

Aircraft Record General Information

Manufacturer Cessna Model 182L
Serial 18259203 Registration Number N 42818
Date of Manufacture 1968

Engine(s) currently installed:

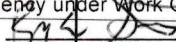
⁰⁹⁻⁰¹⁻⁸⁵
Manufacturer Continental Model IO 520 D Serial 832463-R
Manufacturer _____ Model _____ Serial _____

Propeller(s) currently installed:

Manufacturer Hartzell Model PHC-13YR-1RF
HUB Model _____ Serial FD524B Serial F7691
Blade Model _____ Serial _____ Serial _____ Serial _____
Blade Model _____ Serial _____ Serial _____ Serial _____

LOG BOOK #5

09/02/05 Installed "AIR PLAINS SVCS CORP
300HP Engine Conversion. Continental
IO-520-D Eng and Hartzell prop. (Scimitar)

YEAR 20 <u>21</u> DATE <u>6-10</u>	RECORDING TACH TIME	TODAY'S FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
				<p>Date: 6/10/2021; Aircraft: N42818; Type: CESSNA 182L; S/N: 18259203; Hobbs:; Tach: 4762.71; Total Time:</p> <p>Opened the headliner over the left cockpit door and removed the fuel gauge sender. Removed Floats & Fuel Cells (FFC) manufactured fuel cell FFC P/N 3030-23, S/N CR3984, Cessna Number 1200065-23, from the left wing tank bay. Removed all loose tape in the tank bay and replaced with P/N 231 OD military grade tape. Found a snap in the in the bottom of the tank bay with broken rivet. Riveted the snap back to the bottom of the tank bay using a blind rivet. Installed a replacement fuel bladder tank FFC Mfg. P/N 3030-2, S/N CR4167, Cessna P/N 1200065-23, Construction Type P-2393 meeting TSO-C80. The tank date of manufacture was May 2009. Installed fuel sender and verified that gauge operation by manually moving the sender float arm up and down. Replaced the scot-5 duct in the leading edge and left wing root for the left rear seat passenger vent. Checked the fuel gauge while fueling the aircraft and with 3 gallons of fuel in the tank the fuel gauge is still reading empty. After the three unusable gallons were pumped in the fuel gauge was accurate as measured against the fuel gauge on the truck pumping the fuel in tank. Inspected the tank for fuel leakage after setting full of fuel over night and no leakage is evident at this time. Closed all inspection covers wing root covers and installed the portions of headliner that were opened for this repair. Removed the right cabin door interior panel and found the inside door handle shaft is worn out and the plate bushing assembly that it rides in also worn out. Installed a replacement used serviceable plate bushing assembly P/N 0413195-5, and a used serviceable door latch shaft P/N 1511212-5. Assembled the door latch with the replacement parts and cycled the door lock multiple times with out incident. Installed the interior door panel along with inside door handle and tested the door a second time. The door latch functions properly.</p> <p>These repairs were performed with reference to current Cessna published service information including but not limited to the following manuals: Cessna 100 Series Service Manual P/N D637, Dated 8/01/2019, Cessna Parts Manual P/N P515, Dated 3/15/1974 and other technical data referenced in this work order. The aircraft and/or component identified above was repaired and inspected in accordance with current FAA regulations and was found AIRWORTHY for return to service. Pertinent details of this repair are on file at this agency under Work Order Number 10279.</p> <p>6/10/2021 Roger E. Smith  UW4R585M. Carter Aircraft, Inc. FAA Repair Station, Sebring FL.</p>

YEAR
20
DATE

RECORDING
TACH
TIME

9-12-21
TACH
4791.03

09-12-21 **N42818** **Aircraft Log** A/C TT: 4791.03 Tach: 4791.03

Complied with Annual Inspection Per FAR 43 appendix D. Removed panels needed to inspect airframe. Replaced Bracket air filter with new element # BA8103. Checked all lights for operation. Checked ELT operation in accordance with 91.207D. Operates as designed. Checked warning systems. No defects noted at this time. Filled brake fluid reservoir to proper level. Checked tension on flight cables. Lubed all zerks, flap and pivot points. Performed AD 11-10-09 seat track wear and pin engagement. No defects noted at this time. Next due 4891.03 tach time or Ann. Performed AD20-18-01 Cracks in lower door post strut attachments. Visual inspection per service bulletin. No defects noted at this time. Next due at 5791.03 ACTT / 36 months. Removed front strut and reseal using reseal kit from Cessna. Replaced Anti-Collision bulb with new unit 02-0350433-00.

I certify that this aircraft has been inspected in accordance with the Annual Inspection program and found to be in airworthy condition at this time. Ok to return to service.

David Raper

27147491A

Flite Instruments and Avionics
4927 E Falcon Drive
Mesa, AZ 85215 CS# 6FIR926C 480-516-0308

Reg #	A/C Make	Model	S/N	Tach
42818	Cessna	182L	18259203	2885

I certify that the transponder tests, including data correspondence required by § 91.413, have been performed and found to comply with 14 CFR part 43, appendix F.

Work performed by: [Signature] Date: 09/08/2021
Inspected by: [Signature] Work Order# 2918

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Tucson Aircraft Maintenance
6330 S Aviator Ln Hangar A
Tucson AZ 85735

Make: Textron (Cessna)

Model: 182L

Tail: N42818

S/N: 18259203

Hours: TT: 4824.92 EngTT: 1438.95 Hobbs: 2920.9

Installed grommet in baffling where oil temp probe wiring passes through. Removed battery, tested and re-installed after recharging. Operational test satisfactory.

Signed: *Rick Nelson*

Name: Rick Nelson

Cert: A&P 3253535

Date: 10/23/2021



Tucson Aircraft Maintenance
6330 S Aviator Ln Hangar A
Tucson AZ 85735

Make: Textron (Cessna)

Model: 182L

Tail: N42818

S/N: 18259203

Hours: Tach: 4825.70 Hobbs: 2922.0

Installed new muffler assembly P/N A0750161-28 S/N 218210 with 2 ea P/N A0750161-22/24 clamp kits, 6 ea P/N 652458 no-blo gaskets and 24 MS20500-428 exhaust nuts. Installed 10 ea P/N 654439-1 intake hoses, 6 ea P/N MC0750143-1 heat deflectors and 6 ea P/N 649959 intake gaskets. Installed new P/N 624972-2D0282 hose between flow divider and firewall fitting for fuel pressure gauge. Installed new P/N AE3663161G0116 hose between fuel strainer and electric fuel pump. Installed McFarlane fuel strainer rebuild kit P/N FS-KT-3. Installed new P/N MCC299506-0102 propeller control and MCS1104-3 rod end. Installed new oil pressure switch P/N S3010-1 and new P/N MS28778-4 packing on adapter. Fabricated and installed backstrap between GTX330ES tray and GNS530W tray. Tightened cigar lighter and installed washers and nut on power lead stud. Connected loose G/S antenna lead to GNS530W tray. Tested G/S reception with AV-17, satisfactory. Removed Concorde battery P/N RG35-AXC S/N 40759318. Charged and installed new Concorde battery P/N RG35-AXC S/N 41185780. Ground test ran engine and leak checked, no leaks.

Signed: *Rick Nelson*

Name: Rick Nelson

Cert: A&P 3253535

Date: 11/08/2021

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YEAR 20 DATE	RECORDING TACH TIME	TODAY'S FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
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Tucson Aircraft Maintenance
6330 S Aviator Ln Hangar A
Tucson AZ 85735

Make: Cessna

Model: 182L

Tail: N42818

S/N: 18259203

40885 3045.4

Hours: Tach: 4939.97

Installed new BA8103-1 air filter element. Installed 1 ea P/N 1414116-1 spring on pilot and co-pilot seat adjustment handle. Installed new 1 ea AN6-6A bolt, MS35338-46 lock washer and AN970-6 washer on left wheel fairing.

Signed:


Name: Rick Nelson

Cert: A&P 3253535

Date: 05/28/2022

September 19, 2022. The tachometer time is 4,948.10 hours. The total time of aircraft is 4,948.10 hours. AD72-07-09 (10/17/1974) Fin spar attachment bolt AD was previously complied with 29JUN2020 4679.32 TT and is next due @ 5679.32 total time. AD72-03-03 R3 (10/15/1984) Wing flap actuator AD was previously complied with 5/25/1972 472.25 TT SK150-37B installed has this is a terminating action. AD84-10-01R1 (7/5/1988) Water contamination of fuel system AD previously complied with SK182-85 installed has raised fuel caps this is a terminating action. AD2021-23-12 (12/9/2021) Radio altimeter anomalies AD is not applicable the aircraft is not equipped with a radio altimeter. AD2020-18-01 (11/12/2020) Detect and Address Cracking of the Wing Strut Attach Point AD was previously complied with 9/12/2021 4791.03 TT and is next due 5791.03 TT or 36 Months. AD2022-03-15 (3/21/2022) Ensure the amount of fuel indicated is the amount of fuel available AD is not applicable STC SA01899WI or SA02658SE are not installed. A&P 3789786

David D. [Signature]

YEAR 20____ DATE	RECORDING TACH TIME	TODAY'S FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
				<p>September 19, 2022. The tachometer time is 4,948.10 hours. The total time of aircraft is 4,948.10 hours. The brake lines were secured at gear leg springs with glue and zip ties. The nose strut bottom bolt was replaced with a new bolt and nut. The shimmy damper bolts were safety wired. Both wings lite surface corrosion treated with LPS 3. Right wing leading edge loose wires secured with silicone zip ties. Auxiliary tanks fuel vents were checked. Fuel pumps for auxiliary tanks inspected filters and replaced gaskets. Right hand fuel pump inspection hole was cleaned. The flap Jack screw was lubed. The wires next to the flap motor were secured with zip ties. The flaps were cleaned and lubed. Installed new bearings washers and bushing for both flaps. Left hand wing leading edge wires were secured with silicone and zip ties. Applied silicone to left hand fuel line at trailing edge to prevent chafing. Applied silicone to inboard fuel line next to flap bracket to prevent chafing. The bottom fuselage was cleaned. That tail section was cleaned and lubed. Lite corrosion in tail was treated with LPS 3. The elevator trim mechanism was cleaned and lubed both forward and aft. The pilot and copilot seat bottom pins for up-and-down mechanism were secured. Both pilot copilot seats were lubed. The missing left-hand wingtip screws were replaced with new screws. Applied silicone to scat tubing under instrument panel on pilot side. Secured cables under instrument panel center section. Applied silicone pilot side instrument panel gyro to prevent chafing with nipple fitting. The central vacuum air filter was replaced. Applied silicone to tubing under copilot side ensure panels to avoid chafing with radios and electronics. Secured tachometer cable with zip ties to prevent chafing with wires. Secured flap position cable with zip ties to prevent chafing with cabin heat cable. Right hand door plastic rear replaced missing screws. Stopped drilled small crack in battery box. Installed nipple on master solenoid. Installed nipple on battery positive post. Wires next to and under battery box were secured. Applied silicone to wires under floor panel on copilot side to prevent chafing with floor edge. Cleaned area under counsel.</p> <p>A&P 3789786 </p>

YEAR 20 DATE	RECORDING TACH TIME	TODAY'S FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
				<p>September 19, 2022. The tachometer time is 4,948.10 hours. The total time of aircraft is 4,948.10 hours. The aircraft was inspected and serviced in accordance with Cessna's maintenance manual checklist. The ELT battery was replaced. The ELT was inspected in accordance with FAR 91.207 (d), and the operational check was satisfactory. The new battery replacement date is November 2024. AD77-02-08 (2/1/1977) To Prevent ELT battery failure is not applicable by ELT battery date. AD2011-10-09 (6/17/2011) was complied with by inspection and measurement and is next due at 5048.10 hours. The pilot rudder pedal outside shaft was tightened by installation of washer and retorquing bolt. The pilot rudder pedal inner pivot shaft was repaired with a busing. All the rudder pedals and torque tubes hardware were retorqued. The pilot seat outer seat belt was removed, sent for repair, and reinstalled after repair. The co-pilot seat outer seat belt was removed, sent for repair, and reinstalled after repair. The right rear backseat bolt was replaced with a new bolt. The nose wheel pants were repaired. The fuel selector gear was lubricated. Wires underneath floor pilot side were secured to prevent chafing for fuel line. Secured wires under floor inspection panel on left gear spring. Wires under floor plate copilot side outer were secured. Wires secured under passenger's seat section. Wires under floor baggage compartment section were secured. Stopped drilled cracks left and right strut upper and lower covers. We torqued bolts on rudder pedal copilot side shaft. Both the left and right brake caliper linings were replaced. The O-ring in both the left and right brake calipers were replaced and the brakes were bled. Install placards for fuel capacity. Secured scat tubing and wires with silicone in zip ties both wing root areas. Service the brake reservoirs with hydraulic fluid type 5606. The right wheel fairing was repaired with fiberglass. The left tire was flipped over. The aircraft was lubed and greased. The left main gear leg upper fuselage cover screws were replaced. All control surface cable tension and travels were within limits. The fuel selector plug removed, checked, reinstalled, and safety wired. I certify that this aircraft has been inspected in accordance with an Annual inspection and was determined to be in airworthy condition. A&P 3789786 IA <i>James P. Smith</i></p>

[illegible]

SkyView HDX System Equipment Identification and Location

Use Table 1 to list and locate each SkyView HDX system component installed in this airplane. If the component is installed on an avionics tray, simply enter "avionics tray" under Station Location.

Table 1: SkyView HDX System Equipment List

COMPONENT	PART NUMBER	SERIAL NUMBER	STATION LOCATION	WATERLINE LOCATION	BUTTOCK LINE LOCATION
SV-HDX1100 PRIMARY DISPLAY (LEFT)	503273-000	23894	Left Panel	N/A	N/A
SV-HDX1100 SECONDARY DISPLAY (CENTER)	503273-000			N/A	N/A
SV-HDX1100 SECONDARY DISPLAY (RIGHT)	503273-000	23895	Right Panel	N/A	N/A
SV-HDX800 PRIMARY DISPLAY (LEFT)	503473-000			N/A	N/A
SV-HDX800 SECONDARY DISPLAY (CENTER)	503473-000			N/A	N/A
SV-HDX800 SECONDARY DISPLAY (RIGHT)	503473-000			N/A	N/A
SV-BAT-320 BACKUP BATTERY, PRIMARY DISPLAY (LEFT)	503294-000	31416	Avionics Tray	N/A	N/A
SV-BAT-320 BACKUP BATTERY, SECONDARY DISPLAY (CENTER)	503294-000			N/A	N/A
SV-BAT-320 BACKUP BATTERY, SECONDARY DISPLAY (RIGHT)	503294-000	31417	Avionics Tray	N/A	N/A
EFIS-D10A STANDBY DISPLAY	503296-000	13125	Left Panel	N/A	N/A
SV-MAG-236 REMOTE MAGNETOMETER	503370-000	16268	Left Wing		
SV-OAT-340 OAT SENSOR	503291-000	22723	Left Wing		

COMPONENT	PART NUMBER	SERIAL NUMBER	STATION LOCATION	WATERLINE LOCATION	BUTTOCK LINE LOCATION
SV-GPS-2020 GPS RECEIVER / ANTENNA	503286-000	6118	Top of Cabin		
USB PORT, PANEL MOUNT, PRIMARY DISPLAY (LEFT)	503418-000			N/A	N/A
USB PORT, PANEL MOUNT, SECONDARY DISPLAY (CENTER)	503418-000			N/A	N/A
USB PORT, PANEL MOUNT, SECONDARY DISPLAY (RIGHT)	503418-000			N/A	N/A
SV-ADAHRS-200 ADAHRS	503282-000	16621	Avionics Tray	N/A	N/A
SV-EMS-220 ENGINE MONITORING SYSTEM, ENGINE 1	503365-000	25952	Avionics Tray		
SV-EMS-220 ENGINE MONITORING SYSTEM, ENGINE 2	503365-000				
SV-ARINC-429 ARINC 429 CONVERTER	503378-000	11417	Avionics Tray		
SV-ADSB-472 ADS-B IN RECEIVER	503373-000	1997	Avionics Tray		
SV-XPDR-261 TRANSPONDER	503417-000	14781	Behind Right Panel		
AUTOPILOT SERVO - PITCH	504160-000	58047	Tailcone		
AUTOPILOT SERVO - ROLL	503933-000	28104	Left Wing		
AUTOPILOT SERVO - YAW					
SV-KNOB-PANEL/H KNOB CONTROL PANEL #1 (HORIZONTAL)	503471-000			N/A	N/A

COMPONENT	PART NUMBER	SERIAL NUMBER	STATION LOCATION	WATERLINE LOCATION	BUTTOCK LINE LOCATION
SV-KNOB-PANEL/V KNOB CONTROL PANEL #1 (VERTICAL)	503468-000	15458	Panel	N/A	N/A
SV-KNOB-PANEL/H KNOB CONTROL PANEL #2 (HORIZONTAL)	503471-000			N/A	N/A
SV-KNOB-PANEL/V KNOB CONTROL PANEL #2 (VERTICAL)	503468-000			N/A	N/A
SV-AP-PANEL/H AUTOPILOT CONTROL PANEL (HORIZONTAL)	503465-000			N/A	N/A
SV-AP-PANEL/V AUTOPILOT CONTROL PANEL (VERTICAL)	503463-000	27500	Panel	N/A	N/A
SV-COM-PANEL/H COM CONTROL PANEL (HORIZONTAL)	503460-000			N/A	N/A
SV-COM-PANEL/V COM CONTROL PANEL (VERTICAL)	503456-000	17167	Panel	N/A	N/A
SV-COM-T8 COM TRANSCIEVER	01192-00-03	1735	Behind Left Panel		
SV-BUTTON-APDISC AP DISCONNECT BUTTON	503399-000			N/A	N/A
SV-BUTTON-LEVEL LEVEL BUTTON	503397-000			N/A	N/A
CURRENT SENSOR, 0- 100A	503386-000				

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Tucson Aircraft Maintenance
6330 S Aviator Ln Hangar A
Tucson AZ 85735

Make: Textron (Cessna)

Model: 182L

Tail: N42818

S/N: 18259203

Hours: AFTT:5010.0 TT: 5010.0 Hobbs: 3120.7

Complied with recurring AD 2011-10-09 Inspection of seat tracks, seat roller and locking pins by visual and dimensional inspection per AD instructions. AD 2011-10-09 next due at Tach: 5110.0 or TTAF: 5110.0 or 12 calendar months, whichever occurs first (11/2024). Complied with FAR 91.207(d) inspection by testing "G" and ELT switches. ELT Battery due 11/2024. Complied with AD 83-13-01 (a) by installing placard on instrument panel. Complied with AD 83-13-01 (b) by inspecting IAW AD instructions and installing new P/N 1205253-1 Fuel Cap Alignment Placard on left and right wings. AD 83-13-01 (b) next due in 12 calendar months (11/2024). Installed new BA8103-1 air filter. Disassembled, inspected, cleaned, and reassembled fuel strainer with new MS29513-138 and MS29513-111 o-rings. Installed 1 ea. new flexible brake hoses P/N 350-3-0080 on left and right. Installed 1 ea. new Cleveland P/N 164-02601 brake disc on left and right wheels. Bled and serviced brake systems left and right. Greased all wheel bearings with Aeroshell #5. Installed 1 ea. P/N 154-12000 and P/N 154-12400 molded grease seals on left and right main wheels. Installed 2 ea. P/N 154-13000 molded grease seal on nose wheel. Installed 1 ea. P/N B100142-1 gaskets in left and right fuel cap. Removed Cessna 84C drain valves from left and right wings. Installed Storehouse LLC FDV2 valves using installation instructions 720CM Rev O and IAW STC SA01380CH, with permission to use, and AC 43.13-1B/2B. See FAA Form 337 this date. Weight and balance change negligible. Re-rigged air box cable. Sealed gaps in prop control pass-through at firewall with silicone. Installed 2 ea. new P/N S1819-3 rod ends on left and right aileron bell crank push rods. Replaced left and right, upper and lower strut cuffs with Texas Aeroplastics 31-01-80A strut fairing kit. Engraved and installed new FAR 45.11(e) conforming data plate on aft fuselage forward of the horizontal stabilizer. Lubricated elevator trim tab actuator and chain with Aeroshell 64 grease. Cleaned lubricated control shaft rollers. Installed new P/N AE3663161E0254 PT 10/23 hose between oil pressure port on engine and oil pressure transducer. Relocated oil pressure transducer and secured with Adel clamps to engine mount. Installed new P/N AE3663161E0285 PT 10/23 hose between fuel control and return fitting at firewall. Installed new P/N AE3663161G0190 PT 10/23 hose between electric fuel pump and mechanical fuel pump. Installed new P/N S2416-1-31.0 alternator belt and tensioned. Supported fuel line from aux tank in left and right wing with Adel clamp and bracket to wing rib. Ground test ran, leak checked, no leaks. Operational check satisfactory.

Signed:

Rick Nelson

Name: Rick Nelson

Cert: A&P 3253535

Date: 11/04/2023

YEAR 20 DATE	RECORDING TACH TIME	TODAY'S FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
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Tucson Aircraft Maintenance
6330 S Aviator Ln Hangar A
Tucson AZ 85735

Make: Textron (Cessna)

Model: 182L

Tail: N42818

S/N: 18259203

Hours: AFTT: 5010.0 Tach: 5010.0 Hobbs: 3120.7

- Removed fuel flow transducer P/N FT60 S/N 262768. Installed new fuel flow transducer P/N FT60 S/N 267092 and corrected transducer mounting to be IAW Dynon Skyview HDX and EI FT60 IMs. Verified operation and leak checked, operation satisfactory, no leaks. Removed P/N 0726110-1 fuel transmitter from L/H and R/H tanks. Installed new CIES fuel senders P/N CC284022-0602-501 S/N C9827 L/H and S/N C9828 R/H along with 1 ea. P/N MCS1588-1, MCS1588-2 and 0726002-1 gasket on each side. Installation performed using Installation Document CC-TR-2840-031 Rev F dated 10/20/2020, and IAW STC SA02511SE, with permission to use, and AC 43.13-1B/2B. See FAA form 337 this date. Circuit protection for fuel senders provided by a Klaxon 7274-2-2 2A push-pull circuit breaker on the main bus.
- Connected CIES fuel transmitters with Dynon Skyview HDX, incrementally filled tanks and calibrated fuel levels IAW Dynon Skyview HDX installation manual P/N 103261-000 Rev O dated 07/12/2023. Leak checked after tanks sitting full for several days, no leaks. Weight and balance change negligible. Protected and secured loose wiring in left and right wings, under cabin floor and above headliner with spiral wrap, caterpillar track and zip ties and/or rerouting as appropriate.

Signed:

Cert: A&P 2674482

Name: Ryan R. Klems

Date: 11/04/2023

November 04, 2023 N42818 Tach Time 5010.0 TTAF 5010.0

I certify that this aircraft has been inspected in accordance with an Annual Inspection and was determined to be in airworthy condition. ADs researched and current through November 04, 2023.

Jess Smith

A&P 3069970 IA