

## AVIONICS MAINTENANCE RECORD

AIRCRAFT SERIAL AND REGISTRATION 18280374 N374TC



# MAINTENANCE RECORD

TOTAL TIME IN SERVICE		DESCRIPTION OF THE WORK PERFORMED	AUTHORIZED SIGNATURE CERTIFICATE & NUMBER
HOURS	10ths		
58	6	Removed HS 271C Roll servo- P/N 065-00179-0100 S/N 2489 and Installed Factory Exchange HS271C, P/N 065-00179-0100 S/N 2205. Function Tested Autopilot- Good. See WO 10875 This Station For Details	
		<b>CINEMA AIR AVIONICS</b> FAA APPROVED REPAIR STATION NO. J792880F 2056 Palomar Airport Road Carlsbad, CA 92008	<i>Store</i>
		Removed Failed HS 271C Rollservo S/N 2205 and installed Factory Overhauled HS271C S/N 1965. Function Tested Autopilot- Good. See W. O. # 10909 This station For Details.	
		<b>CINEMA AIR AVIONICS</b> FAA APPROVED REPAIR STATION NO. J792880F 2056 Palomar Airport Road Carlsbad, CA 92008	<i>Shp</i>



# MAINTENANCE RECORD

DATE	TOTAL TIME IN SERVICE		DESCRIPTION OF THE WORK PERFORMED	AUTHORIZED SIGNATURE CERTIFICATE TYPE & NUMBER
	HOURS	10ths		
1999				
12/15	40385 266 769	6 7	Removed & replaced GPS receiver, Type Celling Signal KLN-89B P/N 066-01148-0101. Removed SN 7250, installed SN 80048 Replacement unit has Mod # 111 (01/06) installed as req'd by Cessna Service Bulletin. Calibrated GPS Ramp checked normal.	
			Removed & replaced autopilot computer Type KC-140 P/N 065-00776-5201. Removed SN 2626 installed SN 2779 Replacement unit has Mod 4 (01/06) installed as req'd by Cessna Service Bulletin. Ramp checked normal	
			<b>CINEMA AIR AVIONICS</b> F.A.A. APPROVED REPAIR STATION NO. JRP2000F 2056 Palomar Airport Road Carlsbad, CA 92008 <i>W/O # 11290</i> <i>Monte</i> <i>#2011579</i>	

# MAINTENANCE RECORD

DATE	TOTAL TIME IN SERVICE		DESCRIPTION OF THE WORK PERFORMED	AUTHORIZED SIGNATURE CERTIFICATE TYPE & NUMBER
	HOURS	10ths		
2000				



# MAINTENANCE RECORD

DATE	TOTAL TIME IN SERVICE		DESCRIPTION OF THE WORK PERFORMED	AUTHORIZED SIGNATURE CERTIFICATE TYPE & NUMBER
	HOURS	10ths		
1/10	TACH 805 HOURS 904	1  2	Removed and replaced #1 nav/com Type: Allied Signal KX-155A, p/n 069-01032-0101 Removed: s/n <u>5859</u> , Installed: s/n <u>15664</u> Replacement unit has Hardware Mods #1 thru #8 installed as required by Cessna Service Bulletins. Function checked normal.	
			Removed and replaced #2 nav/com Type: Allied Signal KX-155A, p/n 069-01032-0201 Removed: s/n <u>5763</u> , Installed: s/n <u>15518</u> Replacement unit has Hardware Mods #1 thru #8 installed as required by Cessna Service Bulletins. Function checked normal.	
			<b>CINEMA AIR AVIONICS</b> FAA APPROVED REPAIR STATION NO. JRP2350P 2056 Palomar Airport Road Carlsbad, CA 92008 W/ 11290 #2411579	
1/17	TACH 964 HOURS 1087	.4  .5	Removed and replaced audio panel Type: Allied Signal KMA-26, p/n 066-01155-0201 Removed: s/n <u>2616</u> Installed: s/n <u>3098</u> Replacement unit has Hardware Mod#2 installed as required by Cessna Service Bulletin( ). Function checked normal.	
			<b>CINEMA AIR AVIONICS</b> 2056 PALOMAR AIRPORT RD CARLSBAD, CA REPAIR STATION # AMPR 225C W/ 11290-1 #2411879	



TOTAL TIME  
IN SERVICE

HOURS 10th

DESCRIPTION OF THE WORK PERFORMED



**CINEMA AIR  
AVIONICS**

2056 Palomar Airport Rd.  
Carlsbad, California  
92008

AIRCRAFT MAINTENANCE LOG BOOK ENTRY  
N\_374TC\_\_ DATE\_MAY 22,2000 HOBBS\_914.2\_\_ W.O.#\_11536\_\_

PERFORMED CESSNA SERVICE BULLETIN #SB00-22-01A PERTAINING TO A KAP-140 AUTOPILOT, AND REPAIRED ACCORDING TO HONEYWELL SERVICE BULLETINS, SB KS 271C-5, REV.1, DATED FEB/00 (PRIMARY SERVO), SB KS270C-4, REV.1, DATED FEB/00 (PITCH SERVO) AND SB KS 272C-4, REV.2, DATED FEB/00 (TRIM SERVO). Replaced the pitch servo p/n 065-00178-2200 s/n 1715 with an exchange servo s/n 1803. Function checks- good. SYSTEM GROUND TESTED AND AIRCRAFT RETURNED TO SERVICE.

CINEMA AIR AVIONICS  
2056 PALOMAR AIRPORT ROAD  
CARLSBAD, CA 92008  
FAA APPROVED R.S. #AMPR725C

TECHNICIAN

----- END -----

MAINTENANCE RECORD

DATE TOTAL TIME



# MAINTENANCE RECORD

DATE <u>2000</u> <u>10</u>	TOTAL TIME IN SERVICE	
	HOURS	10ths

DESCRIPTION OF THE WORK PERFORMED

I CERTIFY THAT THE ATC TRANSPONDER TESTS AND INSPECTIONS REQUIRED BY FAR 91.413(a) HAVE BEEN PERFORMED AND FOUND TO COMPLY WITH FAR PART 43, APPENDIX F

TRANSPONDER NO. 1 SERIAL NO. 5477

TRANSPONDER NO. 2 SERIAL NO.                     

DATE May 24, 2000 SIGNATURE [Signature]

CINEMA AIR AVIONICS

FAA CERTIFICATE NO. AMPR725C

2056 'V' PALOMAR AIRPORT RD. WORK ORDER NO. 11640

CARLSBAD, CA 92008

TELEPHONE (760) 438-1042

I CERTIFY THAT THE ALTIMETER AUTOMATIC PRESSURE ALTITUDE REPORTING AND STATIC SYSTEMS TESTS AND INSPECTIONS REQUIRED BY FAR 91.411 HAVE BEEN PERFORMED AND FOUND TO COMPLY WITH FAR PART 43, APPENDIX E. ALTIMETER TESTED TO 20,000 FT.

PILOT ALTIMETER SERIAL NO. 396198

COPILOT ALTIMETER SERIAL NO.                     

DATE May 24, 2000 SIGNATURE [Signature]

CINEMA AIR AVIONICS

FAA CERTIFICATE NO. AMPR725C

2056 'V' PALOMAR AIRPORT RD. WORK ORDER NO. 11640

CARLSBAD, CA 92008

TELEPHONE (760) 438-1042

38-1042

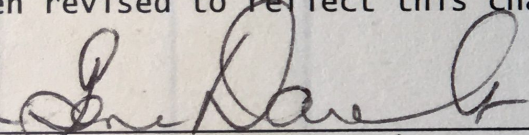
DEC 7, 00	IN SERVICE TOTAL TIME	FACTORY REPAIR OF ADF MODEL KR-87 SN 70028 REINSTALLED - TRANSDUCER'S NORMAL CINEMA AIR AVIONICS AMPR 725C <u>[Signature]</u>
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# MAINTENANCE RECORD

DATE	TOTAL TIME IN SERVICE		DESCRIPTION OF THE WORK PERFORMED
19	HOURS	10ths	
<b>Jet Source Avionics 2056 Palomar Airport Rd. Carlsbad, CA 92008</b> <b>LOG ID# 1 18-December-2000 WO# 11522</b> <b>N374TC S/N 18280374 CESSNA 182S TACH 1151.0</b>			
			Pg 1 / 1

Removed Bendix/King KLN89B GPS (P/N 066-01148-0101 S/N 80048) from aircraft. Installed Bendix/King KLN94 GPS (P/N 069-01034-0101 S/N 1168) per Honeywell Installation manual and in accordance with FAA STC #SA00909WI-D, Dated 11-1-2000, as approved for Beech 95-55 type aircraft. The KLN94 GPS unit is a direct substitute for the above IFR Certified Bendix/King KLN89B GPS. Post-installation ground accuracy and function tests were performed in accordance with the manufacturer's instructions. All systems operated normally. An approved Bendix/King KLN94 GPS Flight Manual Supplement has been installed into the Pilots Operating Handbook. The aircraft weight and balance and equipment list has been revised to reflect this change.


12-18-2000  
 Signed AMPR725C Gene Danenberger, Avionics Insp. Date




# MAINTENANCE RECORD

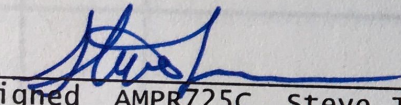
DATE	TOTAL TIME IN SERVICE	DESCRIPTION OF THE WORK PERFORMED
19	HOURS   10ths	

**CINEMA AIR AVIONICS** Jet Source Management 2056 Palomar Airport Rd. Carlsbad, CA 92008  
 LOG ID# 238 23-May-2002 WO# 12964/1 REF# 12964  
 N374TC S/N 18280374 CESSNA 182S

Pg 1 / 1

I CERTIFY THAT THE ALTIMETER, AUTOMATIC PRESSURE ALTITUDE REPORTING, AND STATIC SYSTEMS TESTS AND  
 INSPECTIONS REQUIRED BY FAR 91.411 HAVE BEEN PERFORMED AND FOUND TO COMPLY WITH FAR PART 43,  
 APPENDIX E. ALTIMETER TESTED TO 20,000\_FT.  
 PILOT ALTIMETER SERIAL NO. 396198  
 COPILOT ALTIMETER SERIAL NO.

I CERTIFY THAT THE ATC TRANSPONDER TESTS AND INSPECTIONS REQUIRED BY FAR 91.413(a) HAVE BEEN  
 PERFORMED AND FOUND TO COMPLY WITH FAR PART 43, APPENDIX F.  
 TRANSPONDER NO. 1 SERIAL NO. 5477  
 TRANSPONDER NO. 2 SERIAL NO.

Signed  AMPR725C Steve Jurens

May 23, 2002  
 Date



[illegible]

N 374 TC

Signed

## Make

## Model

Date: 23 Dec 03

S/N:

Details on file under Work Order No. 12880

FAA Approved Repair Station No.  
FDBR221K  
Freedom Avionics Company  
Jeffco Airport  
Broomfield, CO 80021

Date 23 DEC 03 Signature [Signature]  
Freedom Avionics Company CRS No. FDBR221K  
Jeffco Airport  
Broomfield, CO 80021

# MAINTENANCE RECORD

DESCRIPTION OF THE WORK:



# MAINTENANCE RECORD

TOTAL TIME IN SERVICE		DESCRIPTION OF THE WORK PERFORMED	AUTHORIZED CERTIFICATION & NUMBER
HOURS	10ths		
		<p>I certify that the <b>Altimeter, Static System, Encoder and Transponder</b> tests required by FAR Part 91.411 and 91.413 (Part 43 appendix E &amp; F) have been preformed. The Altimeter was tested to <u>20,000</u> feet. WO# <u>51486</u></p> <p>Mile High Avionics Division Arizona Air-Craftsman, Inc. FAA CRS No. ZM3R029M</p> <p>Signature <u>Stuart R. Runy</u> Date <u>12-28-05</u></p>	
		<p>I certify that the <b>Altimeter, Static System, Encoder and Transponder</b> tests required by FAR Part 91.411 and 91.413 (Part 43 appendix E &amp; F) have been preformed. The Altimeter was tested to <u>20,000</u> feet. WO# <u>52200</u></p> <p>Mile High Avionics Division Arizona Air-Craftsman, Inc. FAA CRS No. ZM3R029M</p> <p>Signature <u>Mat Runy</u> Date <u>1-23-08</u></p>	



ME CE 0ths	DESCRIPTION OF THE WORK PERFORMED	AUTHORIZED SIGNATURE, CERTIFICATE TYPE & NUMBER
	5-20-08 Tach 2100.4 altimeter and static check completed to 20,000 ft Stevens Beechcraft see airframe log for entry	

MAINTENANCE RECORD

SMYRNA AIR CENTER

AUTHORIZED SIGNATURE  
CERTIFICATE TYPE  
& NUMBER



## MAINTENANCE RECORD

TOTAL TIME  
IN SERVICE

HOURS 10ths

### SMYRNA AIR CENTER

300 Doug Warpoole Rd.

Smyrna, TN 37167

(615) 459-3337

FAA CRS# RG4R491M

### Transponder Certification

Work Order #: 23138 Acft Reg #: N374TC  
Acft Type: 182S Xponder Pos: Primary  
Xponder P/N: 066-01156-0101 Xponder S/N: 5477

I certify that the ATC Transponder Tests and Inspections required by  
FAR 91.413 were performed and found to comply with Part 43, Appendix F.  
Details of this inspection on file @ Smyrna Air Center.

Date: June 10, 2010  
Signed: David A. Clouse



# MAINTENANCE RECORD

SMYRNA AIR CENTER

615-459-3337

FAA CRS# RG4R491M

## Altimeter / Static System Certification

Work Order #: 23138

Acft Reg #: N374TC

Aircraft Type: 182S

Altimeter Position: Primary

Altimeter P/N: 5934P-3 (United)

Altimeter Serial #: 396198

Signature: \_\_\_\_\_

Date: June 10, 2010

ALTITUDE	READING	TOL.	ALTITUDE	READING	TOL.	FRICTION	TOL.	PRESS	DIFF.(TOL. 25 FT)
-1000	<u>1000</u>	20	14000	<u>13980</u>	100	1k	<u>20</u>	70	28.10 <u>10</u> -1727
0	<u>0</u>	20	16000	<u>15950</u>	110	2k	<u>20</u>	70	28.50 <u>0</u> -1340
500	<u>505</u>	20	18000	<u>17940</u>	120	3k	<u>40</u>	70	29.00 <u>10</u> -863
1000	<u>1010</u>	20	20000	<u>19940</u>	130	5k	<u>40</u>	70	29.50 <u>5</u> -392
1500	<u>1505</u>	25	22000	<del>_____</del>	140	10k	<u>40</u>	80	29.92 <u>0</u> 0
2000	<u>2010</u>	30	24000	<del>_____</del>	155	15k	<u>60</u>	90	30.50 <u>10</u> +531
3000	<u>3010</u>	30	26000	<del>_____</del>	160	20k	<u>60</u>	100	30.90 <u>10</u> +893
4000	<u>3995</u>	35	30000	<del>_____</del>	180	25k	<del>_____</del>	120	30.99 <u>14</u> +974
6000	<u>6000</u>	40	35000	<del>_____</del>	205	30k	<del>_____</del>	140	CASE LK - TOL. 100 fpm <u>15</u>
8000	<u>8000</u>	60	40000	<del>_____</del>	230	35k	<del>_____</del>	160	
10000	<u>9980</u>	80	45000	<del>_____</del>	255	40k	<del>_____</del>	180	
12000	<u>11980</u>	90	50000	<del>_____</del>	280	50k	<del>_____</del>	250	
hysteresis: 1st test point @ 50% max alt. <u>10</u>									after effect - tol. 30 ft. <u>20</u>
2nd test point @ 40% max alt. <u>10</u>									

Encoder MFG \_\_\_\_\_ P/N \_\_\_\_\_ S/N \_\_\_\_\_

This encoder/altimeter and pitot/static system meets the requirements of FAR 91.411 part 43 app. E



## MAINTENANCE RECORD

TOTAL TIME  
IN SERVICE  
HOURS

DESCRIPTION OF THE WORK PERFORMED

AUTHOR  
CERT

LOGBOOK ENTRY FORM

QUALITY AVIONICS

9619 WRIGHT DRIVE

MIDLAND, TEXAS 79711

FAA REPAIR STATION QA9R719J

4/24/2014 CESSNA 182S S/N 18280374 N374TC TACH: 2318.5 HRS  
TESTED ASPEN EFD-1000 PFD DIGITAL AIRDATA ALTIMETER P/N 910-00001-001 S/N 10868  
FOR ACCURACY IN ACCORDANCE WITH FAR 91.411 PART 43 APPENDIX E TO 20KFT.  
TESTED UNITED 5934P-3 STBY ALTIMETER S/N 396198 FOR ACCURACY IN ACCORDANCE  
WITH FAR 91.411 PART 43 APPENDIX E TO 20KFT. AD 74-24-13 AND AD 86-05-02 IS N/A BY  
SERIAL NUMBER. TESTED KING KT-76C TRANSPONDER FOR PROPER OPERATION IN  
ACCORDANCE WITH FAR 91.413 PART 43 APPENDIX F. TESTED TCI SSD120 BLIND ENCODER  
S/N S19177 FOR ACCURACY IN AIRCRAFT IN ACCORDANCE WITH FAR 91.411 PART 43  
APPENDIX E AND FOR ACCURACY CORRELATION IN ACCORDANCE WITH FAR 91.217b.  
TESTED PITOT/STATIC SYSTEM FOR LEAKS IN ACCORDANCE WITH FAR 91.411. DATA  
FORMS ARE ON FILE AT THIS REPAIR STATION REFERENCE WORK ORDER# 8766 DATED  
4/24/14. SIGNED Donald H. Hoyer QUALITY AVIONICS FAA REPAIR STATION QA9R719J



# MAINTENANCE RECORD

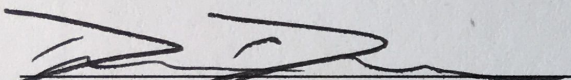
TOTAL TIME SERVICE	DESCRIPTION OF THE WORK PERFORMED	AUTHOR CERT 8
URS 10ths		

G & G Avionics 6002 N. Cedar ave. Lubbock, TX 79403-9971 GG7R357J  
LOG ID# 1499 10-September-2014 WO# 140905 HOBBS 2699.7  
N374TC S/N 18280374 CESSNA 182 S

Pg 1 / 1

Performed roll offset procedure on Aspen EFD1000 Pro  
Adjusted Analog Converter Unit Composite centering potentiometer.  
Performed VOR check , checks ok

Details of work performed are kept on file at this repair station under the work order listed above.

  
GG7R357j Dustin Delano

9/10/14  
Date



# MAINTENANCE RECORD

TOTAL TIME IN SERVICE		DESCRIPTION OF THE WORK PERFORMED	AUTHORIZ CERTIF & I
HOURS	10ths		
		<p>LOGBOOK ENTRY FORM            QUALITY AVIONICS            9619 WRIGHT DRIVE            MIDLAND, TEXAS 79711            FAA REPAIR STATION QA9R719J</p> <p>10/29/2014 CESSNA 182S S/N 18280374 N374TC TACH: 2389 HRS            REMOVED DEFECTIVE TCI SSD-120 BLIND ENCODER S/N S19177 AND INSTALLED NEW            SSD120-30N S/N N21224. TESTED KING KT-76C TRANSPONDER FOR PROPER OPERATION IN            ACCORDANCE WITH FAR 91.413 PART 43 APPENDIX F. TESTED ALTIMETER/ENCODER FOR            ACCURACY CORRELATION IN ACCORDANCE WITH FAR 91.217b. TESTED STATIC SYSTEM            FOR LEAKS IN ACCORDANCE WITH FAR 91.411. REFERENCE QUALITY AVIONICS WORK            ORDER# 8995 DATED 10/29/2014 FOR DETAILS. SIGNED <u>Donald Haefner</u>            QUALITY AVIONICS FAA REPAIR STATION QA9R719J-----</p>	



Test Altitude	Scale Error	Friction Error	Encoder	Leak Acft/Case
-1000	0		-1.5	62018000
0	0		0	
500	0		0	
1000	0	0	1.2	
1500	-10		1.5	
2000	-10	0	2.2	
3000	0	0	3.0	
4000	0		4.2	
5000	0		5.2	
6000	0		6.2	
8000	+10		8.2	
10000	0	0	10.2	
12000	0		12.2	
14000	0		14.2	
15000	0	0	15.2	
16000	0		16.2	
18000	+10		18.2	
20000	+20	0	21.2	
22000				
25000				
30000				
35000				
40000				
45000				
50000				

Hysteresis		
% Altitude Tested	Altitude Tested	Amount Hysteresis
50%	10h	0
40%	8h	0

After Effect	Check Pitot Heat
0	<input type="checkbox"/>

Baro. Scale Error (± 25 ft.)	
28.10	-1727
28.50	-1340
29.00	-863
29.50	-392
29.92	0
30.50	+531
30.90	+893
30.99	+974

Aircraft #	Date Tested:	Technician:	Inspector:
113747C	10/28/16		

Tested in compliance with Appendix E, Part 43, F.A.R 91.411  
 Tested to: 20h ft. W.O. #:

**Precision Static Testing FAA CRS# QJAR856X**

**Form 1A PILOT / FIRST OFFICER / STAND-BY**

AO2 / 08 Jan 2007

Testing Station #QJAR856X  
 Livermore, CA  
 -5904 FAX (925) 449-6081

and static system(s) tests required  
 responder tests, including data  
 § 91.413, have been performed  
 4 CFR Part 43, Appendix E and F.

Time: 2767.7

N 7747C

Date 10/28/16

AUTHORIZED SIGNATURE

# MAINTENANCE RECORD

TOTAL TIME IN SERVICE	
HOURS	10ths

## DESCRIPTION OF THE WORK PERFORMED

AUTHORIZED CERTIFICATION

Aircraft: CESSNA Model: 182S S/N: 18280374 N# 374TC

1. Installed Bendix/King KSN-770 GPS/NAV/COM System in accordance with STCSA11174SC install manual 006-10716-0000



## MAINTENANCE RECORD

TOTAL TIME IN SERVICE	HOURS	10ths	DESCRIPTION OF THE WORK PERFORMED		AUTHORIZED SIGNATURE CERTIFICATE TYPE & NUMBER
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Aircraft: CESSNA Model: 182S S/N: 18280374 N# 374TC

1. Installed Bendix/King KSN-770 GPS/NAV/COM System in accordance with STCSA11174SC install manual 006-10716-0000 rev7.
2. The KSN-770 is installed as # 1 system.
3. The KSN-770 IS interfaced to the ASPEN PFD for navigation using GAMA 429 for GPS, ARINC 429 VOR/LOC information.
4. The KSN-770 is interfaced to the ASPEN ADC for baro corrected for altitude information.
5. The KSN-770 is interfaced to the A/C existing audio panel for audio.
6. The KSN-770 is circuit protected by the Klaxon 7277-2-7.5 Amp circuit breaker labeled " GPS #1 , 7277-2-10 amp com # 1.
7. The KSN-770 is interfaced to the existing Com/Nav antennas.
8. The Airplane Flight Manual Supplement , ( required to be attached to, or remain with , the Approved Flight Manual ) P/N 900-00024-001 Rev H dated Nov-10 -2014 (or later FAA approved revision) is required for this installation.
9. The configuration and check-out log sheet has been completed and is to maintained with the aircraft permanent records.
10. By design, the KSN-770 does not require periodic , preventive. Maintenance of KSN-770 is REQUIRED ONLY ON CONDITON OF FAILURE.
11. The locations of the units were determined to meet the field-of-view requirements without the need of external annuciation.
12. The installation of the GPS does not interfere with the operation of any other equipment installed in the Aircraft in accordance with AC 20-138A Paragraph 22a(3) and AC 20-138A Paragraph 17c(1).
13. The KSN-770 used as the Navigation source annuciator , is installed in the center radio stack , inicationg the current navigation source provided to the CDI/HSI indicator in accordance with AC 20-138A Paragraph 18c.
14. All displays, controls, and annuciators are readable under all normal cockpit conditions and expected ambient light conditions. Night lighting is consistent with other cockpit lighting. System controls and displays have been designed to maximize operational suitability and ease of use in accordance with AC 20-138A Paragraph 22c (1).
15. Operation of the GPS equipment will not adversely affect the performance of other equipment. Operation of the GPS will not interfere with the operation on any other equipment installed in the aircraft in accordance with AC 20-138A Paragraph 17c(1).
16. Location of the GPS display used as a primary flight instrument in the guidance and control of the aircraft, for maneuver anticipation, or for failure/status/integrity annuciation , are located where it is visible to the pilot with the least practicable deviation from the pilot's normal position and line of vision in accordance with AC 20-138A 18d.



# MAINTENANCE RECORD

TE	TOTAL TIME IN SERVICE		DESCRIPTION OF THE WORK PERFORMED	AUTHORIZED CERTIFICA & NUM
HOURS	10ths			
17.			An input of pressure and/or barometric altitude has been provided to the GPS equipment in accordance with AC 20-138A Paragraph 18(2).	
18.			The GPS installed performance ground tests for installation switching and transfer functions and interference from VHF radios has been satisfactorily completed in accordance with AC 20-138A Paragraph 22a(1) and AC 20-138A Paragraph 22a(3).	
19.			Electrical load can be safely controlled or managed within rated limits of the aircraft electrical power supply system per AC 43.13-1B, Paragraph 11-36 (b) and (c).	
20.			Electrical components added are circuit protected in accordance with AC 43.13-1B, Paragraphs 11-47, 11-48, 11-49, and 11-50.	
21.			Wiring used meets all requirements of AC 43.13-1B, Paragraph 11-77(d), and Paragraph 11-66, subparagraphs (b) and (c) and has been installed in accordance with AC 43.13-1B, Paragraph 11-96, subparagraphs a, c, d, f, q, and e.	
22.			Aircraft Weight and Balance was updated and installed in Weight Balance log.	
23.			Aircraft Equipment List updated as per AC 43.13-B Paragraph 10-9.	
24.			Logbook entries complied with.	

This Installation has been inspected and tested in accordance with manufacturers specs and applicable FAR's, and been found airworthy with respect to the work performed and is APPROVED FOR RETURN TO SERVICE.  
 J and R Electronics F.A.A. #XB3R965L  
 Signed: [Signature] Date: 11-8-16

# MAINTENANCE RECORD

Aircraft: CESSNA

Model: 182S

S/N: 18280374

N#: N374TC



## MAINTENANCE RECORD

Aircraft: CESSNA

Model: 182S

S/N: 18280374

N#: N374TC

1. Removed Bendix/King KT-76C transponder.
2. Installed Bendix/King KT-74 transponder using installation manual Bendix/King D201308000037 Rev 0 with reference to STC SA 00765DE. The Bendix/King KT-74 is circuit protected by the existing 5 AMP circuit breaker labeled "XPNDR" located in the aircraft's radio circuit breaker location. Used existing transponder antenna. The altitude information is provided by existing Aspen using RS-232. For continuing airworthiness inspect in accordance with the attached Bendix/King KT-74 Transponder ICA Checklist.
3. The KT-74 is interfaced to the audio panel for TIS function through an unswitched audio input. The KT-74 is interfaced to the KNS-770 #1 using RS-232 information for TIS display and control Peregrine Airspeed switch P/N PA111114-820-1 installed as per STC.
4. Airplane Flight Manual Supplement, Document Number E-BK-13-0009 Rev B dated 02-02-2015 (or later revision) will be located in the supplemental section of the aircraft flight manual, is required for this installation.
5. Transponder tested by Precision Instruments CRS QJAR856X and complies with appendix F of part 43 in accordance with FAR 91.413. Static systems tested by Precision Instruments CRS QJAR856X and complies with paragraph (a) of appendices E of part 43 in accordance with FAR 91.411. Automatic Pressure altitude reporting equipment and ATC Transponder system integration was tested by Precision Instruments CRS QJAR856X and complies with paragraph (c) of appendix E per part 43 in accordance with FAR 91.411.
6. Electrical load can be safely controlled or managed within rated limits of the aircraft's electrical power-supply system per AC 43.13-1B, Paragraph 11-36 (b) and (c).
7. Electrical Components added are circuit protected in accordance with AC 43.13-1B, Paragraphs 11-47, 11-48, 11-49 and 11-50.
8. Wiring used meets all requirements of AC 43.13-1B, Paragraph 11-77(d), and Paragraph 11-66, subparagraphs (b) and (c) and has been installed in accordance with AC 43.13-1B, Paragraph 11-96, subparagraphs a, c, d, f, q, and ee.
9. Aircraft Weight and Balance re-computed in accordance with AC43.13-1B Paragraph 10-16.
10. Aircraft Equipment List updated as per AC43.13-1B Paragraph 10-19.
11. Logbook entries complied with.

-----END-----

This Installation has been inspected and tested in accordance with manufacturers specs and applicable FAR's, and been found airworthy with respect to the work performed and is APPROVED FOR RETURN TO SERVICE  
J and R Electronics F.A.A. #XB3R965L

Signed: JRW

Date: 11-8-16



Test Altitude	Scale Error	Friction Error	Encoder	Leak Acft/Case
-1000	-10		0	Hysteresis
0	-10		0	
500	-10		0	
1000	-10	-	0	
1500	-10		0	% Altitude Tested
2000	-10	-	0	50 % 10000 0
3000	-10	-	0	40 % 8000 0
4000	-10		0	After Effect (+/- 30 ft)
5000	-10		0	Check Pitot Heat
6000	-10		0	Baro. Scale Error (+/- 25 ft)
8000	-10		0	28.10 -1727 -
10000	-10	-	0	28.50 -1340 -10
12000	-10		0	29.00 -863 -10
14000	-10		0	29.50 -392 -10
15000	-10		0	29.92 -0 -10
16000	-10	-	0	30.50 1531 -10
18000	-10		0	30.90 1893 -10
20000	-10	-	0	30.99 1974 -10
22000				Aircraft #:
25000				N374TC
30000				Date Tested:
35000				10-15-18
40000				Technician:
45000				Inspector:
50000				

Tested in compliance with Appendix E, Part 43, F.A.R. 91.411

Tested to: 20000 ft W.O. # 18368

J and R Electronics FAA 464-90 / XB3R965L

☒ PILOT / ☐ FIRST OFFICER / ☐ STANDBY

## MAINTENANCE RECORD

THE WORK PERFORMED

AUTHOR  
CE

TRANSPONDER(S) HAVE BEEN TESTED AND INSPECTED AND FOUND TO COMPLY WITH APPENDIX F OF PART 45 I.A.W. FAR 91.413

DATE 10-15-18 W.O.# 18368

SIGNED

J & R ELECTRONICS. FAA CRS #XB3R965L

333 West Jack London Blvd #141

Livermore, CA 94551

MAINTENANCE RECORD

TOTAL TIME  
IN SERVICE