUFACTURER

NAME	ADDRESS

B. PRODUCTION BASIS *(Check applicable item)*

- [ ] PRODUCTION CERTIFICATE *(Give production certificate number)*
- [ ] TYPE CERTIFICATE ONLY
- [ ] APPROVED PRODUCTION INSPECTION SYSTEM

C. GIVE QUANTITY OF CERTIFICATES REQUIRED FOR OPERATING NEEDS:

DATE OF APPLICATION	NAME AND TITLE *(Print or type)*	SIGNATURE

VII. SPECIAL FLIGHT PERMIT PURPOSES OTHER THAN PRODUCTION FLIGHT TEST

A. DESCRIPTION OF AIRCRAFT

REGISTERED OWNER	ADDRESS
BUILDER *(Make)*	MODEL
SERIAL NUMBER	REGISTRATION MARK

B. DESCRIPTION OF FLIGHT

FROM	TO	
VIA	DEPARTURE DATE	DURATION

C. CREW REQUIRED TO OPERATE THE AIRCRAFT AND ITS EQUIPMENT

- [ ] PILOT
- [ ] CO-PILOT
- [ ] NAVIGATOR
- [ ] OTHER *(Specify)*

D. THE AIRCRAFT DOES NOT MEET THE APPLICABLE AIRWORTHINESS REQUIREMENTS AS FOLLOWS:

E. THE FOLLOWING RESTRICTIONS ARE CONSIDERED NECESSARY FOR SAFE OPERATION *(Use attachment if necessary)*

F. **CERTIFICATION**—I hereby certify that I am the registered owner (or his agent) of the aircraft described above; that the aircraft is registered with the Federal Aviation Administration in accordance with Section 501 of the Federal Aviation Act of 1958, and applicable Federal Aviation Regulations; and that the aircraft has been inspected and is airworthy for the flight described.

DATE	NAME AND TITLE *(Print or type)*	SIGNATURE

VIII. AIRWORTHINESS DOCUMENTATION (FAA use only)

- [X] A. Operating Limitations and Markings in Compliance with FAR 91.31 as Applicable
- [ ] B. Current Operating Limitations Attached
- [ ] C. Data, Drawings, Photographs, etc. *(Attach when required)*
- [X] D. Current Weight and Balance Information Available in Aircraft
- [ ] E. Major Repair and Alteration, FAA 337 *(Attach when required)*
- [X] F. This Inspection Recorded in Aircraft Records
- [ ] G. Statement of Conformity, FAA Form 317 *(Attach when required)*
- [ ] H. Foreign Airworthiness Certification for Import Aircraft *(Attach when required)*
- [ ] I. Previous Airworthiness Certificate Issued in Accordance with FAR _____ CAR _____ *(Original attached)*
- [X] J. Current Airworthiness Certificate Issued in Accordance with FAR 21.183(a) per 21.273 *(Copy attached)*

FAA AIRCRAFT REGISTRY
CAMERA NO. 3 DATE: 8-16-85

UNITED STATES OF AMERICA
DEPARTMENT OF TRANSPORTATION—FEDERAL AVIATION ADMINISTRATION

STANDARD AIRWORTHINESS CERTIFICATE

1. NATIONALITY AND REGISTRATION MARKS	2. MANUFACTURER AND MODEL	3. AIRCRAFT SERIAL NUMBER	4. CATEGORY
N1833D	BEECH AIRCRAFT CORP. F33A	CE-952	UTILITY

5. AUTHORITY AND BASIS FOR ISSUANCE

This airworthiness certificate is issued pursuant to the Federal Aviation Act of 1958 and certifies that, as of the date of issuance, the aircraft to which issued has been inspected and found to conform to the type certificate therefor, to be in condition for safe operation, and has been shown to meet the requirements of the applicable comprehensive and detailed airworthiness code as provided by Annex 8 to the Convention on International Civil Aviation, except as noted herein.

Exceptions:

NONE

6. TERMS AND CONDITIONS

Unless sooner surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator, this airworthiness certificate is effective as long as the maintenance, preventative maintenance, and alterations are performed in accordance with Parts 21, 43, and 91 of the Federal Aviation Regulations, as appropriate, and the aircraft is registered in the United States.

DATE OF ISSUANCE	FAA REPRESENTATIVE	DESIGNATION NUMBER
Apr. 23, 1981	D. H. Porteous	DOA, PC#8

Any alteration, reproduction, or misuse of this certificate may be punishable by a fine not exceeding $1,000, or imprisonment not exceeding 3 years, or both. THIS CERTIFICATE MUST BE DISPLAYED IN THE AIRCRAFT IN ACCORDANCE WITH APPLICABLE FEDERAL AVIATION REGULATIONS.

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